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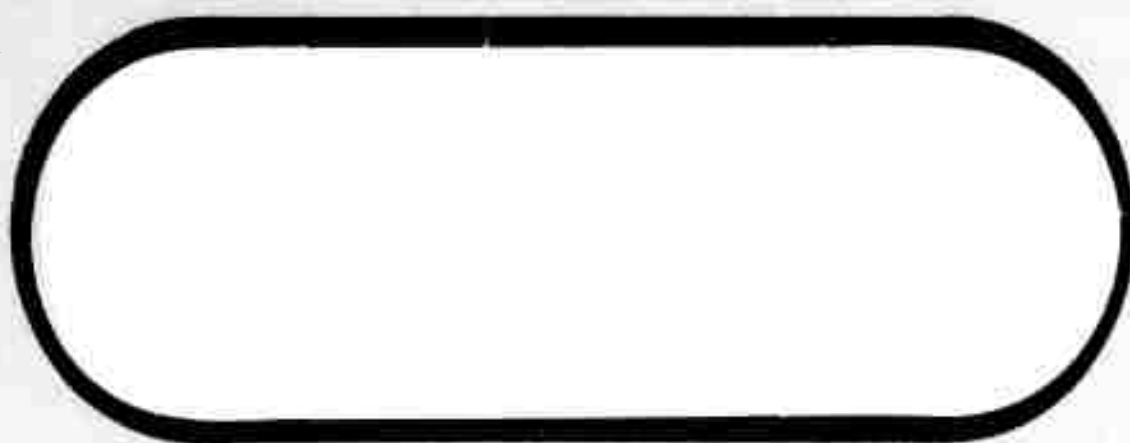
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SPO # 203

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This report was prepared in accordance with the requirements of CCN 35 to Contract AF33(657)-7132 and is submitted in compliance with paragraph B(1.1.3.1.1)1.1 of Statement of Work System 620A, Exhibit 620A-62-2, dated 26 January 1962, revised 1 August 1962.



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## 1.0 SUMMARY

Aerodynamic noise data obtained from a transonic wind tunnel test of an X-20/Titan IIIC model are reported. Measurements made at 28 test points on the X-20 payload only were designed to define the aerodynamic noise environments resulting from separated flow and fluctuating shock phenomena associated with transonic flight in the Mach number range 0.60 - 1.08. Angles of attack and yaw were varied within  $\pm 4^\circ$ . Reported data include overall and one-third octave band sound pressure levels and space-correlation coefficients in the frequency range 40 - 2500 cps. These have been evaluated for a number of representative test conditions selected from a total of 202 test runs.

No analyses of reported data are incorporated in the present document which is essentially a test data report intended to provide data for subsequent analyses. The latter will be conducted by X-20 Structures and Materials Technology relative to specific studies of interest, e.g., comparison of measured data with predicted aerodynamic noise environments at specific locations on the X-20 configuration. Brief descriptions of instrumentation and basic test procedures are presented.

This test represents the Boeing Company's portion of the X-20/Titan IIIC buffet test program conducted jointly with the Martin-Marietta Company. The test was conducted during the period of 22 April 1963 through 1 May 1963 at the NASA/Ames 14 foot wind tunnel facility.

## 2.0 INTRODUCTION

Aerodynamic noise (buffeting) is anticipated to be the primary cause of X-20 structural vibration during flight. Early in 1962 The Boeing Company conducted wind tunnel tests to measure the fluctuating aerodynamic pressures on a model X-20 abort vehicle (data reported in Boeing Document D2-80713). Analyses of test data indicated that measured sound levels were within the acoustic design environment for the external surface of the X-20 glider. Data were lacking, however, which established the effect of the Titan III booster configuration on the acoustic environment at aft sections of the glider during the transonic boost phase of flight.

Mercury flight data measured at the booster/capsule adapter shoulder indicated the existence of rms fluctuating pressure magnitudes of the order of 10% to 20% of the free stream dynamic pressure during the transonic portion of flight (NASA Technical Note D-997). Similar results were observed in wind tunnel tests, employing various payload shapes, conducted at the NASA/Ames Research Center (NASA Technical Notes X-503, X-646). Extrapolation of these data to the X-20 configuration resulted in an increased acoustic design environment for the X-20 glider-booster transition section.

In view of the above considerations, the test program described in the present document was conducted to aid in further defining the acoustic environment associated with the X-20.

Honeywell Model LAR 7400 tape recorder (large rack-mounted recorder to the left of the figure). The second tape recorder shown (to the right) in Figure 6 is a 14-channel Ampex Model CP 100 used to re-record the test data for data reduction at Boeing-Seattle in compliance with a requirement specifying retention of all original data tapes at the Ames Research Center.

### 3.3. Calibration

#### 3.3.1 Frequency Response

A block diagram of instrumentation used for frequency response calibration of the Ames transducer systems is shown in Figure 7. Calibration was performed by a closed-coupler, pressure calibration technique, using a wedge-shaped pressure coupler.

The Ames transducer and a 1/4 inch diameter Bruel & Kjaer Model 4136 condenser microphone, employed as a secondary standard, were mounted with their diaphragms flush with the internal cavity of the closed-coupler calibrator. The standard microphone was calibrated by an electrostatic actuator and found to have a flat frequency response from 40 through 2500 cps. The sensitivity of the standard microphone was checked prior to each calibration with a Bruel & Kjaer pistonphone at a level of 124 db and a frequency of 250 cps.

The acoustic driver unit was excited by a sinusoidal sweep signal; a constant sound pressure level of 105 db re 0.0002 microbar, as indicated by the reference microphone, was maintained by use of compressor circuitry over a 40 - 2500 cps frequency range. Traces of the output signals of the Ames and reference transducers were obtained on a Bruel & Kjaer graphic level recorder for direct comparison of frequency response of the two transducers to a common sound pressure level input.

Five of the 28 Ames transducers used in this test program were individually calibrated by the method described. The frequency response characteristics of these five transducer systems are shown in Figure 8 (a-e). These curves indicate some dropping off of response at frequencies above 500 cps for two transducers; compensation for this response characteristic for the transducer systems not individually calibrated was accomplished by applying a frequency response correction as determined from Figure 9, which shows the average response correction curve for the five transducers individually calibrated.

### 3.3.2 Phase Shift Determination

A block diagram of the instrumentation to determine the phase relation between transducer systems used for evaluation of space-correlation coefficients is shown in Figure 10.

Two Ames transducers were mounted in the closed-coupler calibrator, which imposed an in-phase dynamic pressure with single frequency excitation of the acoustic driver unit. The output signals of the transducers were amplified and recorded on magnetic tape by the Honeywell Model IAR 7400 recorder.

The recorder signals were played back, with the signal from one transducer directed to one axis of an oscilloscope and the signal from the other transducer directed through a phase shifter to the other axis of the oscilloscope. The phase shifter was adjusted until the Lissajous figure on the oscilloscope indicated no phase shift between the two signals; the phase angle was then read directly on the phase shifter dial.

The phase angle between the two systems forming each pair used for space correlation evaluations is shown in Table I. Specific transducer system - tape recorder channel assignments were maintained throughout the test program. These assignments are listed in Table II, which also shows the corresponding Ampex CP 100 recorder channels on which data for space-correlation measurements were copied.

### 3.3.3 Absolute Sound Pressure Level Calibration

Transducers were calibrated with reference to an absolute sound pressure level after mounting of the model in the wind tunnel. A battery-operated, transistorized acoustic calibrator, especially fabricated for this application to provide a compact, portable unit, imposed a sound pressure level of 150 db re 0.0002 microbar at a frequency of 1000 cps at the transducer diaphragm.

Daily static pressure calibration,\* using an input gage pressure of -1.0 psi, was performed on all transducers by a member of the Ames Research Center Staff. For transducers that were mounted in the model and inaccessible to the dynamic pressure calibrator, the static pressure calibration data were applied. This procedure was based on consistent agreement noted between results of both static and dynamic pressure calibration of other transducers.

### 3.3.4 Vibration Response of Transducers

The vibration response characteristics of typical Ames transducers were evaluated prior to testing. Results of previous X-20 model tests (Martin Marietta Co. Inertia Compensated Balance Tests) conducted in the 14-foot transonic wind tunnel indicated that a maximum overall vibration environment of 1g rms could be expected. Based on these findings, all reported acoustic data are at least 8 db above the level corresponding to the maximum signal due to vibration response.



### 3.4 Procedures

#### 3.4.1 Pre-test procedures

Daily calibration was performed on all transducers on the mounted model, as described in the preceding section, and the system response to the absolute sound pressure level calibration signal was recorded. Carrier amplifiers were balanced prior to recording all calibration signals and before each test run.

#### 3.4.2 Test Conditions

The test conditions were selected to cover the range of interest and planned in a sequence allowing minimum delays for model changes. Complete print-outs of model attitude and tunnel parameter values were obtained during each test run which was conducted for a minimum duration of one minute under stable conditions. Test runs (identified by correlation numbers), corresponding tunnel Mach numbers, and angles of attack and yaw defining model attitudes are listed in Table III (a). Nominal values of tunnel parameters corresponding to given Mach number conditions are shown in Table III (b).

Test signals from the 28 transducers were monitored, attenuated if necessary to prevent overload, and recorded on the Model LAR 7400 tape recorder for one minute of test run under stable conditions.

#### 3.4.3 Re-recording of Test Data

As mentioned, all test data were re-recorded on the Ampex Model CP 100 in compliance with the requirement for all original data tapes to remain at the Ames Research Center. Additional considerations associated with the recording and re-recording procedures include the following:

A characteristic frequency response drop at 2500 cps was noted on all channels of the Honeywell Model LAR 7400 tape recorder; Table IV lists the appropriate correction applied to sound pressure level data for the one-third octave band centered on 2500 cps to compensate for the frequency response of each data channel of the LAR 7400 recorder. No additional correction for tape recorder frequency response is required, based on the observation of a flat frequency response throughout the 40 - 2500 cps range of interest for all channels of the Ampex CP 100 recorder.

Phase angle shifts, of interest to space-correlation evaluations, were significantly affected by the re-recording procedure. To check for phase shift, a common signal was recorded on all channels of the Model LAR 7400 tape recorder, played back, and re-recorded by the Ampex CP 100 recorder and played back through a phase detection circuit. Figure 11 is a block diagram of the instrumentation used for checking the phase shift of recorded and re-recorded data through the two tape recorders on the various pairs of data channels used for space-correlation evaluations. Table V lists the phase shift values associated with the indicated pairs of LAR 7400 and CP 100 data channels.

### 3.5 Data Reduction

#### 3.5.1 Sound Pressure Level Data

Acoustic instrumentation used for reducing the taped test data to obtain sound pressure level values in one-third octave bands of frequencies is shown in Figure 12. Spectrum analysis of the acoustic data was accomplished by playing the data tapes through one-third octave band filters and obtaining traces of the true rms output voltage signals on a graphic level recorder. The traces were referenced to



the voltage trace corresponding to the 150 db calibration signal to determine one-third octave band sound pressure level values. These values defined a sound pressure level spectrum in the frequency range 40 - 2500 cps after appropriate corrections were made for frequency response of the transducer system and tape recorder. Summation of the one-third octave band levels yielded an overall sound pressure level in the frequency range of interest.

In some cases, however, only the overall sound pressure levels were determined by playback of data tapes through a broad band filter passing all frequencies in the 40 - 2500 cps range. Since these values are not corrected for frequency response, they may reflect slightly different values for overall sound pressure level data relative to the overall sound pressure level value computed by summation of component one-third octave band sound pressure level data. The computed values are distinguished by an asterisk following the tabulated overall sound pressure level data. The difference in values of overall sound pressure level determined by the two methods described will be greatest for spectra which peak at frequencies where system and tape recorder response begins to fall off.

### 3.5.2 Space Correlation Data

Acoustic instrumentation used for determining voltage values required for evaluation of space-correlation coefficients (R) is shown in Figure 13.

Signals from two transducers, "a" and "b", used to form one pair for space correlation coefficient evaluations, were played-back through a sum and difference amplifier which performed a

vector addition and subtraction of the two signals. The magnitudes of the signals of the two systems were equalized by amplifier gain controls, and the space correlation coefficient, R, was evaluated by the expression,

$$R = \frac{\left( \frac{\overrightarrow{E_a + E_b}}{\overrightarrow{E_a - E_b}} \right)^2 - 1}{\left( \frac{\overrightarrow{E_a + E_b}}{\overrightarrow{E_a - E_b}} \right)^2 + 1} \quad \begin{matrix} |E_a| = |E_b| \\ \text{(rms)} \end{matrix}$$

where  $\overrightarrow{E_a}$ , voltage output of transducer system "a"  
 $\overrightarrow{E_b}$ , voltage output of transducer system "b".

#### 4.0 RESULTS

##### 4.1 Sound Pressure Level Data

Sound pressure level data in one-third octave bands of frequency, corrected for attenuator settings and system frequency response characteristics, are reported in Table VI and Figure 14. Overall sound pressure level data evaluated by playback of data tapes through a broad band filter, as described in the preceding section, are shown in Table VII. Note that tabulated values of overall sound pressure level which were computed by summation of component one-third octave band levels are identified by an asterisk.

Wind tunnel background noise levels for the Mach number range 0.7 - 1.1 are shown in Figure 15 for comparison of measured noise data relative to background noise. Data shown in Figure 15 were measured by the Ames Research Center Staff during previous tests conducted in the 14-foot transonic wind tunnel.

#### 4.2 Space-Correlation Coefficient Evaluations

Space-correlation coefficients evaluated for selected test conditions and pairs of data systems are shown in Table VIII. Figure 16 shows space-correlation coefficients for various one-third octave bands in the 40 - 2500 cps range considered. It should be noted that considerable phase shifts were introduced by the instrumentation; this seriously limits the usefulness of the measured space-correlation coefficients. Therefore, no space-correlation data are reported at frequencies where the phase shift exceeds 20°.

#### 4.3 Transient Pressure Fluctuations

A small portion of the acoustic data indicated large transient pressure fluctuations. Typical oscillograph time history traces of the peak fluctuating pressure measured are shown in Figure 17. The large transient pressure phenomenon was observed from measurements made from the cylindrical transition section but the data are not reported in the acoustic test results. The values of transient condition are indicated in Figures 17b and 17c. Similar transient data have been reported in other wind tunnel tests (NASA D-1633). The major portion of the acoustic data measured is of the type shown in Figure 17a and is considered to be most representative of the acoustic loading on the glider/transition structure. However, Boeing X-20 Structures will evaluate the significance of the transient phenomenon (Figure 17b and 17c) in detail.

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NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
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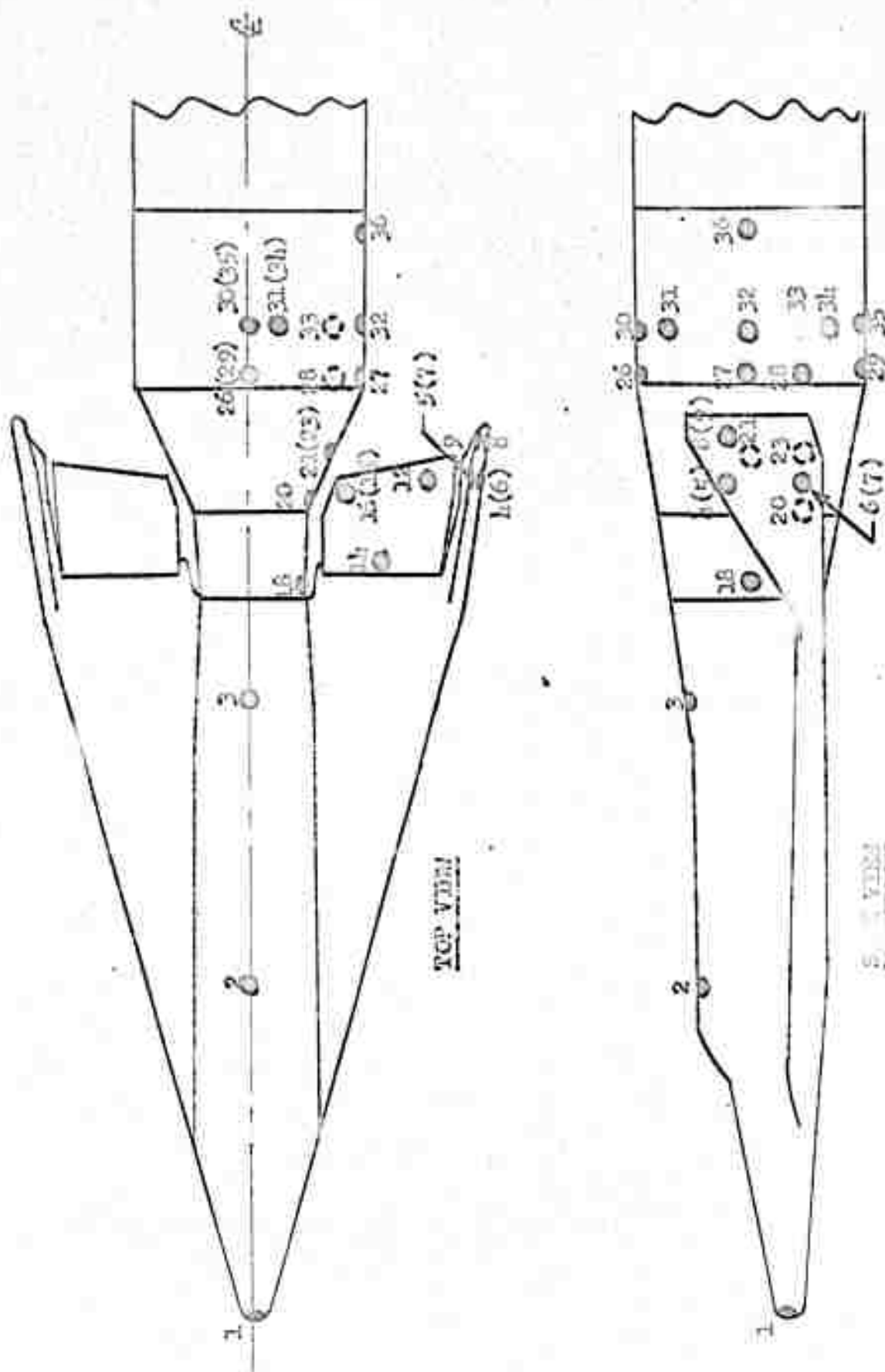
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Figure 2. Transducer Locations.

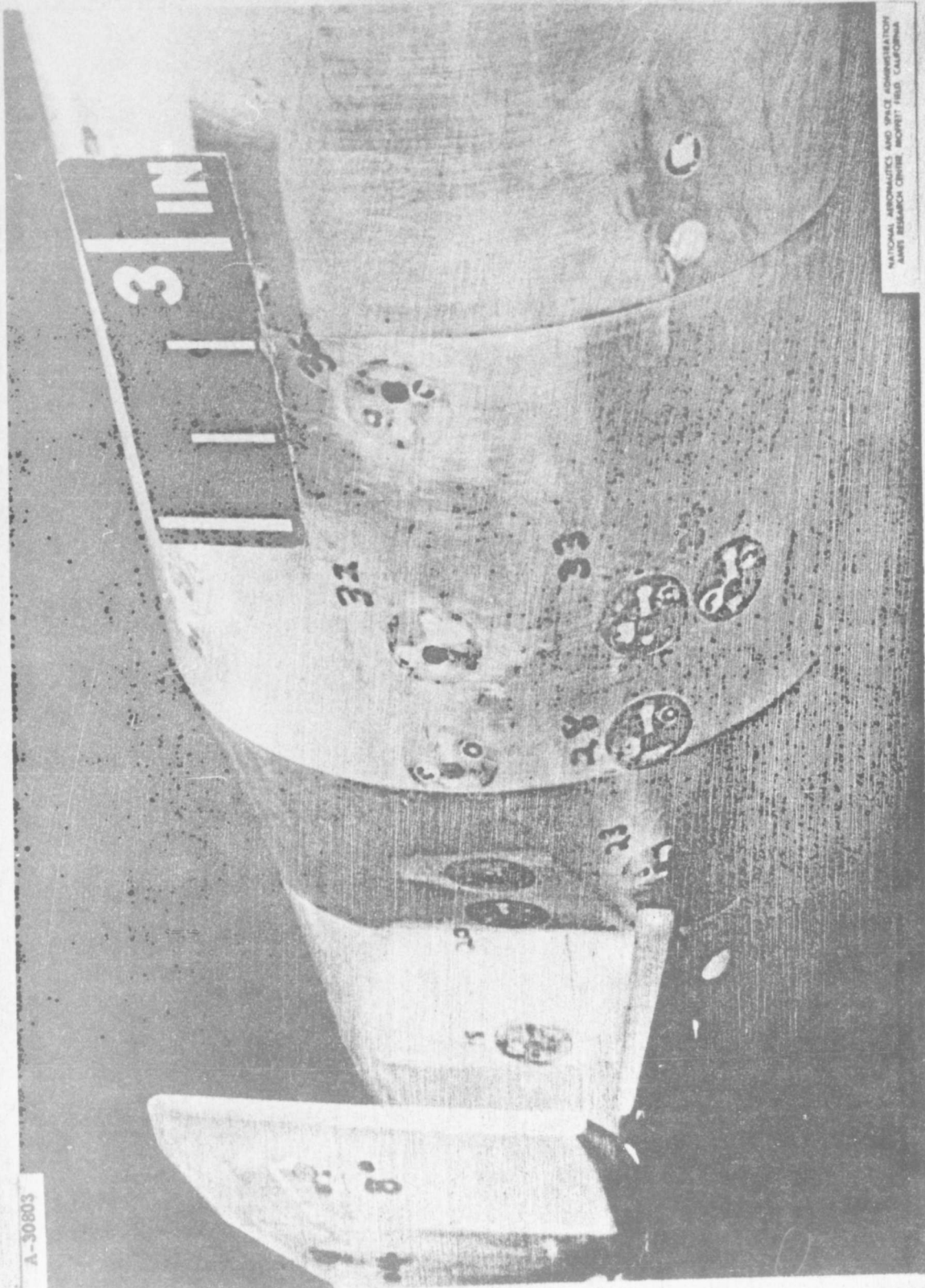
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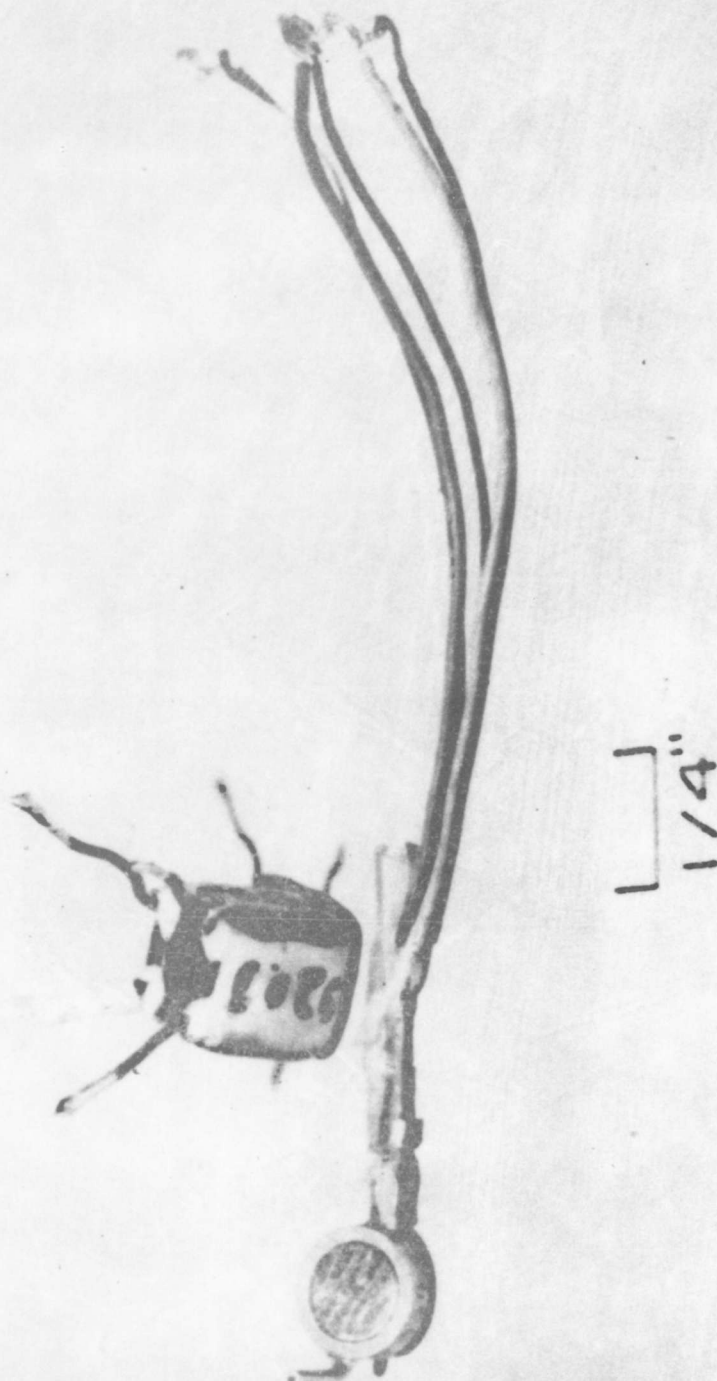
Figure 3. X-20/624A model, showing flush-mounted transducers.

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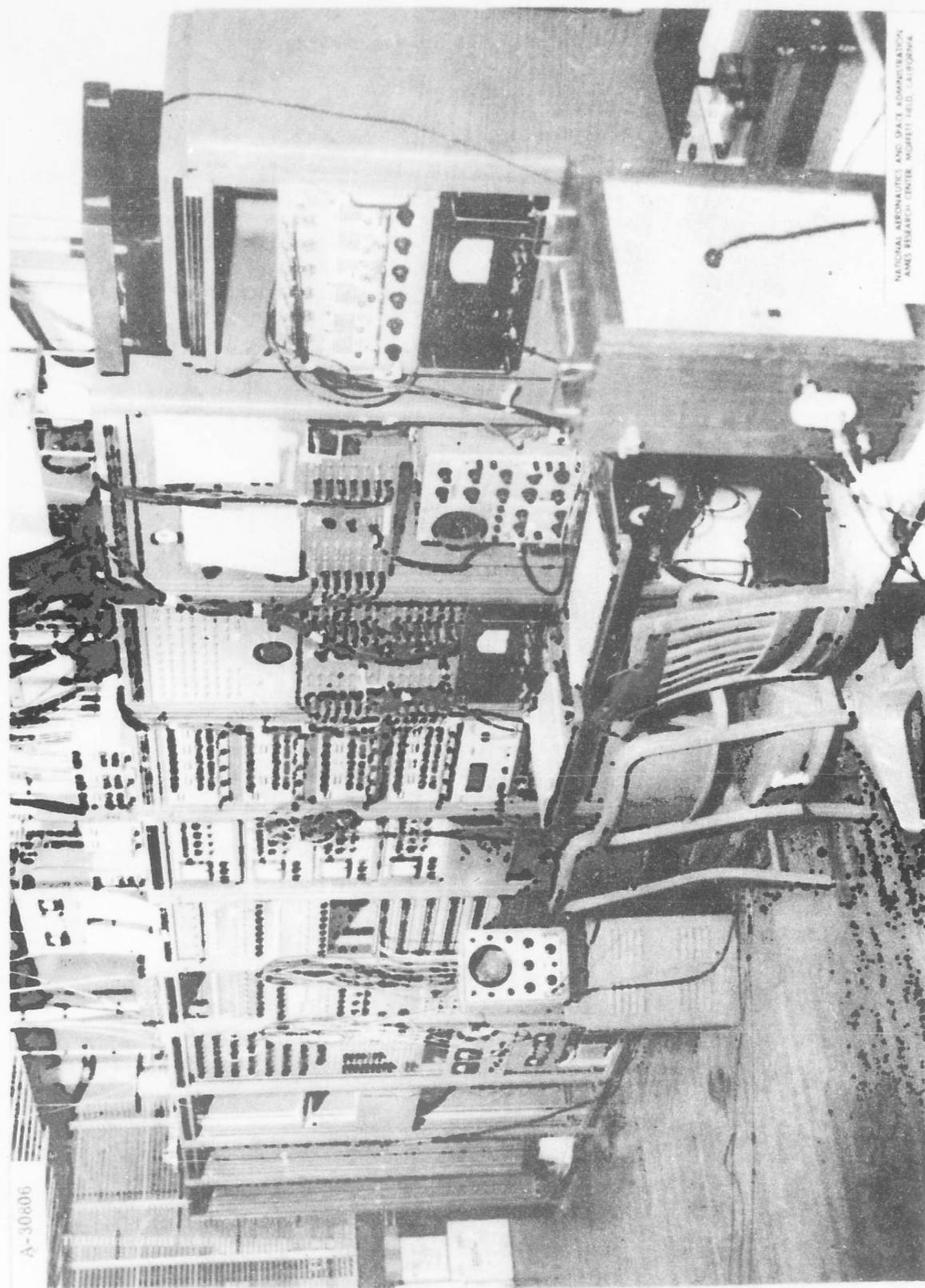




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CALC			REVISED	DATE	Figure 1. Ames pressure transducer and external bridge.	
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NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
 AMES RESEARCH CENTER, MOUNTAIN VIEW, CALIFORNIA

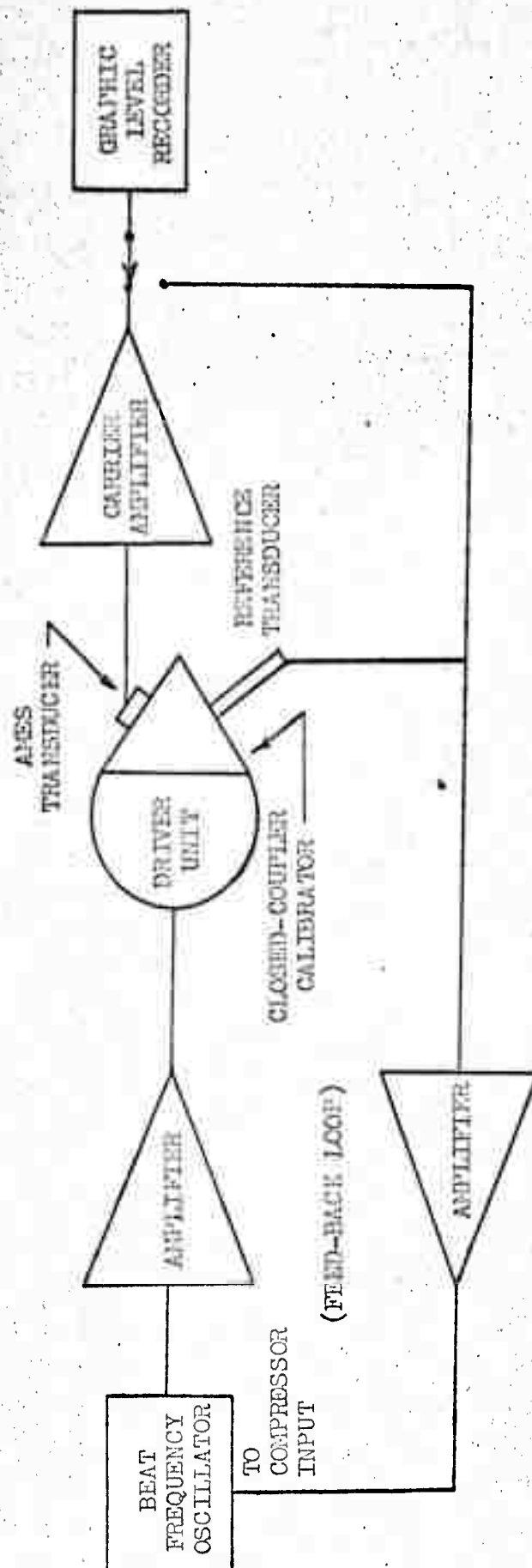


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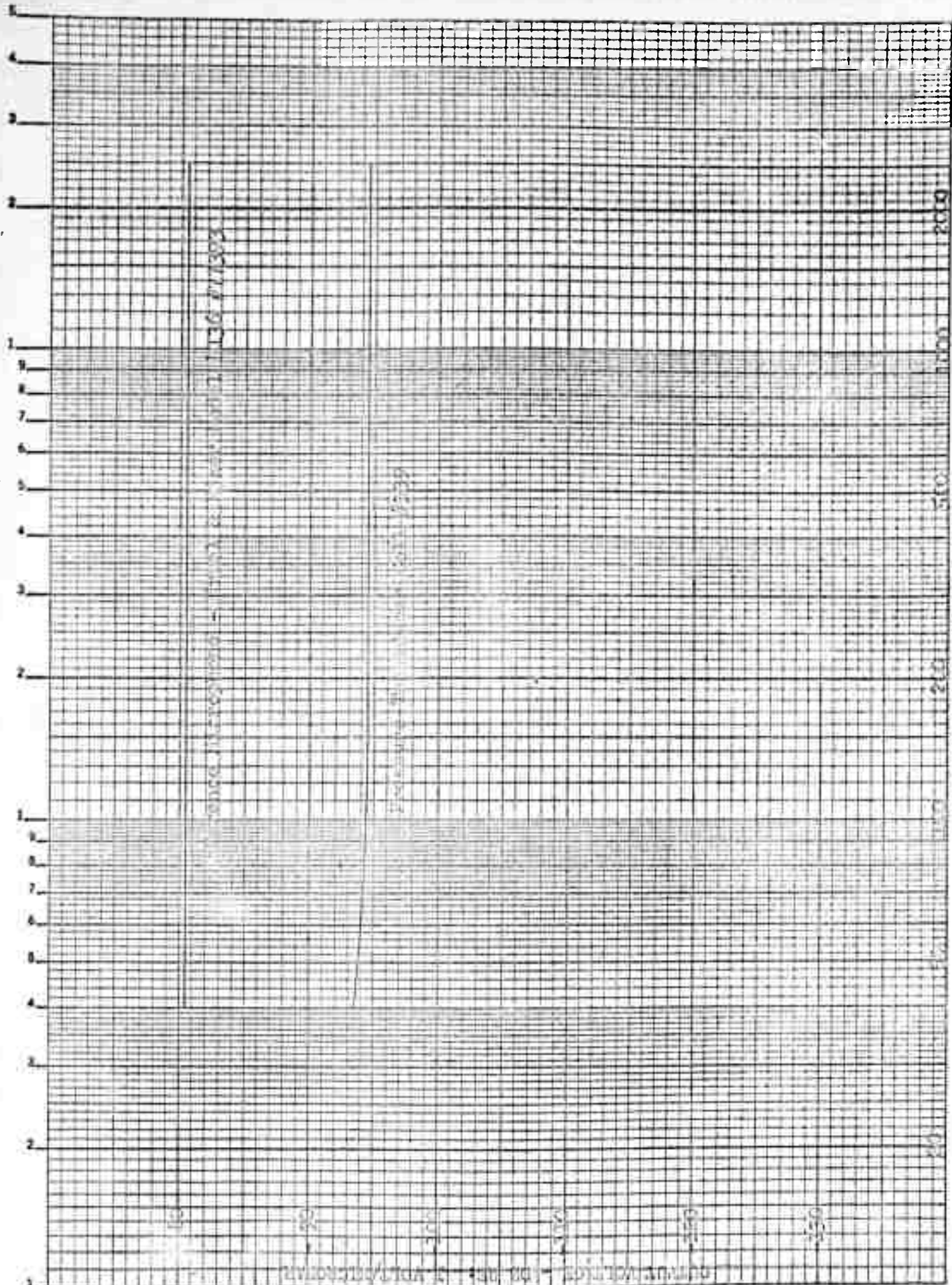
CALC			REVISED	DATE	Figure 1. Support instrumentation for transducer and signal monitoring..	
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CALC			REVISED.	DATE	Figure 7. Block diagram of equipment -- Transducer calibration.	
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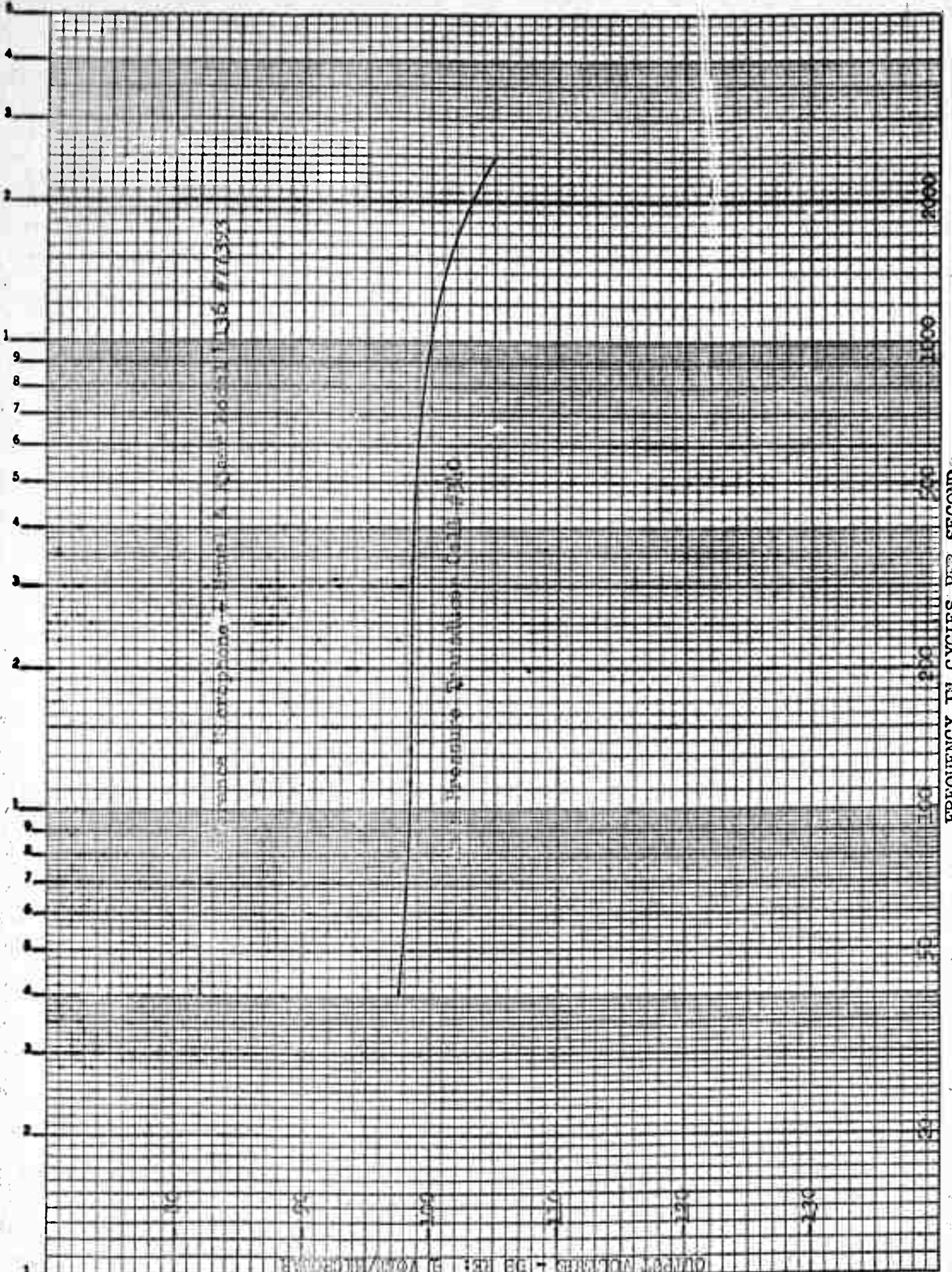


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CALC			REVISED	DATE	Figure 8(a). Response of Ames Pressure Transducer Conn. #339, with Carrier Amplifier.	T2-2648
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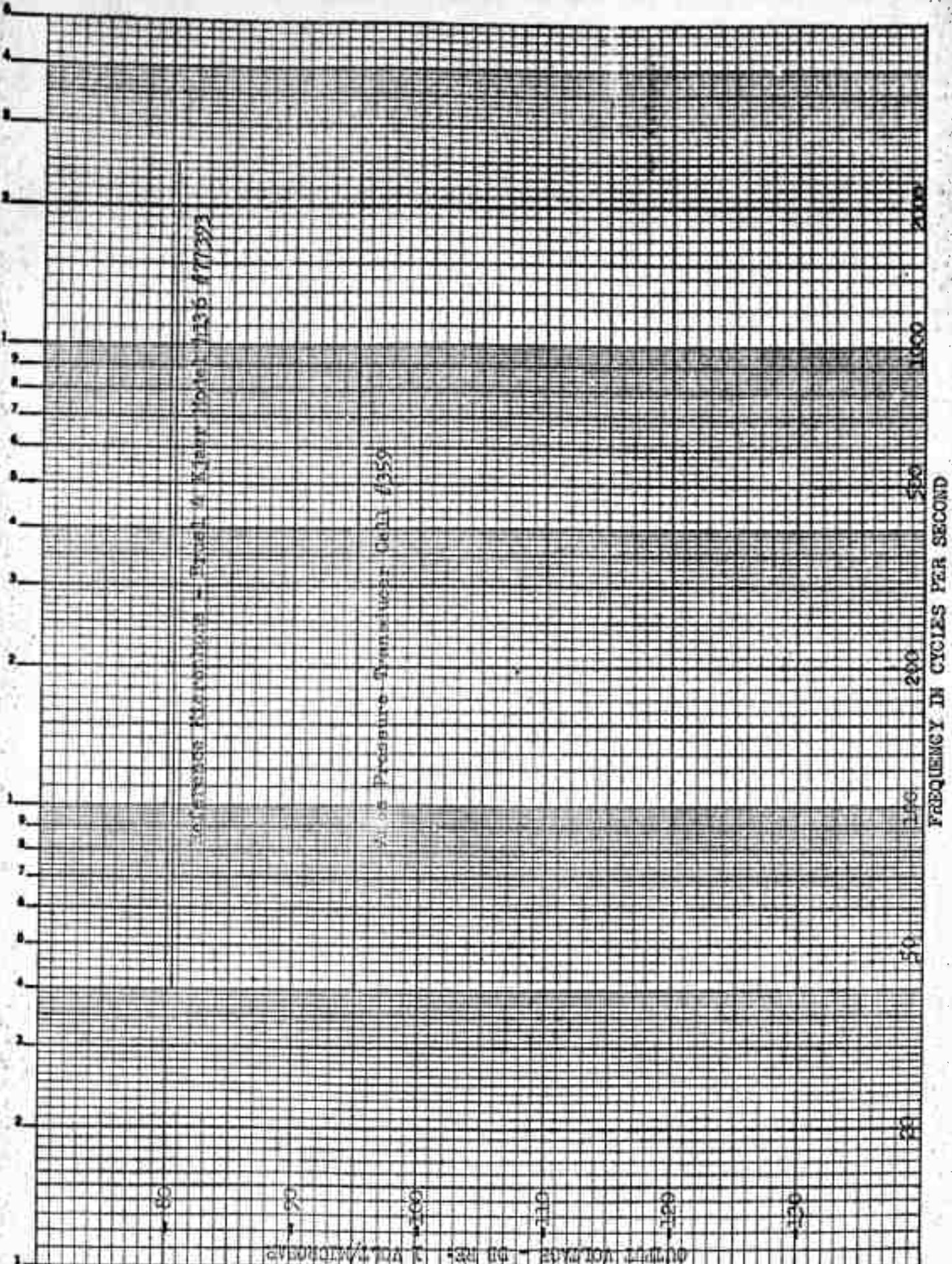
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Figure 8(b). Response of Ames Pressure Transducer Cell #340, with Carrier Amplifier.

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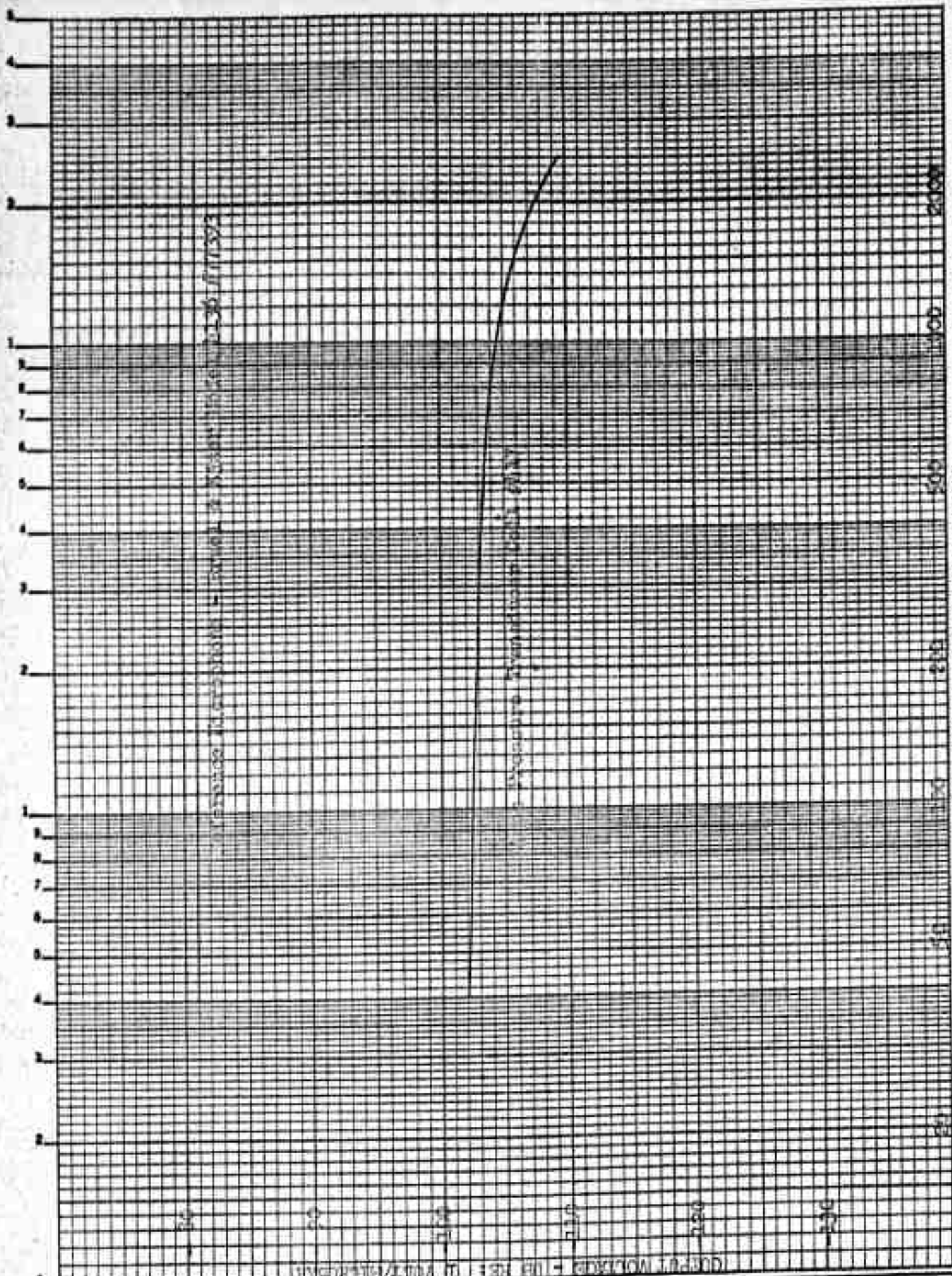
Figure 8(c). Response of Ames Pressure Transducer Cell #359, with Carrier Amplifier.

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Figure 8(d). Response of Ames Pressure Transducer Cell #417, with Carrier Amplifier.

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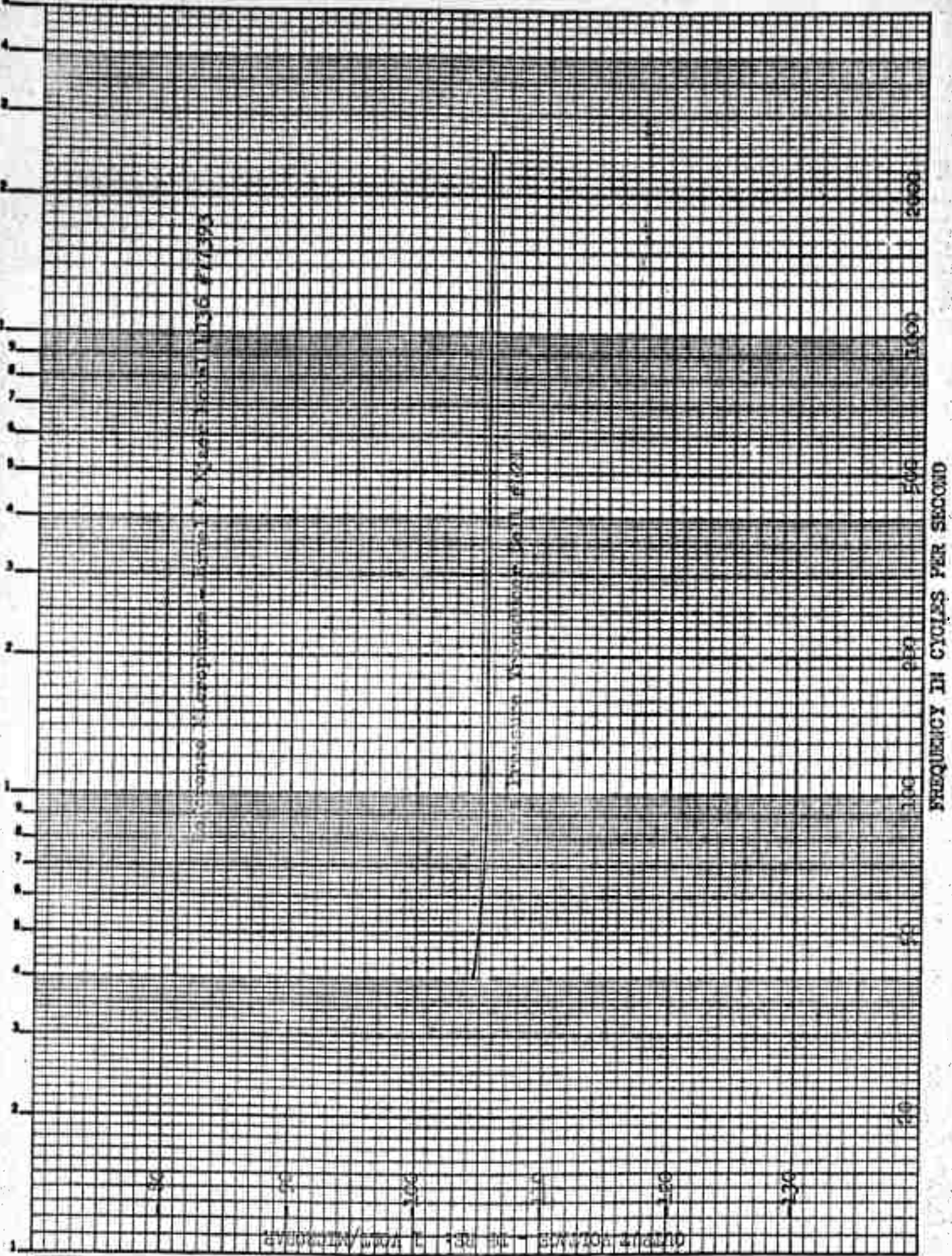
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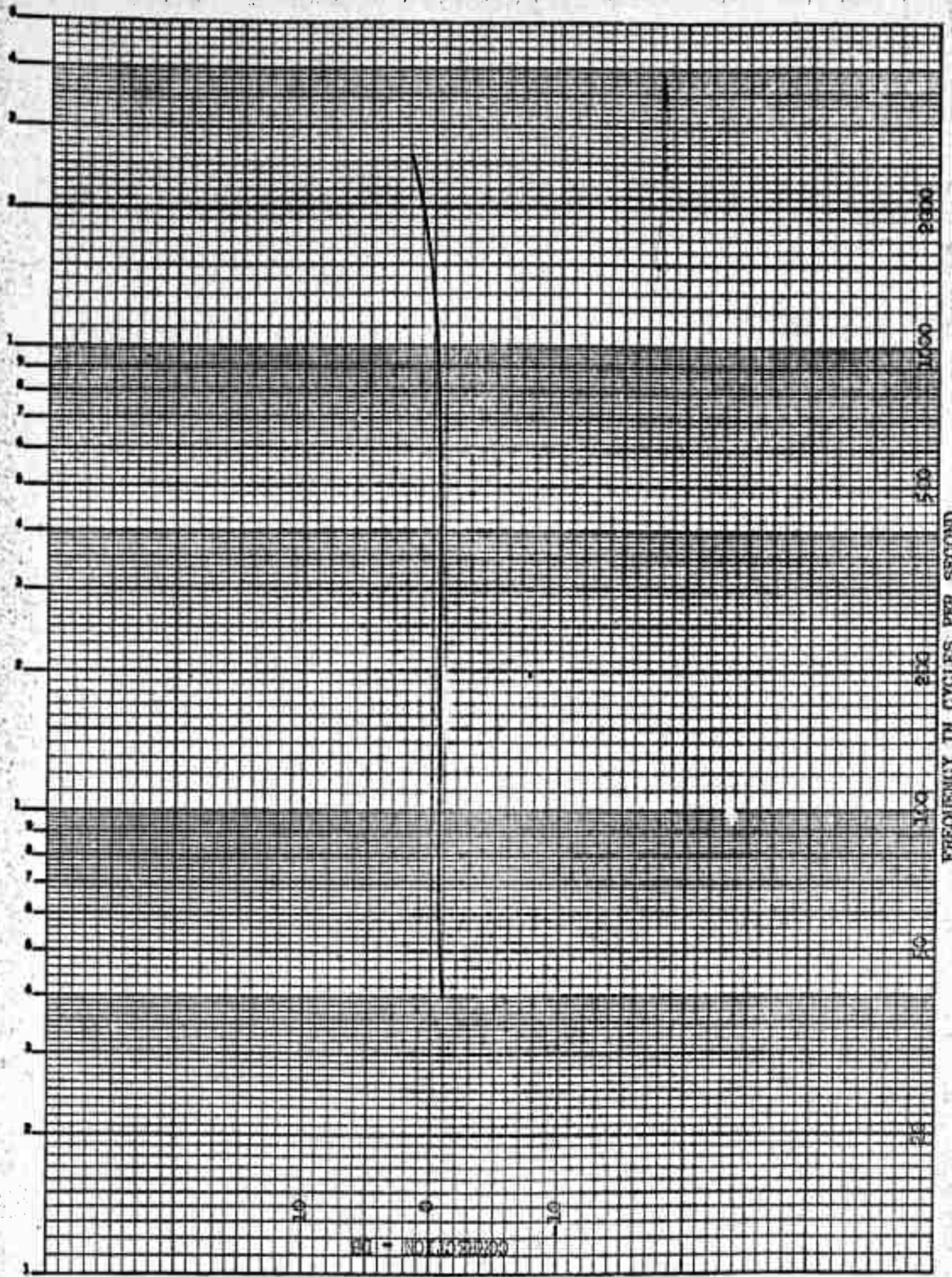


CALC			REVISED	DATE	Figure 8(e). Response of Ames Pressure Transducer Cell #421, with Carrier Amplifier. <b>BOEING AIRPLANE COMPANY</b>	c 5/5
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Figure 9. Average correction curve for dynamic pressure response of Ames pressure transducer cells.

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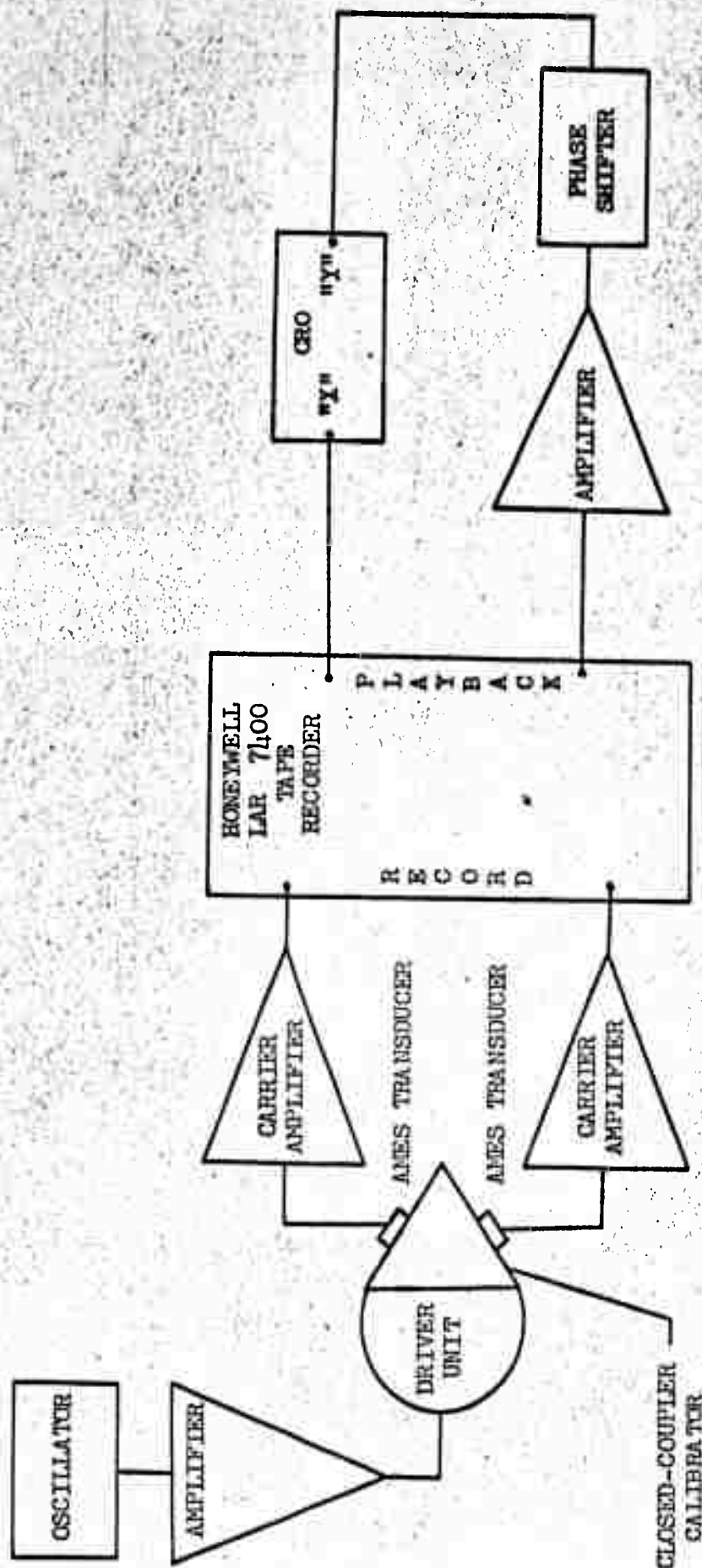
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CALC			REVISED.	DATE	Figure 10. Block diagram of equipment -- Phase check of transducer signals through LAR 7400 recorder.	
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CALC			REVISED.	DATE	Table I. Phase shift through transducer systems and LAR 7400 recorder.	
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REQ 10388

REFERENCE TRANSDUCER	Test Point Channel	4 14	4 14	4 14	14 20	14 20	27 7	27 7	27 7	27 7	27 7	27 7	32 15	32 15	32 15	32 15	32 15
TEST TRANSDUCER	Test Point Channel	5 10	8 16	9 8	12 22	16 6	26 3	28 25	29 27	29 27	21 23	30 13	31 17	34 19	35 9	36 21	
Frequency		Phase Angle *															
100 cps		0°	3°	0°	0°	4°	1°	4°	1°	4°	5°	4°	0°	0°	5°	0°	2°
500		1	2	0	0	18	2	18	2	18	18	14	0	16	15	0	16
800		2	0	0	0	24	2	25	2	25	22	23	0	20	20	0	23
1000		2	4	0	0	27	4	29	4	29	26	26	0	29	26	0	28
2000		1	3	0	2	37	3	41	3	41	31	35	4	34	39	7	34
2500		2	2	0	7	—	13	—	13	—	—	—	9	—	—	2	—
3000		3	3	0	6	—	14	—	14	—	—	—	13	—	—	2	—

\* Includes transducer and LAR 7400 recorder (See Figure 10).

ORIGINAL DATA TAPE  
HONEYWELL LAR 7400

TAPE COPY  
AMPEX CP 100

<u>CHANNEL</u>	<u>TEST POINT</u>	<u>CHANNEL</u>
1	1	-
2	20	-
3	26*	3
4	23	-
5	18	-
6	16*	6
7	27*	7
8	9*	8
9	35*	9
10	5*	10
11	33	-
12	6	-
13	30*	13
14	14*	4
15	32*	5
16	8*	6
17	31*	7
18	7	-
19	34*	1
20	14*	10
21	36*	11
22	12*	12
23	21*	3
24	15	-
25	28*	5
26	2	-
27	29*	9
28	3	-

\* Systems used for space correlation measurements.

CALC			REVISED.	DATE	Table II. Test point - data channel assignments.	
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				* Positive Angle of Attack, Nose Up				** Positive Angle of Yaw, Nose Right			
CALC	CHECK	APPD	APPD	CORRELATION NUMBER	MACH NUMBER	ANGLE OF* ATTACK ( $\alpha$ )	ANGLE OF** YAW ( $\beta$ )	CORRELATION NUMBER	MACH NUMBER	ANGLE OF* ATTACK ( $\alpha$ )	ANGLE OF** YAW ( $\beta$ )
				301	0.60	0	0	338	0.90	-4	0
				302	0.70	0	0	339	0.90	-2	0
				303	0.75	0	0	340	0.90	0	0
				304	0.80	0	0	341	0.90	2	0
				305	0.80	4	0	342	0.90	4	0
				306	0.80	-4	0	343	0.92	4	0
				307	0.85	-4	0	344	0.92	2	0
				308	0.85	0	0	345	0.92	0	0
				309	0.85	4	0	346	0.92	-2	0
				310	0.90	0	0	347	0.92	-4	0
				311	0.90	4	0	348	1.00	-4	0
				312	0.90	-4	0	349	1.00	0	0
				313	0.95	-4	0	350	1.00	4	0
				314	0.95	0	0	351	1.08	4	0
				315	0.95	4	0	352	1.08	0	0
				316	1.00	4	0	353	1.08	-4	0
				317	1.00	0	0	355	0.80	0	4
				318	1.00	-4	0	356	0.80	0	-4
				319	1.08	-4	0	357	0.85	0	-4
				320	1.08	0	0	358	0.85	0	4
				321	1.08	4	0	359	0.90	0	4
				323	0.825	4	0	360	0.90	0	-4
				324	0.825	4	2	361	0.95	0	-4
				325	0.825	0	0	362	0.95	0	4
				326	0.825	-2	0	363	1.00	0	4
				327	0.825	-4	0	364	1.00	0	-4
				328	0.86	-4	0	365	1.08	0	-4
				329	0.86	-2	0	366	1.08	0	-4
				330	0.86	0	0	367	0.825	0	4
				331	0.86	2	0	368	0.825	0	2
				332	0.86	4	0	369	0.825	0	0
				333	0.88	4	0	370	0.825	0	0
				334	0.88	2	0	371	0.825	0	-4
				335	0.88	0	0	372	0.875	0	-4
				336	0.88	-2	0	373	0.875	0	-2
				337	0.88	-4	0	374	0.875	0	0

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Table III. Test Conditions.  
a. Tunnel Mach number and model attitude.

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CORRELATION NUMBER		MACH NUMBER	ANGLE OF* ATTACK ( $\alpha$ )	ANGLE OF** YAW ( $\beta$ )	CORRELATION NUMBER		MACH NUMBER	ANGLE OF* ATTACK ( $\alpha$ )	ANGLE OF** YAW ( $\beta$ )
375	0.875	0	2	413	0.85	4	4		
376	0.875	0	4	414	0.90	4	4		
377	0.900	0	4	415	0.90	-4	4		
378	0.900	0	2	416	0.95	-4	4		
379	0.900	0	0	417	0.95	4	4		
380	0.900	0	-2	418	1.00	4	4		
381	0.900	0	-4	419	1.00	-4	4		
382	0.925	0	-4	420	1.08	-4	4		
383	0.925	0	-2	421	1.08	4	4		
384	0.925	0	0	423	0.825	4	4		
385	0.925	0	2	424	0.825	2	0		
386	0.925	0	4	425	0.825	0	-2		
387	0.975	0	4	426	0.825	-4	-4		
388	0.975	0	2	427	0.825	-4	-4		
389	0.975	0	0	428	0.85	-2	0		
390	0.975	0	-2	429	0.85	2	4		
391	0.975	0	-4	430	0.85	4	4		
392	0.900	0	-4	431	0.85	2	0		
393	0.900	0	-2	432	0.85	4	4		
394	0.900	0	2	433	0.875	4	2		
395	0.900	0	4	434	0.875	0	0		
397	0.80	4	-4	435	0.875	-4	-4		
398	0.80	-4	-4	436	0.875	-4	-4		
399	0.85	-4	-4	437	0.875	-4	-4		
400	0.85	4	-4	438	0.875	-4	-4		
401	0.90	4	-4	439	0.825	-4	-4		
402	0.90	-4	-4	440	0.825	-4	-2		
403	0.95	-4	-4	441	0.825	2	0		
404	0.95	4	-4	442	0.825	4	4		
405	1.00	4	-4	443	0.825	4	2		
406	1.00	-4	-4	444	0.85	4	4		
407	1.08	-4	-4	445	0.85	2	0		
408	1.08	4	-4	446	0.85	0	-2		
410	0.80	4	4	447	0.85	-2	-2		
411	0.80	-4	4	448	0.85	-4	-2		
412	0.85	-4	-4	449	0.875	-4	-4		

Table III. Test Conditions.  
a. Tunnel Mach number and model attitude. (continued)

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\* Positive Angle of Attack, Nose Up - \*\* Positive Angle of Yaw, Nose Right

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				CORRELATION NUMBER	MACH NUMBER	ANGLE OF ATTACK ( $\alpha$ )	ANGLE OF** YAW ( $\beta$ )
				187	0.98	0	+
				188	0.98	+	+
				189	1.00	+	+
				190	1.00	+	+
				191	1.00	+	+
				192	1.02	+	+
				193	1.02	+	+
				194	1.02	+	+
				195	1.04	+	+
				196	1.04	+	+
				197	1.04	+	+
				198	1.06	+	+
				199	1.06	+	+
				500	1.06	+	+
				501	1.08	+	+
				502	1.08	+	+
				503	1.08	+	+
				504	0.725	+	+
				505	0.70	+	+
				506	0.65	+	+
				507	0.70	+	+
				508	0.75	+	+
				150	0.875	2	-
				151	0.875	2	-
				152	0.875	2	-
				153	0.875	2	-
				155	0.75	+	+
				156	0.75	+	+
				157	0.75	+	+
				158	0.80	+	+
				159	0.80	+	+
				160	0.80	+	+
				161	0.80	+	+
				162	0.82	+	+
				163	0.82	+	+
				164	0.82	+	+
				165	0.84	+	+
				166	0.84	+	+
				167	0.84	+	+
				168	0.86	+	+
				169	0.86	+	+
				170	0.86	+	+
				171	0.88	+	+
				172	0.88	+	+
				173	0.88	+	+
				174	0.90	+	+
				175	0.90	+	+
				176	0.90	+	+
				177	0.92	+	+
				178	0.92	+	+
				179	0.92	+	+
				180	0.94	+	+
				181	0.94	+	+
				182	0.94	+	+
				183	0.96	+	+
				184	0.96	+	+
				185	0.96	+	+
				186	0.98	+	+

\* Positive Angle of Attack, Nose Up - \*\* Positive Angle of Yaw, Nose Right

CALC			REVISED	DATE	Table III. Test Conditions. a. (Concluded)	
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M	TOTAL PRESSURE (PSF)	STATIC PRESSURE (PSF)	DYNAMIC PRESSURE (PSF)	REYNOLDS NUMBER PER FT. 10 <sup>6</sup>	TOTAL TEMPERATURE °R	STATIC TEMPERATURE °R	VELOCITY FPS
0.60	2100	1653	415.0	3.37	546.8	510.2	663
0.65		1583	467.6	3.31	581.4	536.2	737
0.70		1518	520.7	3.65	558.9	509.0	787
0.725		1479	545.6	3.41	598.1	541.1	878
0.75		1456	566.3	3.73	565.5	509.3	824
0.8		1381	618.8	3.8	574.1	508.9	885
0.825		1333	635.8	3.83	573.1	504.4	909
0.85		1314	662.7	3.89	575.8	503.3	933
0.86		1297	671.8	3.83	584.5	509.1	951
0.875		1272	683.6	3.87	582.2	504.7	965
0.88		1269	688.3	3.86	585.7	507.1	971
0.9		1246	704.1	3.89	586.2	504.7	990
0.92		1271	719.1	3.93	584.6	499.6	1007
0.925		1209	720.1	3.91	586.8	501.4	1013
0.94		1189	735.0	4.06	572.9	486.9	1016
0.95		1178	742.6	3.88	595.6	504.7	1045
0.96		1164	748.9	4.01	581.6	491.3	1041
0.975		1134	753.0	3.83	601.1	505.2	1073
1.0		1113	780.0	3.94	596.8	497.3	1094
1.02		1088	778.0	4.0	590.6	489.4	1103
1.04		1061	800.0	3.99	594.2	488.9	1125
1.06		1037	811.2	3.93	602.4	492.4	1150
1.08		1002	818.3	3.95	599.0	485.8	1167

CALC			REVISED	DATE	Table III. Test Conditions. b. TUNNEL TEST SECTION PARAMETERS (NOMINAL VALUES)	T2-2648
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TAPE RECORDER TRACK NUMBER  
(Data Channel)

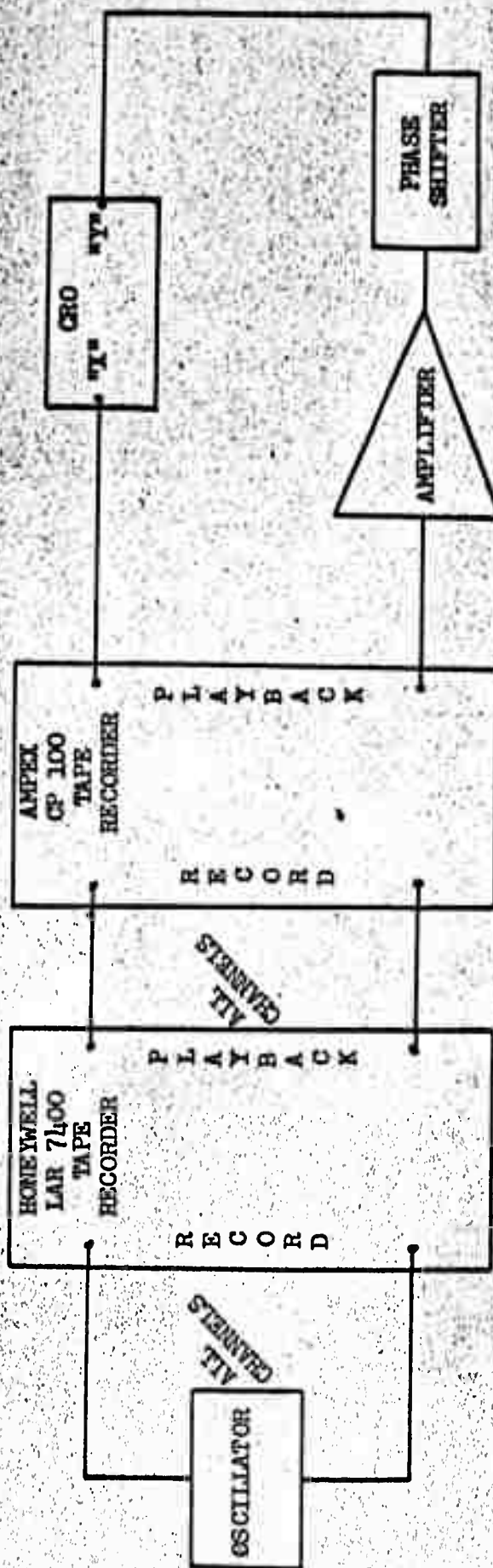
DB CORRECTION\*

1	4.0 db
2	3.0
3	2.0
4	1.0
5	2.5
6	0.5
7	1.5
8	3.0
9	1.0
10	1.5
11	1.0
12	1.5
13	1.5
14	1.0
15	3.0
16	1.5
17	2.0
18	2.0
19	3.0
20	4.0
21	3.0
22	2.0
23	3.0
24	4.0
25	1.0
26	2.0
27	1.5
28	2.0

\*Correction for tape recorder response at 2500 cps to be added to sound pressure level values for the one-third octave band centered on 2500 cps only.

CAIC			REVISED	DATE	Table IV. Correction for Honeywell LAR 7400 Tape Recorder Response at 2500 cps.	
CHECK						
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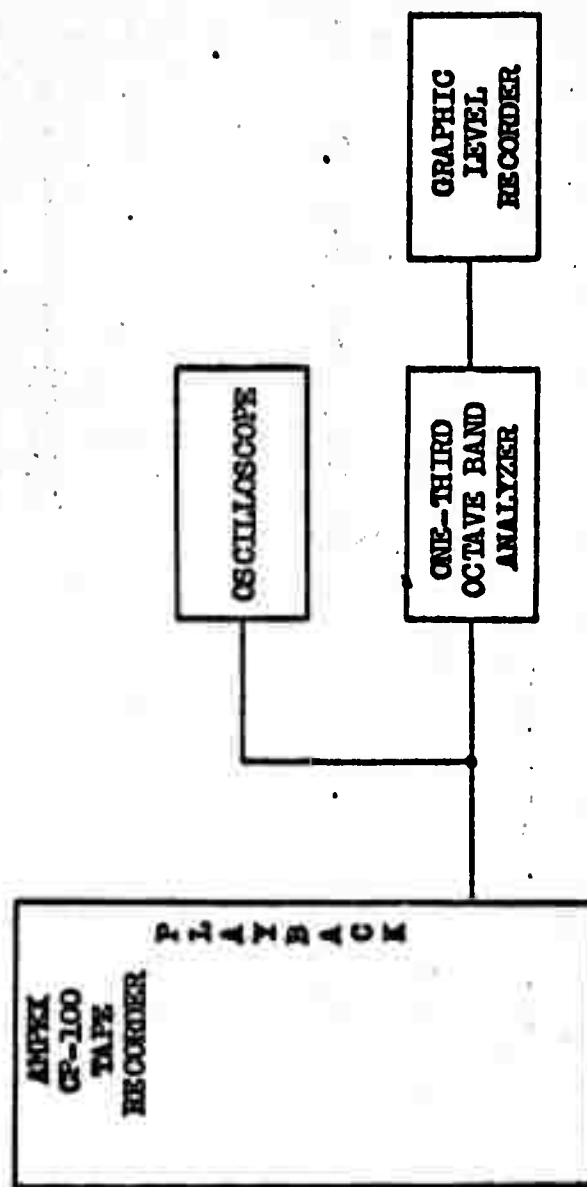


CALC			REVISED.	DATE	Figure 11. Block diagram of equipment -- Determination of phase difference through tape recorders.	
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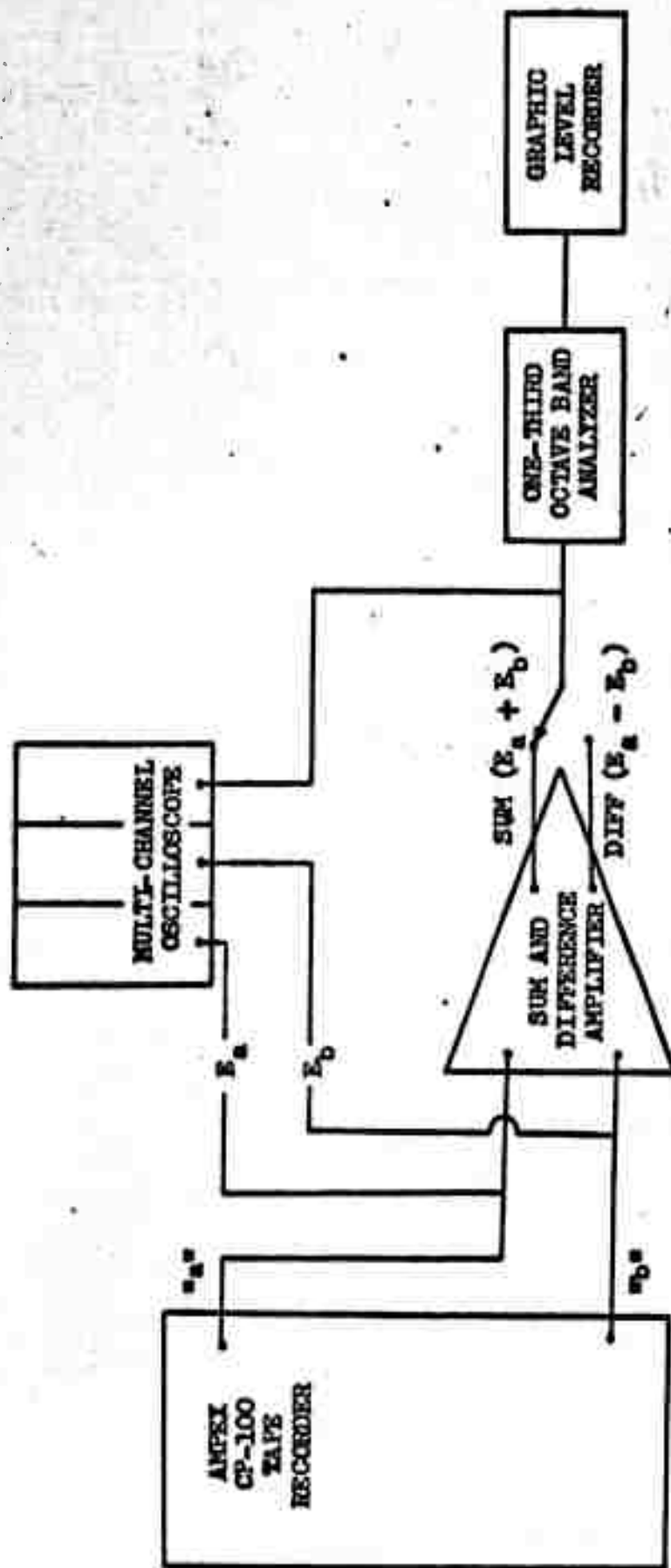
REFERENCE TRACK	IAR 7400	14	14	14	20	7	7	7	15	15	15	15	15
	CP 100	4	4	4	10	7	7	7	5	5	5	5	5
TEST TRACK	IAR 7400	10	16	8	22	6	25	27	13	17	19	9	21
	CP 100	10	6	8	12	6	5	9	13	7	1	9	11
Frequency							Phase	Angle					
100 cps	3.5°	2.5°	8.0°	0°	2.5°	0°	0°	0°	2.5°	0°	2.5°	2.5°	2.5°
500 cps	10.0	3.5	9.0	2.5	9.0	6.5	5.5	4.5	9.0	0	5.5	6.5	8.0
800 cps	16.0	5.5	13.0	7.0	14.5	9.5	9.0	8.0	16.0	3.5	8.5	9.5	13.0
1000 cps	19.5	7.5	15.5	9.0	19.0	11.5	10.0	10.5	17.0	4.5	9.0	10.5	17.5
2000 cps	46.5	19.0	38.0	18.0	36.5	20.0	21.5	18.0	35.5	13.0	19.0	24.0	34.0
2500 cps	57.5	26.5	48.0	21.5	49.5	24.5	27.5	24.5	41.5	16.0	23.5	29.0	40.0
3000 cps	62.0	29.5	—	28.5	58.5	27.0	26.0	37.0	44.0	21.0	23.5	36.5	55.0

CALC			REVISED.	DATE	Table V. Phase shift through tape recorders.	
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CALC			REVISED.	DATE	Figure 12. Block Diagram of Data Reduction Equipment - One-Third Octave Band Sound Pressure Level Data. T2-2648	
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CALC			REVISED	DATE	Figure 13. Block diagram of data reduction equipment -- Space correlation data.	
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Table VI. ACOUSTIC TEST RESULTS - SOUND PRESSURE LEVELS

Transducer Location: #2

Correlation No.	305	309	311	323	461	467	468	473	474	485	486
Mach No.	0.8	0.85	0.9	0.825	0.8	0.84	0.86	0.88	0.9	0.96	0.98
$\alpha$	+4	+4	+4	+4	+4	+4	+4	+4	+4	+4	+4
$\beta$	0	0	0	0	-4	-4	-4	-4	-4	-4	-4
ONE THIRD OCTAVE BAND SOUND PRESSURE LEVELS - DB RE: .0002 MICROBAR											
40	125.0	133.5	131.0	129.0	127.0	133.0	137.5	137.0	131.0	120.5	120.0
50	126.0	135.5	133.0	130.5	129.0	134.5	138.5	138.0	132.0	120.5	120.5
63	127.5	135.0	133.5	131.0	129.5	134.5	138.5	138.0	133.5	122.5	122.0
80	128.0	136.5	134.5	132.5	131.0	136.0	140.5	140.0	135.0	119.0	119.5
100	128.5	137.0	135.5	133.5	131.0	136.0	141.0	140.0	135.5	118.5	118.0
125	129.0	138.0	136.5	134.0	131.5	137.0	142.0	139.5	136.5	118.5	118.0
160	131.5	140.0	137.5	136.5	133.0	138.0	142.0	139.0	137.0	118.0	118.0
200	135.0	143.5	138.5	140.5	136.0	141.0	143.5	139.5	138.0	118.0	119.0
250	133.0	141.5	139.0	137.5	134.0	139.0	142.0	140.0	139.0	118.0	119.0
320	135.0	141.0	140.5	137.0	134.5	139.0	142.5	141.0	140.5	119.0	119.5
400	134.5	141.0	141.5	136.5	135.0	139.0	142.5	141.5	141.5	120.5	120.5
500	135.0	141.0	142.0	137.0	135.5	138.5	143.0	142.5	142.5	122.5	122.0
630	135.5	142.0	142.5	138.0	135.5	139.5	143.5	144.0	144.5	122.5	122.5
800	137.0	143.0	143.5	139.0	137.0	141.0	145.0	146.0	147.0	125.0	125.5
1000	138.5	143.0	143.0	140.0	138.0	141.5	145.0	147.0	147.0	128.5	128.0
1250	141.0	143.5	143.5	141.5	141.0	142.5	145.5	147.0	146.5	132.0	132.0
1600	144.5	145.0	145.0	144.5	145.0	145.0	146.0	146.5	146.5	137.5	137.0
2000	147.5	147.5	147.0	147.0	148.5	149.5	149.0	148.5	148.5	144.5	144.5
2500	149.0	150.0	150.5	149.0	149.0	151.5	150.5	150.0	150.0	140.5	141.0
Z SPL	153.5	156.0	155.5	154.0	153.5	156.0	157.0	156.5	156.5	147.0	146.5

ONE THIRD OCTAVE BAND CENTER FREQUENCY

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Table VI. ACOUSTIC TEST RESULTS - SOUND PRESSURE LEVELS (continued)

Transducer Location: #7

Correlation No.	490	495	496	501	502
Mach No.	1.0	1.04	1.04	1.08	1.08
$\alpha$	0	-4	0	-4	0
$\beta$	-4	-4	-4	-4	-4

ONE THIRD OCTAVE BAND SOUND PRESSURE LEVELS - DB RE: .0002 MICROBAR

	40	50	63	80	100	125	160	200	250	320	400	500	630	800	1000	1250	1600	2000	2500	XSPL
ONE THIRD OCTAVE BAND CENTER FREQUENCY	113.0	113.0	114.5	114.0	114.5	115.0	116.0	117.0	118.5	119.5	120.5	122.0	123.5	126.0	128.0	130.5	135.5	144.5	142.0	146.5
	119.5	120.0	119.5	121.0	121.5	122.0	123.0	124.0	125.0	126.5	127.5	128.0	129.0	131.0	133.0	136.5	142.0	152.5	152.0	155.0
	113.0	113.5	114.0	114.0	114.5	116.0	116.0	117.5	119.0	120.0	121.0	122.5	123.5	127.0	128.5	131.0	135.5	146.0	142.5	148.0
	121.0	121.5	122.5	124.5	125.5	126.5	127.0	127.5	128.5	129.5	130.5	131.5	132.5	134.5	137.0	141.5	150.5	160.0	150.5	160.5
	114.0	114.5	115.0	116.0	116.0	117.0	117.5	118.5	120.0	121.0	122.5	123.5	125.5	127.5	129.0	131.5	135.5	146.0	145.0	149.0

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**Transducer Location: #8**

Correlation No.	L57	L65	L72
Mach No.	0.75	0.84	0.88
$\alpha$	-4	-4	-4
$\theta$	-4	-4	-4

	L57	L65	L72
40	120.5	136.5	130.5
50	122.5	139.5	132.5
63	125.0	140.5	133.0
80	123.5	141.0	133.5
100	123.5	142.0	134.0
125	121.5	143.0	135.0
160	128.0	146.5	136.5
200	131.0	151.5	139.0
250	131.0	149.0	137.0
320	133.5	149.5	136.0
400	133.5	148.0	135.5
500	134.0	144.0	134.5
630	135.0	140.5	132.5
800	137.0	140.0	134.0
1000	138.0	140.0	135.0
1250	142.0	142.5	137.5
1600	146.0	147.0	140.5
2000	145.0	147.5	142.5
2500	141.0	142.5	138.0
$\Sigma$ SPL	151.0	158.5	149.5

ONE THIRD OCTAVE BAND CENTER FREQUENCY



Table VI. ACOUSTIC TEST RESULTS - SOUND PRESSURE LEVELS (continued)

Transducer Location: #12									
Correlation No.	489	495	500	501					
Mach No.	1.0	1.04	1.06	1.08					
$\alpha$	-4	-4	-4	-4					
$\beta$	-4	-4	-4	-4					
ONE THIRD OCTAVE BAND SOUND PRESSURE LEVELS - DB RE: .0002 MICROBAR									
40	112.0	113.5	118.5	114.0					
50	112.0	114.0	119.5	115.0					
63	113.0	114.0	120.5	116.0					
80	114.0	115.5	121.5	117.0					
100	114.5	116.5	122.0	117.0					
125	117.0	117.5	123.0	119.5					
160	116.0	117.5	123.0	119.5					
200	118.5	118.5	124.0	120.0					
250	120.0	120.0	124.5	121.5					
320	120.5	121.5	124.5	122.5					
400	122.0	123.0	125.5	123.5					
500	123.5	124.0	126.0	124.5					
630	125.0	124.5	126.0	125.5					
800	127.5	127.0	127.5	128.0					
1000	130.5	129.5	129.5	131.0					
1250	134.0	133.0	132.5	134.5					
1600	139.0	138.0	138.5	143.0					
2000	148.0	147.5	150.5	152.0					
2500	147.0	146.5	144.0	142.5					
$\Sigma$ SPL	151.0	150.5	151.5	153.0					



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Table VI. ACOUSTIC TEST RESULTS - SOUND PRESSURE LEVELS (continued)

Transducer Location: #14

Correlation No.	L04	L08	L91	L92	L97
Mach No.	0.95	1.08	1.0	1.02	1.04
$\alpha$	+4	+4	+4	+4	+4
$\beta$	-4	-4	-4	-4	-4

ONE THIRD OCTAVE BAND SOUND PRESSURE LEVELS - DB RE: .0002 MICROBAR

	40	50	63	80	100	125	160	200	250	320	400	500	630	800	1000	1250	1600	2000	2500	ESPL
	117.0	118.0	118.0	119.0	120.5	121.0	122.0	123.5	124.5	125.0	126.0	127.0	128.0	130.0	132.0	133.5	137.0	141.5	142.5	146.5
	121.0	125.0	125.0	127.5	128.5	129.0	130.5	131.5	133.0	134.0	135.0	136.0	137.0	138.0	139.0	140.5	142.0	146.0	148.0	152.5
	119.0	119.5	120.0	121.5	122.5	123.5	124.5	126.0	127.0	128.0	129.0	130.0	131.0	132.5	133.5	134.5	136.5	143.5	145.0	149.0
	120.0	121.0	121.0	123.0	124.0	125.5	126.0	127.5	128.5	129.5	130.5	131.0	132.0	133.0	134.0	135.0	137.0	140.0	145.0	150.0

ONE THIRD OCTAVE BAND CENTER FREQUENCY

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**Transducer Location: #16**

ONE THIRD OCTAVE BAND CENTER FREQUENCY



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Table VI. ACOUSTIC TEST RESULTS - SOUND PRESSURE LEVELS (continued)

Transducer Location: #20									
Correlation No.	356	357	360	361					
Mach No.	0.8	0.85	0.9	0.95					
$\alpha$	0	0	0	0					
$\delta$	-1	-1	-1	-1					
ONE THIRD OCTAVE BAND SOUND PRESSURE LEVELS - DB RE: .0002 MICROBAR									
40	122.0	124.0	125.0	121.0					
50	123.5	124.5	127.0	122.5					
63	123.5	125.0	127.0	122.5					
80	124.5	127.0	129.0	124.0					
100	125.5	128.0	130.0	125.0					
125	126.0	129.0	131.0	126.0					
160	127.0	129.5	131.5	126.5					
200	129.5	131.0	132.5	128.0					
250	129.0	131.5	133.5	128.5					
320	130.5	132.5	135.0	129.5					
400	131.0	133.5	136.0	130.5					
500	131.0	134.0	136.5	131.0					
630	131.5	134.5	137.0	131.5					
800	132.5	136.0	138.0	133.0					
1000	133.0	136.5	138.5	133.5					
1250	134.0	137.5	139.5	134.0					
1600	134.5	138.0	141.0	135.0					
2000	135.5	138.5	142.5	137.5					
2500	136.0	138.5	141.5	137.5					
$\Sigma$ SPL	144.0	147.0	149.5	145.0					

Table VI.

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Correlation No.	356	357	360	361
Mach No.	0.8	0.85	0.9	0.95
$\alpha$	0	0	0	0
$\beta$	+	+	+	+

ONE THIRD OCTAVE BAND SOUND PRESSURE LEVELS - DB RE: .0002 MICROBAR

40	118.5	121.0	121.5	117.5
50	120.5	123.5	123.5	119.0
63	120.5	122.5	124.0	119.5
80	121.5	124.0	125.0	120.5
100	122.5	124.5	125.5	121.5
125	123.5	126.0	127.0	122.5
160	125.5	127.0	128.0	123.0
200	129.0	129.5	129.0	124.5
250	128.5	129.5	130.0	125.5
320	129.5	130.0	131.5	126.5
400	129.5	130.5	132.5	128.5
500	129.5	131.5	133.5	129.5
630	130.0	133.5	134.5	131.0
800	131.5	134.5	136.0	133.0
1000	133.0	135.5	137.5	134.0
1250	136.0	138.0	139.5	135.5
1600	137.5	138.5	142.0	138.0
2000	141.0	143.0	145.0	142.5
2500	149.0	144.5	147.0	144.0
3500	150.5	149.0	151.0	148.0

ONE THIRD OCTAVE BAND CENTER FREQUENCY

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Table VI. ACOUSTIC TEST RESULTS - SOUND PRESSURE LEVELS (continued)

Transducer Location: #27

Correlation No.	458	460	461	462	463	464	465	466	467	468	469	470	471
Mach No.	0.8	0.8	0.8	0.82	0.82	0.82	0.84	0.84	0.84	0.86	0.86	0.86	0.88
$\alpha$	-4	0	+4	+4	0	-4	-4	0	+4	+4	0	-4	-4
$\beta$	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4
ONE THIRD OCTAVE BAND SOUND PRESSURE LEVELS - DB RE: .0002 MICROBAR													
40	141.5	138.5	138.0	139.5	140.0	140.0	141.0	141.0	140.5	143.0	141.0	143.5	144.0
50	142.5	140.5	140.5	141.0	140.5	142.0	142.5	141.5	142.5	144.5	142.0	145.5	145.5
63	141.5	141.0	140.5	141.5	141.0	142.0	143.5	142.5	143.5	144.0	143.0	145.5	145.5
80	143.0	142.0	142.0	143.0	142.5	144.0	144.5	144.0	144.5	146.0	145.0	147.0	147.0
100	144.5	143.5	143.5	144.5	144.5	145.0	145.5	145.0	146.0	147.5	146.0	148.5	148.5
125	145.5	144.5	144.0	145.0	145.0	146.5	147.0	146.0	146.5	148.0	147.0	149.0	149.5
160	146.5	145.0	145.5	146.0	146.0	147.0	148.0	147.0	148.0	149.5	148.0	150.0	150.0
200	148.0	147.0	147.0	148.0	148.0	149.0	149.5	148.5	149.0	150.5	149.0	151.0	151.5
250	149.0	148.5	148.5	149.0	149.0	150.0	151.0	149.5	150.5	151.0	150.0	152.0	152.0
320	150.5	150.5	151.0	151.5	151.0	151.5	152.5	151.5	152.5	153.0	152.0	153.5	153.5
400	151.5	151.5	152.5	152.5	152.5	153.0	153.5	152.5	153.0	153.5	153.0	153.5	153.0
500	152.0	152.0	153.0	153.0	153.0	152.5	153.0	153.0	153.0	153.5	153.0	153.0	153.5
630	152.0	152.0	152.0	152.0	152.5	151.5	151.5	152.0	151.5	151.5	151.5	151.0	151.5
800	151.5	150.5	150.0	150.0	150.5	149.5	150.0	150.0	149.5	149.5	149.5	150.0	149.5
1000	147.0	148.0	147.5	147.5	147.5	147.0	148.0	147.5	147.5	147.5	147.5	148.0	147.5
1250	145.5	146.5	146.5	146.5	146.5	146.0	146.5	146.5	146.5	147.0	147.0	146.5	146.0
1600	145.0	146.0	146.0	146.0	146.0	145.0	145.5	146.0	146.0	146.0	146.0	145.5	145.5
2000	145.0	145.5	145.0	145.0	145.5	145.5	146.0	145.5	145.5	145.5	145.5	145.5	145.5
2500	143.5	144.0	144.0	144.0	144.0	145.0	145.0	144.5	144.5	144.0	144.0	144.5	144.0
$\Sigma$ SPL	161.0	161.0	161.0	161.0	161.0	161.0	162.0	162.0	162.0	162.0	162.0	162.5	163.0

ONE THIRD OCTAVE BAND CENTER FREQUENCY

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## Table VI.

**Transducer Location: #27**

Correlation No.	l72	l73	l76	l82
Mach No.	0.88	0.88	0.9	0.94
$\infty$	0	+4	-4	-4
$\theta$	-4	-4	-4	-4

ONE THIRD OCTAVE BAND SOUND: PRESSURE LEVELS - DB RE: .0002 MICROBAR									
40	143.0	145.5	148.5	126.0					
50	144.5	147.5	149.5	127.0					
63	145.0	147.0	150.0	128.5					
80	146.5	148.0	151.0	129.0					
100	148.0	149.0	151.5	130.0					
125	149.0	150.0	152.0	130.5					
160	149.5	150.5	152.0	132.0					
200	150.5	151.0	152.5	132.5					
250	151.5	152.0	153.5	133.5					
320	153.0	153.0	154.5	135.0					
400	154.0	153.5	154.5	136.0					
500	153.5	153.0	153.5	136.5					
630	151.5	151.0	151.0	136.5					
800	149.5	149.0	149.0	137.5					
1000	147.5	147.5	148.0	138.0					
1250	146.5	146.5	146.5	138.5					
1600	145.5	145.5	145.5	139.0					
2000	145.0	145.0	145.5	139.5					
2500	144.0	143.5	144.0	141.0					
$\Sigma$ SPL	162.5	163.0	164.0	149.0					

ONE THIRD OCTAVE BAND CENTER FREQUENCY									
40	1000	1250	1600	2000	2500	3150	4000	5000	6300

ONE THIRD OCTAVE BAND CENTER FREQUENCY

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Transducer Location: #28

ONE THIRD OCTAVE BAND CENTER FREQUENCY





**Transducer Location: #30**

Correlation No.	356	357	362	363	364	365
Mach No.	0.8	0.85	0.95	1.0	1.0	1.08
$\alpha$	0	0	0	0	0	0
$\theta$	-4	-4	+4	+4	-4	-4
ONE THIRD OCTAVE BAND SOUND PRESSURE LEVELS - DB RE: .0002 MICROBAR						
40	120.5	121.0	111.0	111.0	111.0	137.5
50	122.0	126.0	112.5	112.0	112.5	138.0
63	123.0	127.5	113.0	113.0	113.0	139.0
80	123.5	128.0	115.0	115.0	115.0	140.5
100	124.5	130.5	116.5	116.5	116.0	141.0
125	128.0	134.0	116.5	116.5	116.5	142.0
160	131.0	136.0	117.0	117.5	117.0	142.0
200	135.5	139.5	118.0	118.0	117.5	143.0
250	137.0	138.0	118.5	118.5	118.0	143.0
320	138.5	137.0	119.0	118.5	118.5	144.0
400	138.5	135.5	119.0	118.5	118.0	144.0
500	137.0	135.0	118.5	118.5	117.5	143.5
630	136.0	135.5	119.5	119.5	118.0	144.0
800	136.0	137.0	119.5	119.0	118.0	144.5
1000	138.0	139.5	118.5	118.5	117.0	144.0
1250	142.0	141.5	118.0	118.0	116.5	144.5
1600	145.0	149.0	118.0	118.0	117.0	145.5
2000	144.0	147.5	118.0	118.0	117.0	145.5
2500	140.5	143.5	149.0	150.5	147.5	145.0
$\Sigma$ SPL	151.0	154.0	160.5	160.5	160.0	155.5

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**Transducer Location: #31**

Correlation No.	l67	l68	l73	l80
Mach No.	0.84	0.86	0.88	0.94
$\alpha$	+4	+4	+4	+4
$\theta$	-4	-4	-4	-4

ONE THIRD OCTAVE BAND SOUND PRESSURE LEVELS - DB RE: .0002 MICROBAR				
	l67	l68	l73	l80
40	131.0	134.5	134.5	119.5
50	132.0	136.0	136.0	121.0
63	133.5	136.5	136.5	121.0
80	135.0	137.5	138.0	121.5
100	136.0	138.5	138.0	122.0
125	137.0	139.5	139.0	123.0
160	138.0	140.0	139.5	124.0
200	140.0	141.5	140.5	125.5
250	140.0	142.0	141.0	127.0
320	141.5	143.0	142.0	128.0
400	141.5	143.5	142.0	128.5
500	141.0	142.5	142.0	128.5
630	140.0	142.0	141.5	129.5
800	139.5	142.0	141.5	131.0
1000	139.5	142.0	142.0	131.5
1250	141.0	143.0	143.0	133.5
1600	145.0	146.0	146.0	136.0
2000	147.5	148.0	149.5	141.0
2500	145.5	147.5	147.5	139.5
$\Sigma$ SPL	154.0	155.5	156.0	145.5

ONE THIRD OCTAVE BAND CENTER FREQUENCY				
	l67	l68	l73	l80
40	131.0	134.5	134.5	119.5
50	132.0	136.0	136.0	121.0
63	133.5	136.5	136.5	121.0
80	135.0	137.5	138.0	121.5
100	136.0	138.5	138.0	122.0
125	137.0	139.5	139.0	123.0
160	138.0	140.0	139.5	124.0
200	140.0	141.5	140.5	125.5
250	140.0	142.0	141.0	127.0
320	141.5	143.0	142.0	128.0
400	141.5	143.5	142.0	128.5
500	141.0	142.5	142.0	128.5
630	140.0	142.0	141.5	129.5
800	139.5	142.0	141.5	131.0
1000	139.5	142.0	142.0	131.5
1250	141.0	143.0	143.0	133.5
1600	145.0	146.0	146.0	136.0
2000	147.5	148.0	149.5	141.0
2500	145.5	147.5	147.5	139.5
$\Sigma$ SPL	154.0	155.5	156.0	145.5

ONE THIRD OCTAVE BAND CENTER FREQUENCY

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**Transducer Location: #32**

Correlation No.	356	357	358	406
Mach No.	0.8	0.85	0.85	1.0
$\alpha$	0	0	0	-4
$\beta$	-4	-4	+4	-4

ONE THIRD OCTAVE BAND SOUND PRESSURE LEVELS - DB RE: .0002 MICROBAR				
40	137.5	137.0	136.5	129.0
50	139.5	138.5	138.5	130.5
63	140.0	139.5	138.0	131.5
80	141.0	140.5	139.5	130.5
100	142.0	141.5	140.5	132.0
125	143.0	142.5	141.5	132.5
160	144.5	143.0	142.5	133.5
200	146.0	145.0	144.0	134.0
250	146.0	145.5	144.5	134.5
320	147.5	147.0	146.0	136.5
400	148.5	148.5	147.0	136.5
500	149.5	149.5	148.0	139.5
630	149.5	149.5	148.5	137.5
800	150.0	150.0	149.0	138.5
1000	149.0	149.5	148.5	139.5
1250	149.5	150.0	148.5	140.0
1600	150.0	150.5	148.5	141.0
2000	150.5	151.5	148.5	141.0
2500	153.0	153.5	150.0	145.5
FSPL	160.5	161.0	159.5	151.5

ONE THIRD OCTAVE BAND CENTER FREQUENCY
40
50
63
80
100
125
160
200
250
320
400
500
630
800
1000
1250
1600
2000
2500
FSPL

ONE THIRD OCTAVE BAND CENTER FREQUENCY

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Table VI. ACOUSTIC TEST RESULTS - SOUND PRESSURE LEVELS (continued)

Transducer Location: #33

Correlation No.	161	167	168	169	170	171	172	173	174	175	176	180
Mach No.	0.8	0.81	0.86	0.86	0.86	0.88	0.88	0.88	0.9	0.9	0.9	0.94
$\alpha$	+	+	+	0	-	-	0	+	+	0	-	+
$\beta$	-	-	-	-	-	-	-	-	-	-	-	-

ONE THIRD OCTAVE BAND SOUND PRESSURE LEVELS - DB RE: .0002 MICROBAR

40	131.5	110.5	113.0	139.0	140.5	140.0	139.5	142.0	141.0	139.5	139.0	125.5
50	133.5	112.0	114.5	141.0	141.5	142.0	141.5	143.5	143.5	141.0	140.5	125.5
63	134.0	112.5	115.0	141.0	142.0	142.0	141.5	144.0	143.0	141.5	140.5	126.0
80	135.5	113.5	116.0	143.0	143.0	143.0	143.5	146.0	143.5	143.0	141.5	127.0
100	136.5	115.0	117.0	144.5	144.0	144.0	144.5	146.5	145.0	143.0	142.0	127.5
125	136.5	115.5	118.0	144.5	145.0	144.5	145.0	147.0	144.5	144.0	143.5	128.0
160	137.0	116.0	119.0	145.0	145.5	145.0	146.0	148.0	146.0	144.5	143.5	128.0
200	139.0	116.5	119.5	146.0	147.0	146.5	147.5	148.5	146.5	146.0	145.0	129.0
250	139.5	116.5	120.0	147.0	148.0	147.5	148.0	149.5	147.0	147.0	146.0	129.5
320	141.0	117.5	120.0	148.0	148.5	149.0	149.0	150.5	148.5	148.5	148.0	131.0
400	142.0	117.5	119.5	148.0	148.5	149.0	149.0	150.5	149.0	149.5	148.5	131.5
500	142.0	116.5	118.5	147.0	147.5	147.5	148.5	149.5	148.5	149.0	148.0	132.0
630	141.5	115.5	117.0	146.0	146.5	148.5	147.5	148.5	148.5	148.5	147.5	132.5
800	141.5	115.0	117.0	145.5	146.0	147.5	147.5	148.0	148.0	148.0	147.5	133.5
1000	141.0	115.0	116.0	145.0	146.0	146.5	146.5	147.0	147.5	147.0	146.5	134.0
1250	142.0	114.5	115.5	145.0	146.0	146.5	146.0	146.5	146.5	146.5	146.5	134.5
1600	143.5	115.0	115.5	145.5	146.5	147.0	146.0	146.5	146.0	146.5	146.5	138.0
2000	143.5	115.5	115.5	145.5	147.0	147.5	146.0	146.0	146.0	146.5	147.0	141.5
2500	143.0	115.5	115.5	146.0	147.5	147.5	146.5	145.5	145.5	146.0	146.5	137.0
ESPL	153.0	158.5	160.5	158.0	159.0	159.5	160.0	160.5	159.0	159.5	158.5	146.5

ONE THIRD OCTAVE BAND CENTER FREQUENCY



Table VI. ACOUSTIC TEST RESULTS - SOUND PRESSURE LEVELS (continued)

Transducer Location: #34

Correlation No.	358	362	469	472
Mach No.	0.85	0.95	0.86	0.88
$\alpha$	0	0	0	0
$\theta$	+4	+4	-4	-4

ONE THIRD OCTAVE BAND SOUND PRESSURE LEVELS - DB RE: .0002 MICROBAR

	124.0	117.0	133.0	135.0
40	124.0	117.0	133.0	135.0
50	126.5	118.5	135.0	137.0
63	126.5	119.0	135.0	136.0
80	127.0	119.5	135.5	137.5
100	128.5	120.0	137.0	137.5
125	130.0	121.5	137.0	137.5
160	131.0	122.0	137.5	138.0
200	135.0	123.5	138.5	139.0
250	135.0	124.5	139.0	139.5
320	135.0	125.5	139.5	140.0
400	134.0	126.5	140.0	139.5
500	132.5	127.0	139.0	139.0
630	131.5	128.5	138.5	139.0
800	132.0	130.0	138.5	139.5
1000	134.5	131.5	139.0	140.0
1250	138.5	133.5	141.0	141.5
1600	144.0	137.0	145.5	145.5
2000	146.0	142.0	148.0	148.5
2500	145.5	142.5	147.0	147.5
3200	151.5	147.0	151.5	155.5

ONE THIRD OCTAVE BAND CENTER FREQUENCY

ONE THIRD OCTAVE BAND CENTER FREQUENCY

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Table VI. ACOUSTIC TEST RESULTS - SOUND PRESSURE LEVELS (continued)

Transducer Location: #36

Correlation No.	360	361	455	462	467	468	469	470	471	472	473	474	475
Mach No.	0.9	0.95	0.75	0.82	0.84	0.86	0.86	0.86	0.88	0.88	0.88	0.9	0.9
$\alpha$	0	0	+4	+4	+4	+4	0	-4	-4	0	+4	+4	0
$\theta$	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4

ONE THIRD OCTAVE BAND SOUND PRESSURE LEVELS - DB RE: .0002 MICROBAR

|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|--|

Table VI. ACOUSTIC TEST RESULTS - SOUND PRESSURE LEVELS (concluded)

Transducer Location: #36

Correlation No.	476	477	478	479	481	482	485				
Mach No.	0.9	0.92	0.92	0.92	0.94	0.94	0.96				
$\alpha$	-4	-4	0	+4	0	-4	+4				
$\beta$	-4	-4	-4	-4	-4	-4	-4				
ONE THIRD OCTAVE BAND SOUND PRESSURE LEVELS - DB RE: .0002 MICROBAR											
40	136.0	141.5	139.5	139.5	136.0	135.5	121.5				
50	138.0	143.0	141.0	141.0	137.0	137.5	121.5				
63	137.0	141.5	140.5	141.5	137.0	137.5	122.5				
80	139.0	142.5	142.0	142.5	138.0	139.5	123.5				
100	140.0	142.0	142.5	143.0	138.5	140.5	124.5				
125	140.5	141.5	142.5	143.0	139.0	141.0	125.0				
160	141.0	142.0	142.5	143.5	139.0	141.5	125.5				
200	142.0	142.5	143.5	144.5	140.0	142.5	127.5				
250	143.0	143.0	143.5	144.5	140.0	142.5	128.0				
320	143.5	143.0	144.0	144.5	139.5	141.5	128.5				
400	144.0	143.0	143.5	144.5	140.0	141.5	129.5				
500	144.5	143.0	144.0	144.5	140.5	142.0	130.5				
630	145.0	143.5	144.0	144.0	141.0	142.0	131.5				
800	146.5	145.0	145.0	145.5	142.5	143.5	133.0				
1000	148.5	147.0	147.0	147.0	144.0	145.0	135.0				
1250	151.0	149.0	149.5	149.5	147.0	148.0	137.5				
1600	153.0	151.5	152.0	152.0	149.5	150.0	141.5				
2000	154.0	153.5	154.0	154.0	152.0	153.0	146.5				
2500	155.5	155.0	156.5	156.5	155.0	155.0	147.0				
$\Sigma$ SPL	161.0	160.0	161.0	161.0	158.5	159.5	151.0				

ONE THIRD OCTAVE BAND CENTER FREQUENCY

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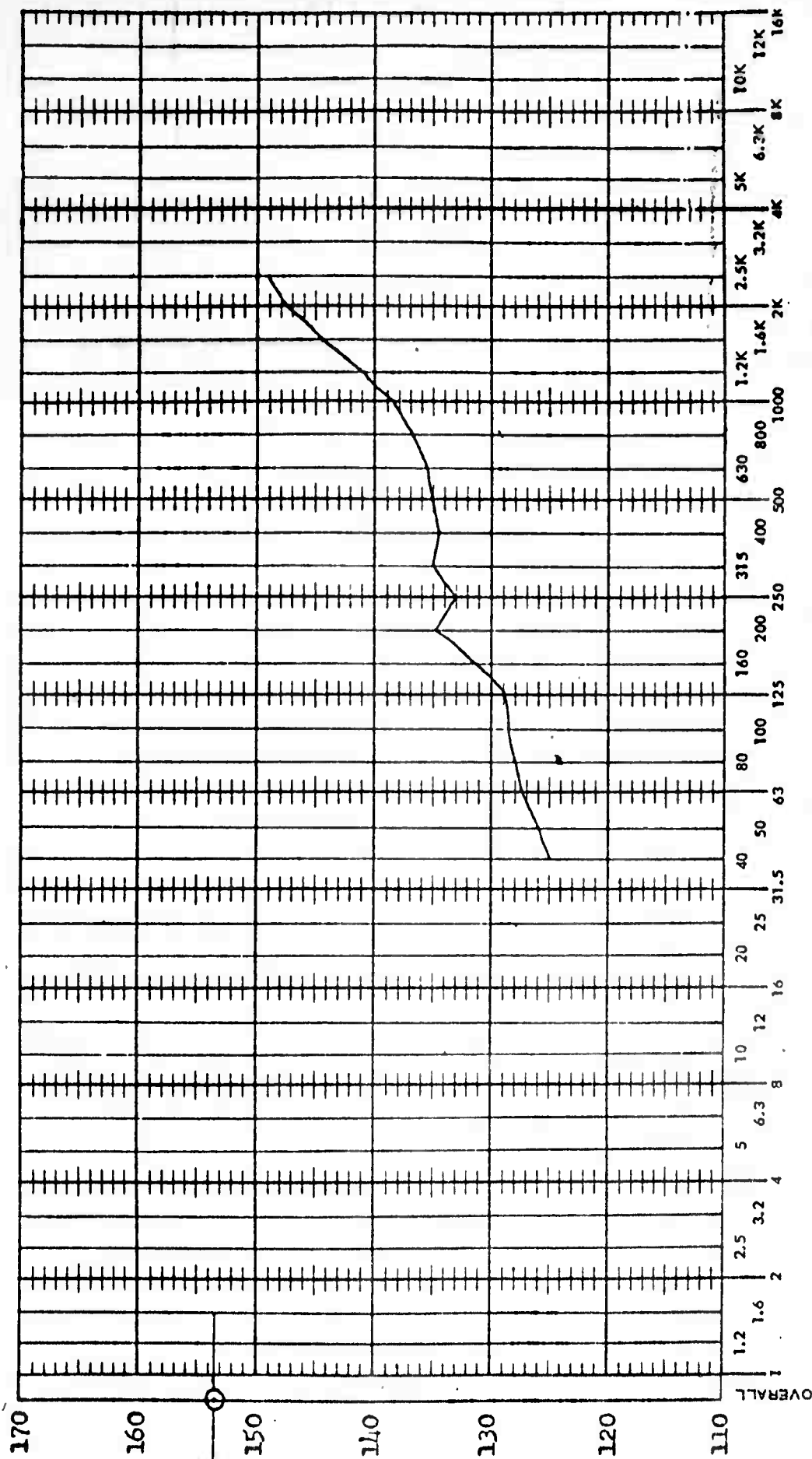
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



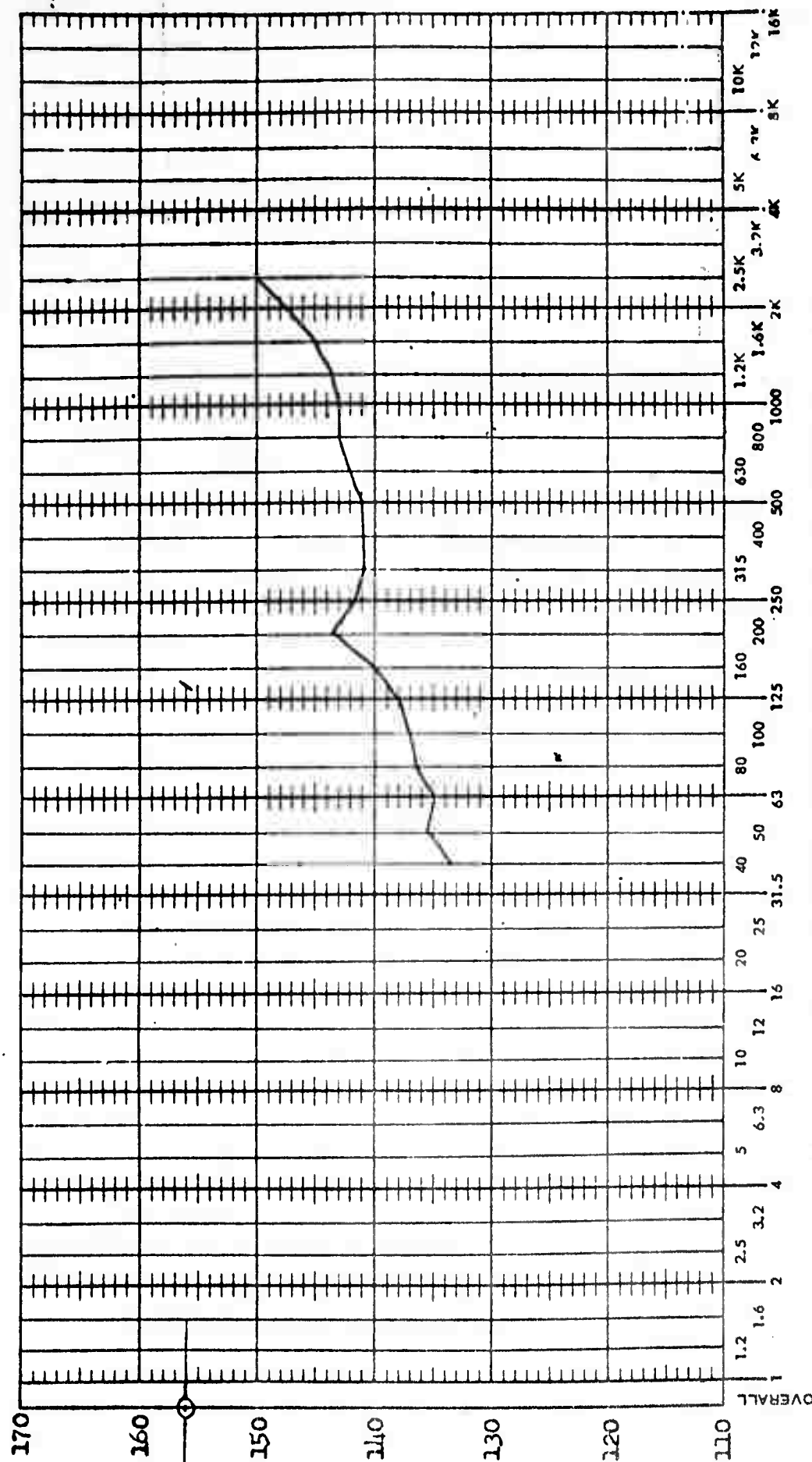
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #2 Mach No. 0.8 Correlation No. 305  $\alpha = 4^\circ$   $\beta = 0^\circ$

Figure 14. One-third octave band sound pressure levels.



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #2 Mach No. 0.85 Correlation No. 309

$\alpha = 7^\circ$

$\beta = 0^\circ$

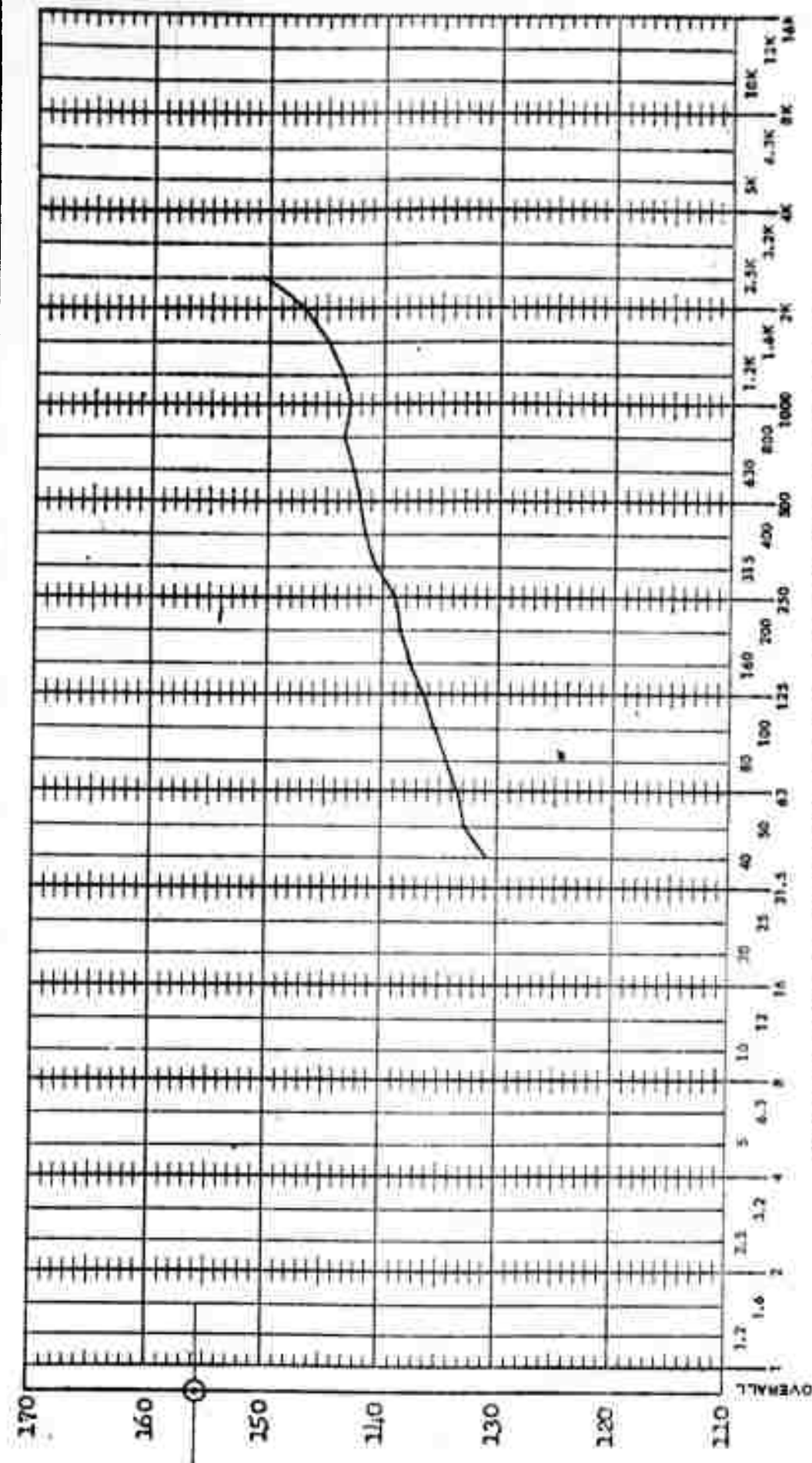
Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #2 Mach No. 0.9 Correlation No. 311

$\alpha = 4^\circ$   $\delta = 0^\circ$

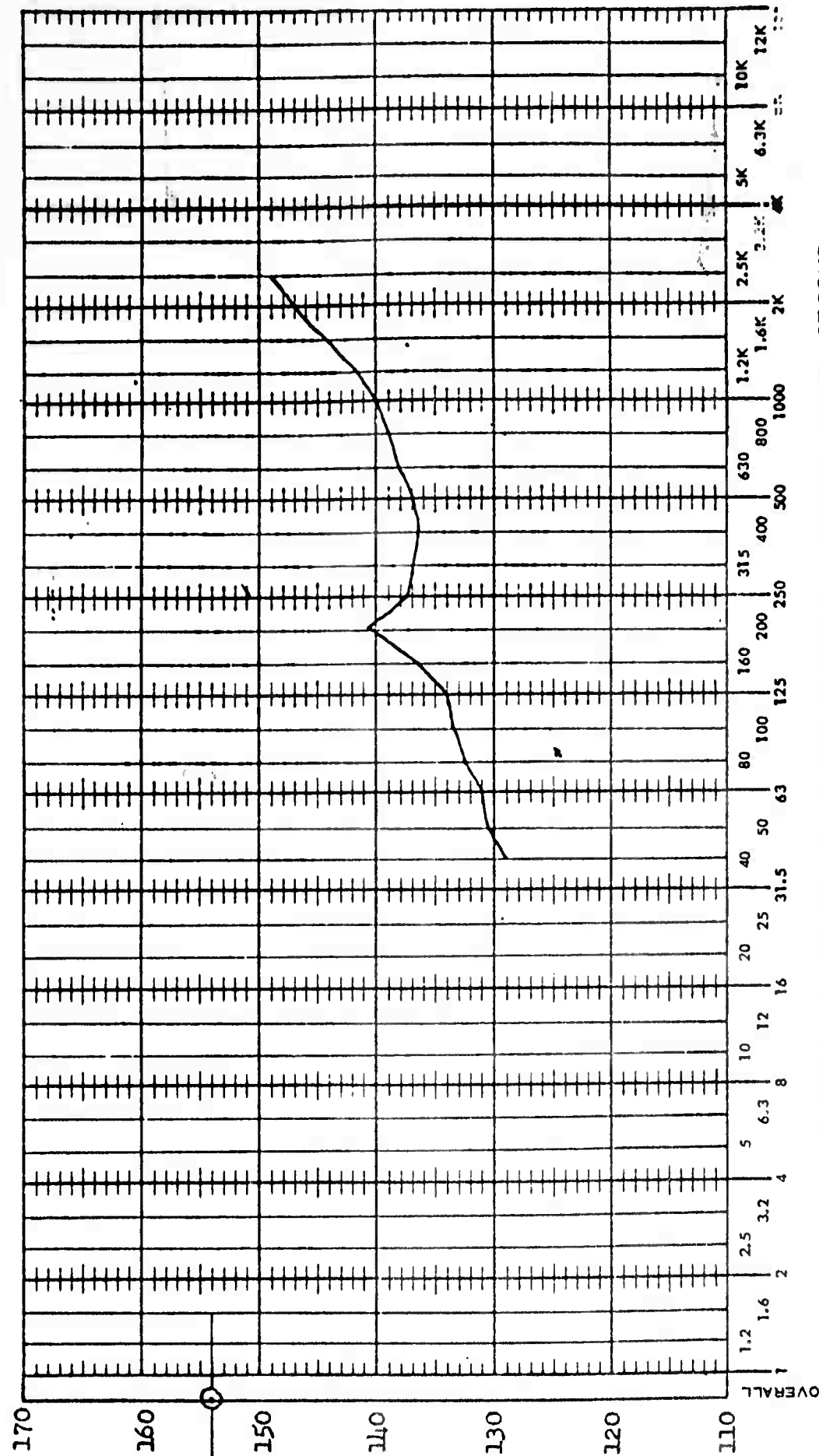
Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #2 Mach No. 0.825 Correlation No. 323

$\alpha = 7^\circ$   $\beta = 0^\circ$

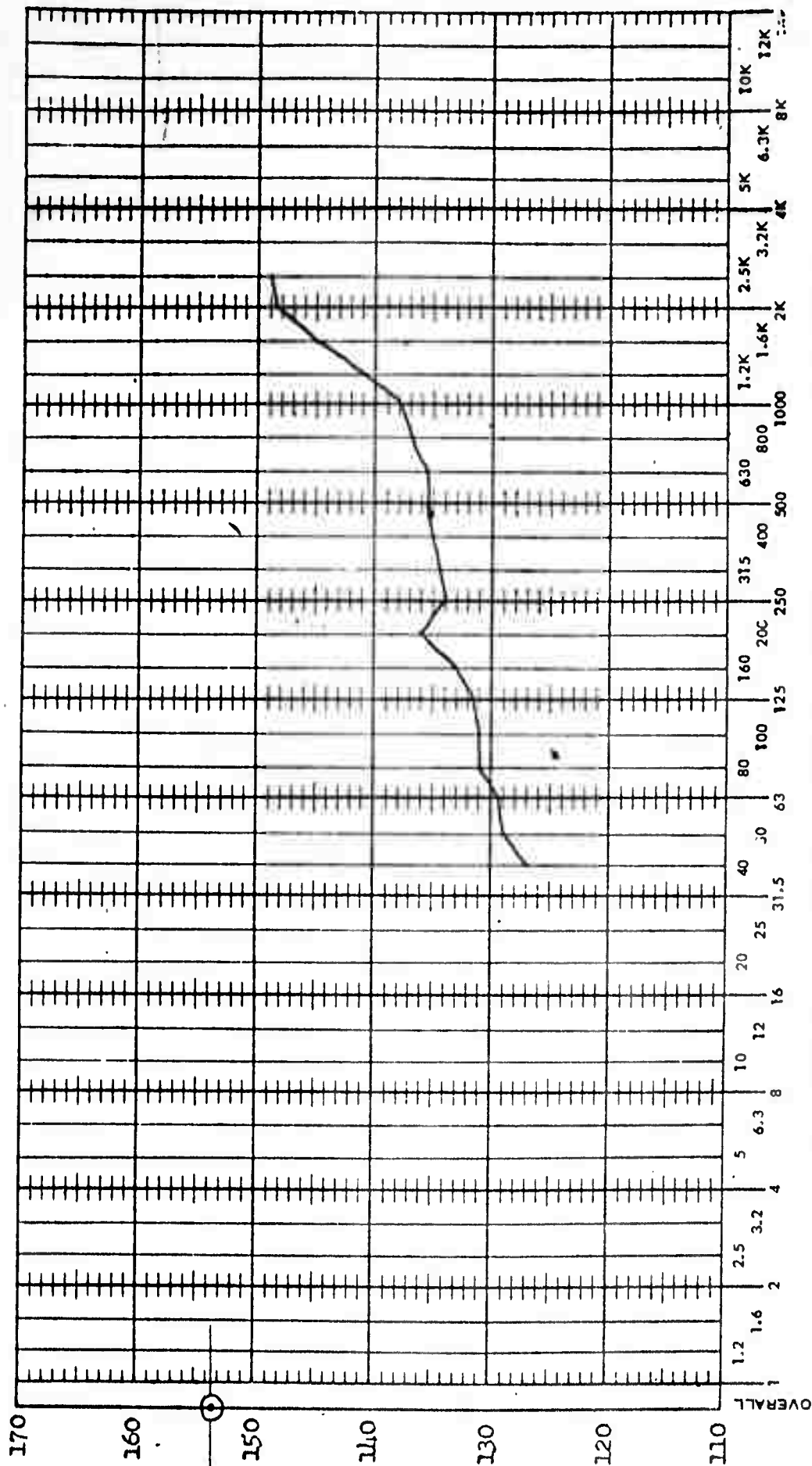
Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #2 Mach No. 0.8 Correlation No. 461

Figure 14 (Continued)

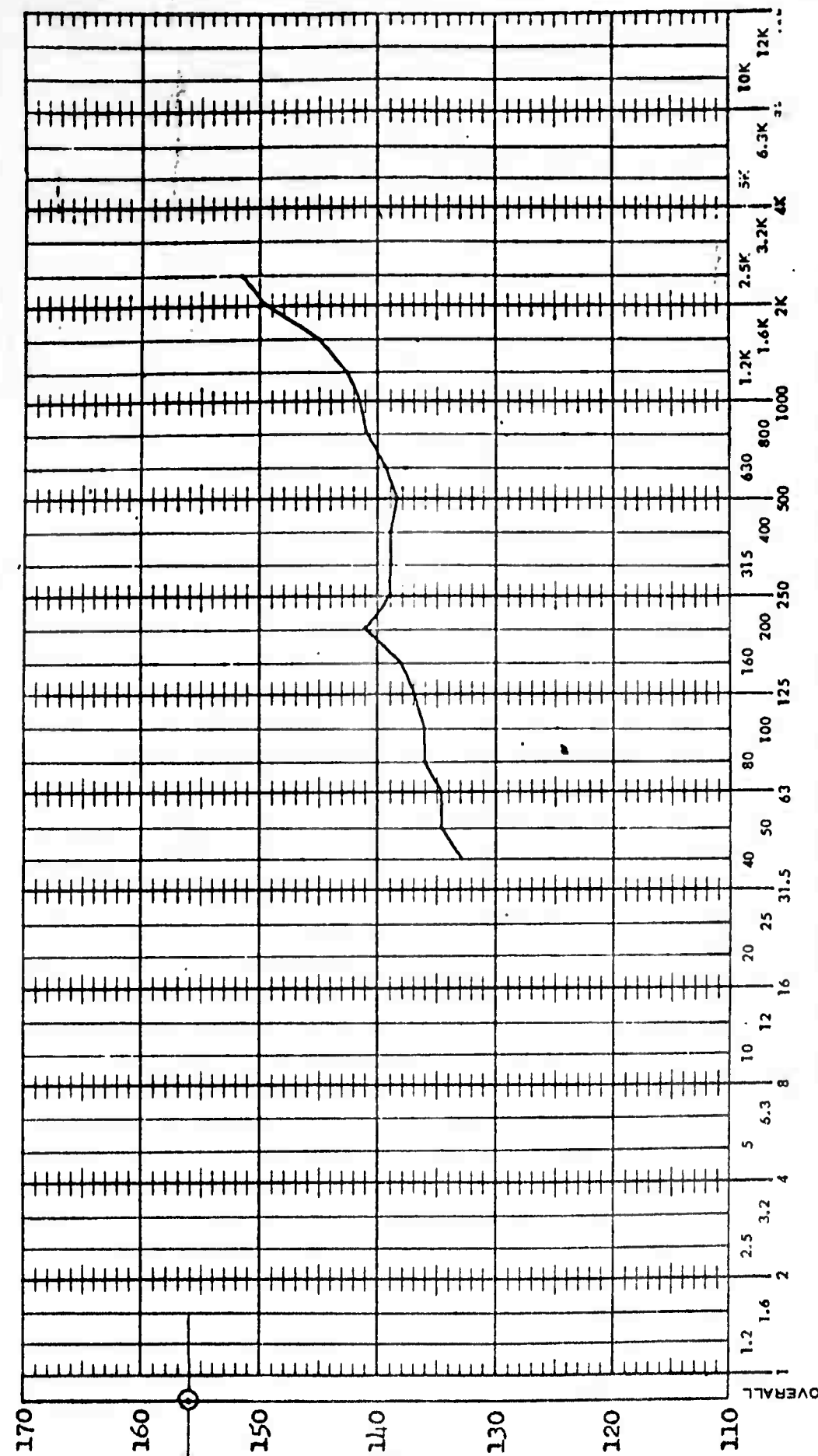
$\alpha = 4^\circ$   $\beta = -4^\circ$

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #2 Mach No. 0.84 Correlation No. 467

$\alpha = 4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

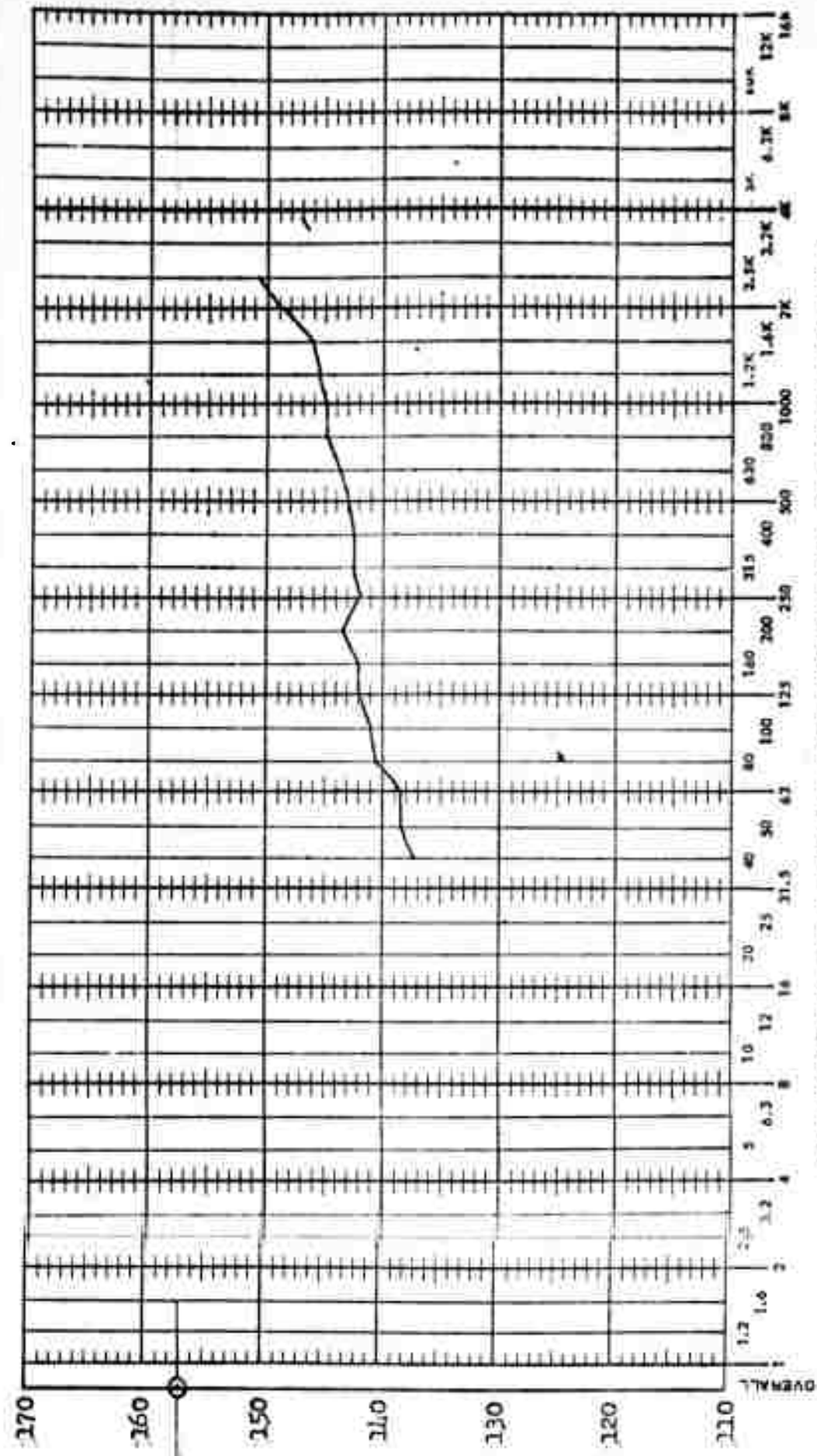
BOEING

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR

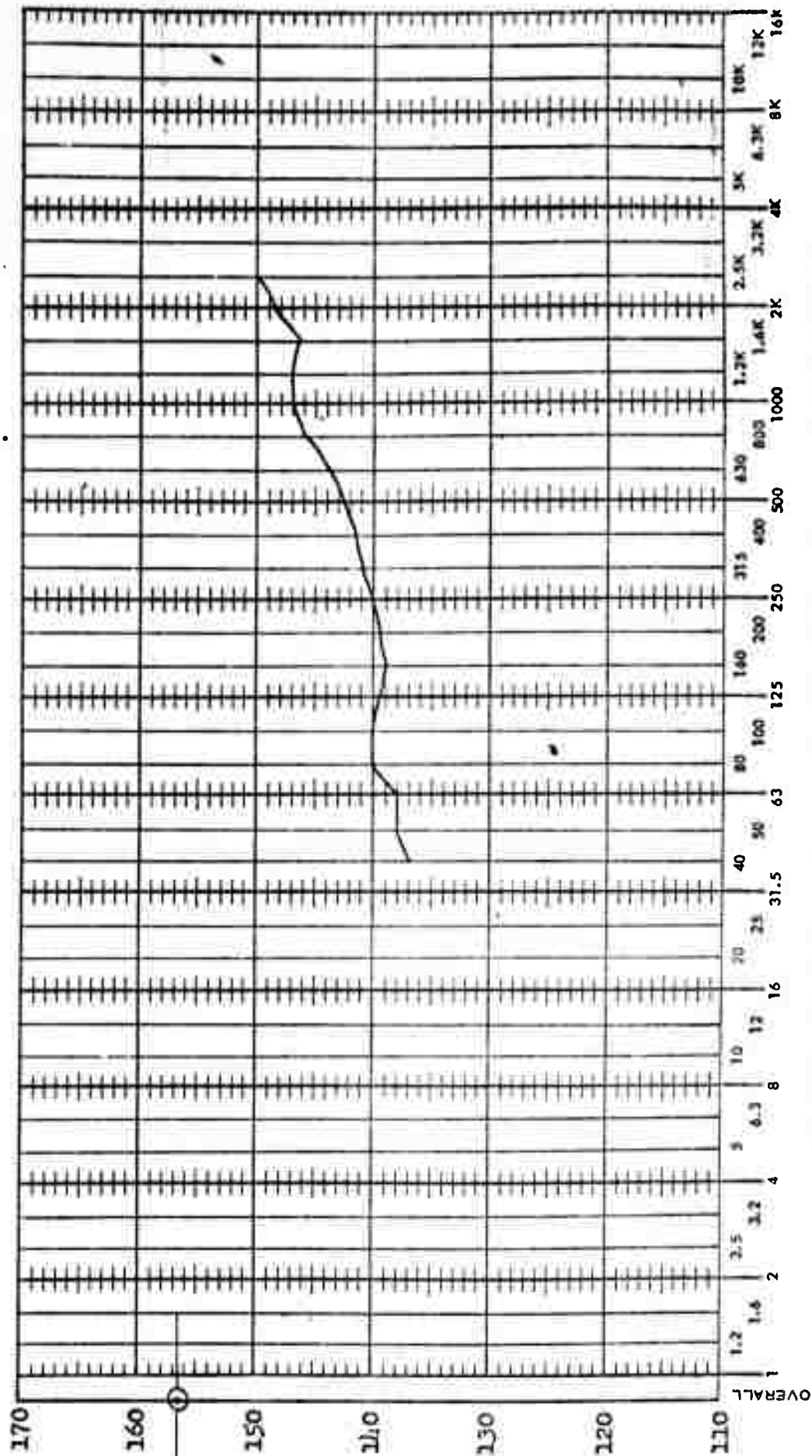


ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #2 Each No. 0.86 Correlation No. 468  $\alpha = 4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #2 Mach No. 0.88 Correlation No. 473

$\alpha = 4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

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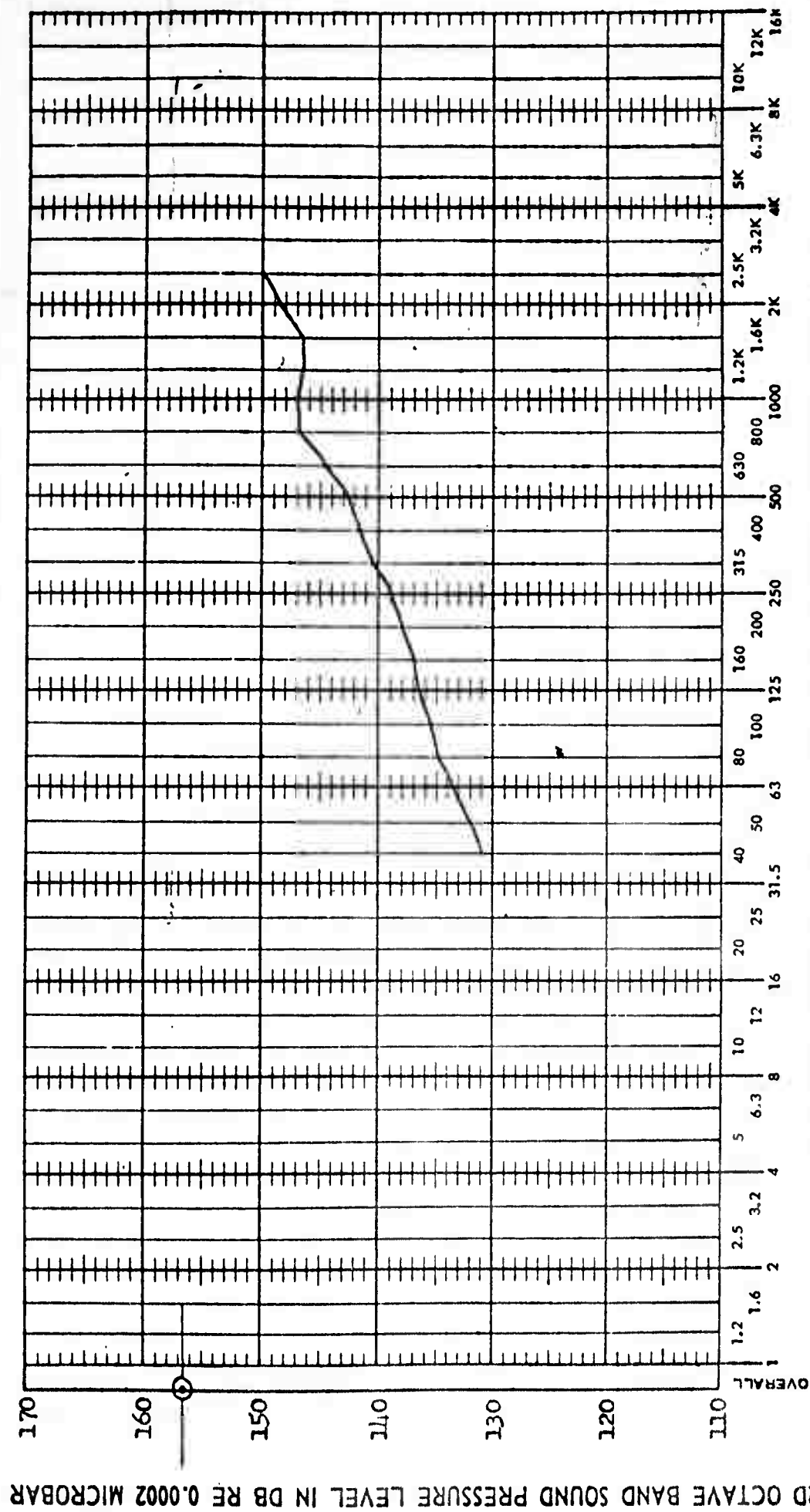


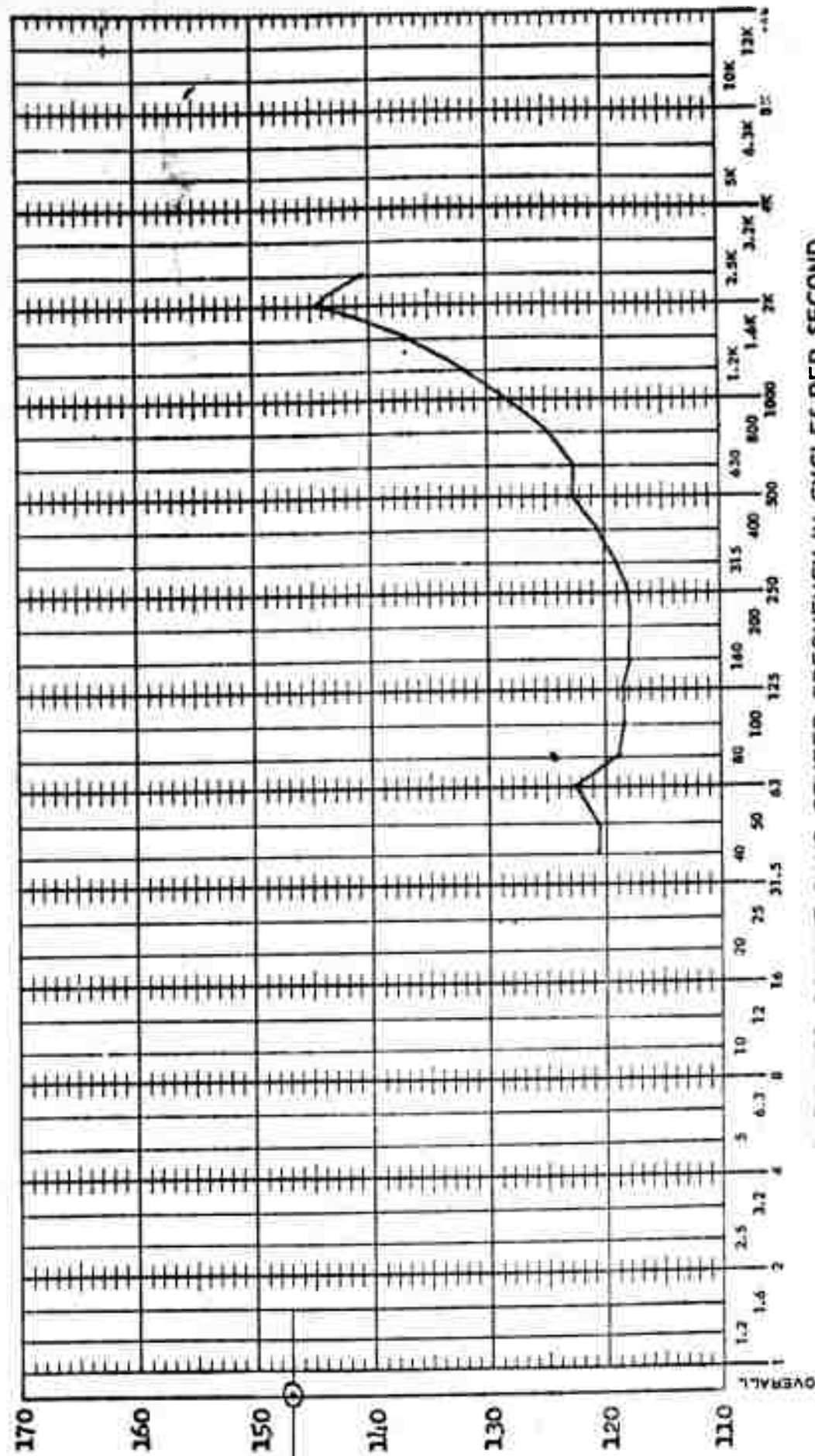
Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #2 Mach No. 0.96 Correlation No. 1485

$\alpha = 4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

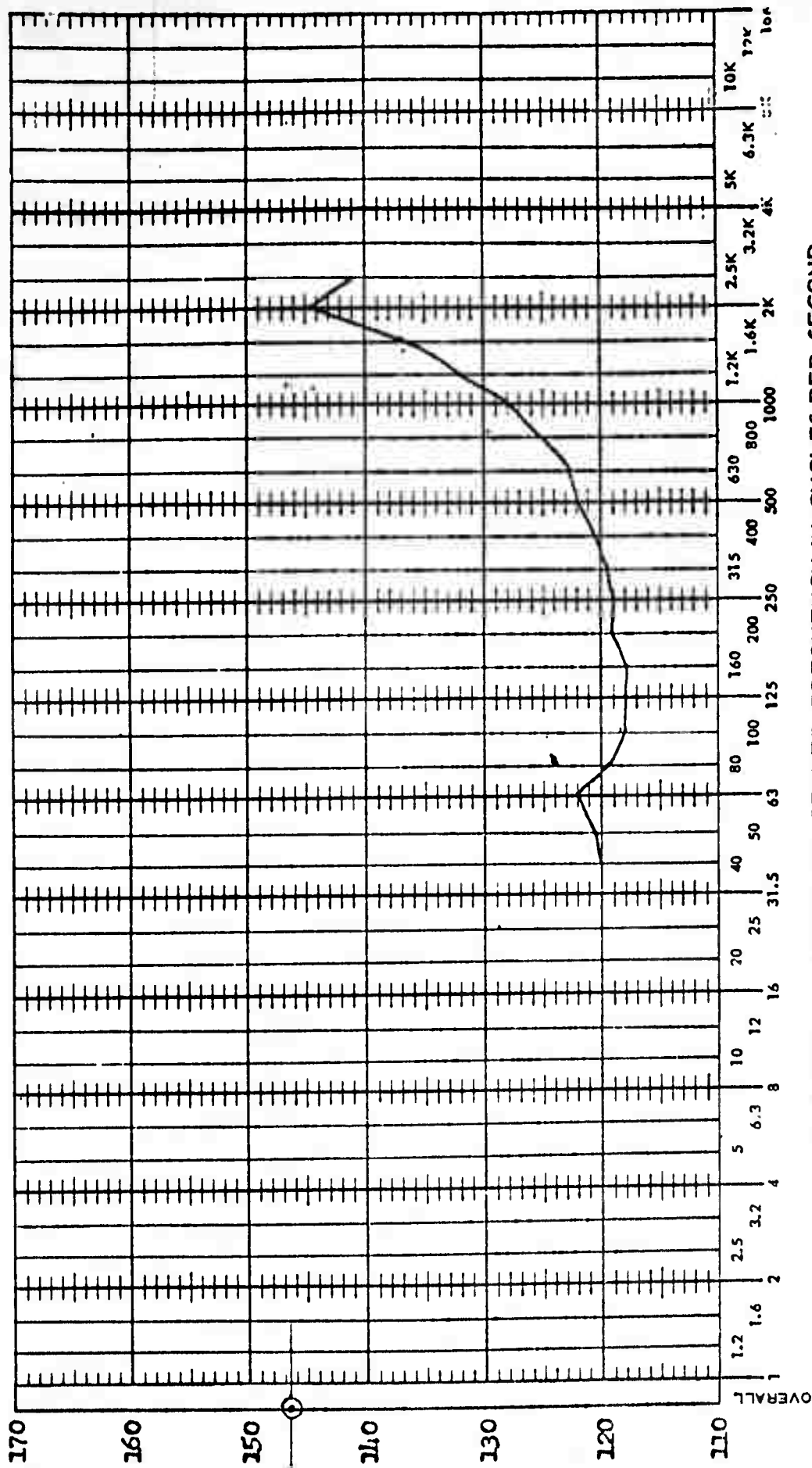
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #2 Mach No. 0.98 Correlation No. 486

$\alpha = 4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

BOEING

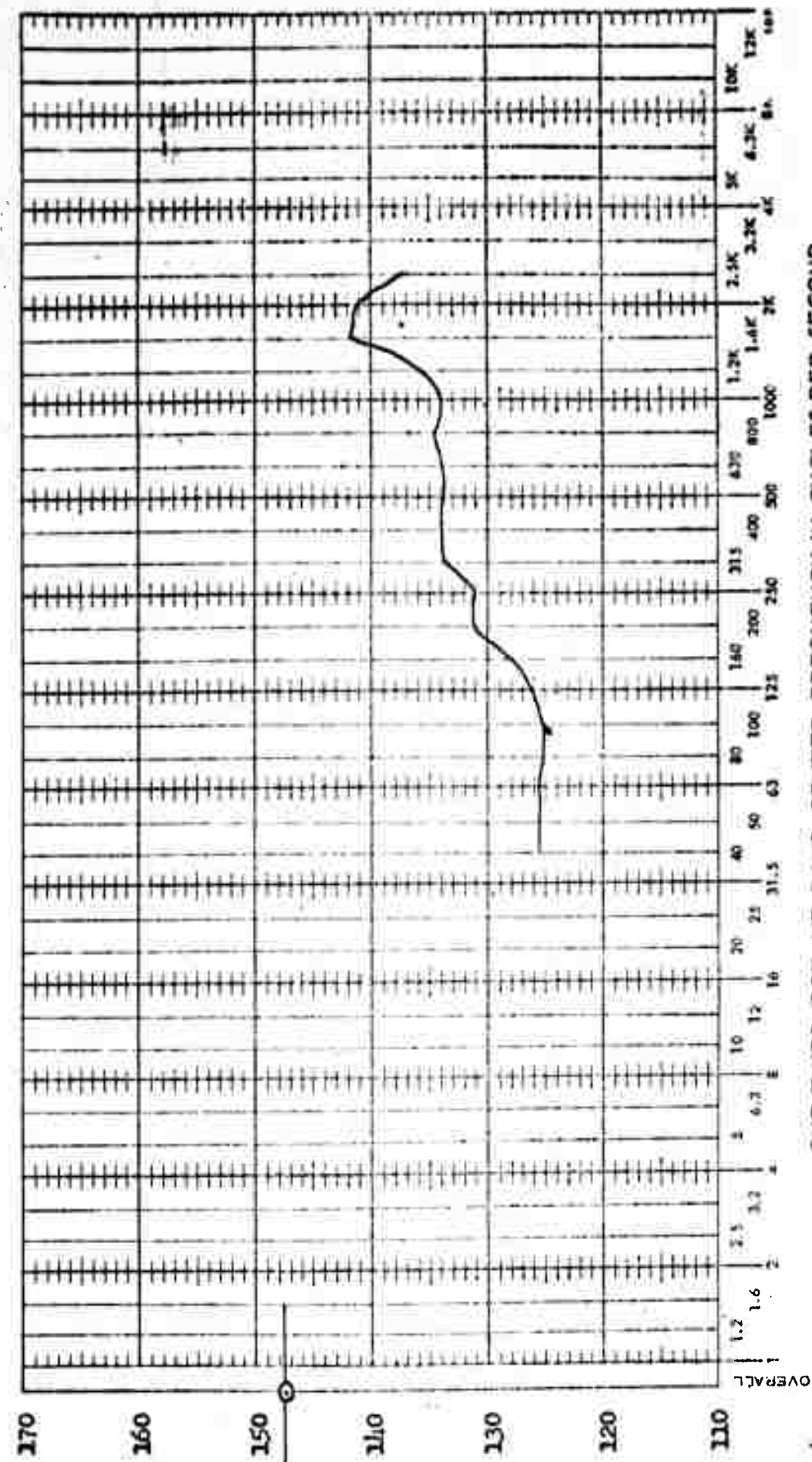
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #1 Mach No. 0.75 Correlation No. 457

$\alpha = -4^\circ$   $\beta = -4^\circ$

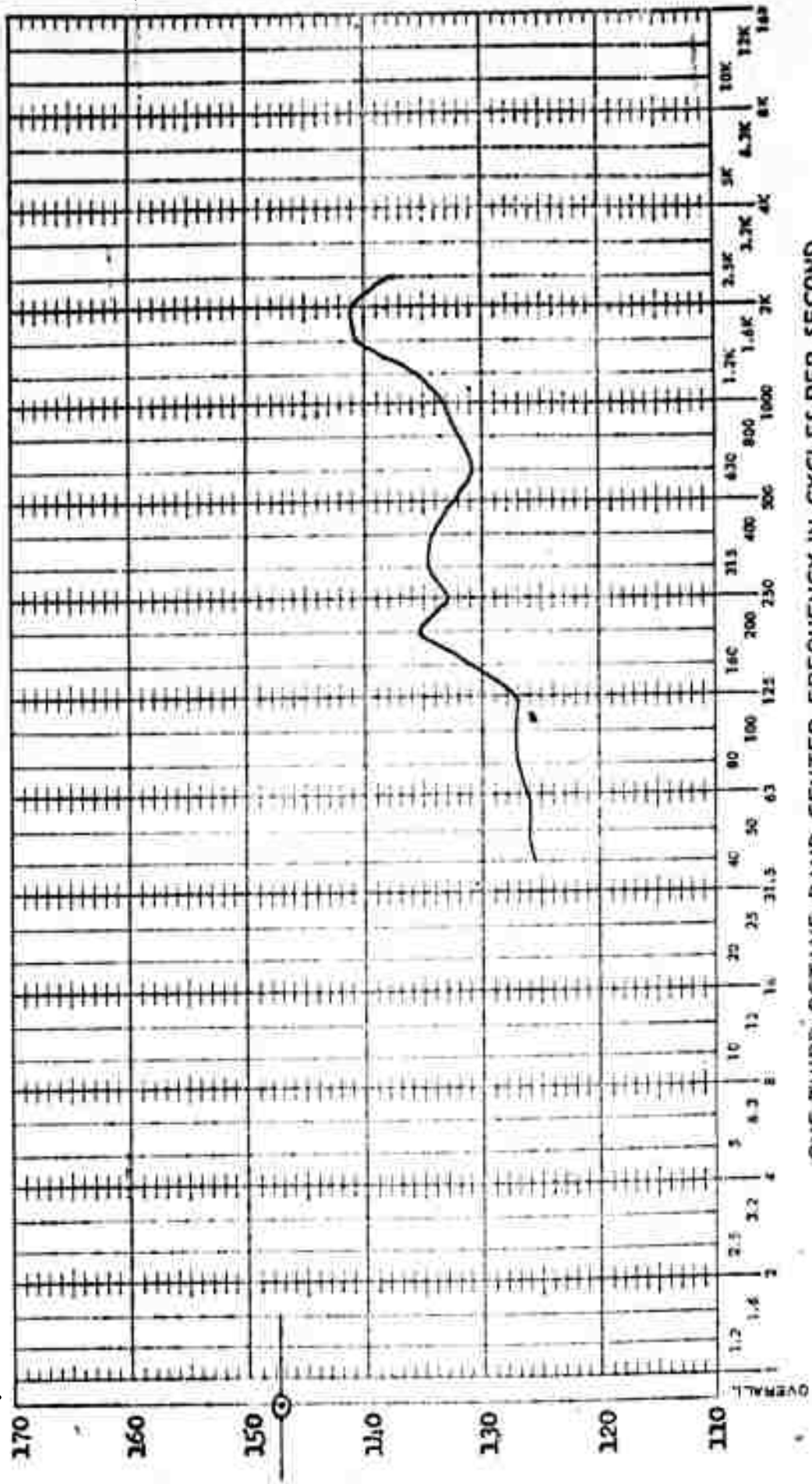
Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



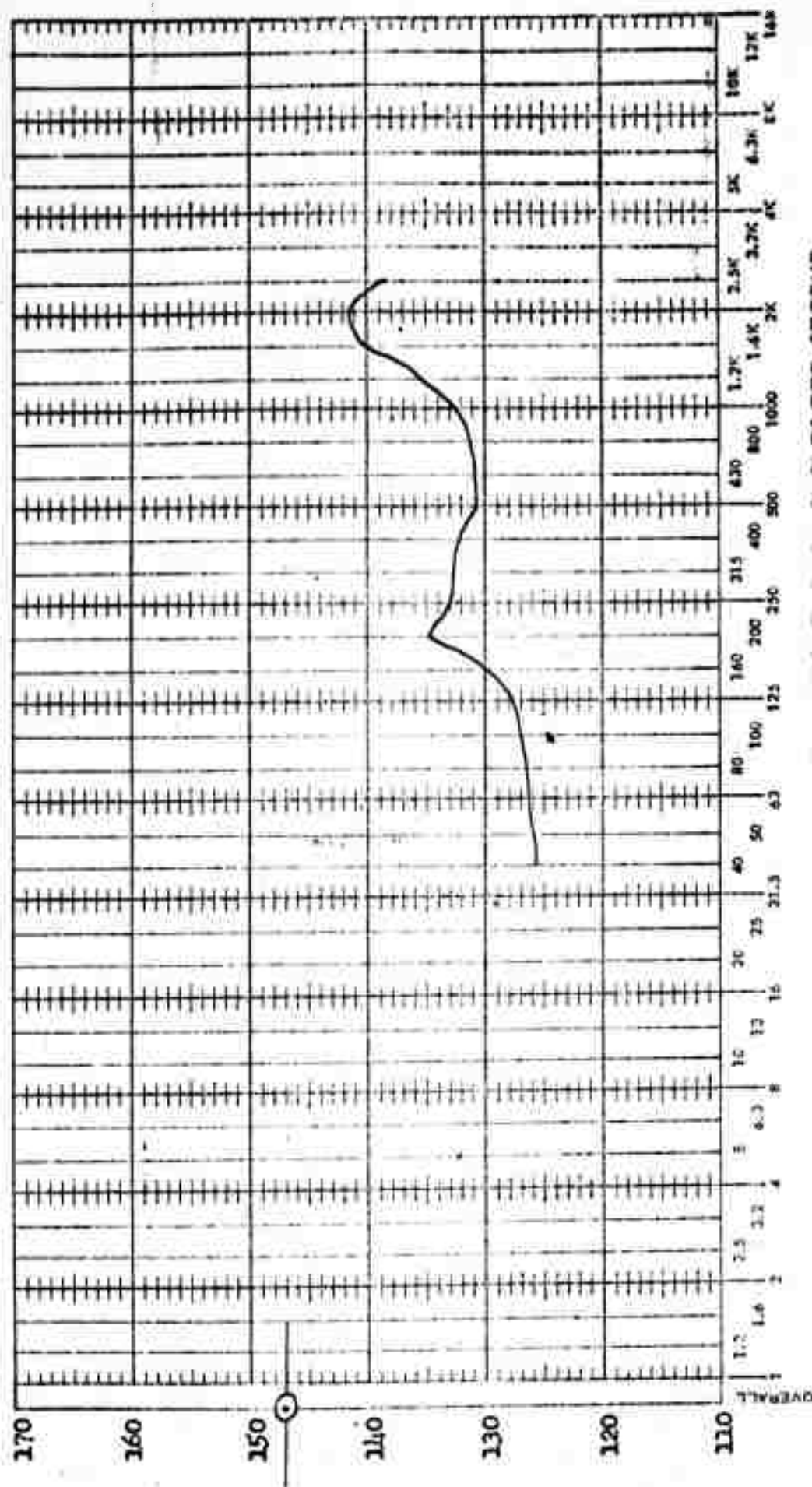
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #1 Mach No. 0.82 Correlation No. 164

$\alpha = -4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

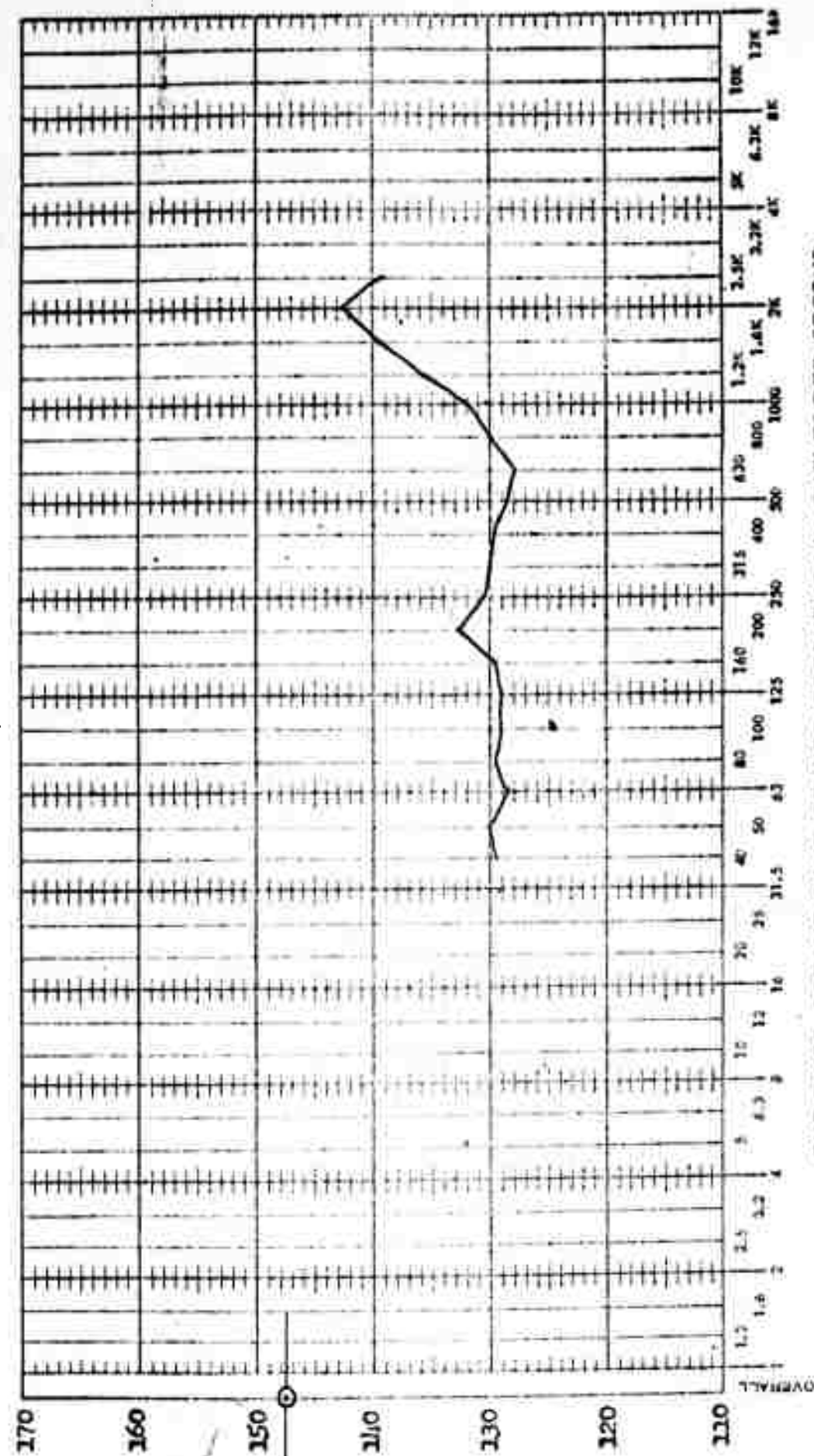
Test Point #/1 Nach No. 0.84 Correlation No. 465

$$\alpha = -4^\circ \quad \beta = -4^\circ$$

Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #4 Mach No. 0.88 Correlation No. 471

$\alpha = -4^\circ$   $\beta = -9^\circ$

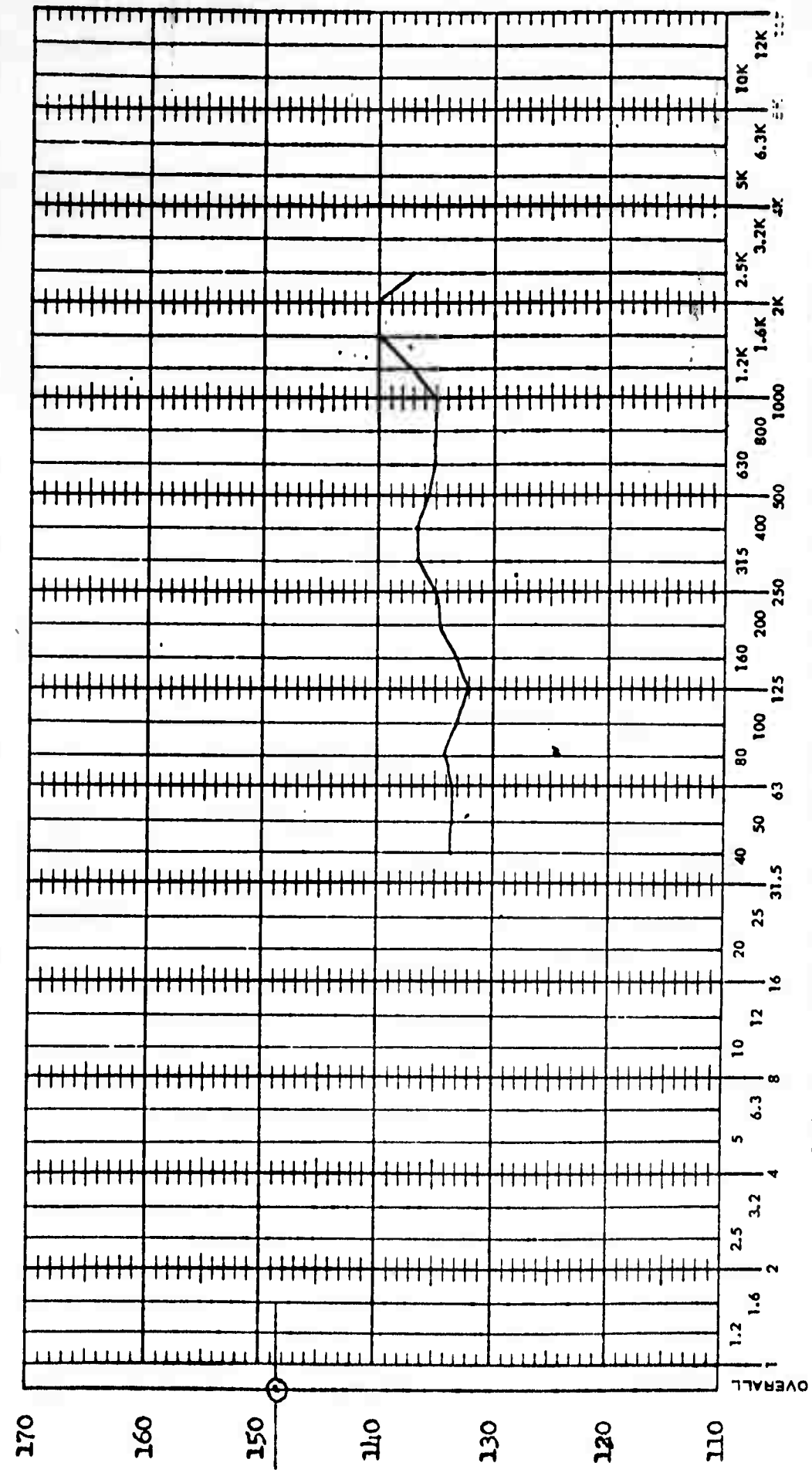
Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #5 Mach No. 0.75 Correlation No. 457

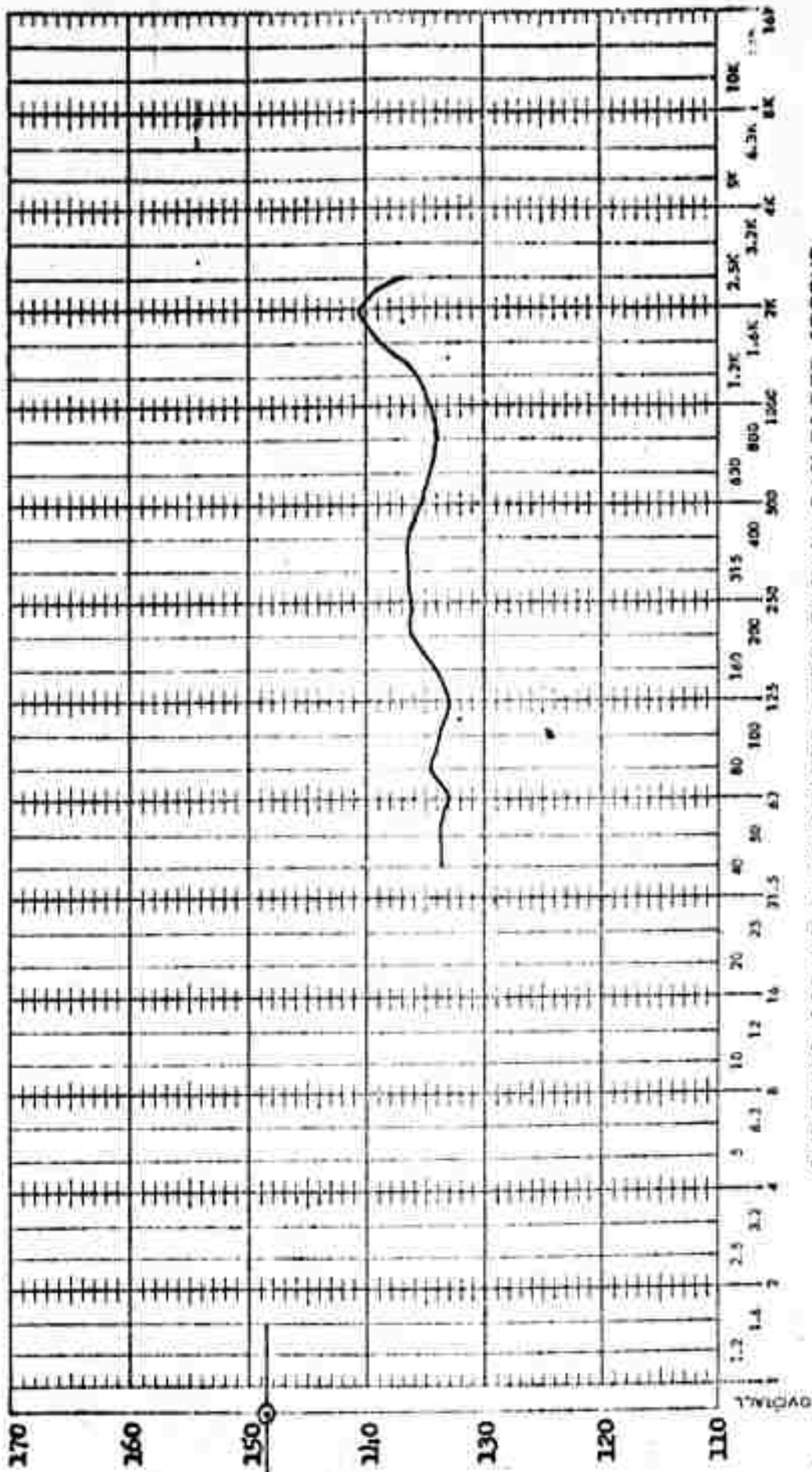
$\alpha = -2^\circ$   $\beta = -7^\circ$

Figure 14 (Continued)

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

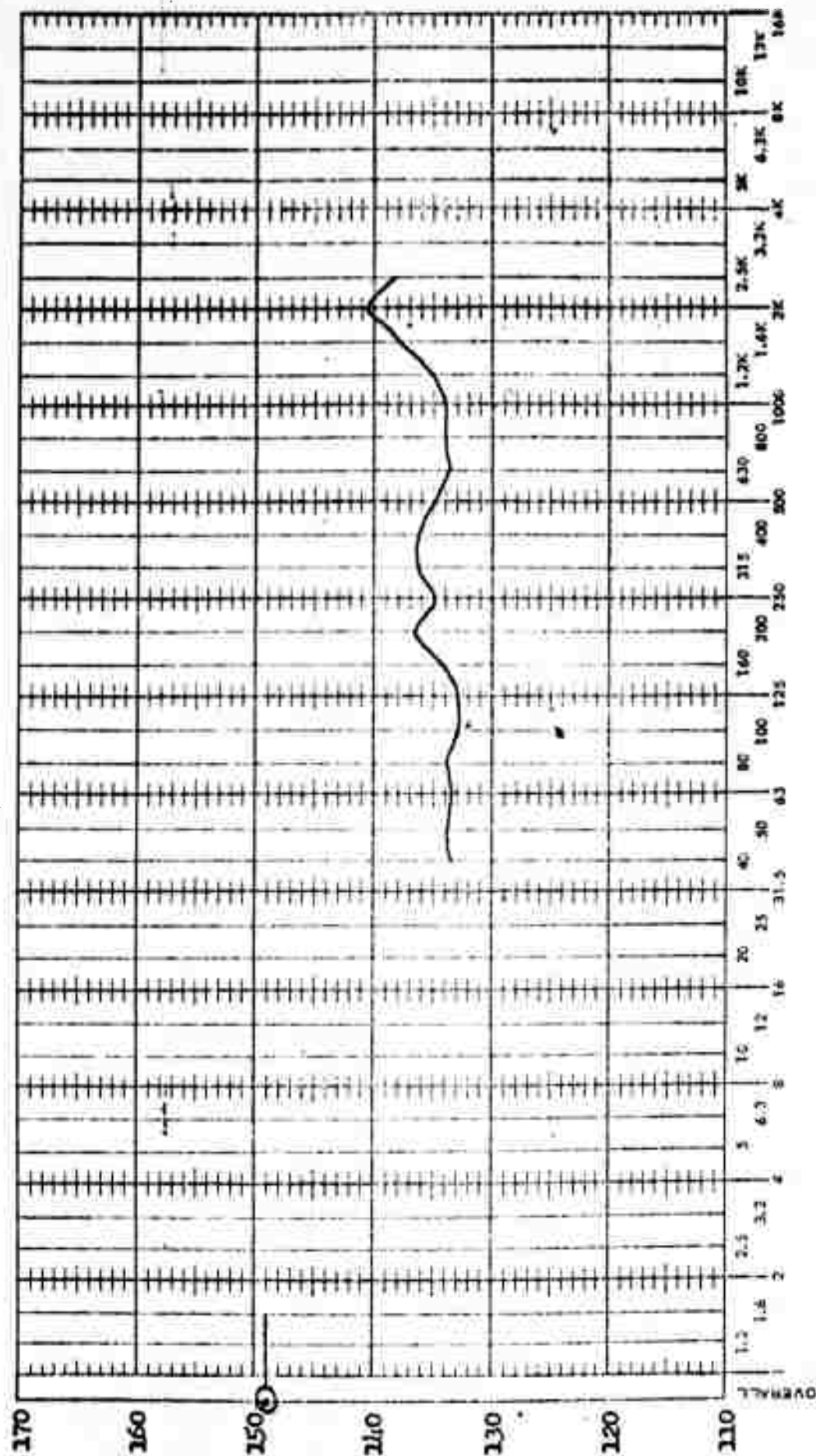
Test Point #5 Mach No. 0.8 Correlation No. 458

$\alpha = -2^\circ$   $\beta = -2^\circ$

Figure 14 (Continued)



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #5 Mach No. 0.82 Correlation No. 464

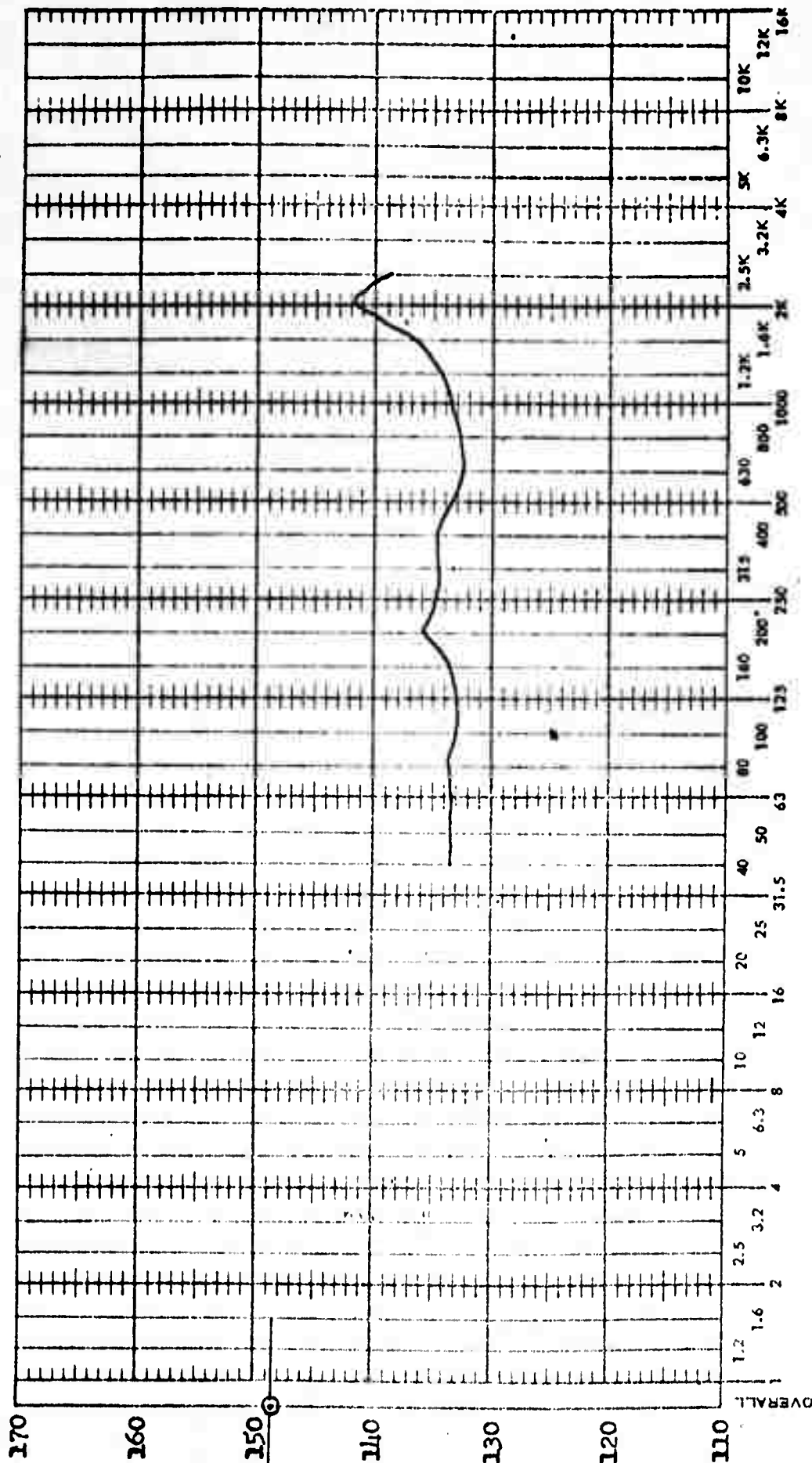
$\alpha = -4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



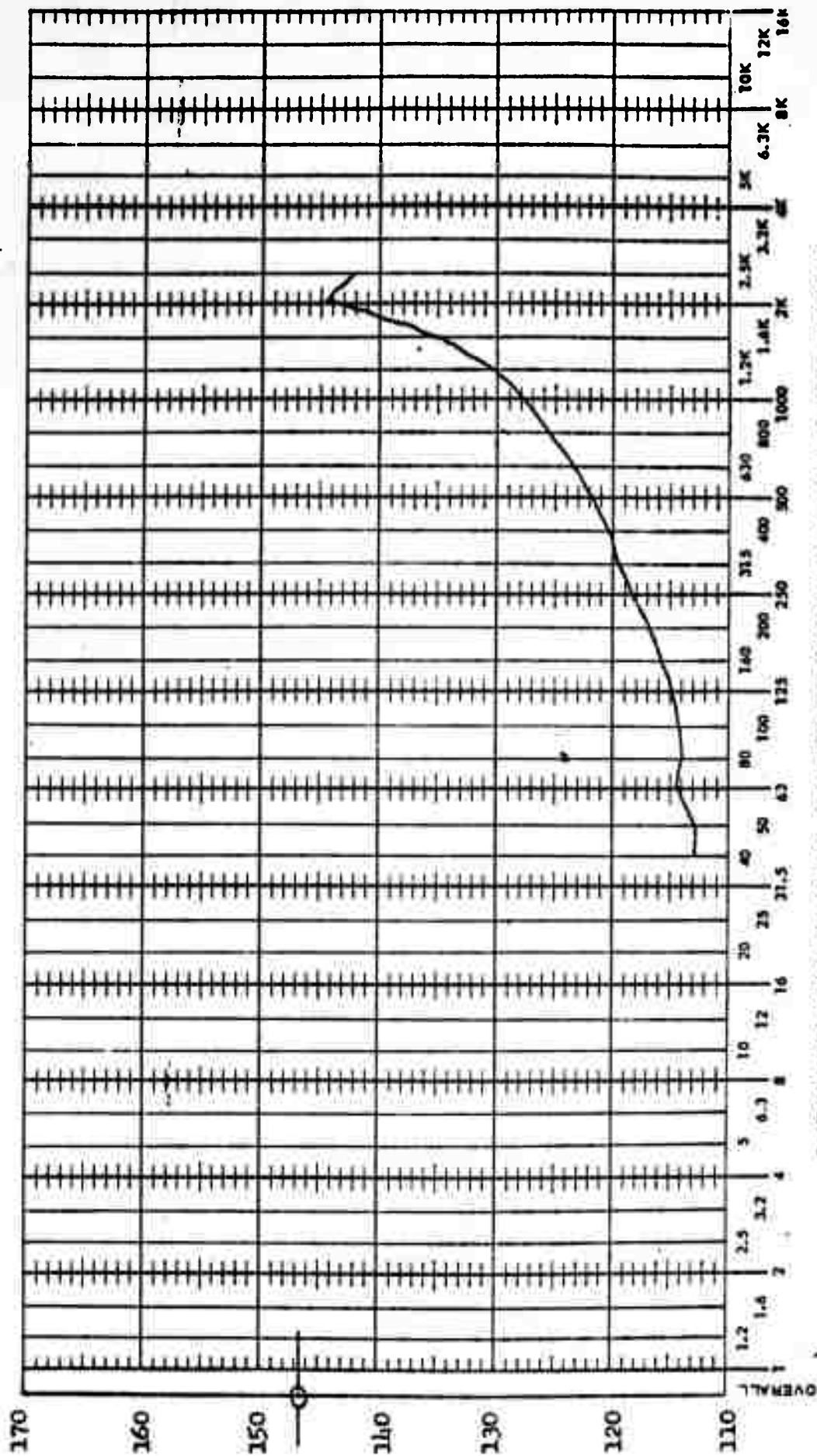
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #5 Mach No. 0.84 Correlation No. 465

Figure 14 (Continued)

$\alpha = -4^\circ$   $\beta = -7^\circ$

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #7 Mach No. 1.0 Correlation No. 490

Figure 14 (Continued)

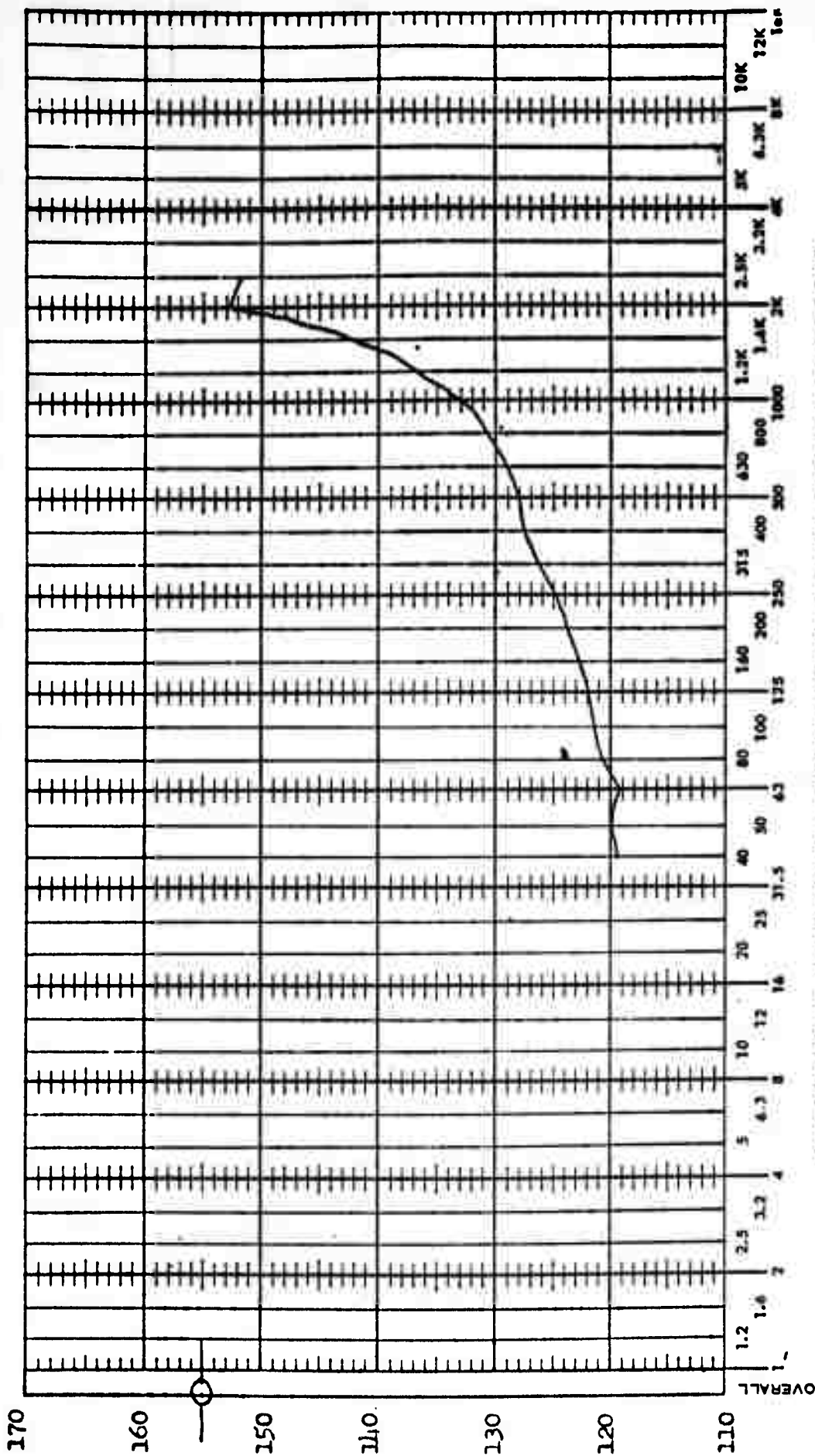
$\alpha = 0^\circ$   $\beta = -4^\circ$

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #7 Mach No. 1.01, Correlation No. 495

Figure 14 (Continued)

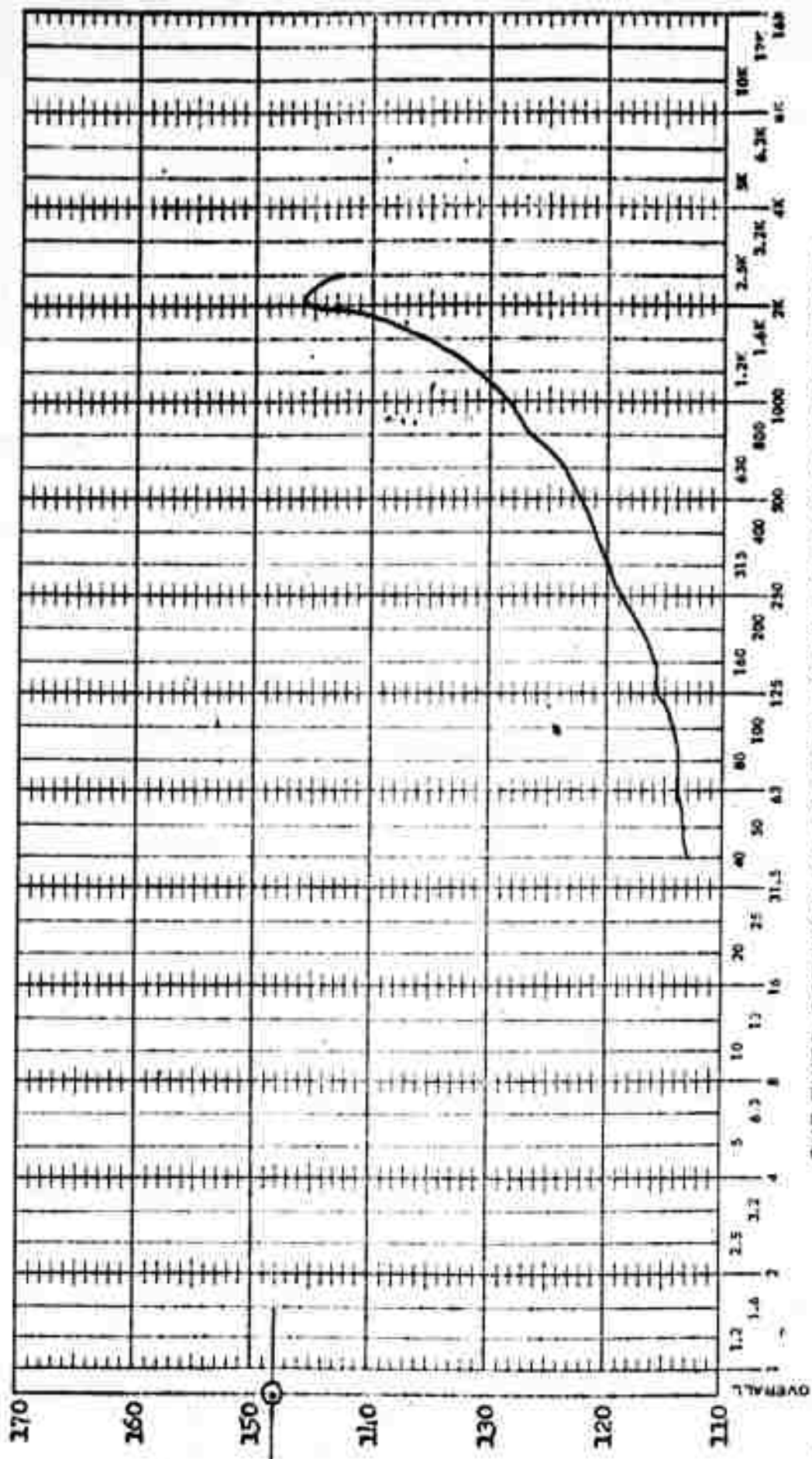
$\alpha = -4^\circ$   $\beta = -4^\circ$

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



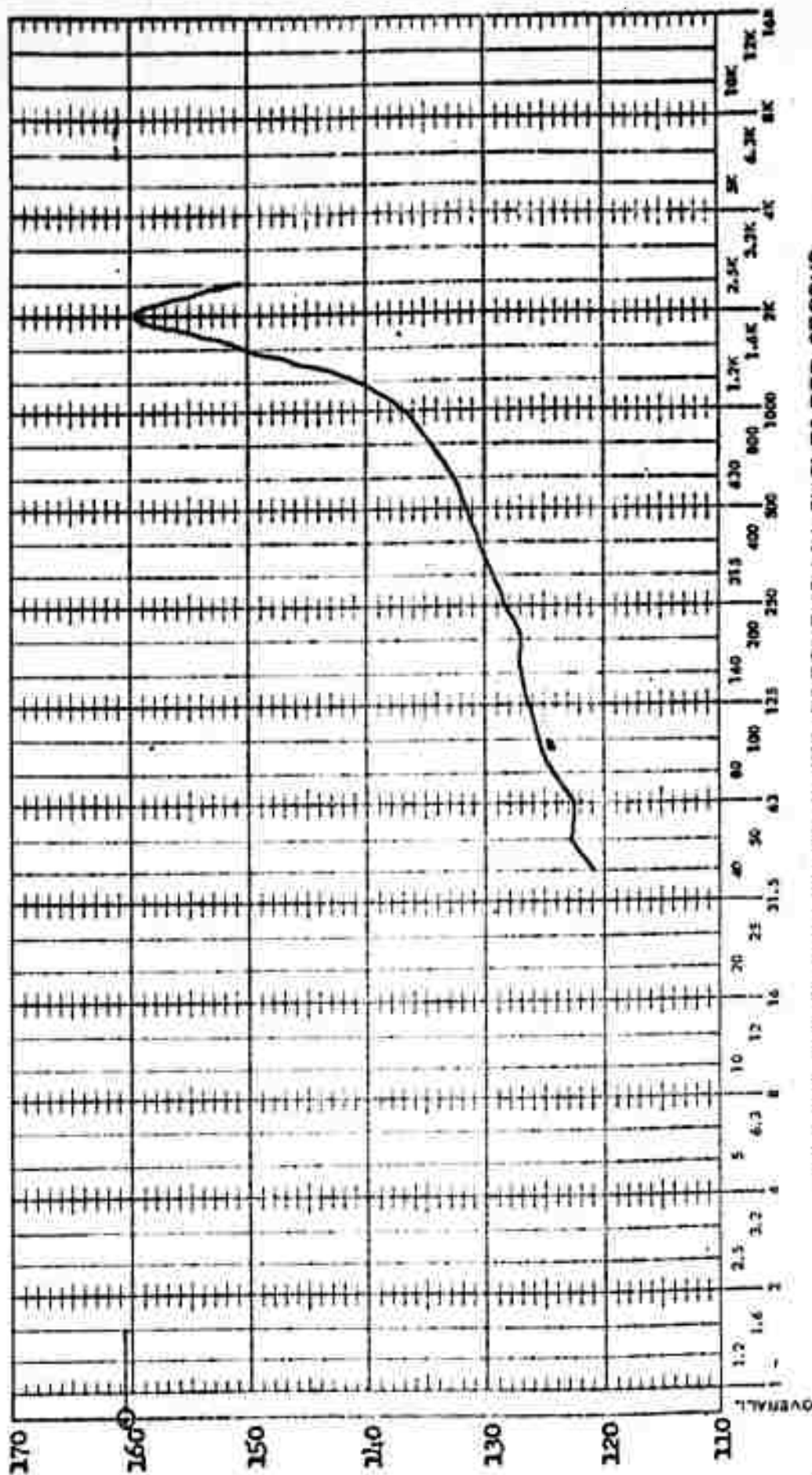
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #7 Mach No. 1.04 Correlation No. 496

Figure 1b (Continued)

$\alpha = 0^\circ$   $\beta = -4^\circ$

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

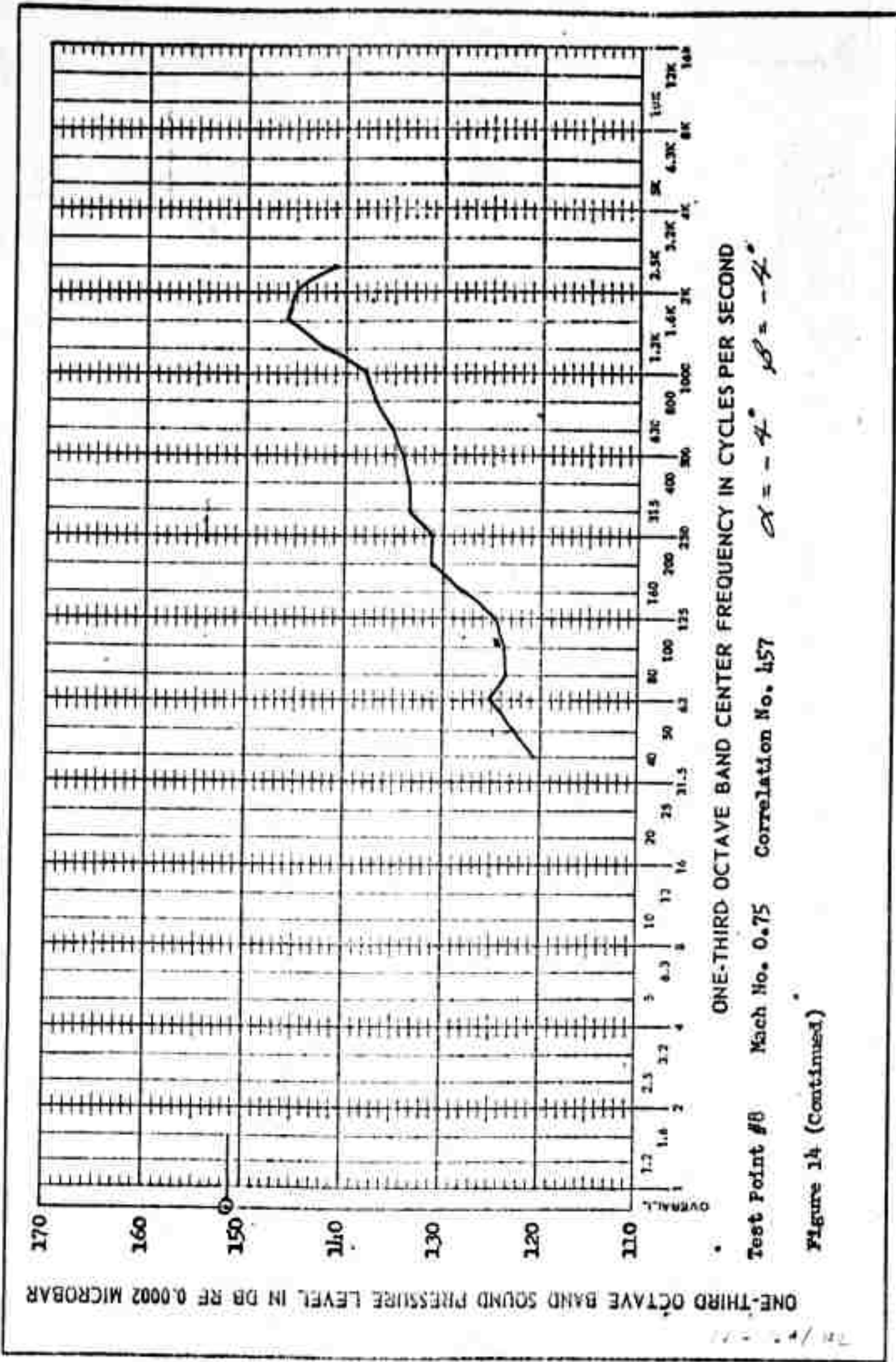
Test Point #7 Mach No. 1.08 Correlation No. 501

$\alpha = -2^\circ$   $\beta = -4^\circ$

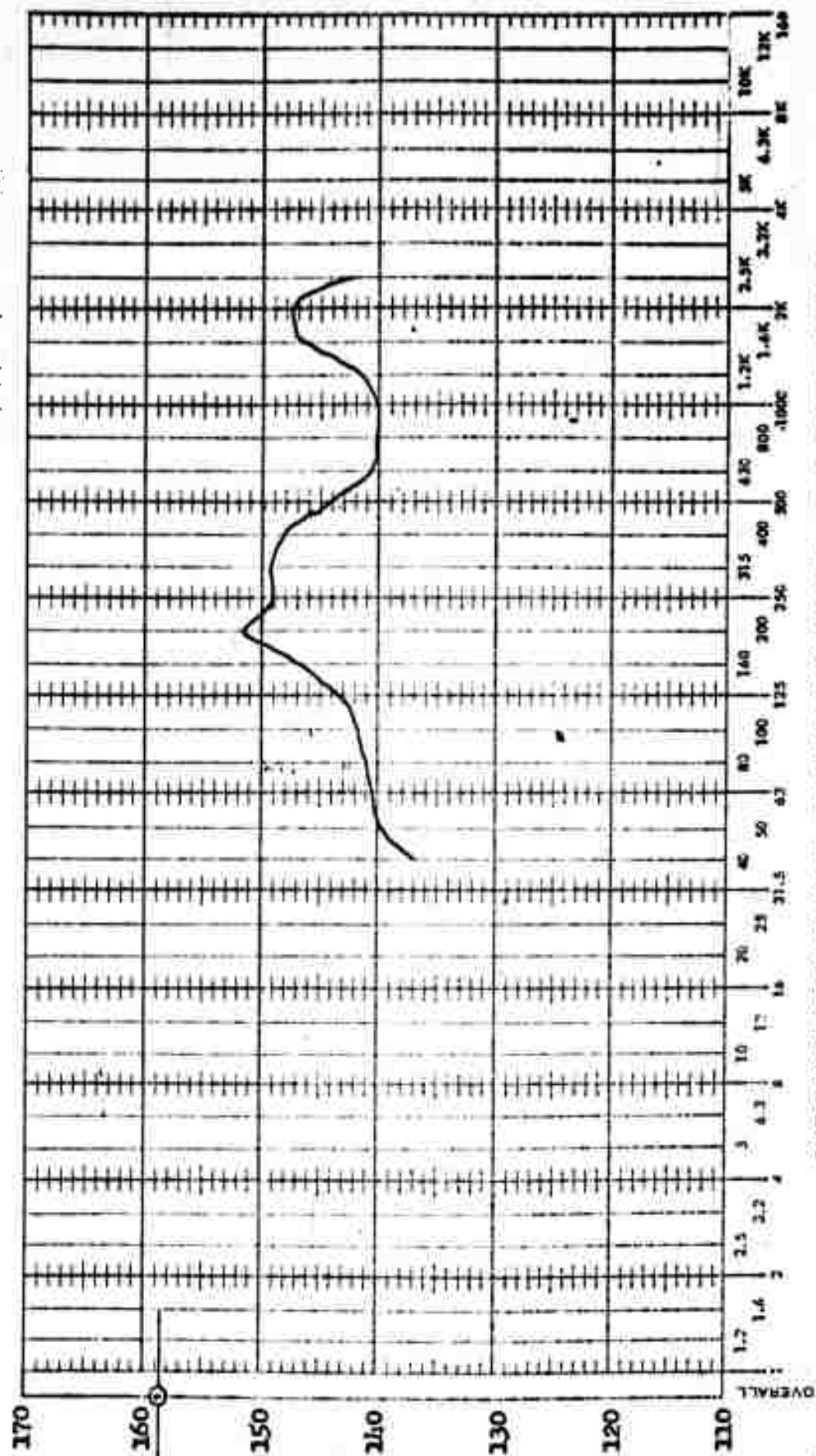
Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

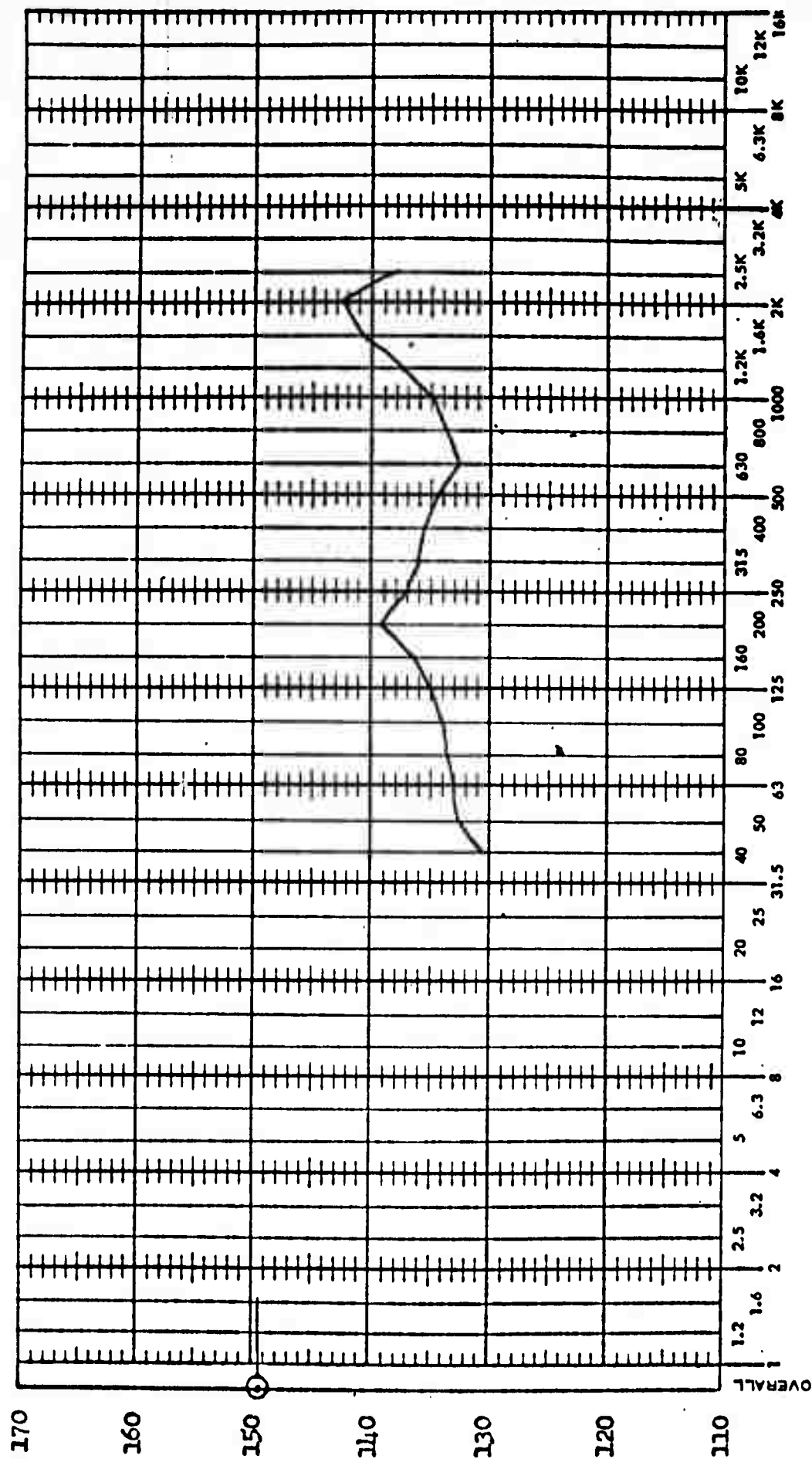
Test Point #8 Mach No. 0.84 Correlation No. 165

$\alpha = -4^\circ$

Figure 14 (Continued)



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #8 Mach No. 0.88 Correlation No. 471

Figure 14 (Continued)

$\alpha = -4^\circ$   $\beta = -4^\circ$

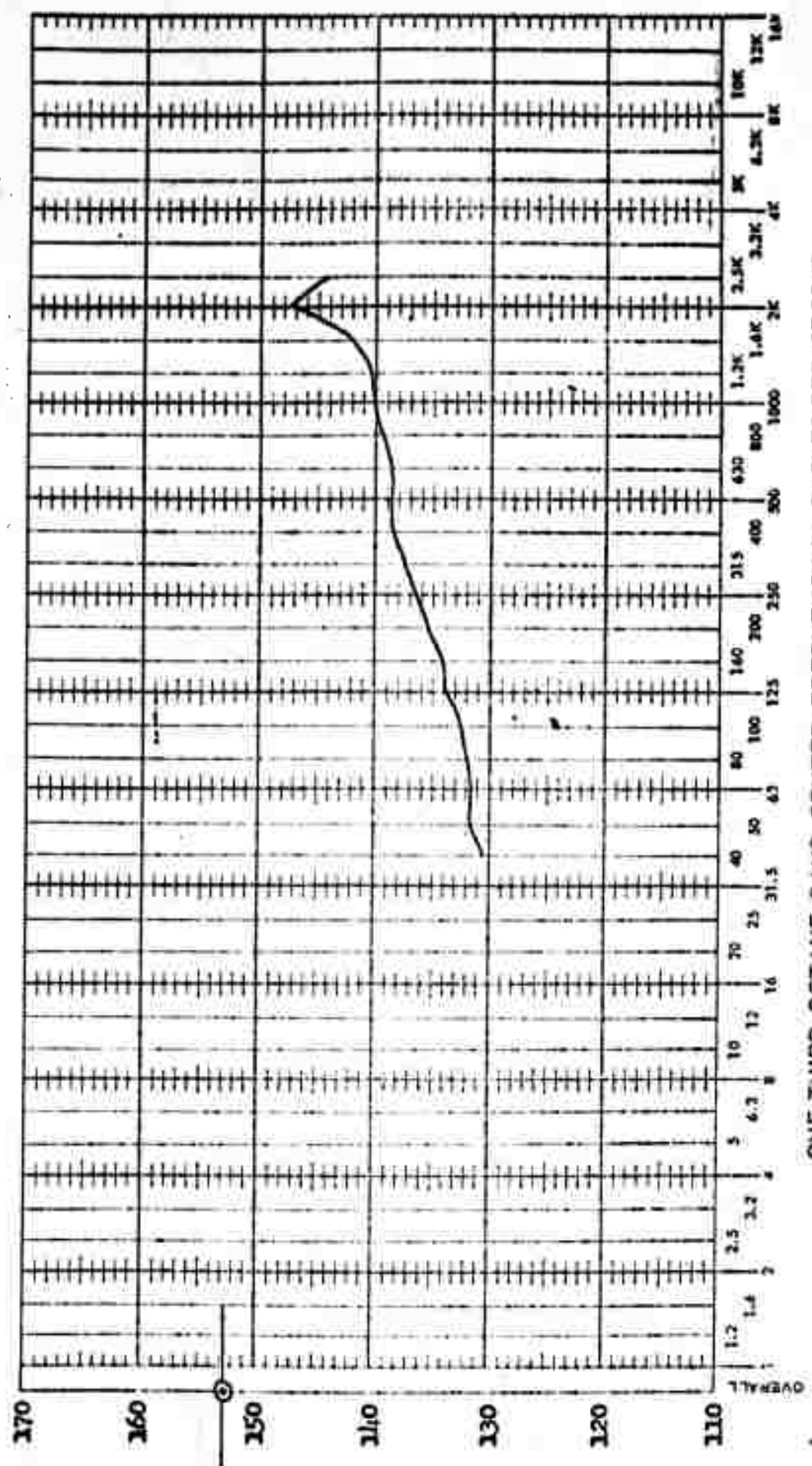
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



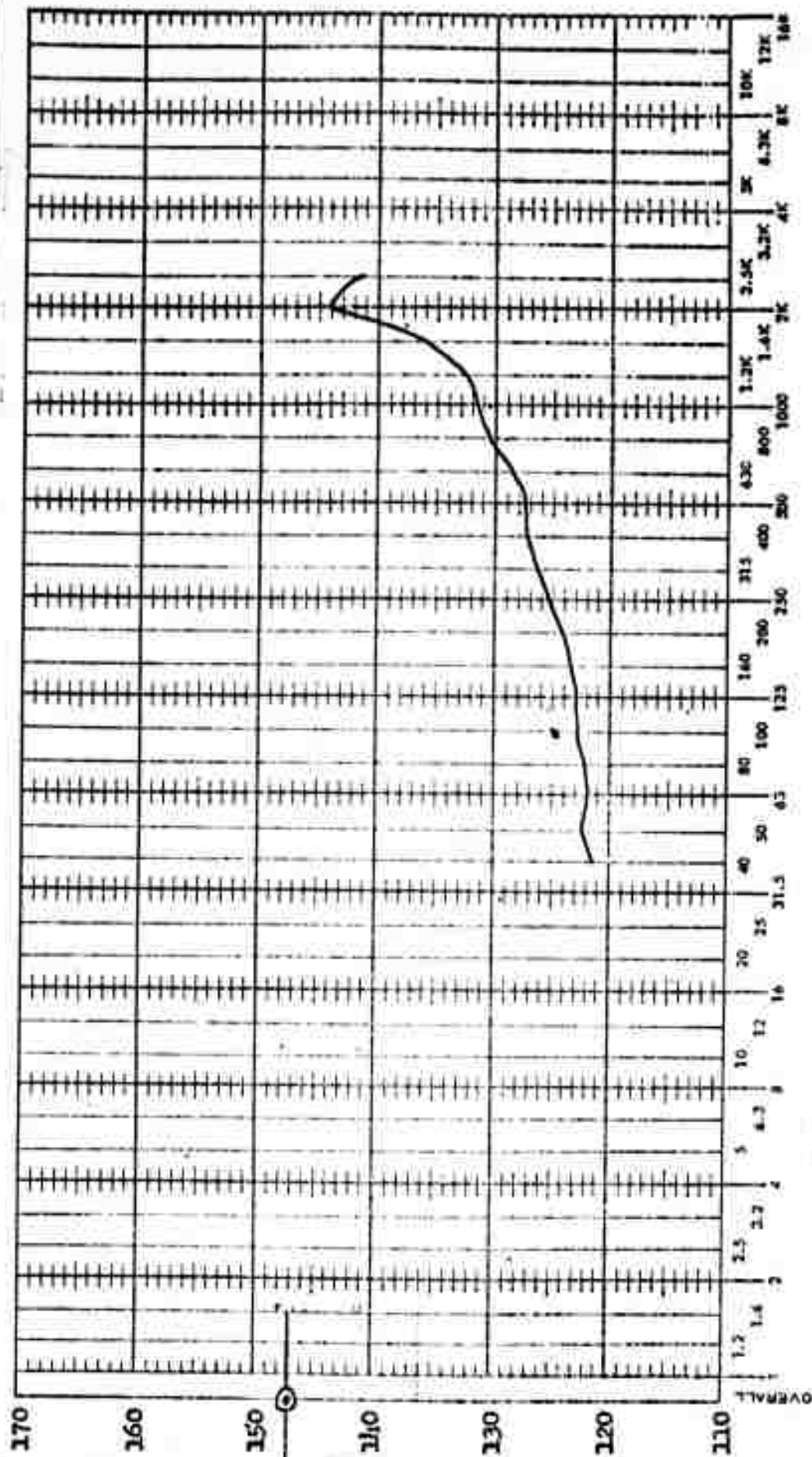
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #9 Mach No. 1.06 Correlation No. 198

Figure 14 (Continued)

$\alpha = 4^\circ$   $\beta = -4^\circ$

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #9 Mach No 1.06 Correlation No. 499

$\alpha = 0$   $\beta = -4$

Figure 14 (Continued)

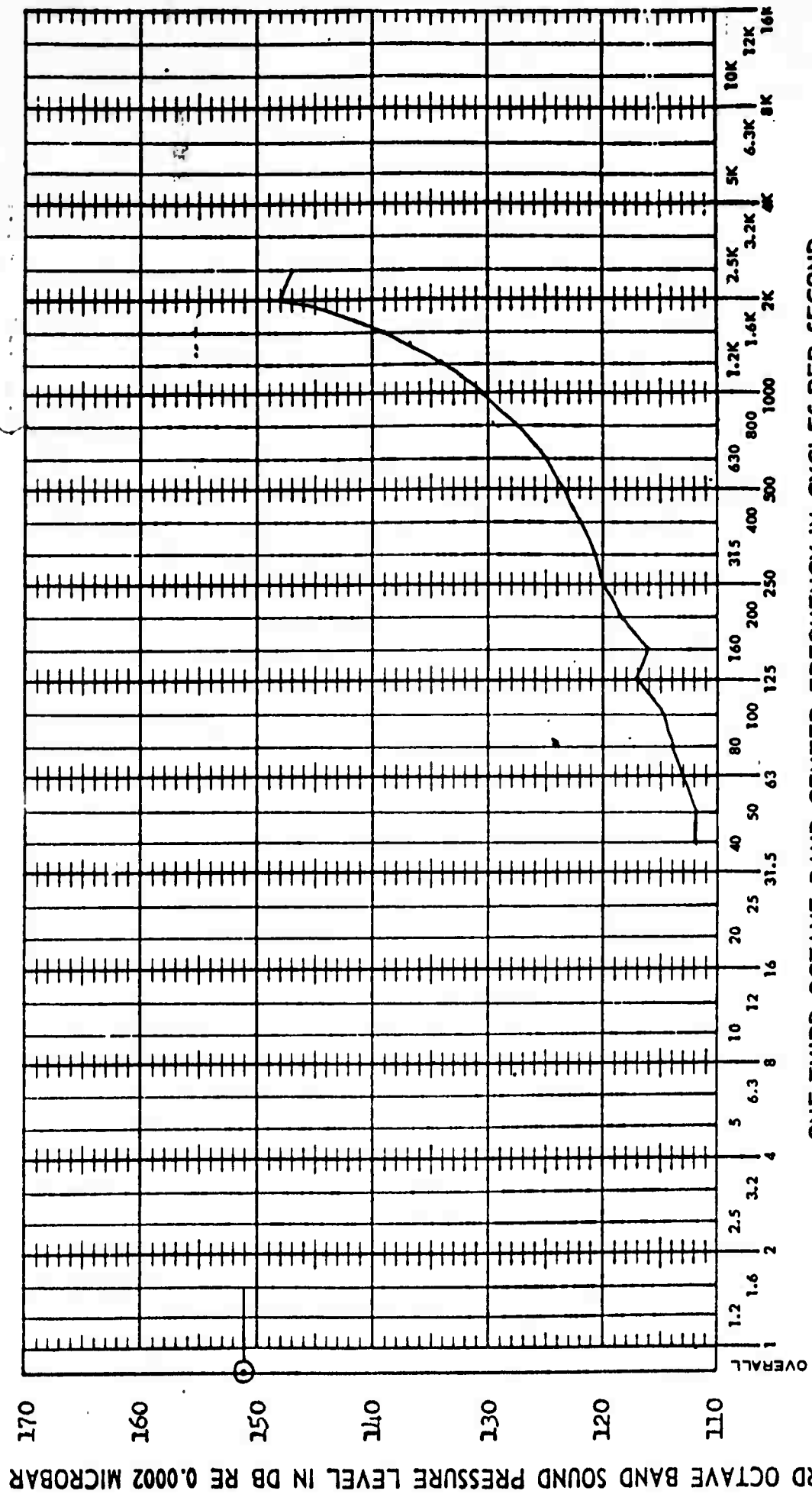
BOEING

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$\alpha = -4$   $\beta = -2$

Test Point #12 Mach No. 1.0 Correlation No. 489

Figure 14 (Continued)

BOEING

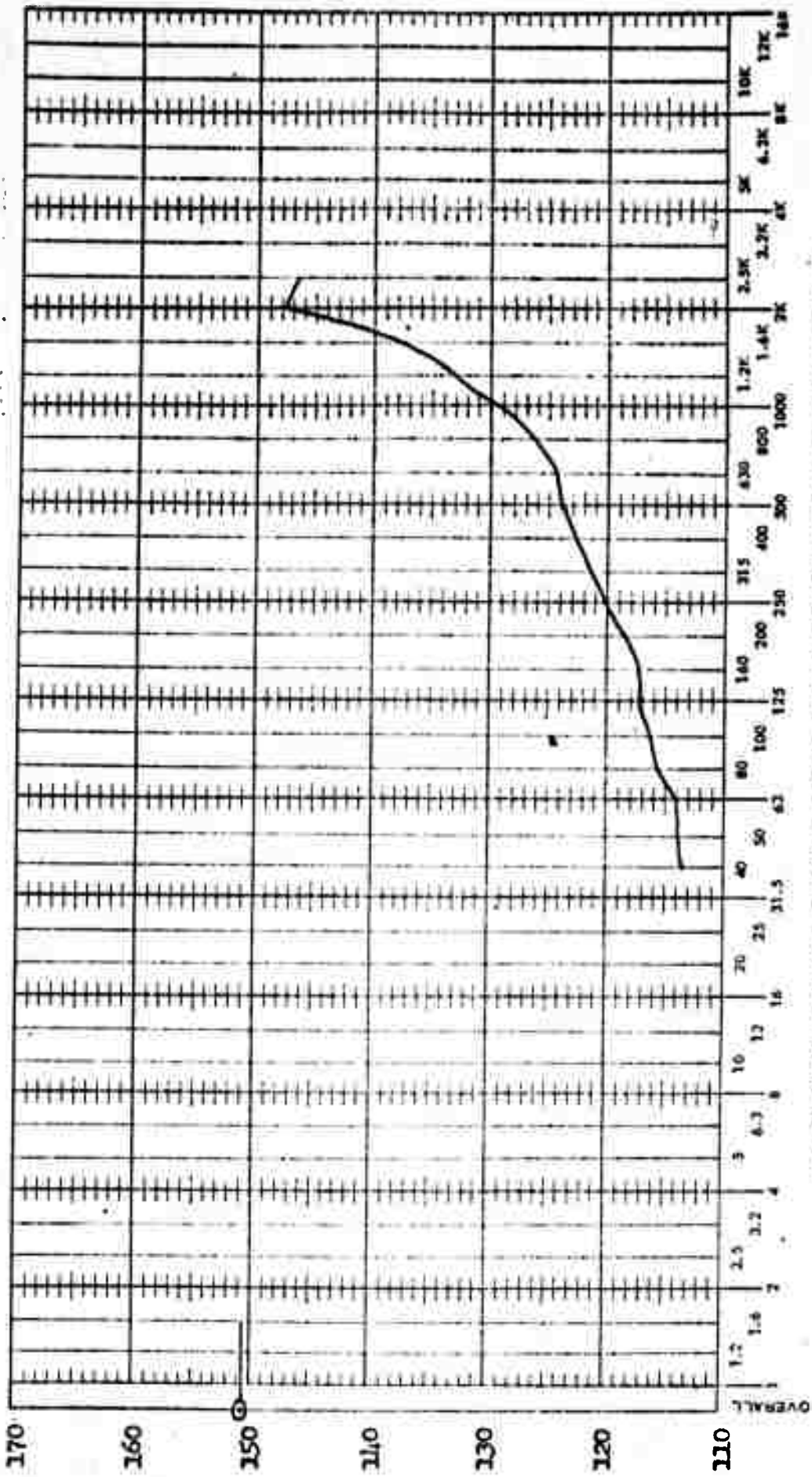
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #12 Mach No. 1.04 Correlation No. 495

Figure 14 (Continued)

$\alpha = -4^\circ$   $\beta = -4^\circ$

BOEING

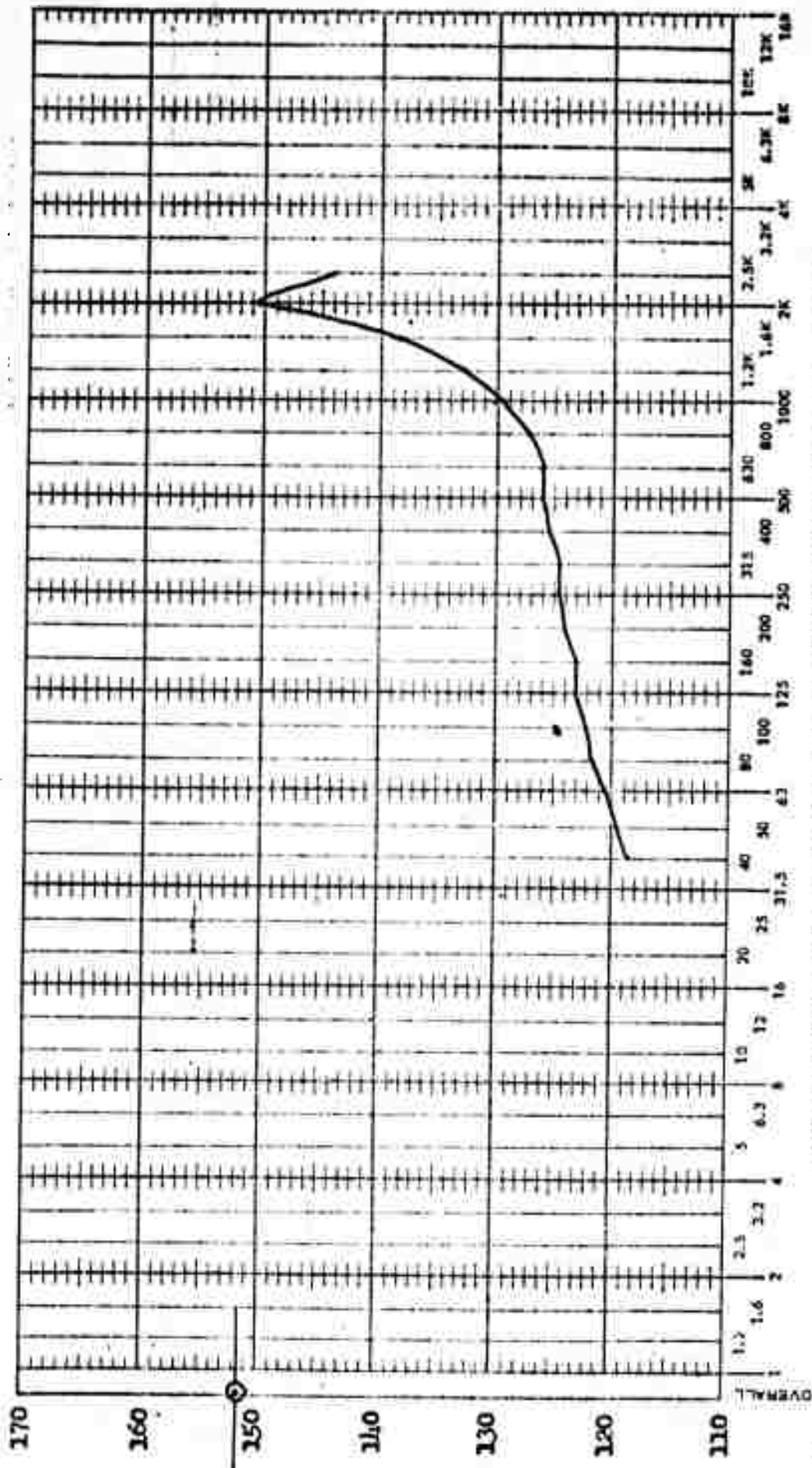
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #12 Mach No. 1.06 Correlation No. 500

$\delta = -4^\circ$   $\delta = -7^\circ$

Figure 14 (Continued)

BUENING

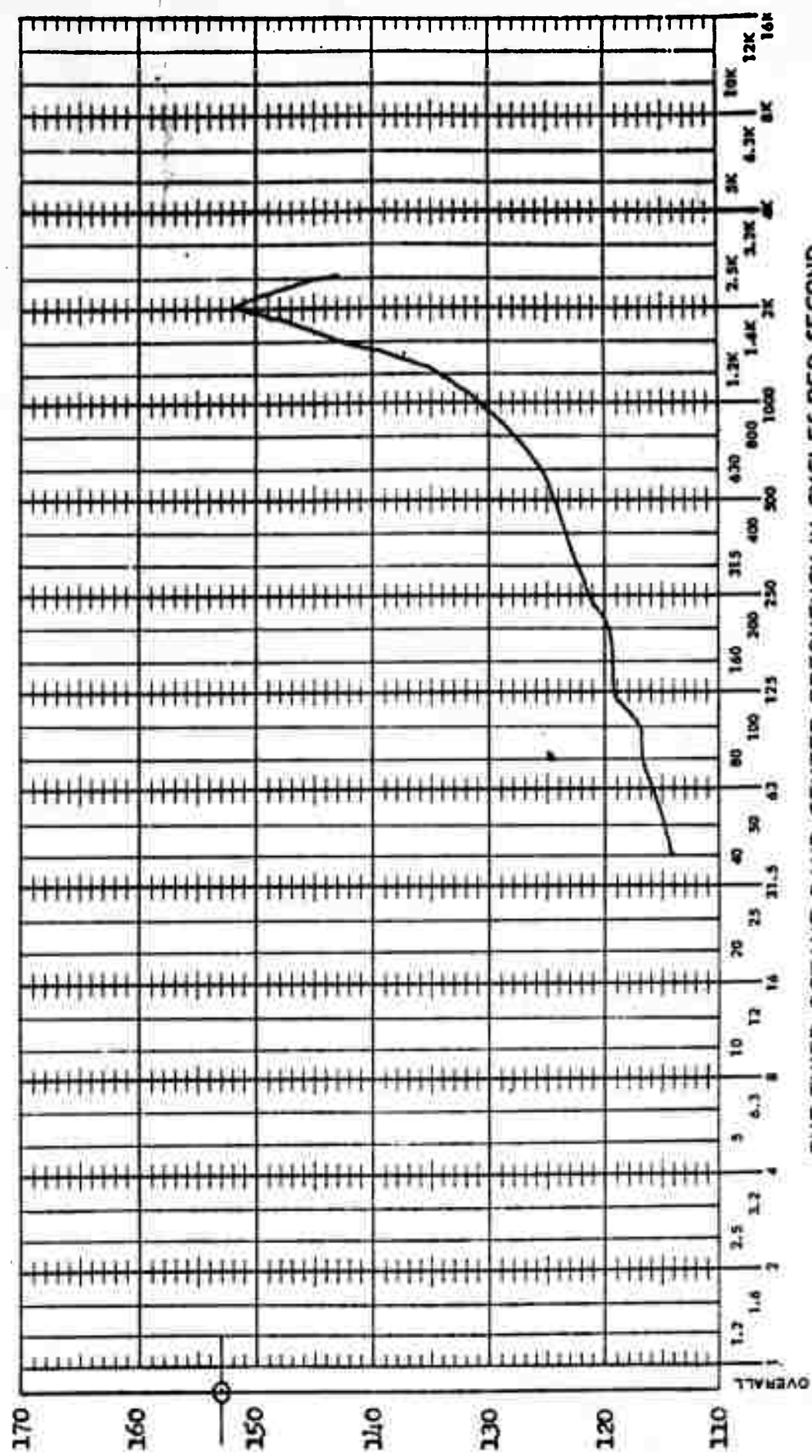
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

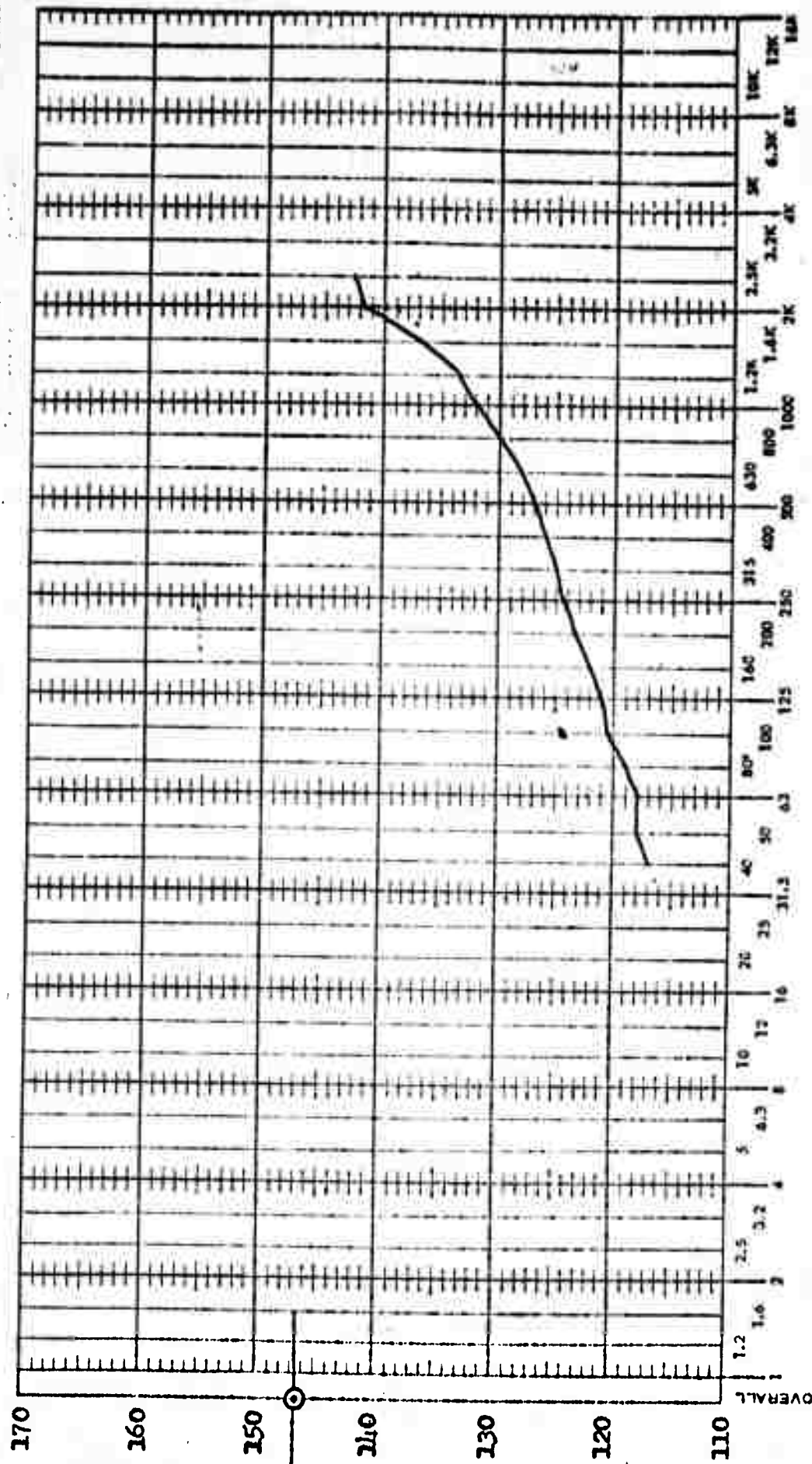
Test Point #12 Mach No. 1.08 Correlation No. 501

$\alpha = -4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #14 Mach No. 0.95 Correlation No. 401

Figure 14 (Continued)

$\alpha = 4^\circ$   $\beta = -4^\circ$

BOEING

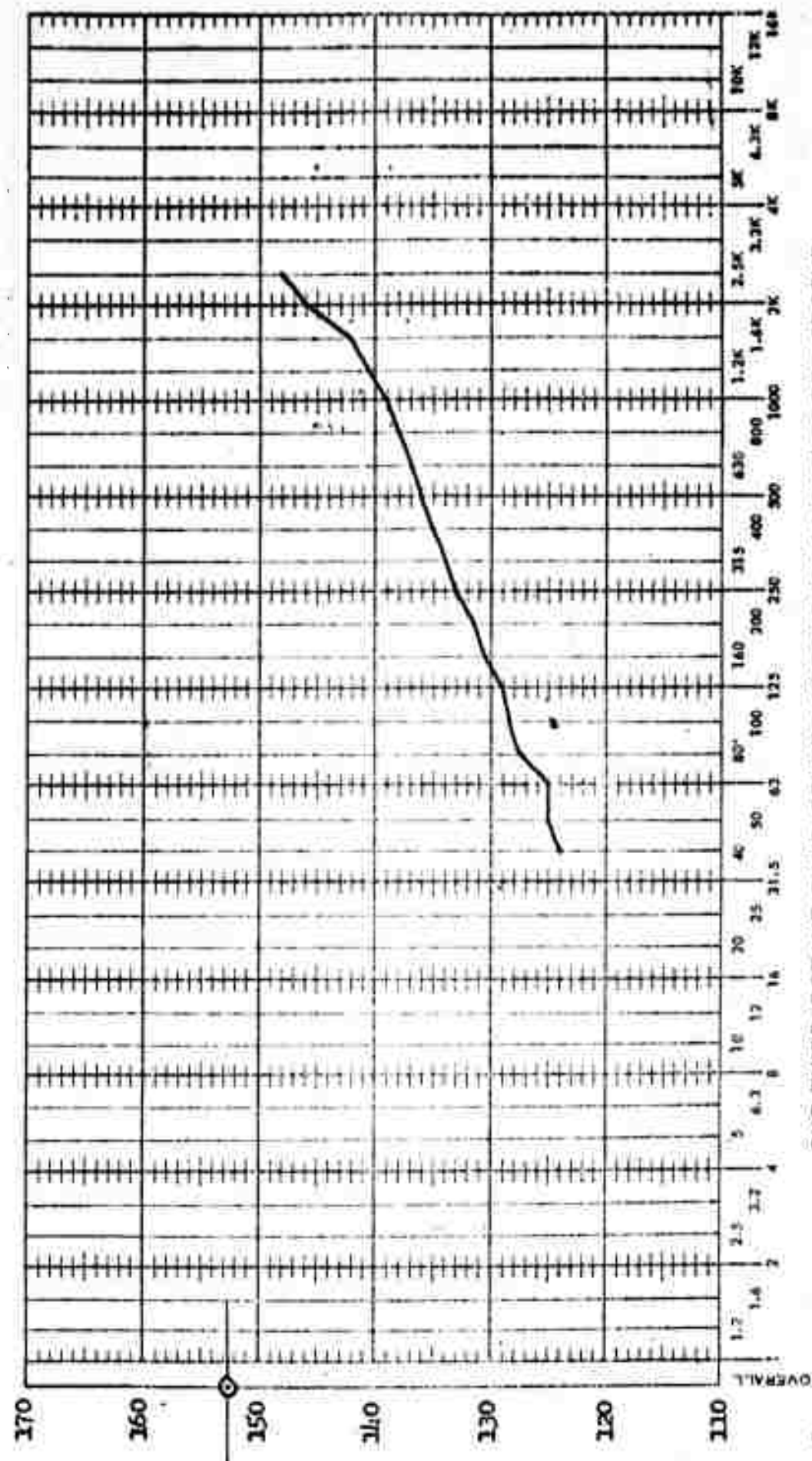
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

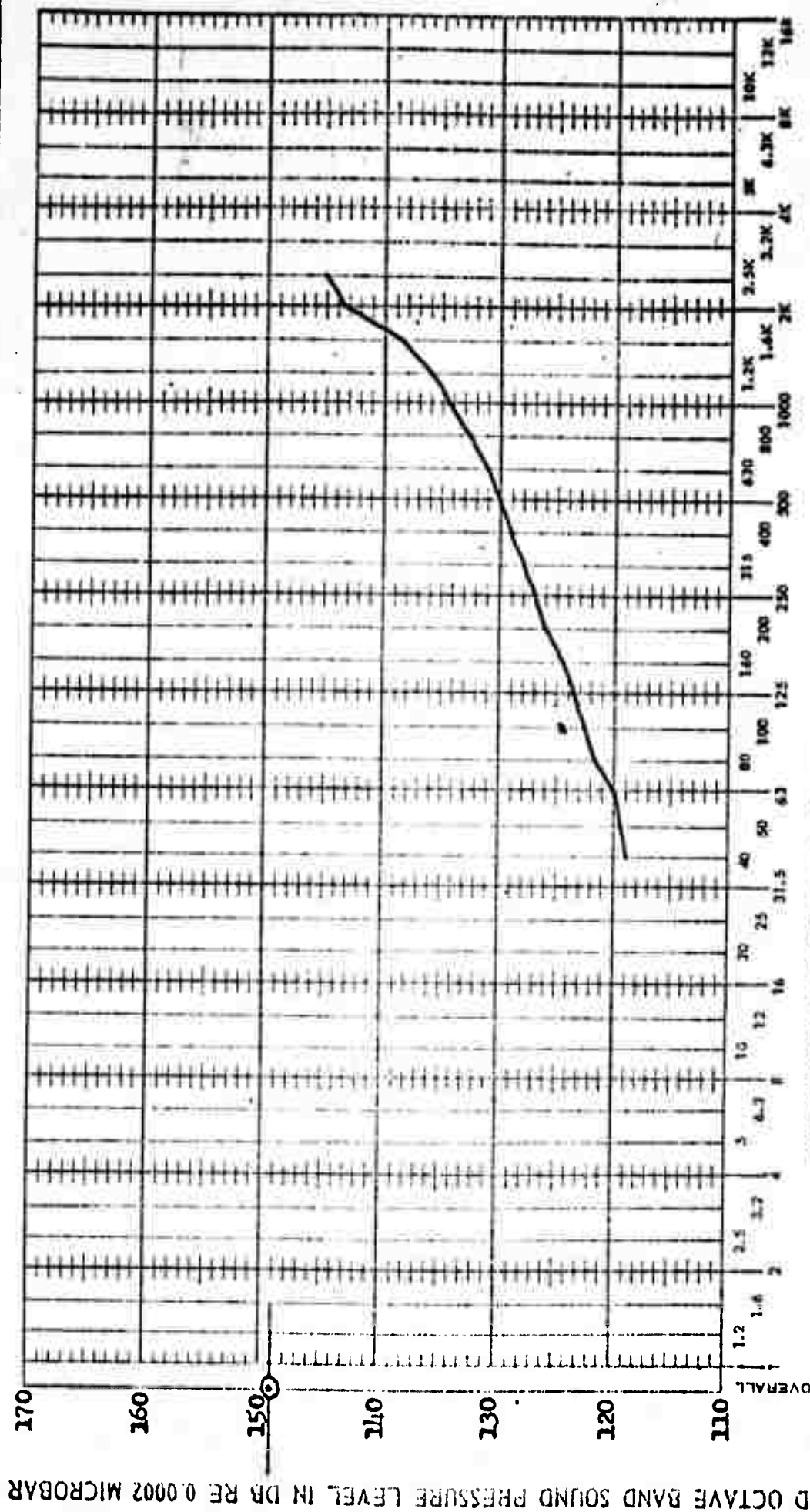
Test Point #74 Mach No. 1.06 Correlation No. 408

$\alpha = 4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

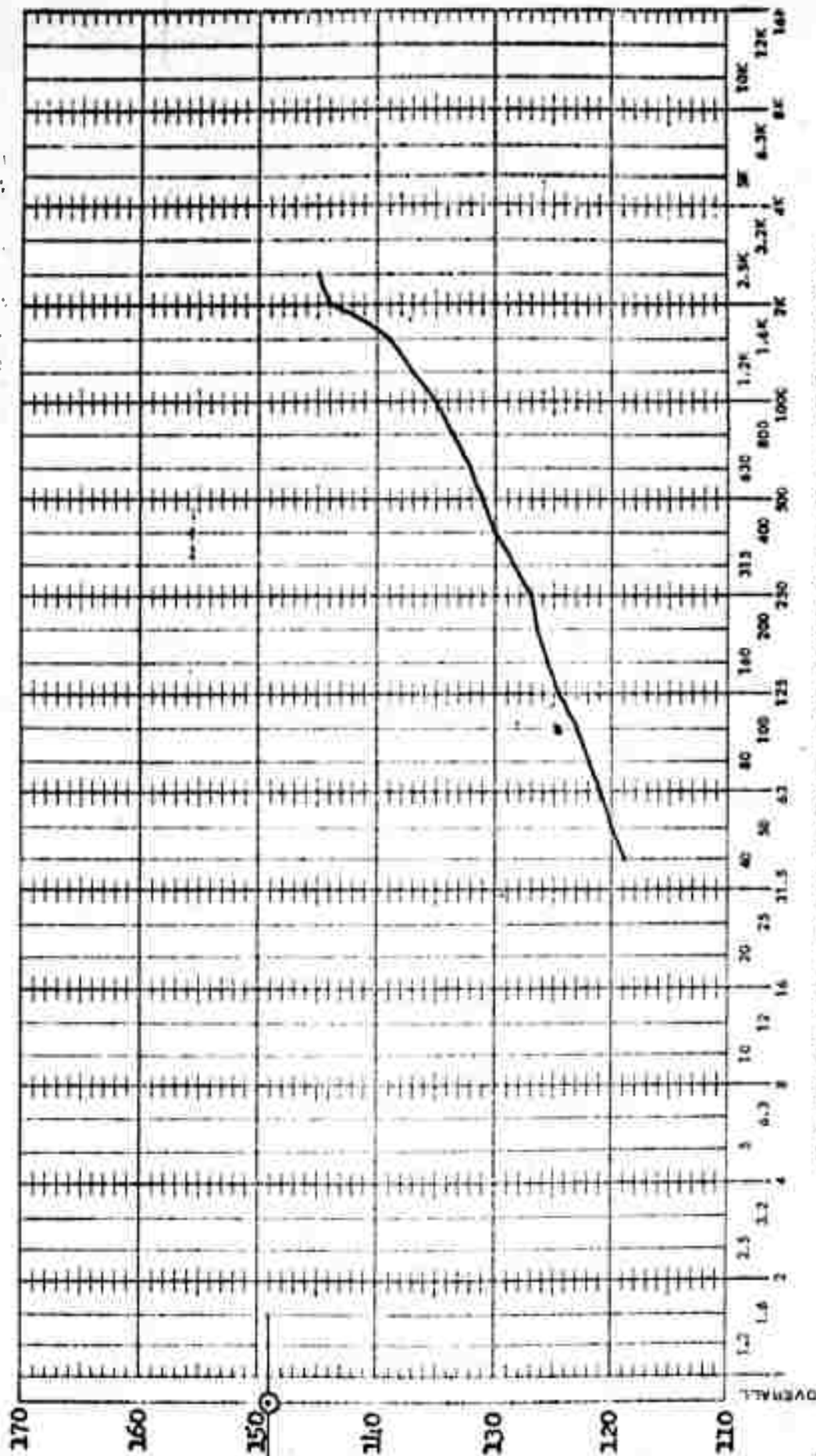
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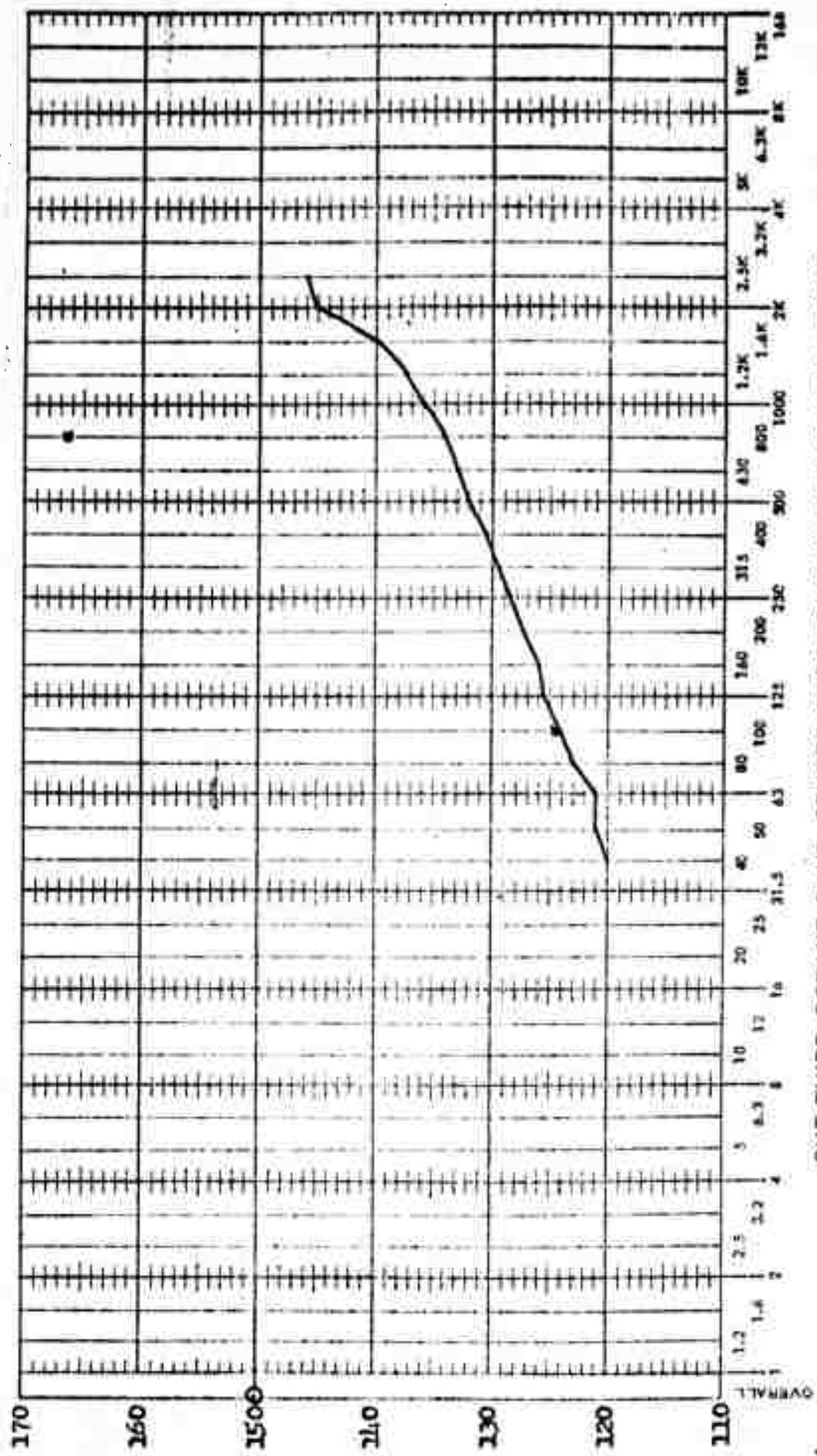




ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

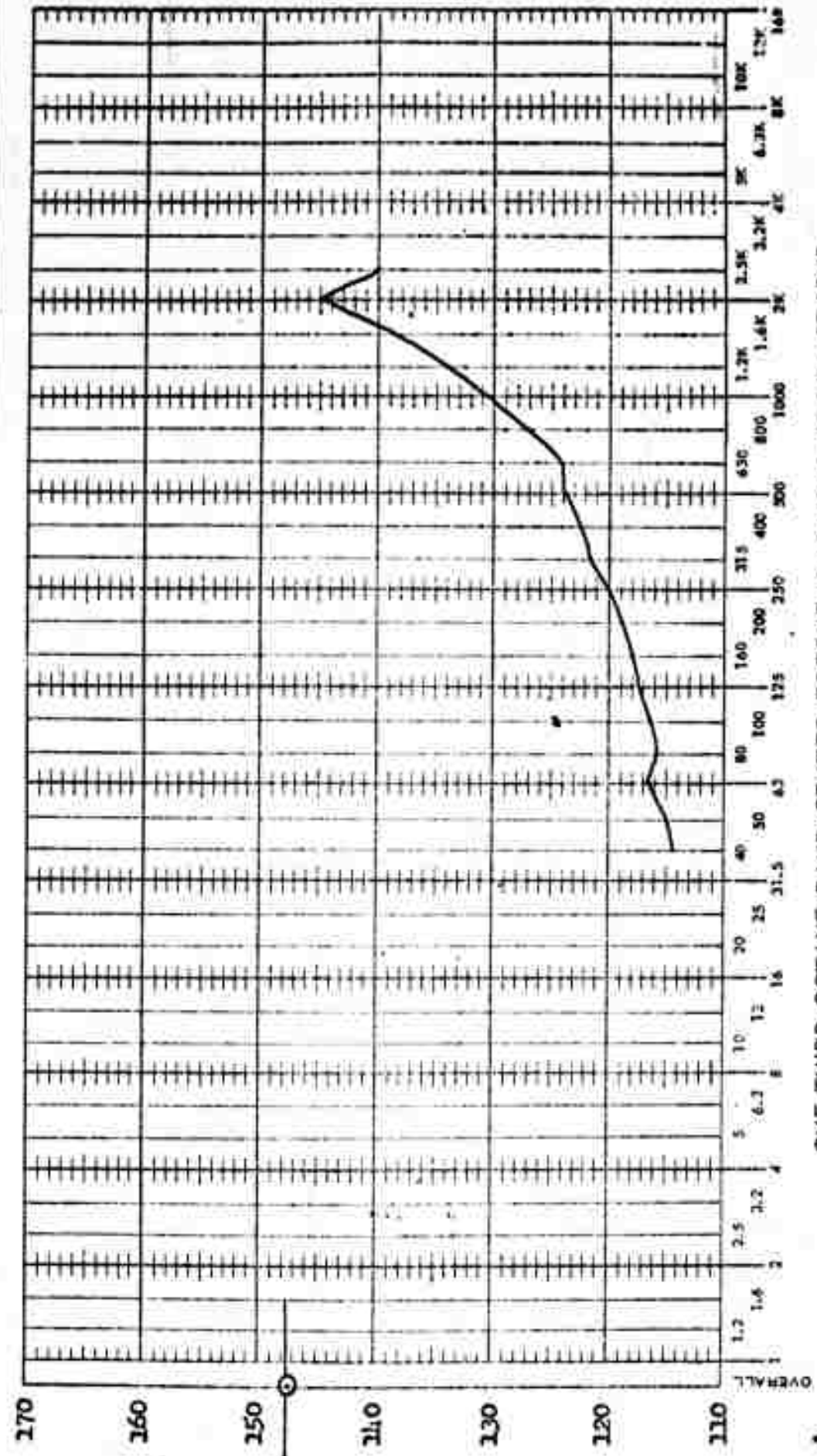
Test Point #114 Mach No. 1.04 Correlation No. 1497

$\alpha = 1^\circ$   $\beta = -2^\circ$

Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



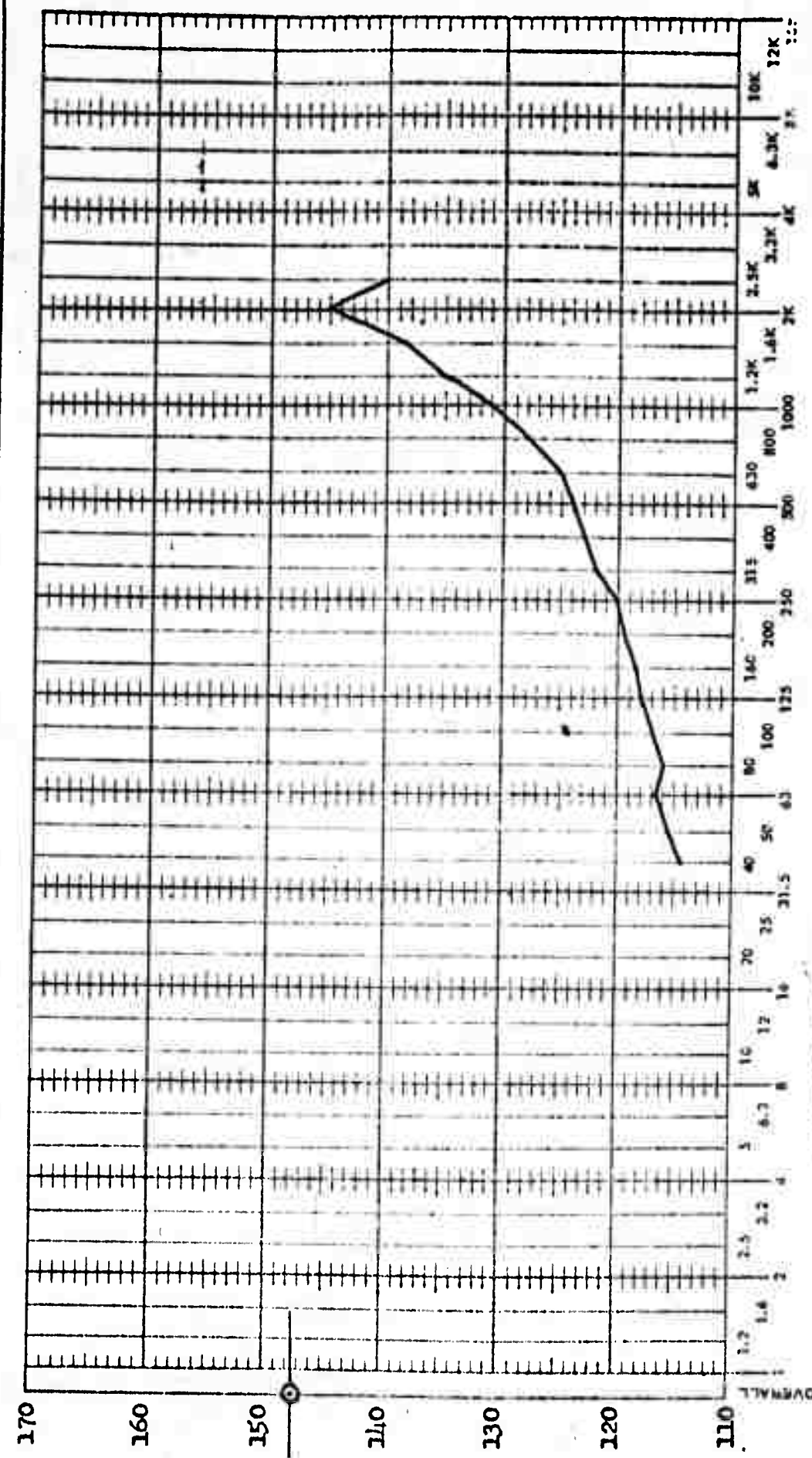
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #16 Mach No. 1.0 Correlation No. 491

$\alpha = 7^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #16 Mach No. 1.02 Correlation No. L92

$\alpha = 4^\circ$   $\beta = -4^\circ$

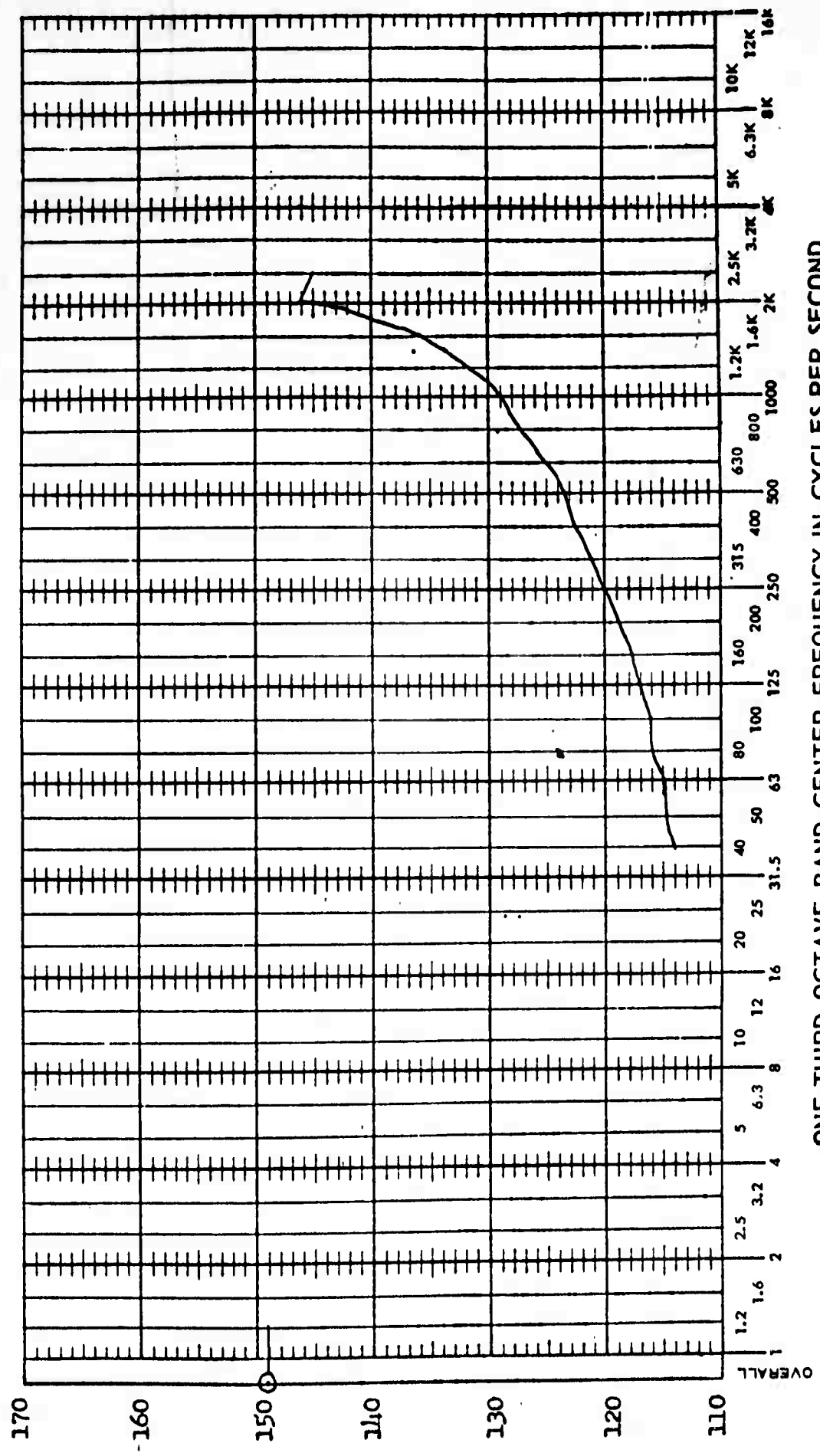
Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #7 Mach No. 1.08 Correlation No. 502

Figure 14 (Continued)

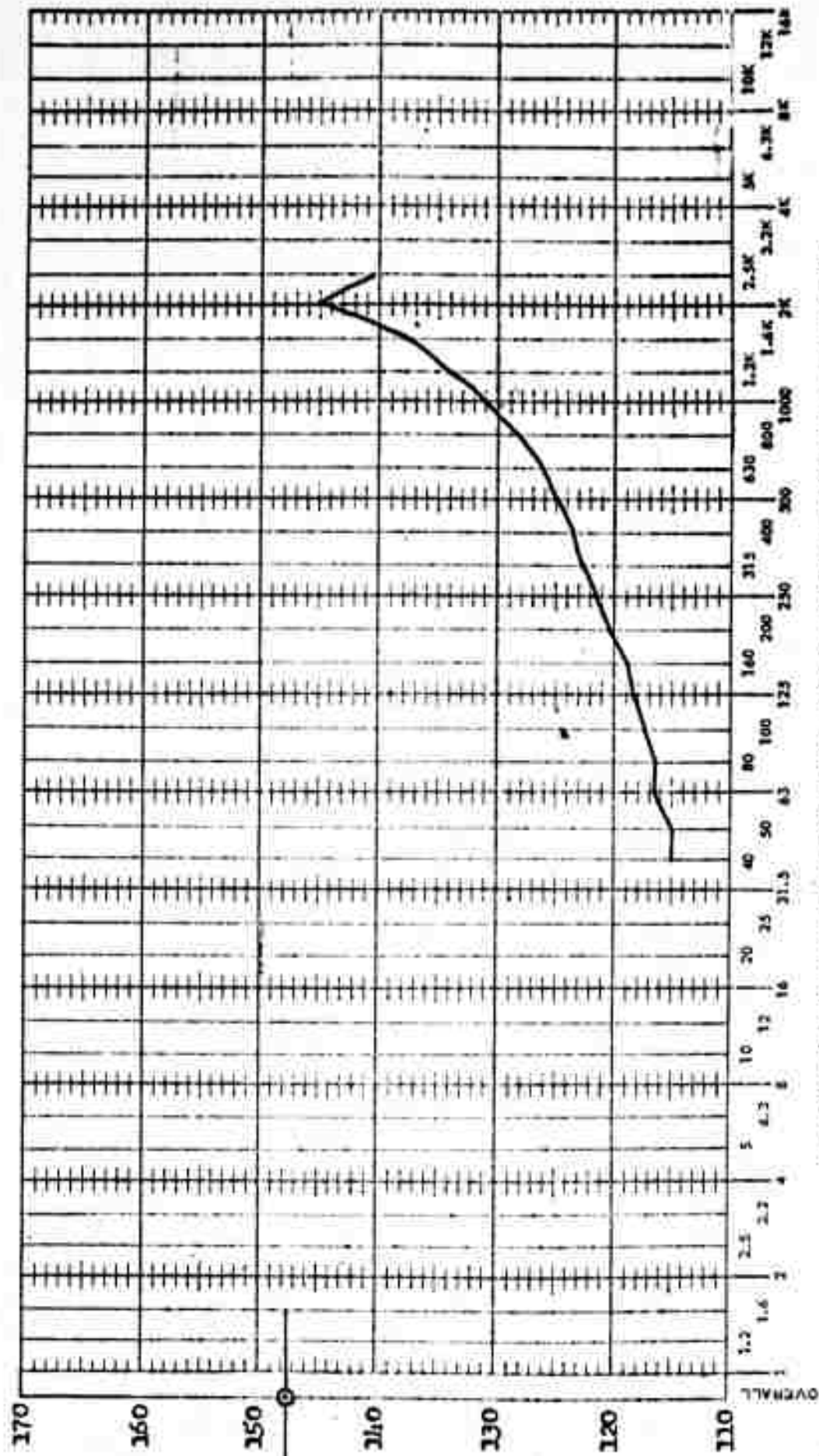
$\alpha = 0^\circ$   $\beta = -2^\circ$





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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #16 Mach No. 1.01 Correlation No. 497

(Figure 14 (Continued))

$\alpha = 4^\circ$   $\beta = -4^\circ$

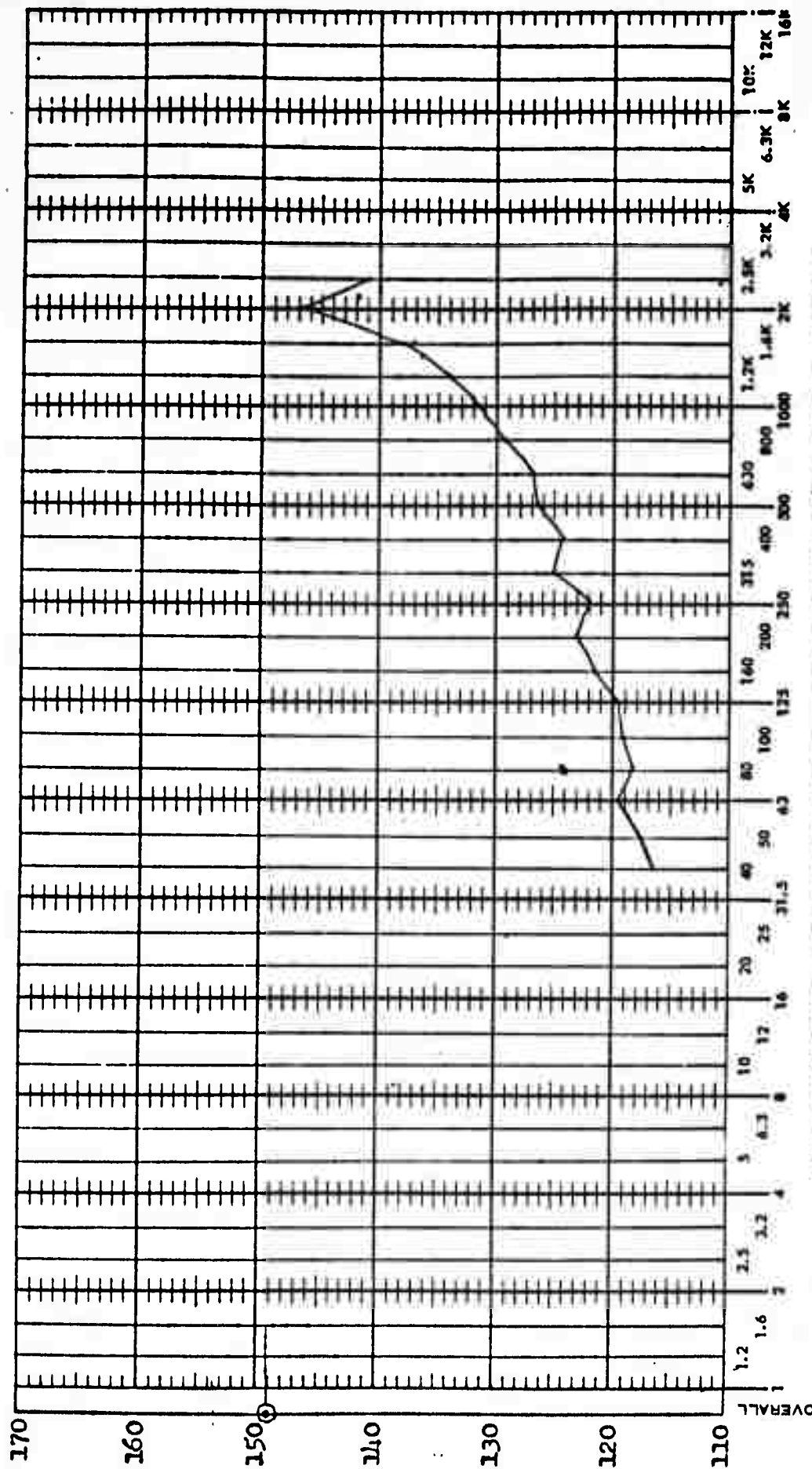
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #18 Mach No. 0.95 Correlation No. 315

Figure 14 (Continued)

$\alpha = 4^\circ$   $\beta = 0^\circ$

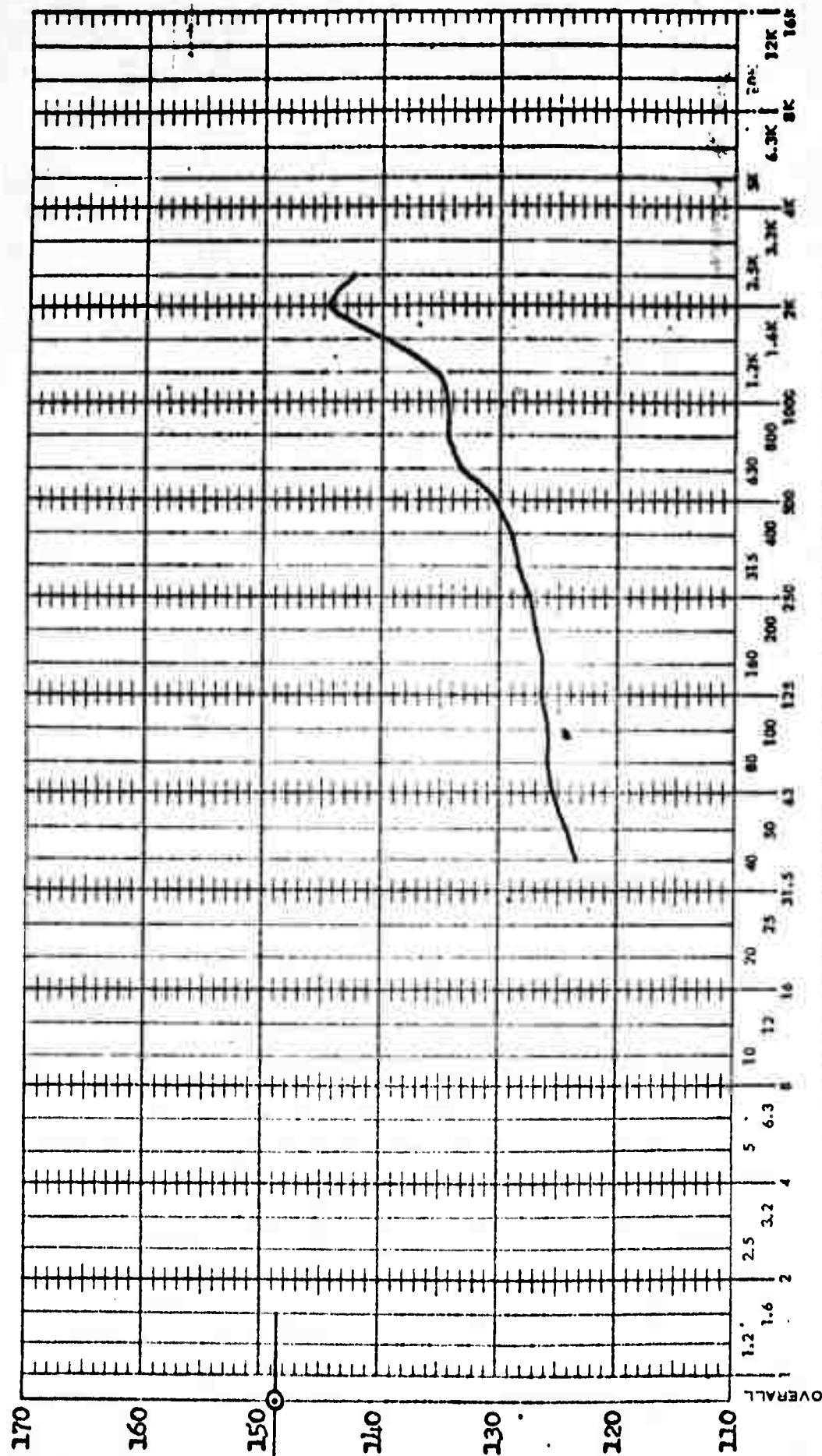
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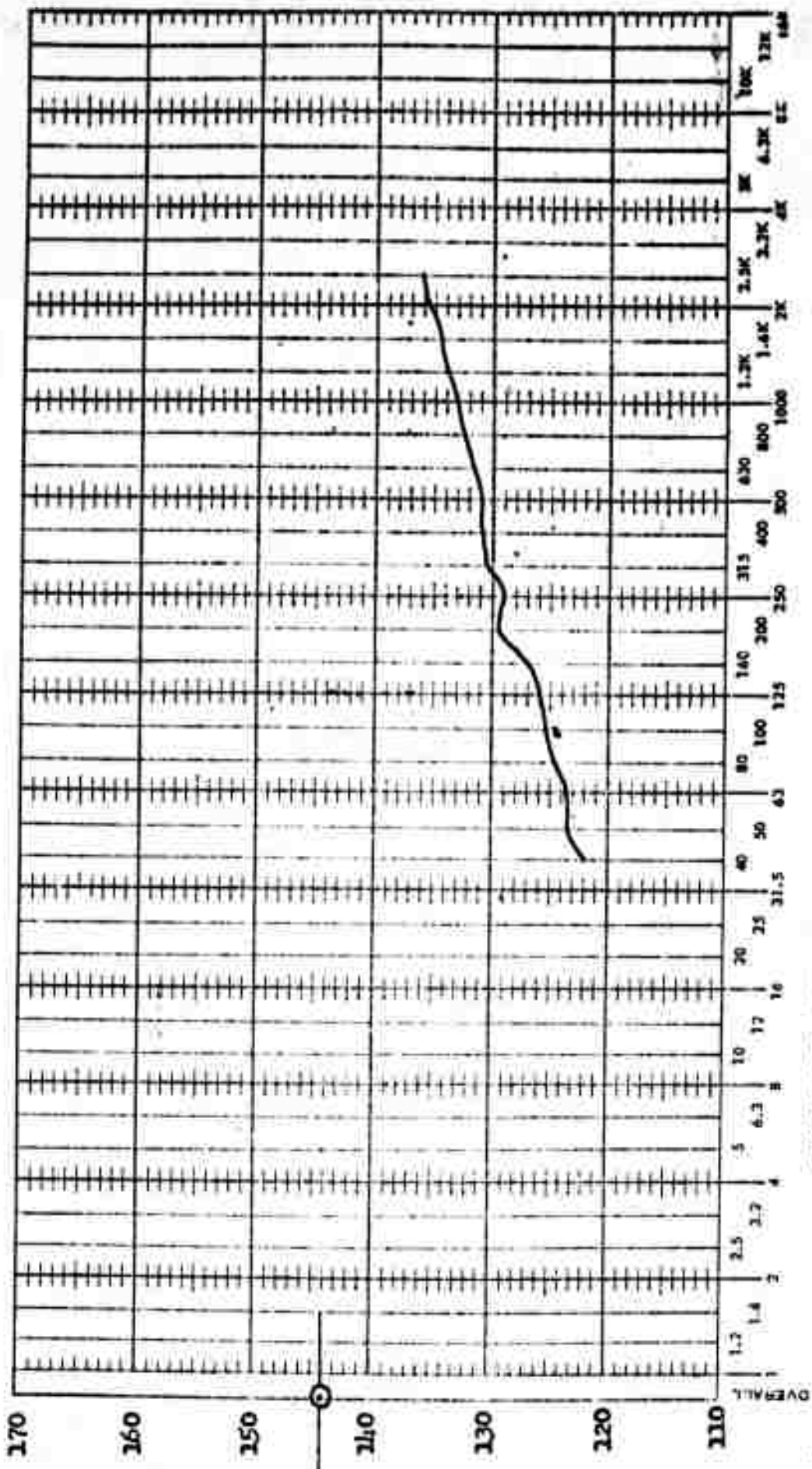


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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

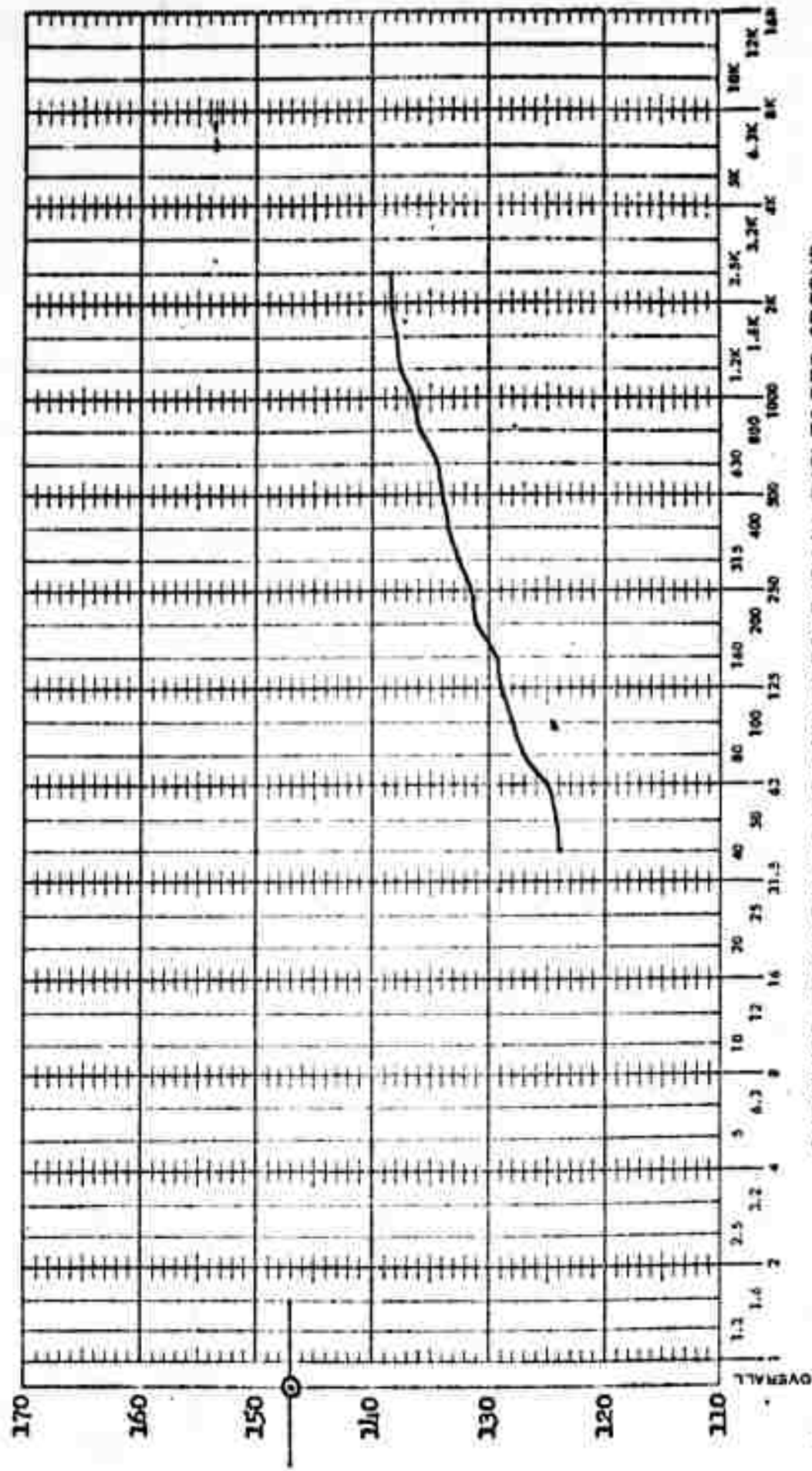
Test Point #20 Mach No. 0.8 Correlation No. 356

$\alpha = 0$   $\beta = -4$

Figure 14 (Continued)



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #20 Mach No. 0.85 Correlation No. 357

Figure 14 (Continued)

$\alpha = 0^\circ$   $\beta = -4^\circ$

BOEING

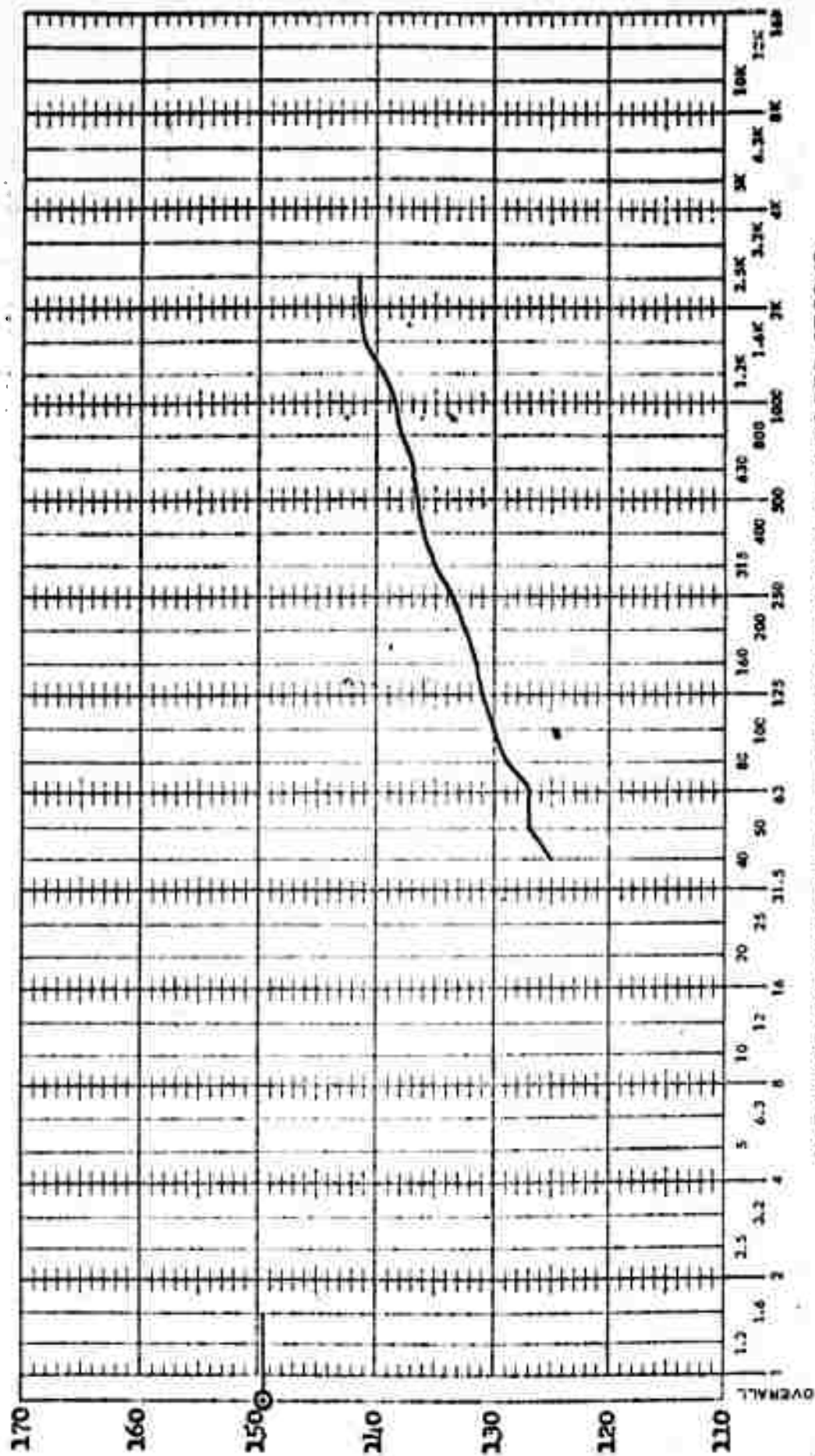
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #20 Mach No. 0.9 Correlation No. 360

$\alpha = 0$   $\beta = -7$

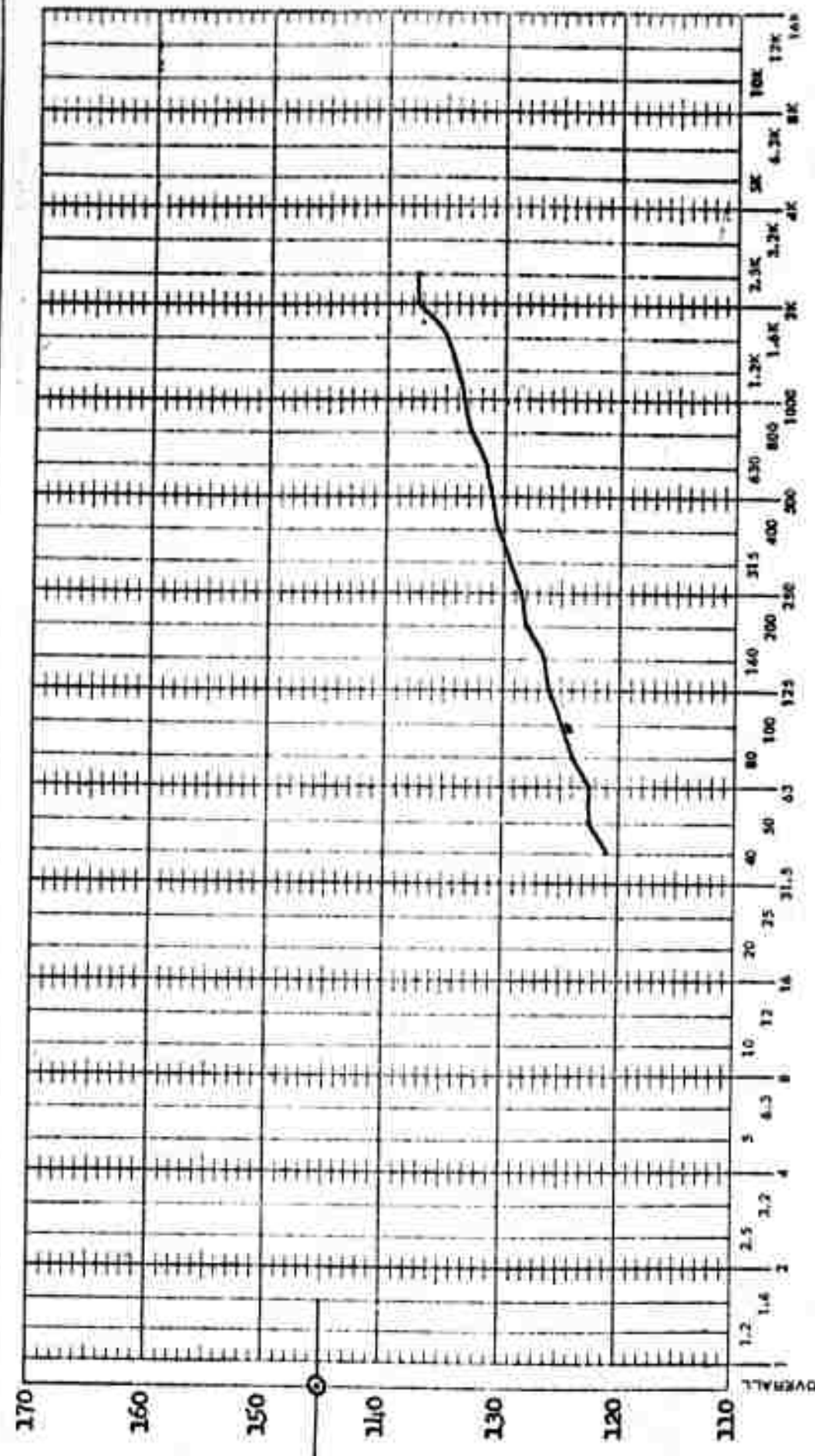
Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #20 Mach No. 0.95 Correlation No. 361

$\alpha = 0^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

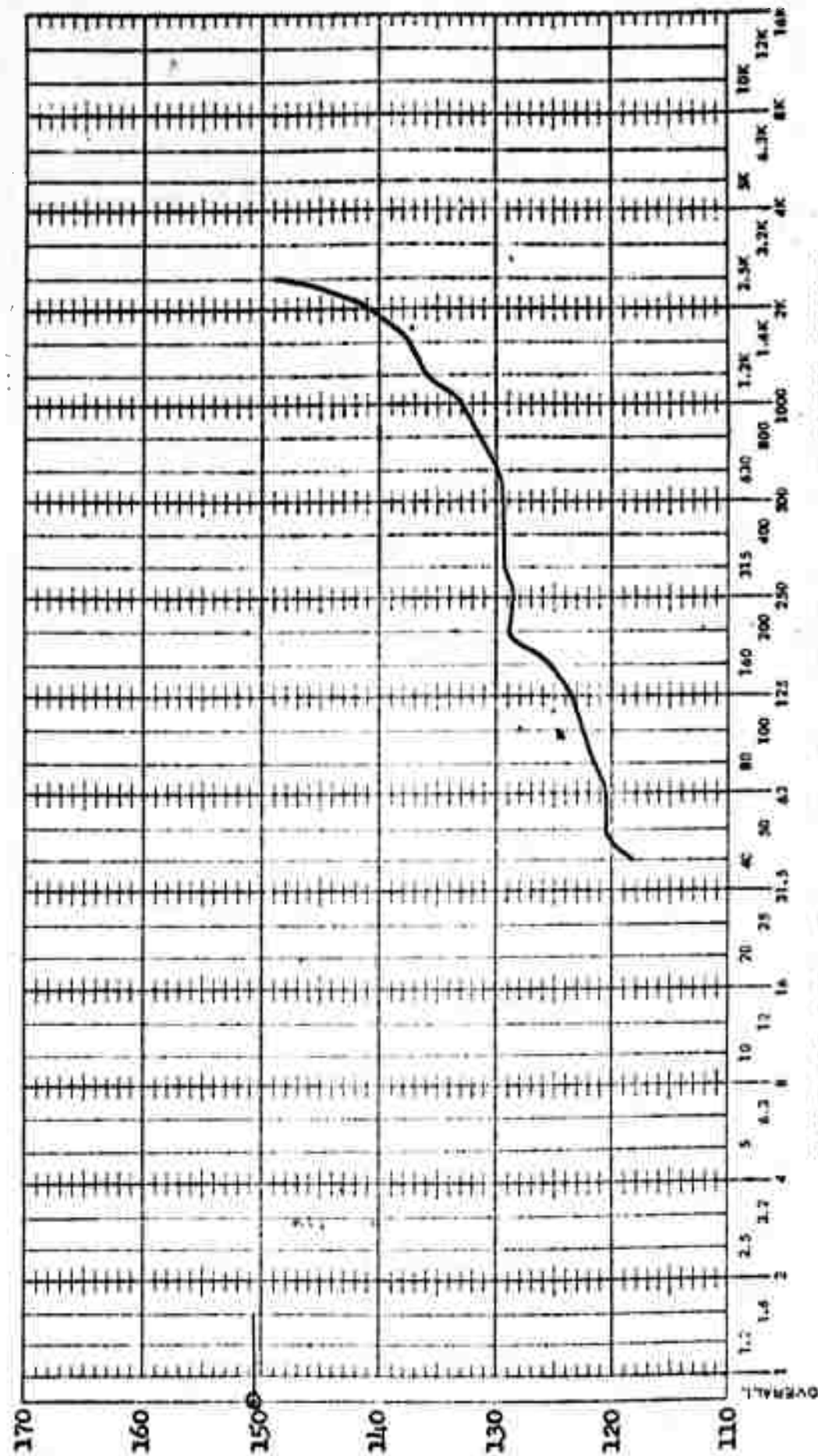
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #21 Mach No. 0.8 Correlation No. 356

$\alpha = 0^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

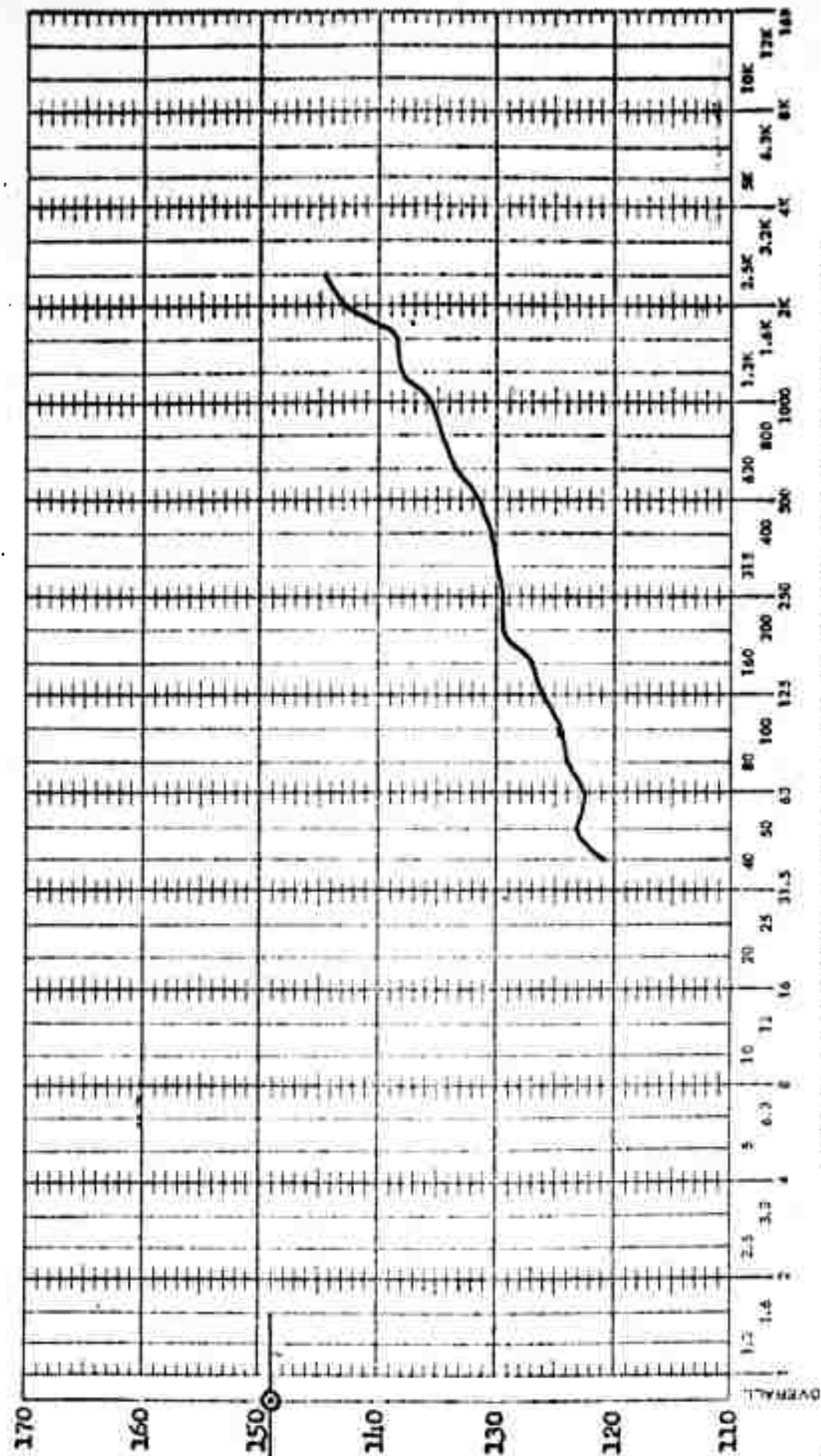
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #21 Mach No. 0.85 Correlation No. 357

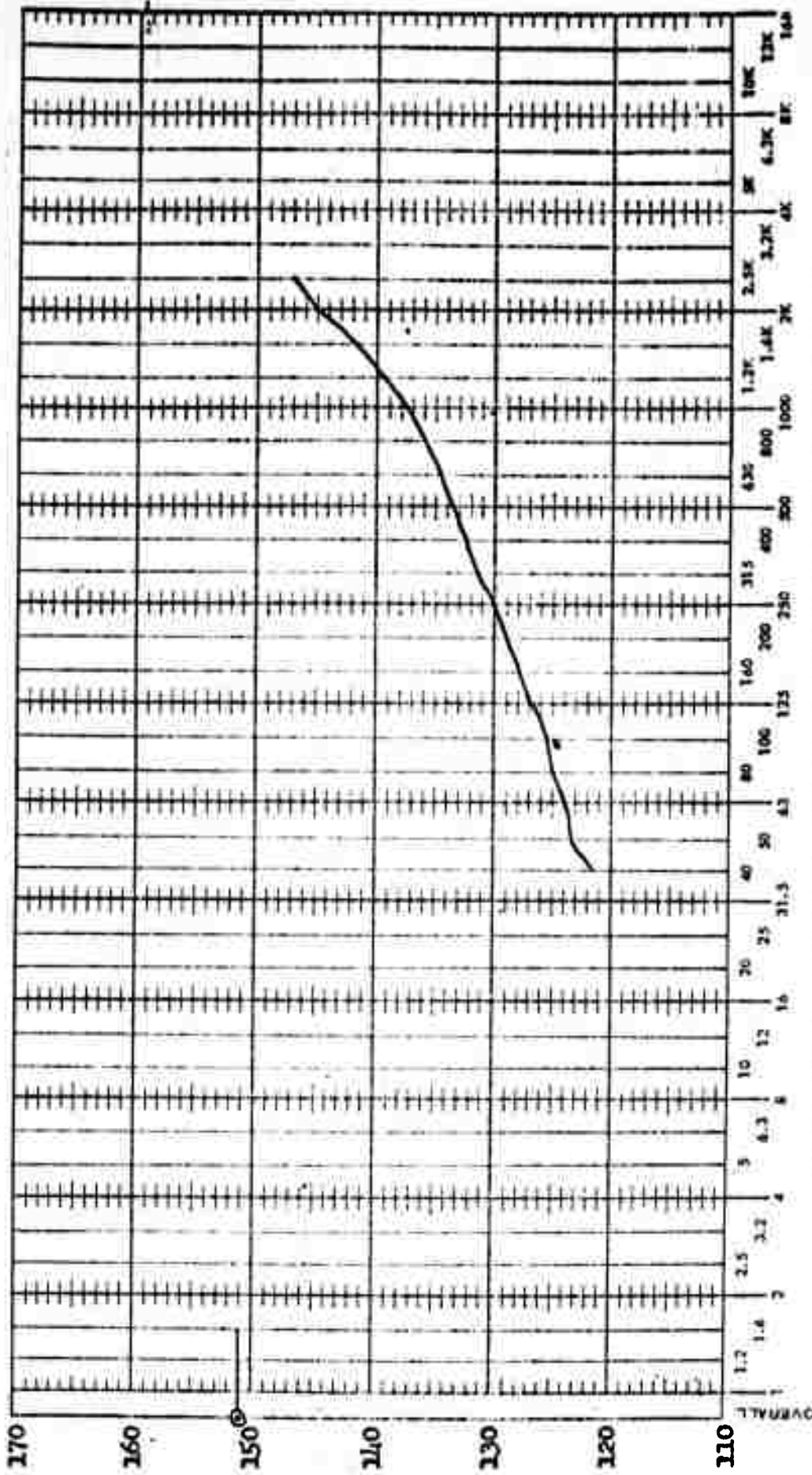
$\alpha = 0$   $\beta = -4$

Figure 14 (Continued)



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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

$\alpha = 0^\circ$   $\beta = -4^\circ$

Test Point #21 Mach No. 0.9 Correlation No. 360

Figure 14 (Continued)

BOEING

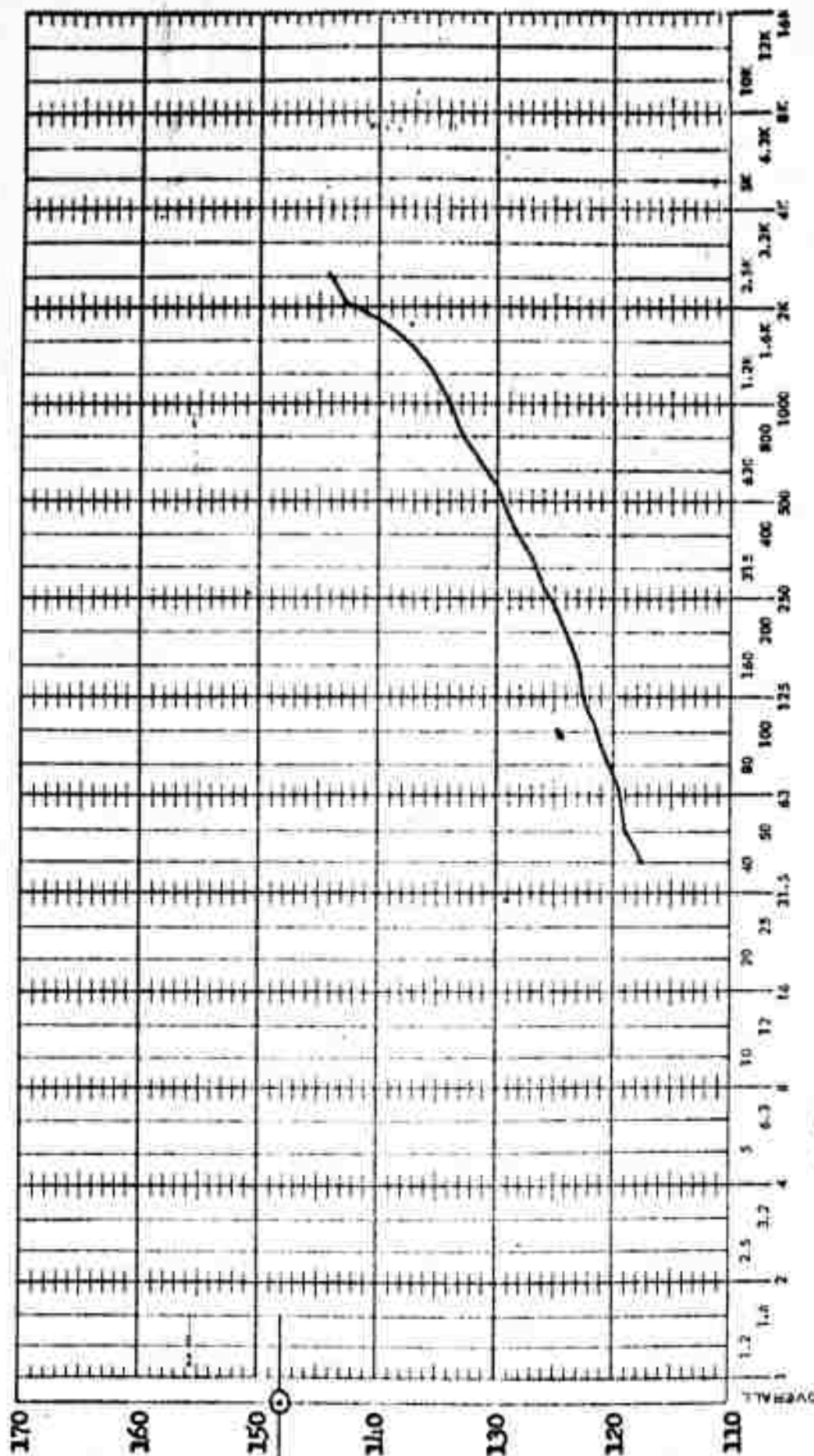
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #21 Mach No. 0.95 Correlation No. 361

Figure 14 (Continued)

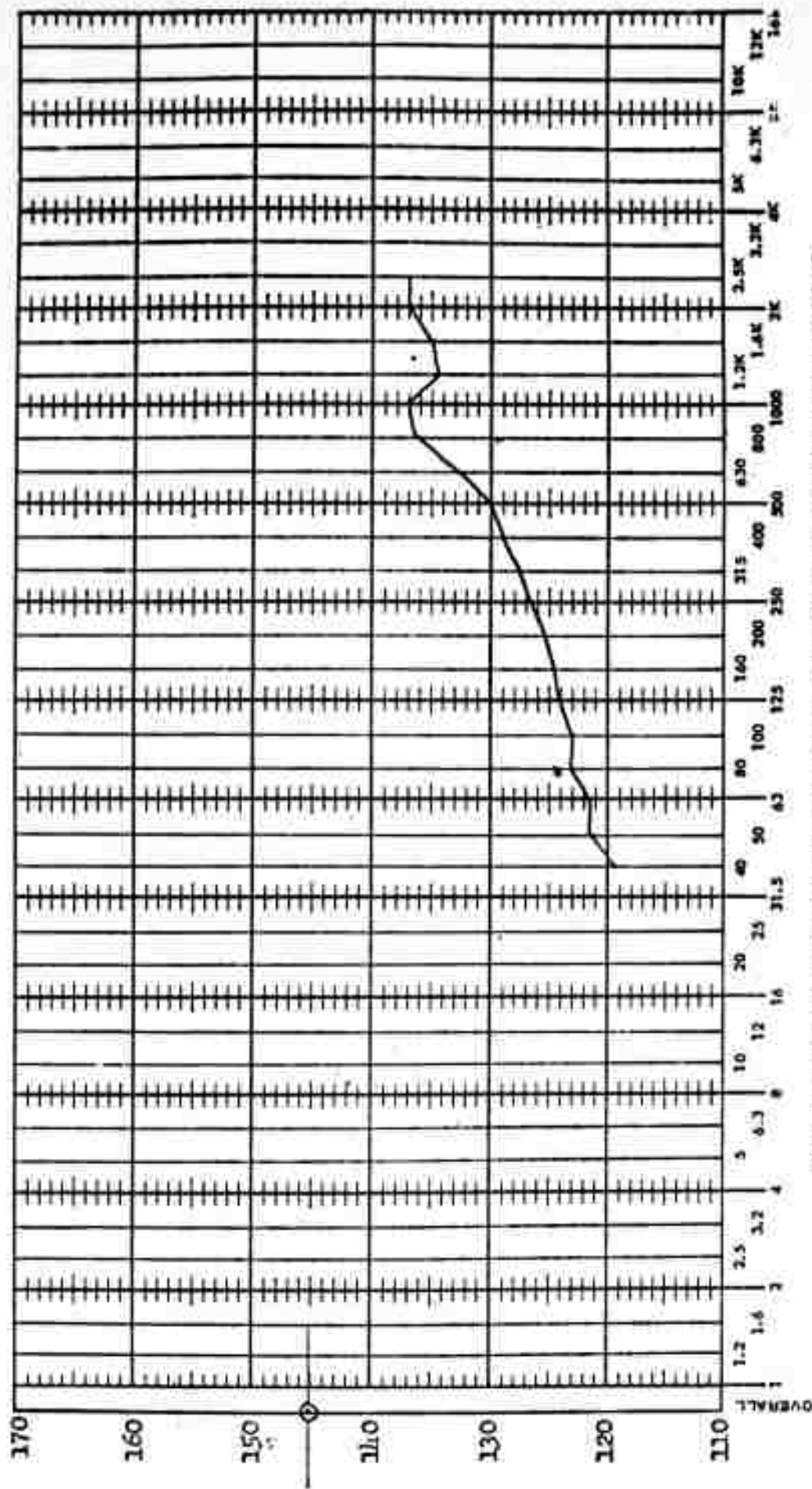
$\alpha = 0$   $\beta = -4$

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #23 Mach No. 0.9 Correlation No. 359

$\alpha = 0^\circ$   $\beta = 4^\circ$

Figure 14 (Continued)

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NO. T2-2648

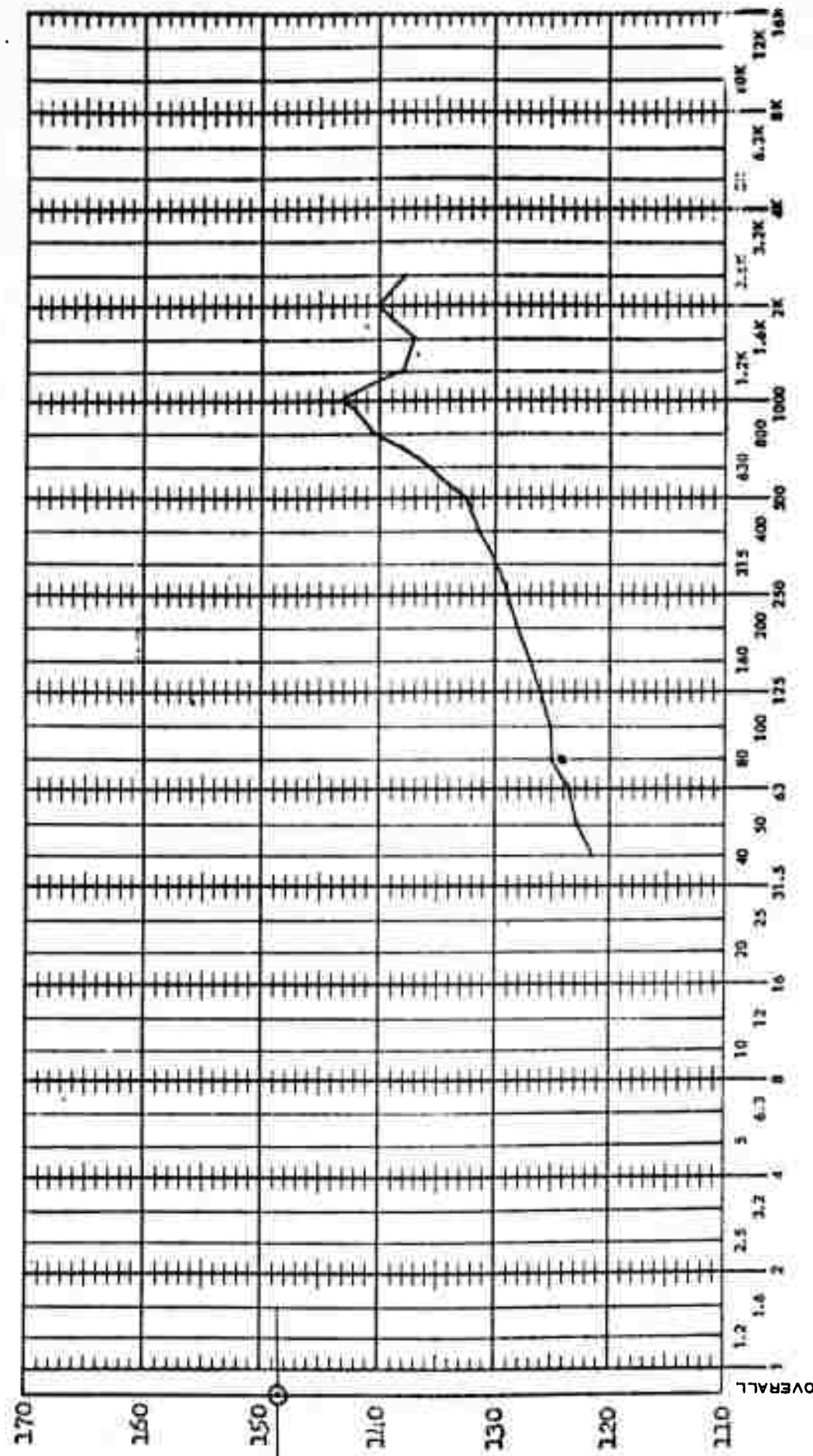
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #23 Mach No. 1.0 Correlation No. 363

$\alpha = 0$   $\beta = 4^\circ$

Figure 14 (Continued)

BOEING

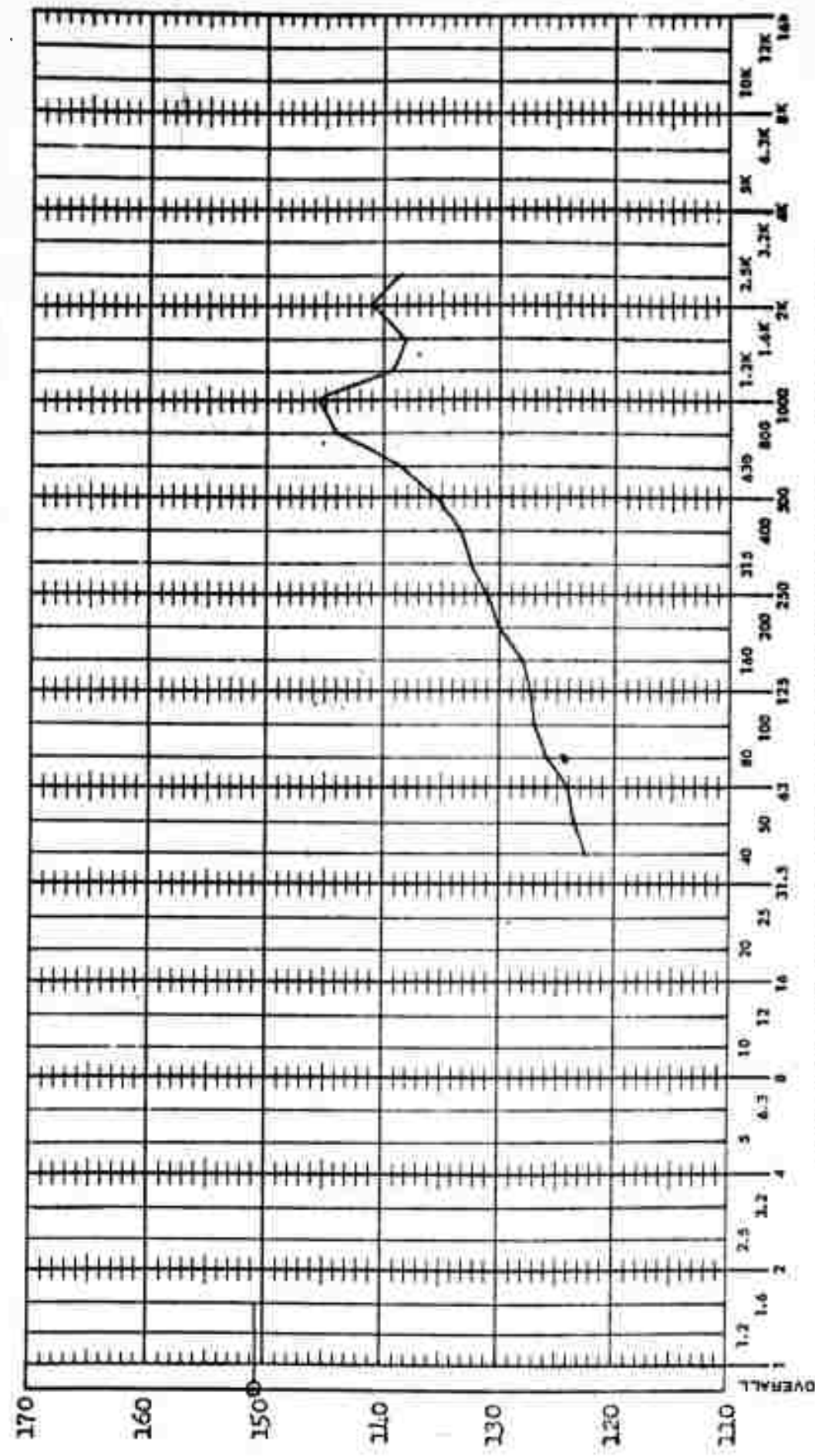
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #23 Mach No. 1.05 Correlation No. 366

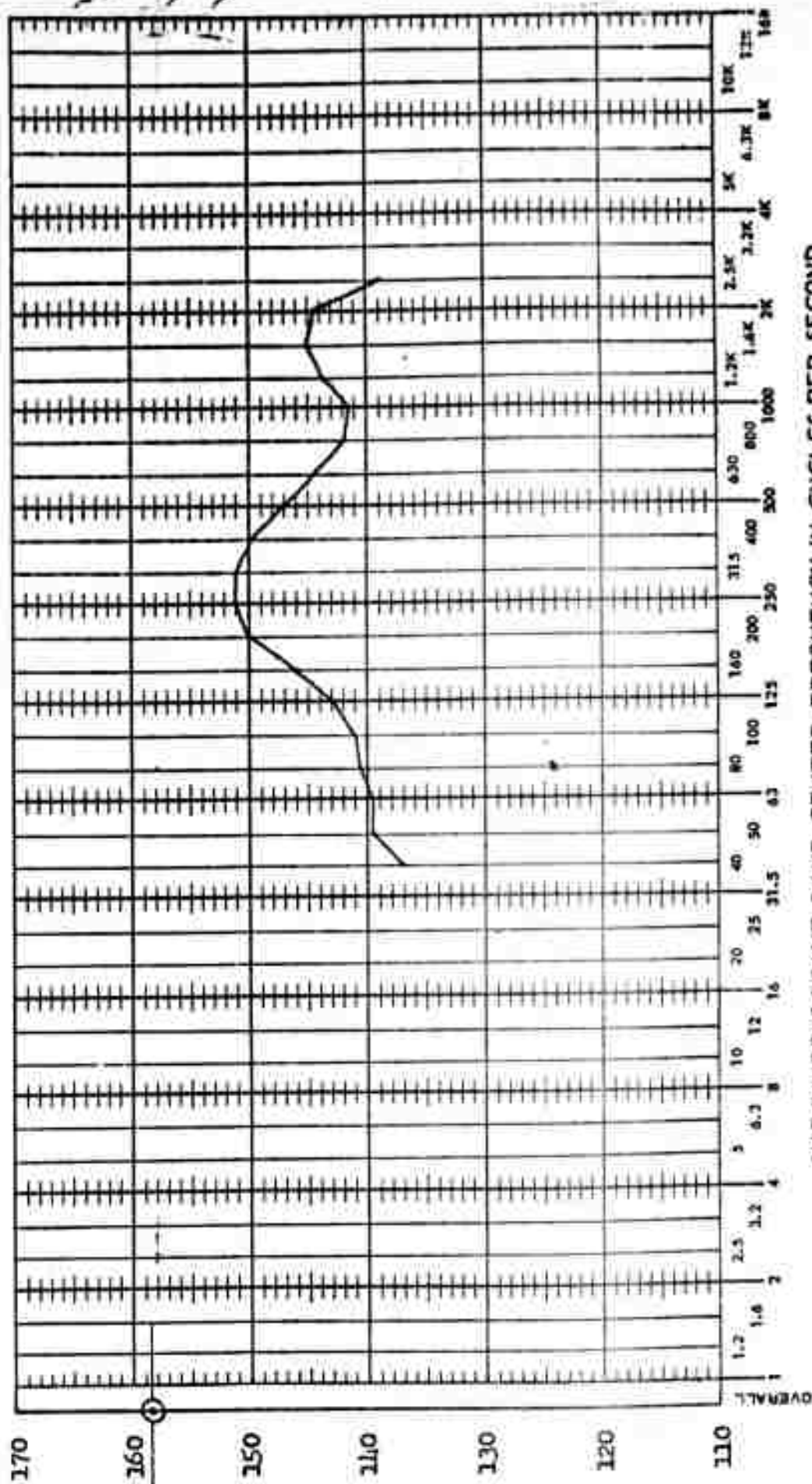
$\alpha = 0^\circ$   $\beta = 4^\circ$

Figure 14 (Continued)





ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #26 Mach No. 0.8 Correlation No. 356

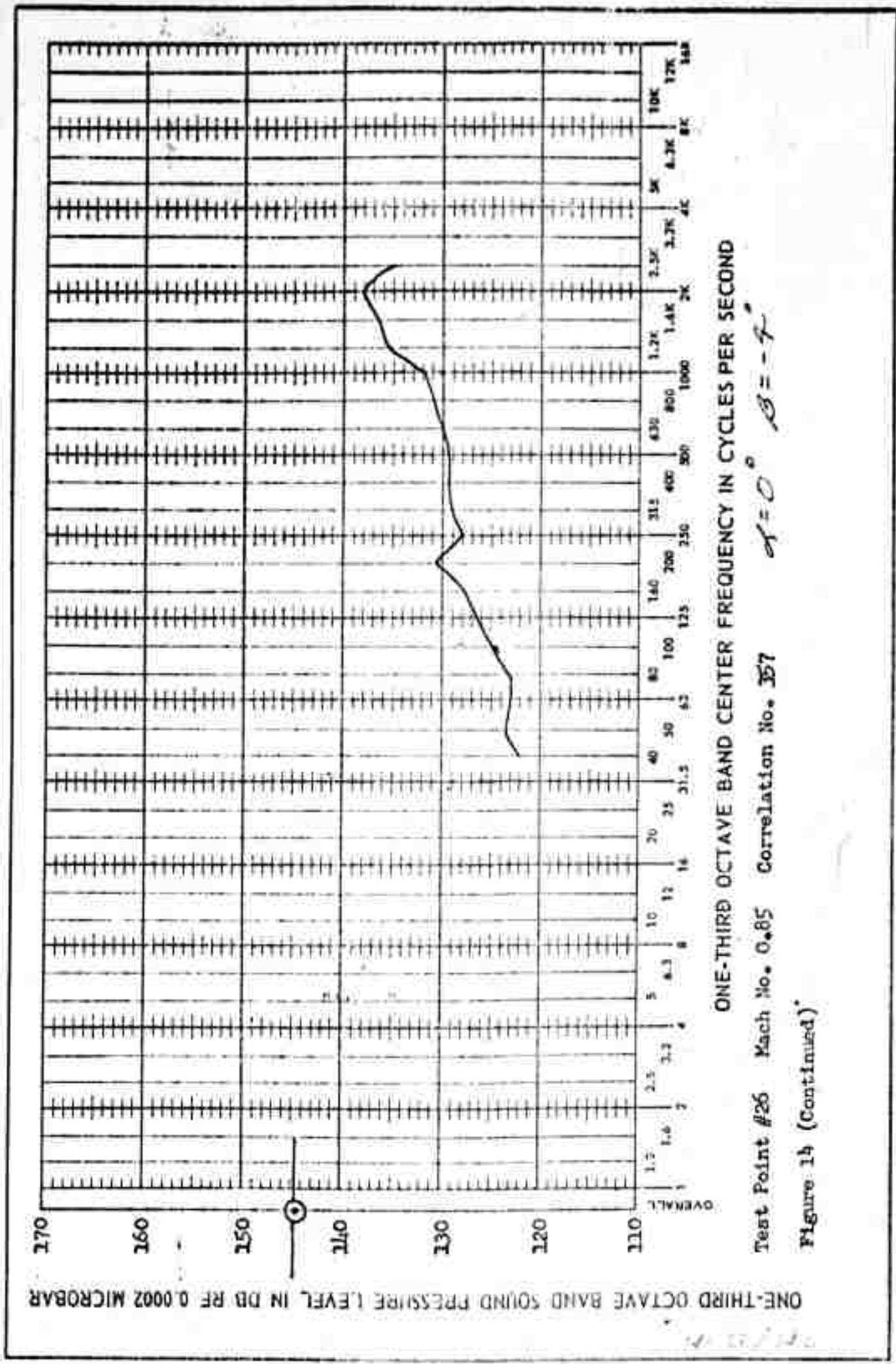
Figure 14 (Continued)

BOEING

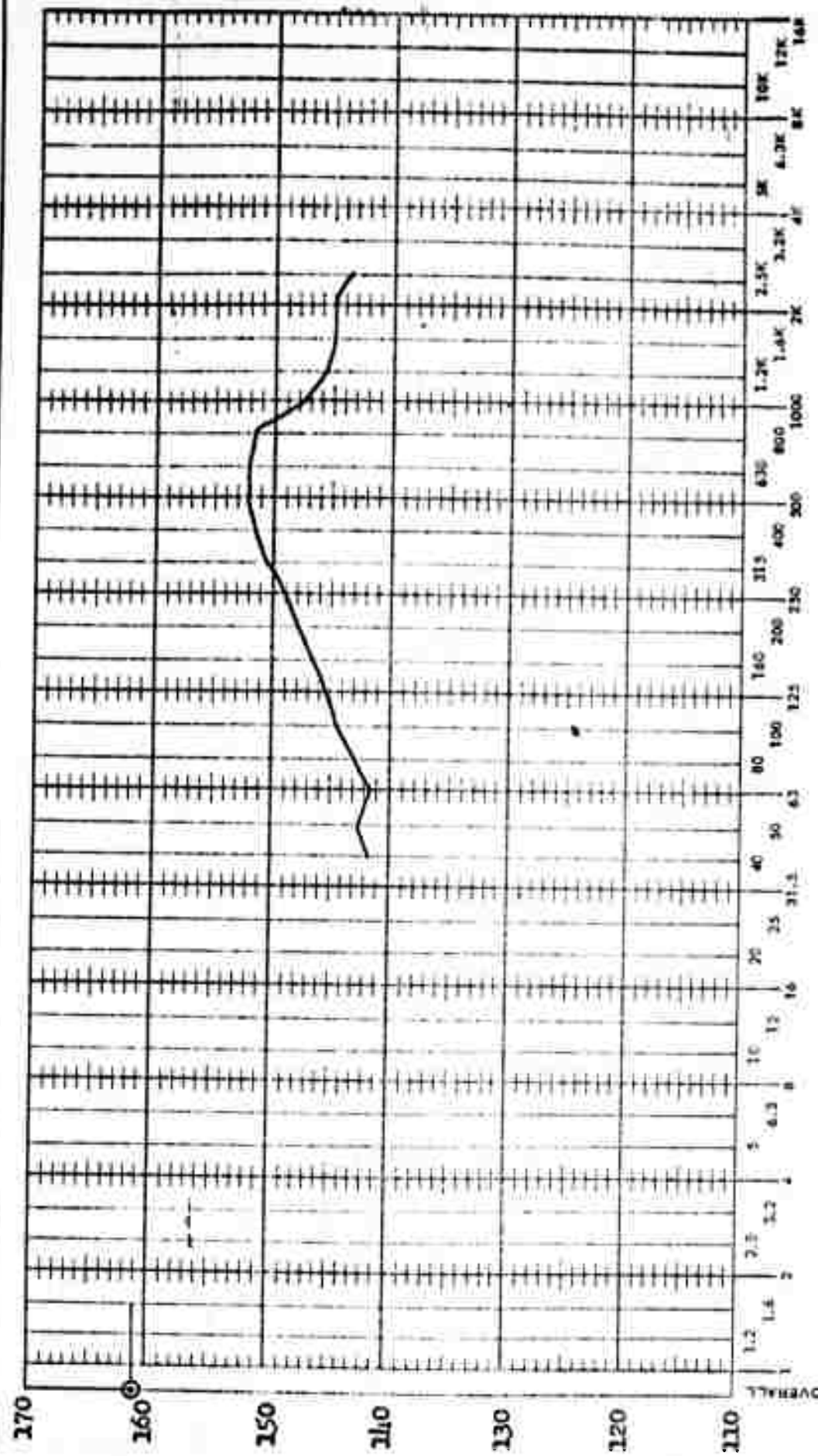
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

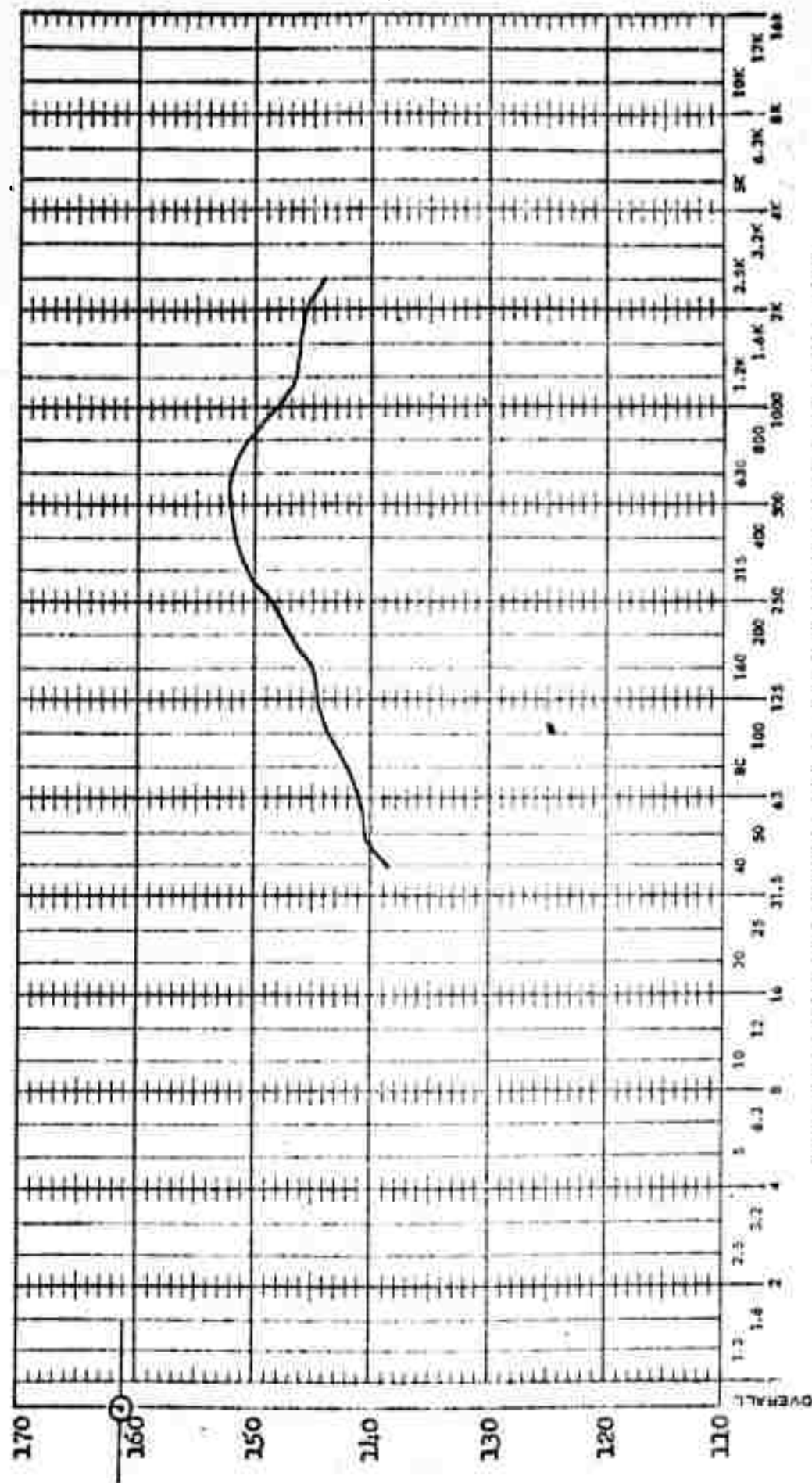
Test Point #27 Mach No. 0.5 Correlation No. 458

Figure 14 (Continued)

$\alpha = -4^\circ$   $\beta = -7^\circ$



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #27 Mach No. 0.8 Correlation No. 140

 $\beta = 0$   $\beta' = -4^\circ$ 

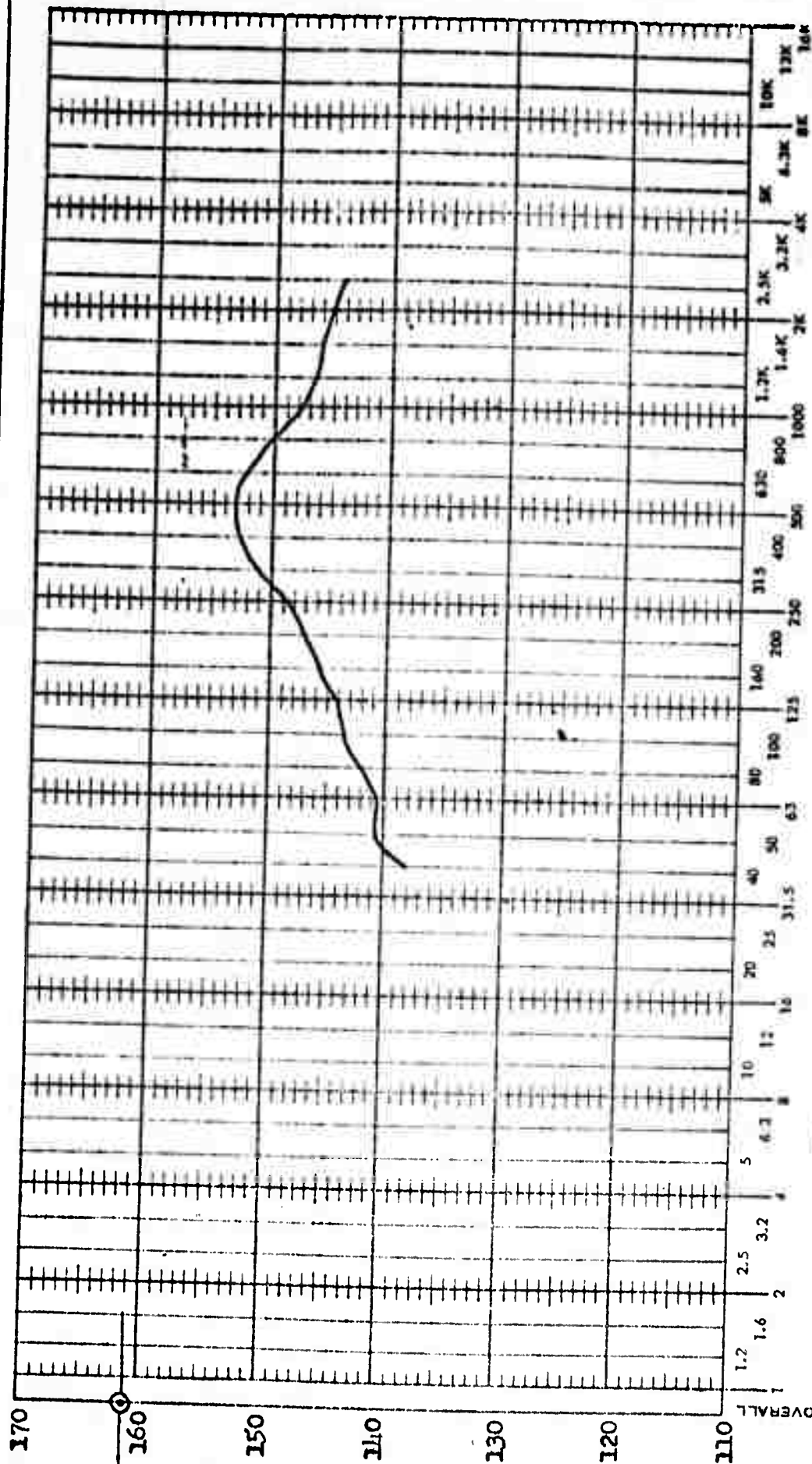
Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #27 Mach No. 0.8 Correlation No. 161

Figure 14 (Continued)

$\alpha = 4^\circ$   $\beta = -4^\circ$

ENDING

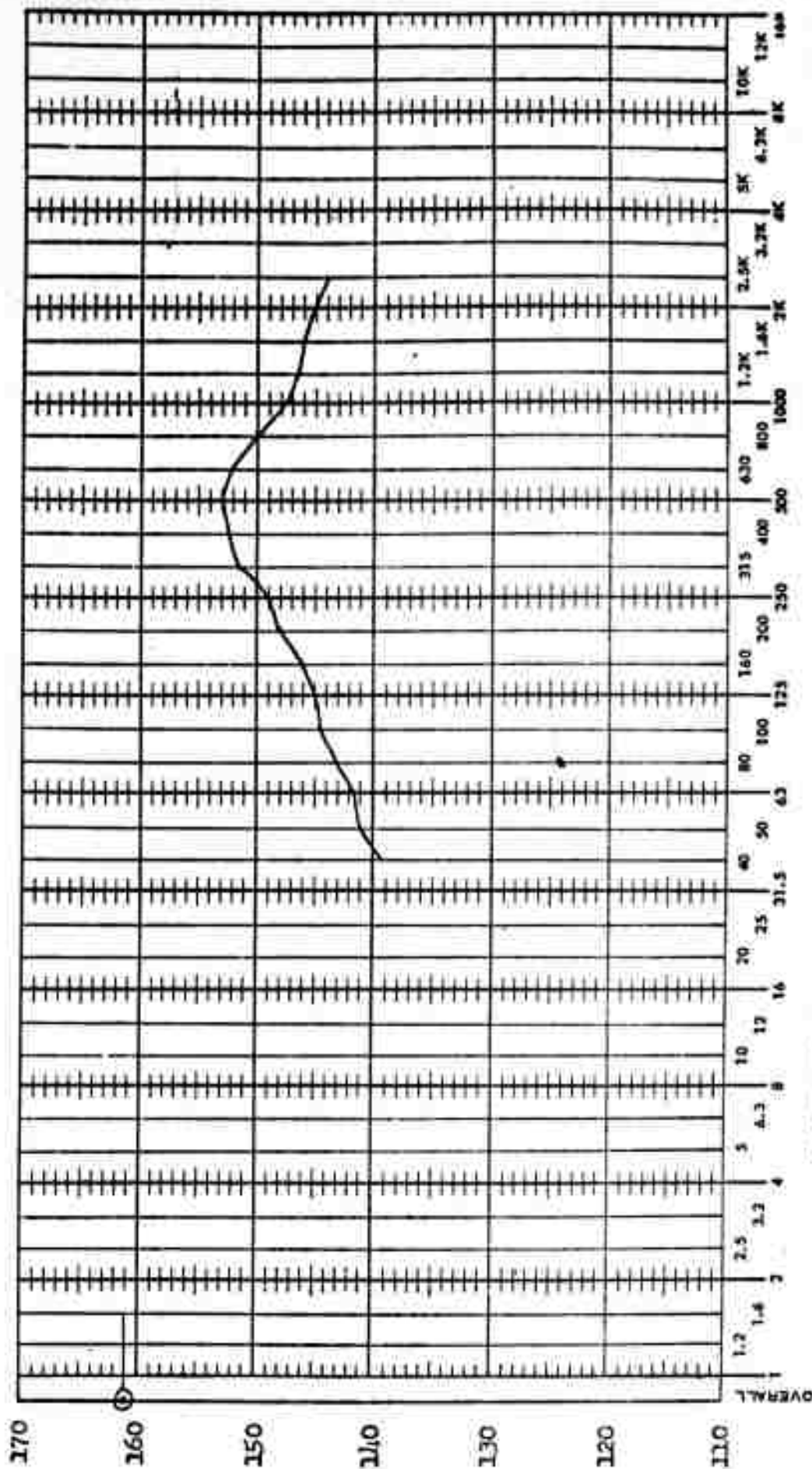
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #27 Mach No. 0.82 Correlation No. 462

$\alpha = 4^\circ$   $\beta = 4^\circ$

Figure 14 (Continued)

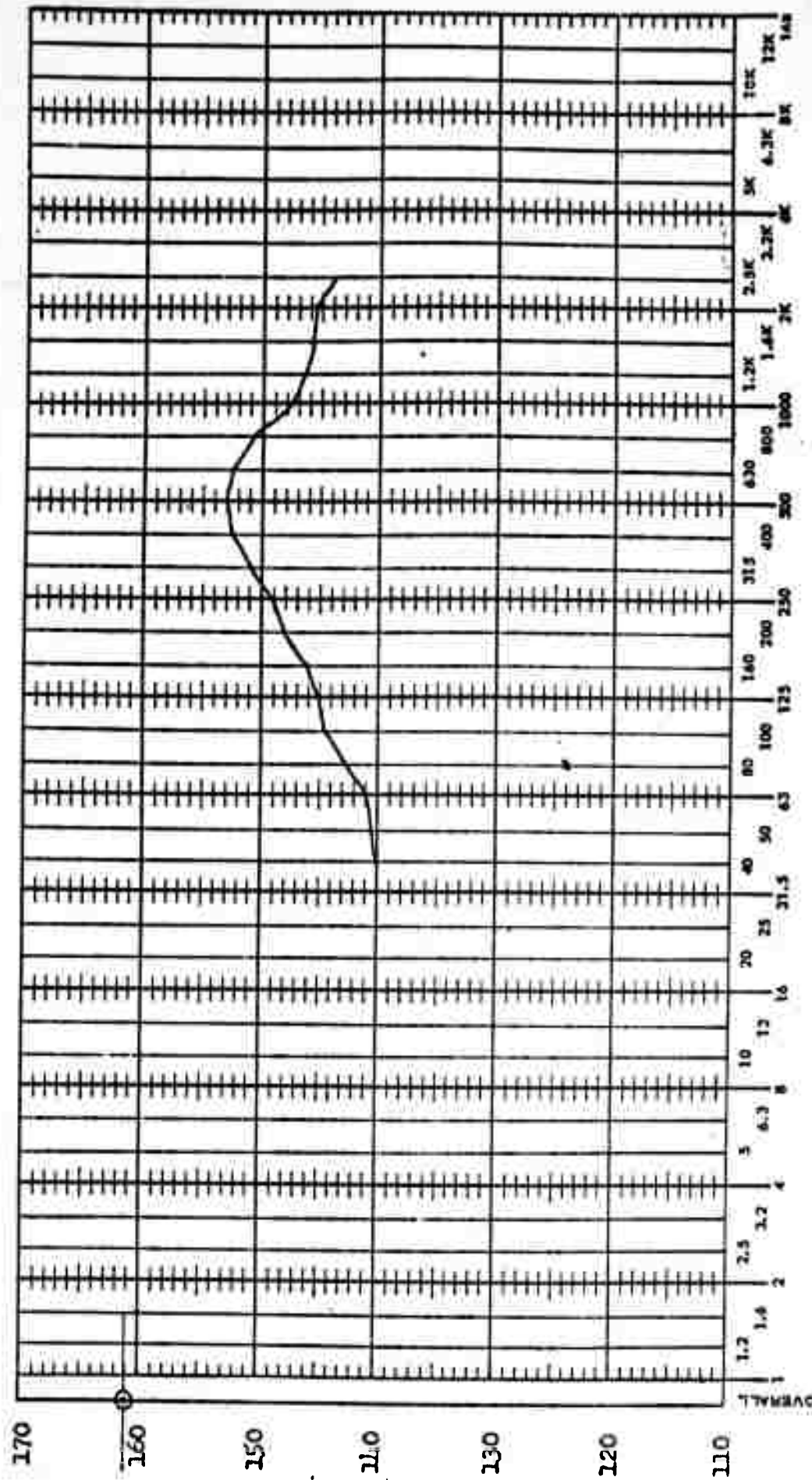
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #27 Mach No. 0.82 Correlation No. 1463

Figure 14 (Continued)

$\alpha = 0$   $\beta = 7$

BOEING

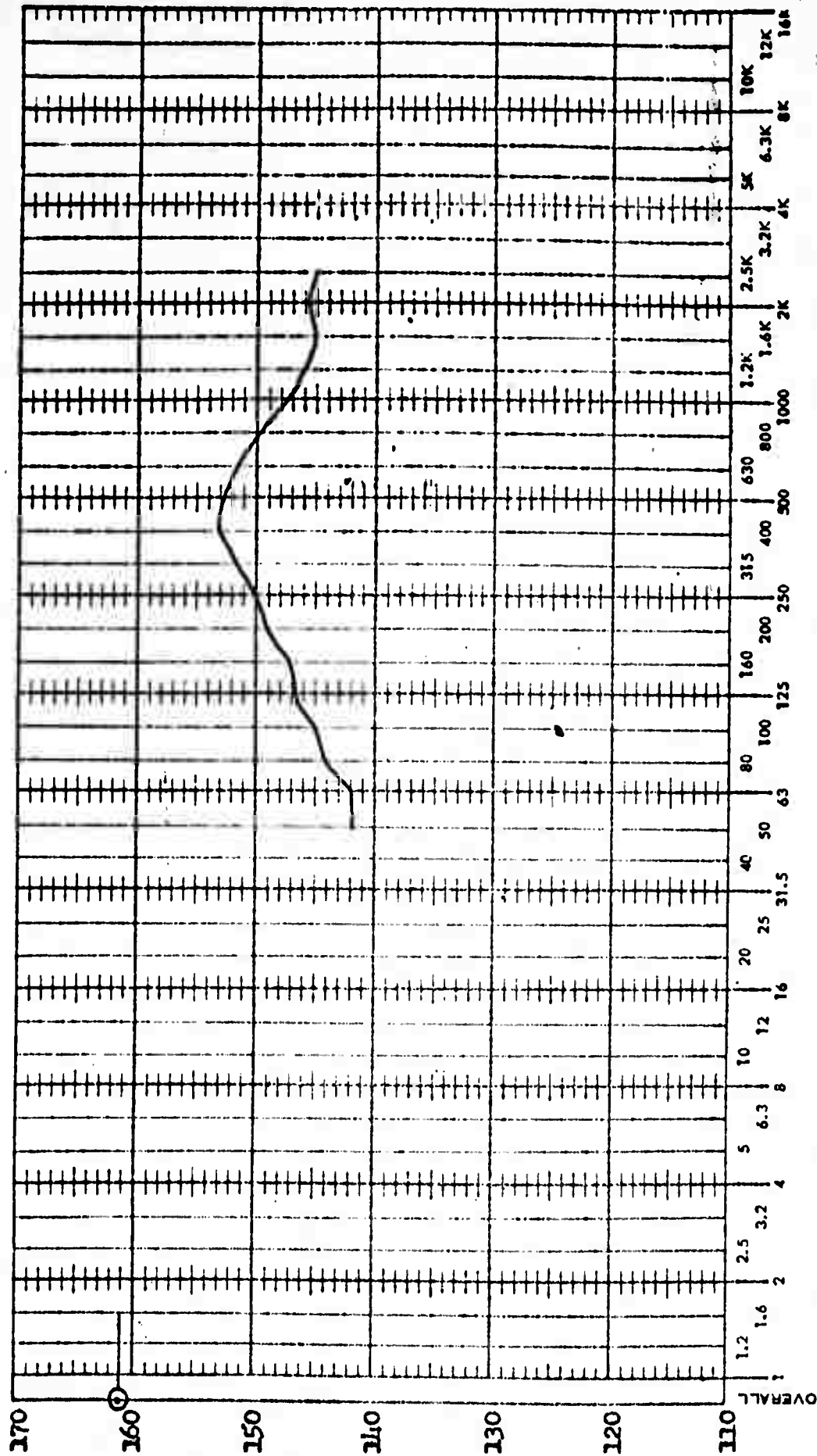
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #27 Mach No. 0.82 Correlation No. 161

Figure 14 (Continued)

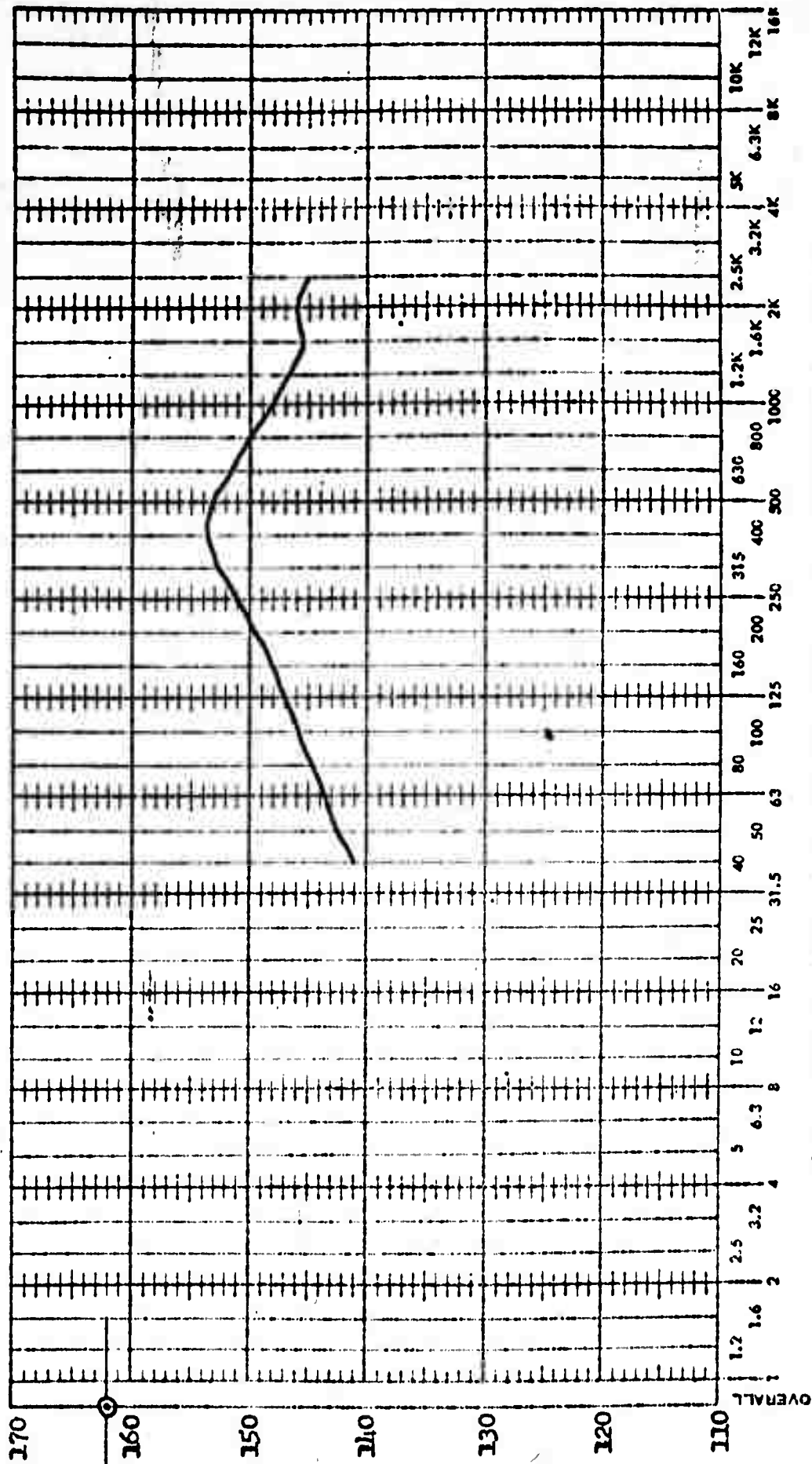
$\alpha = 4^\circ$   $\beta = -4^\circ$

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #27 Mach No. 0.84 Correlation No. 465

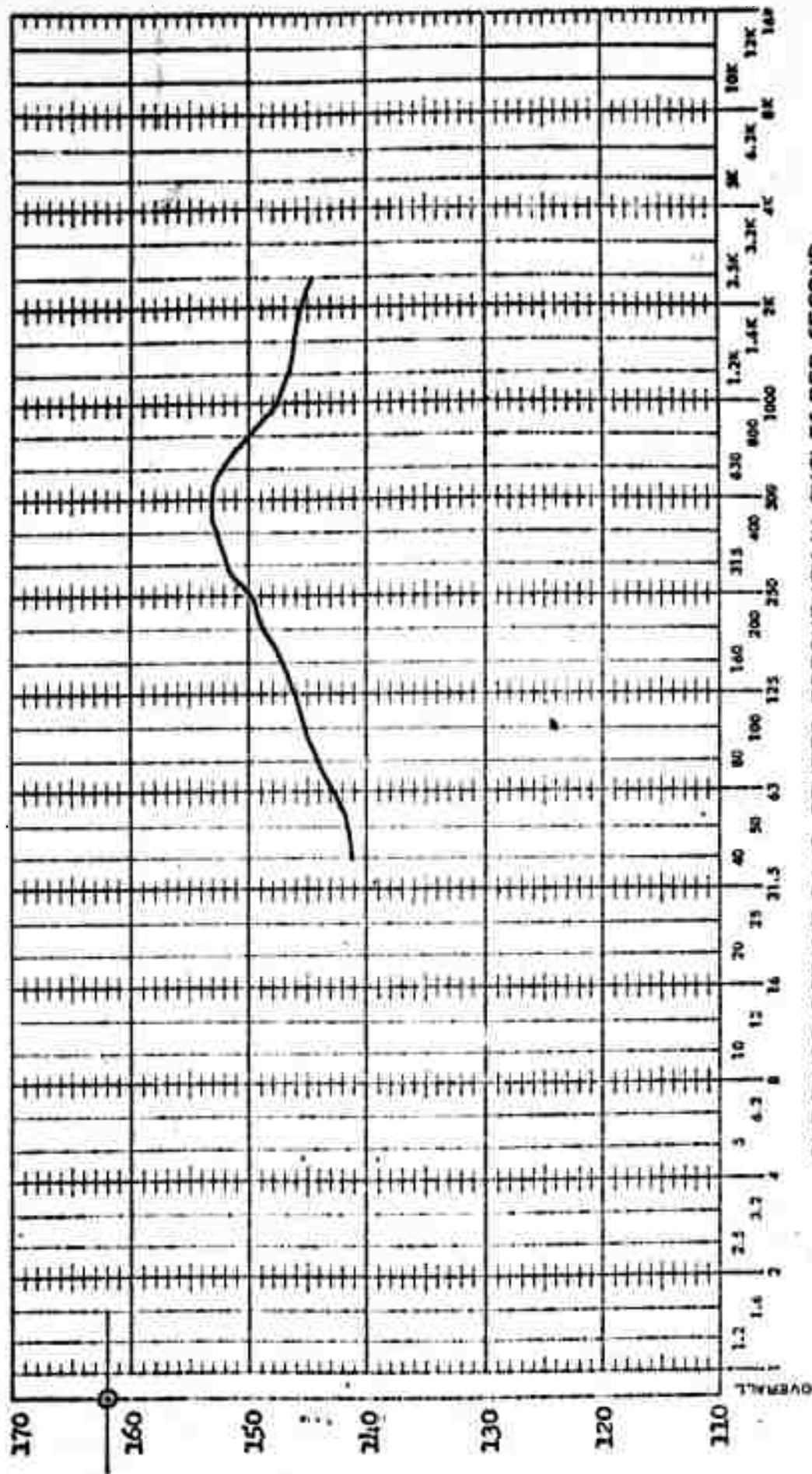
$\alpha = 4^\circ$

$\beta = -4^\circ$

Figure 14 (Continued)



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #27 Mach No. 0.84 Correlation No. 166

$\alpha = 0^\circ$   $\beta = 4^\circ$

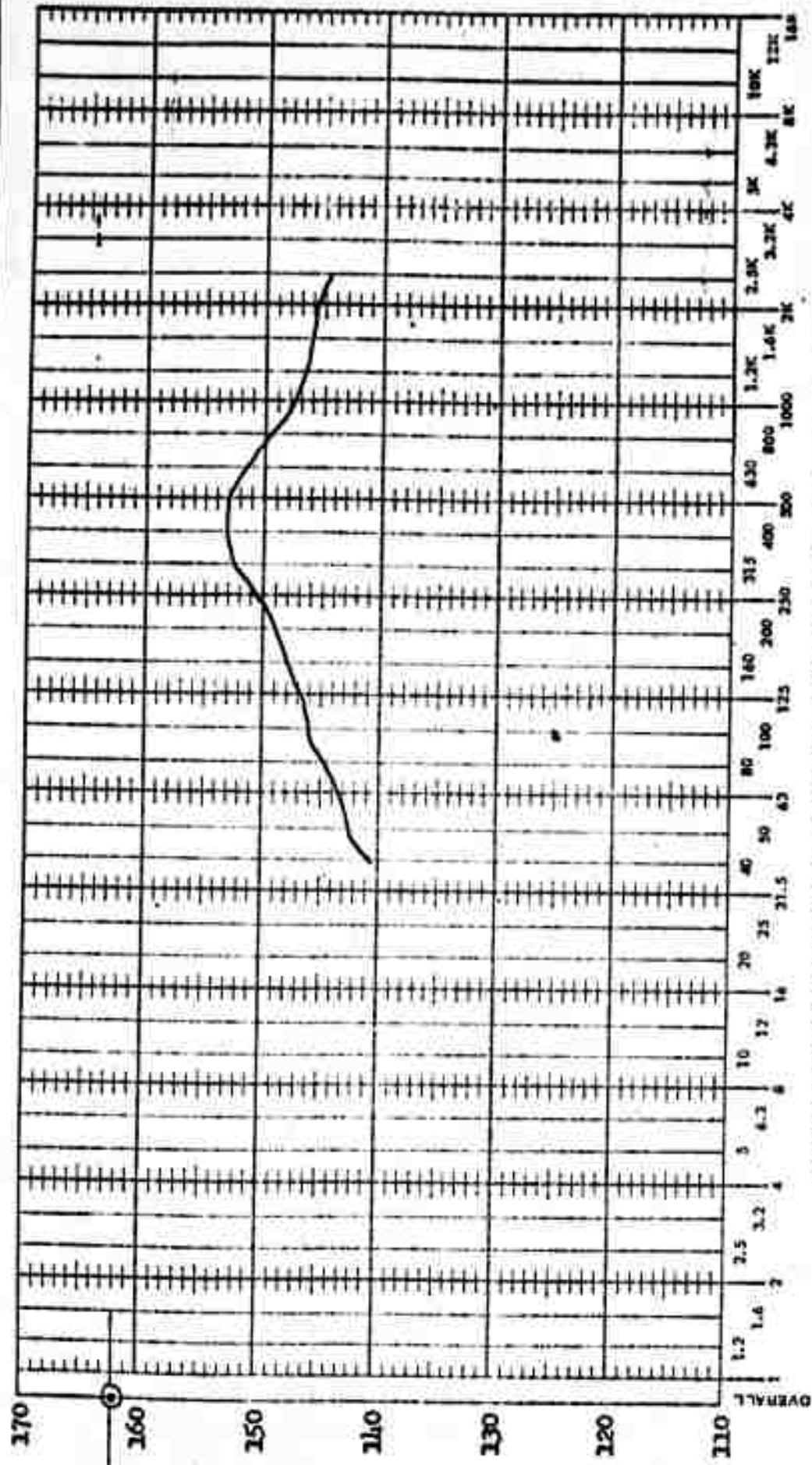
Figure 14 (Continued)





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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #27 Mach No. 0.81 Correlation No. 467

$\alpha = 4^\circ$   $\beta = 7^\circ$

Figure 14 (Continued)

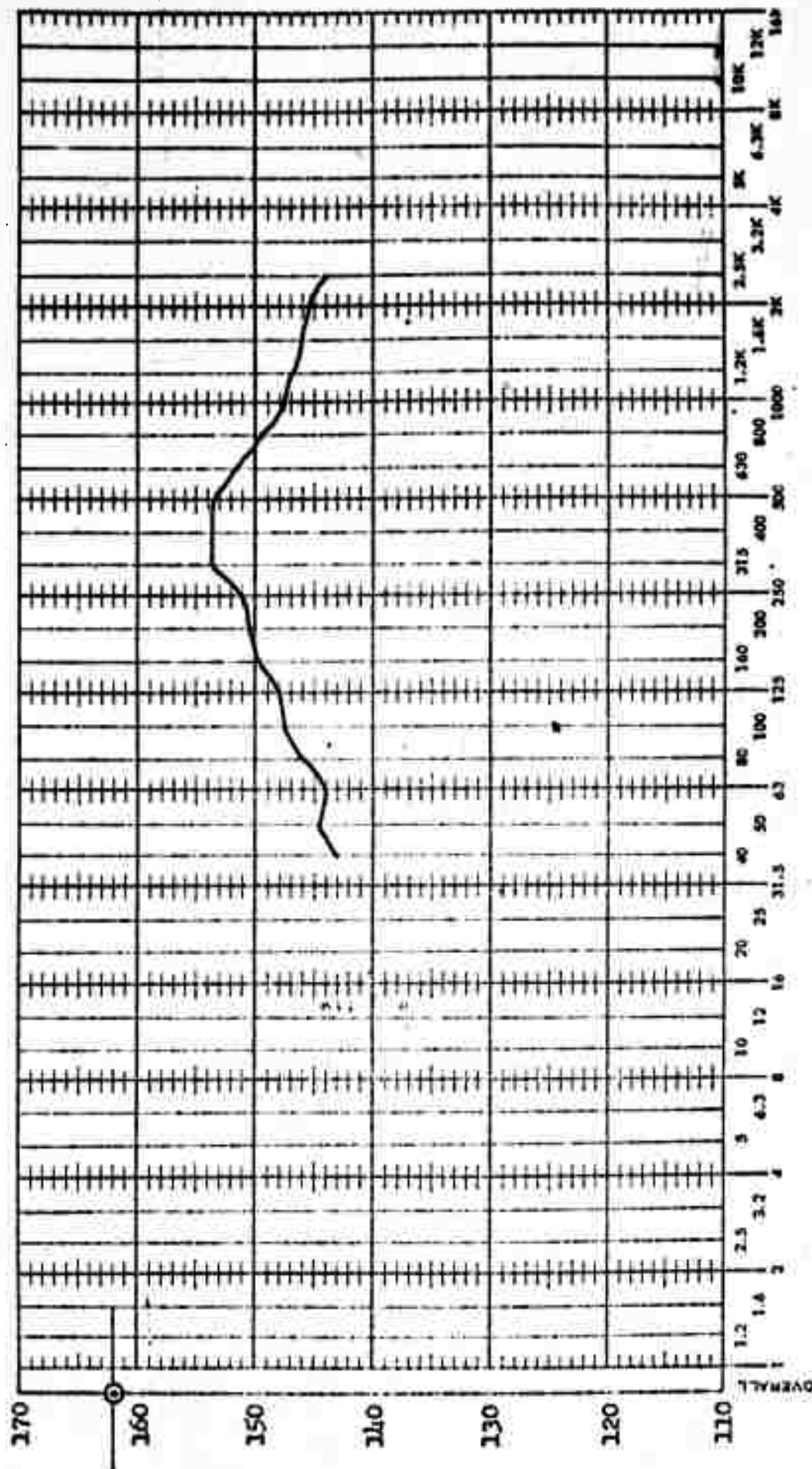
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #27 Mach No. 0.86 Correlation No. 1468

Figure 14 (Continued)

$\alpha = 4^\circ$   $\beta = -4^\circ$

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR

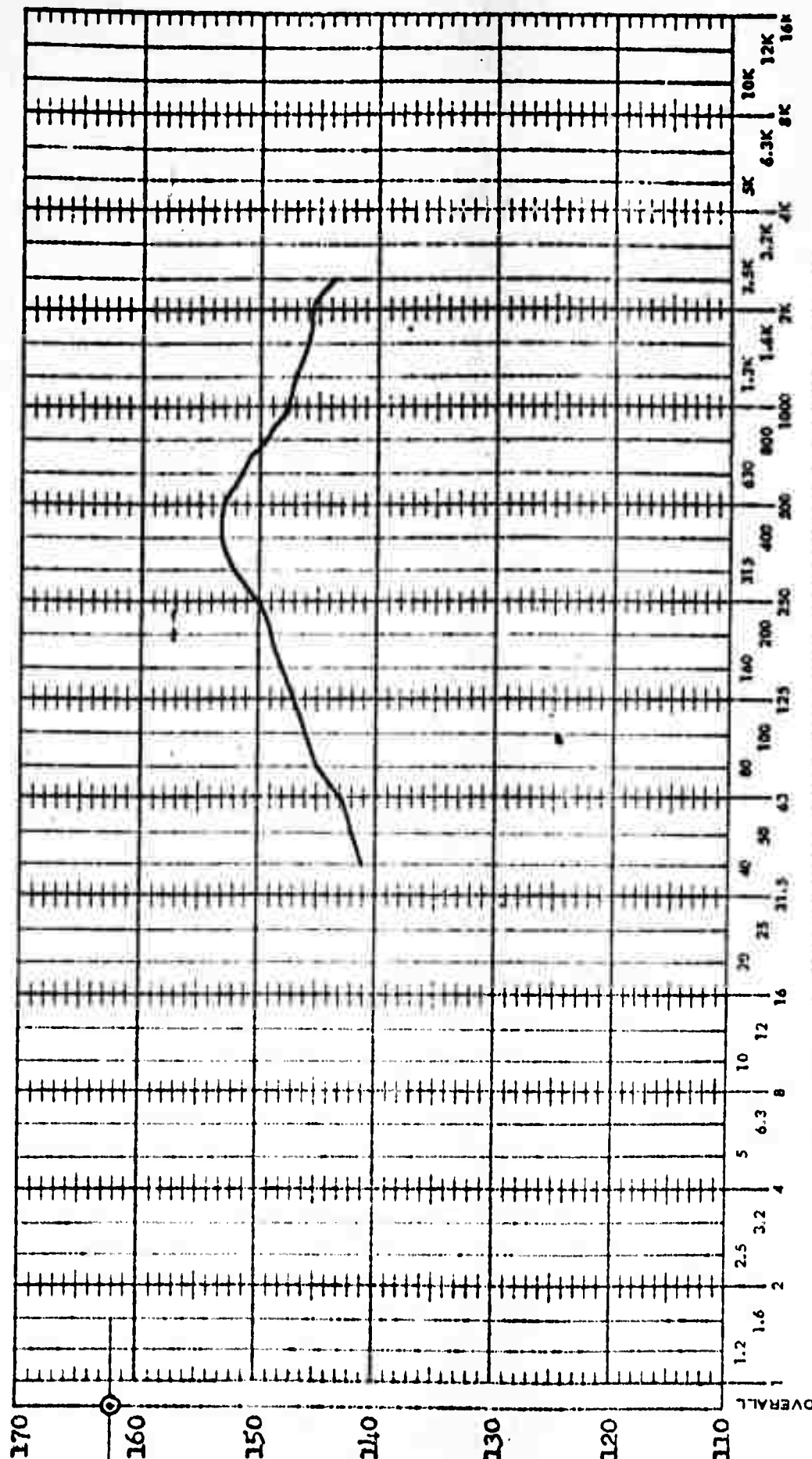
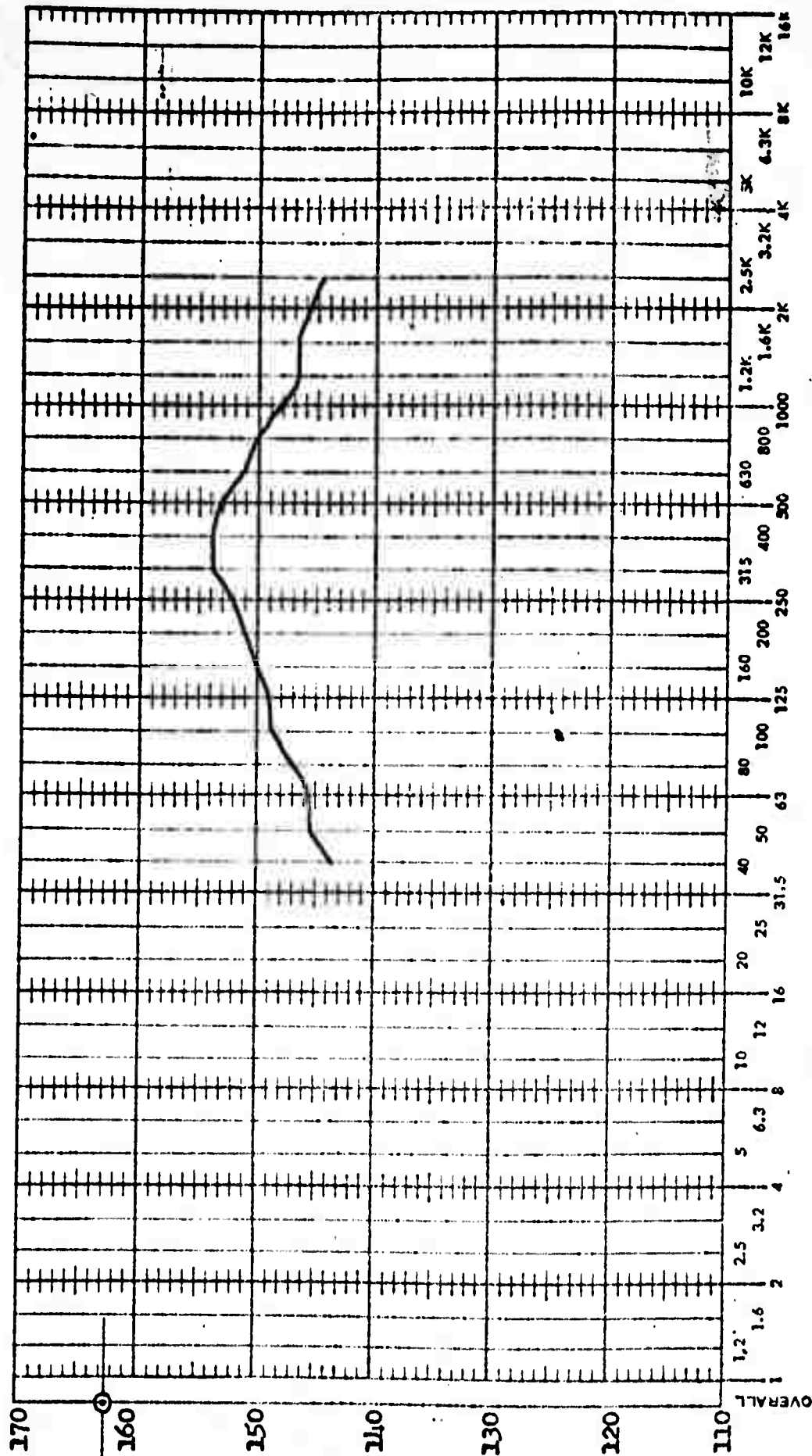


Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #27 Mach No. 0.86 Correlation No. 470

$\alpha = 4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

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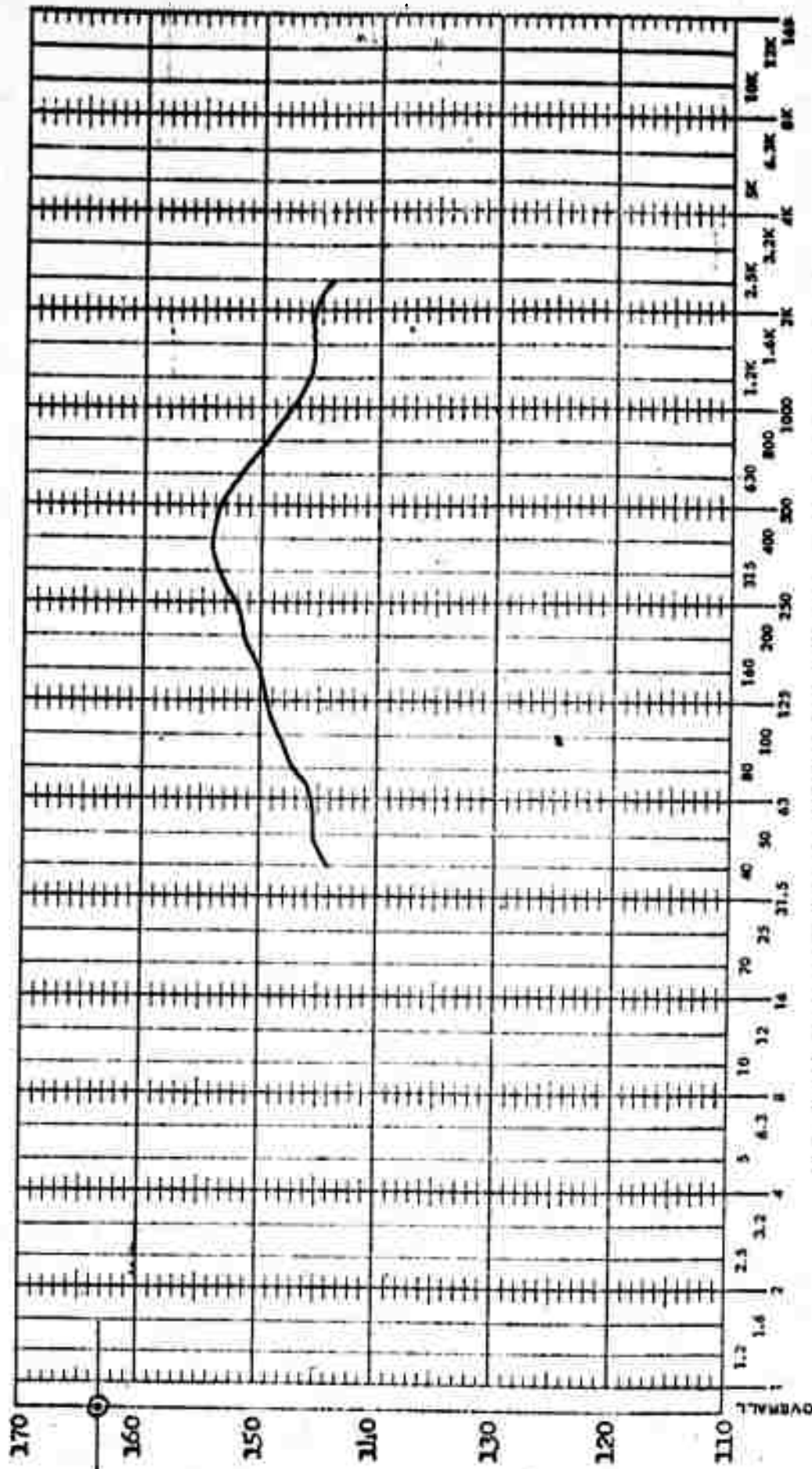
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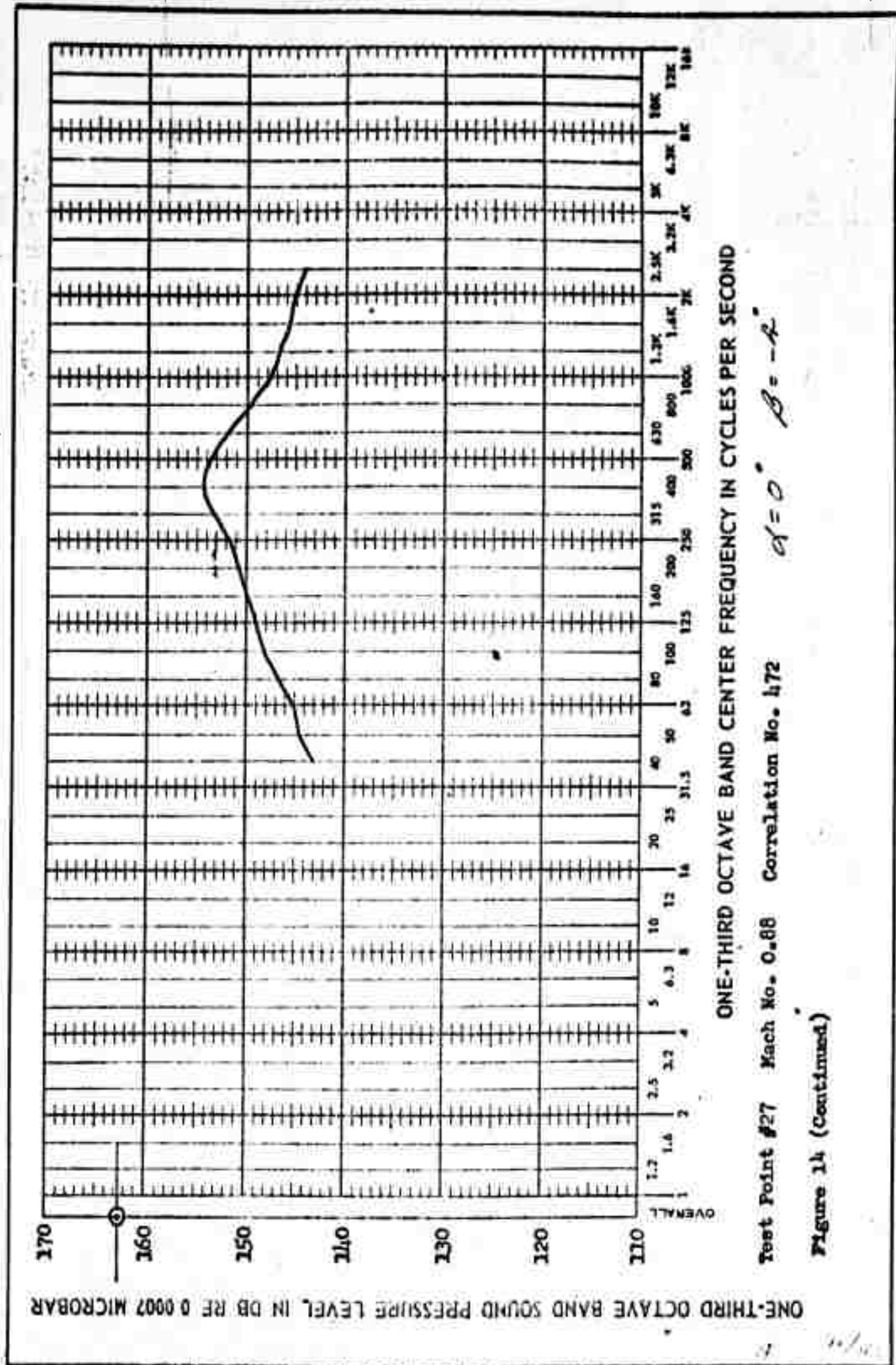
144

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR





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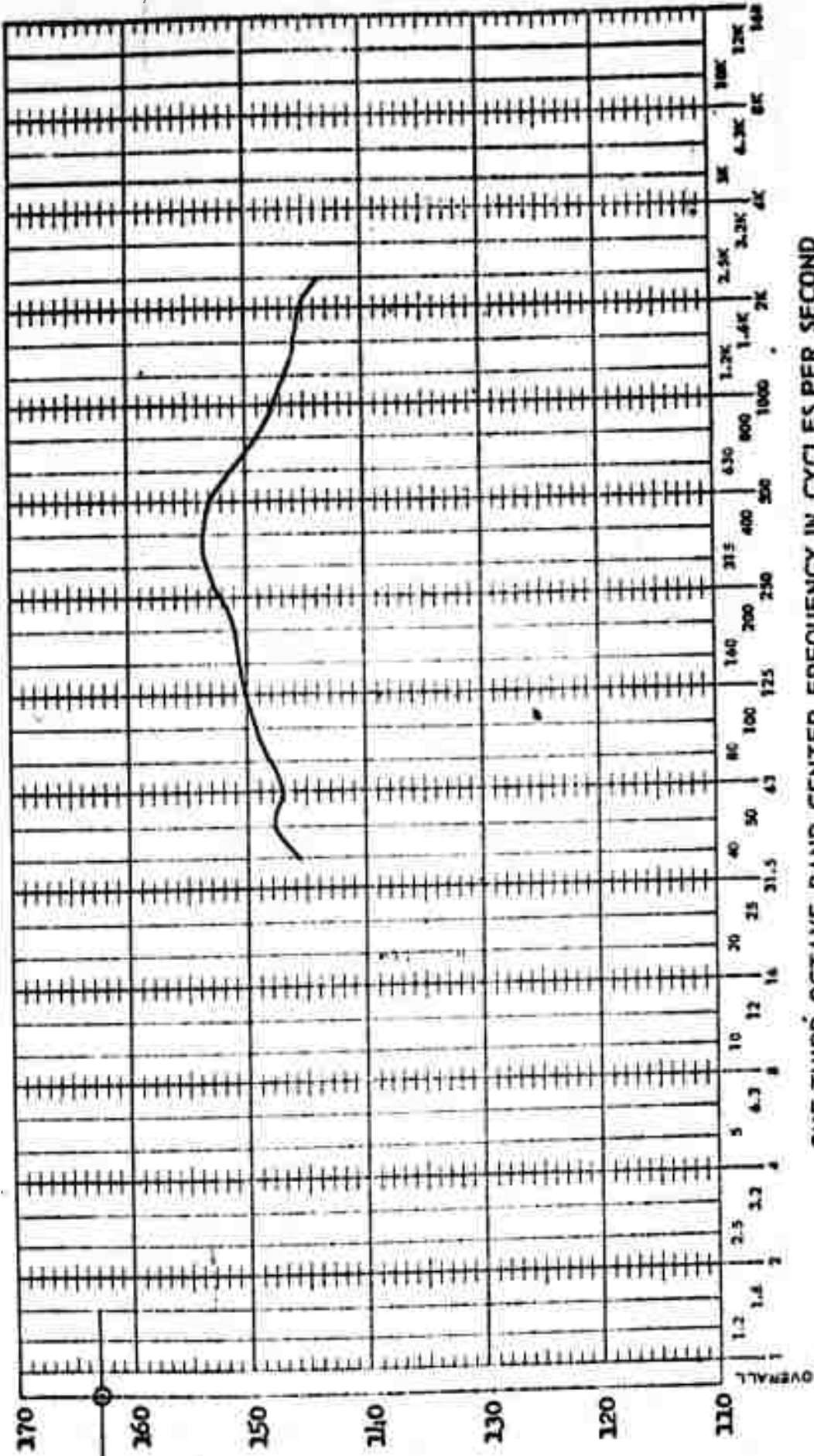


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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB. RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #27 Mach No. 0.88 Correlation No. 473

Figure 14 (Continued)

$\alpha = 4^\circ$   $\beta = -2^\circ$

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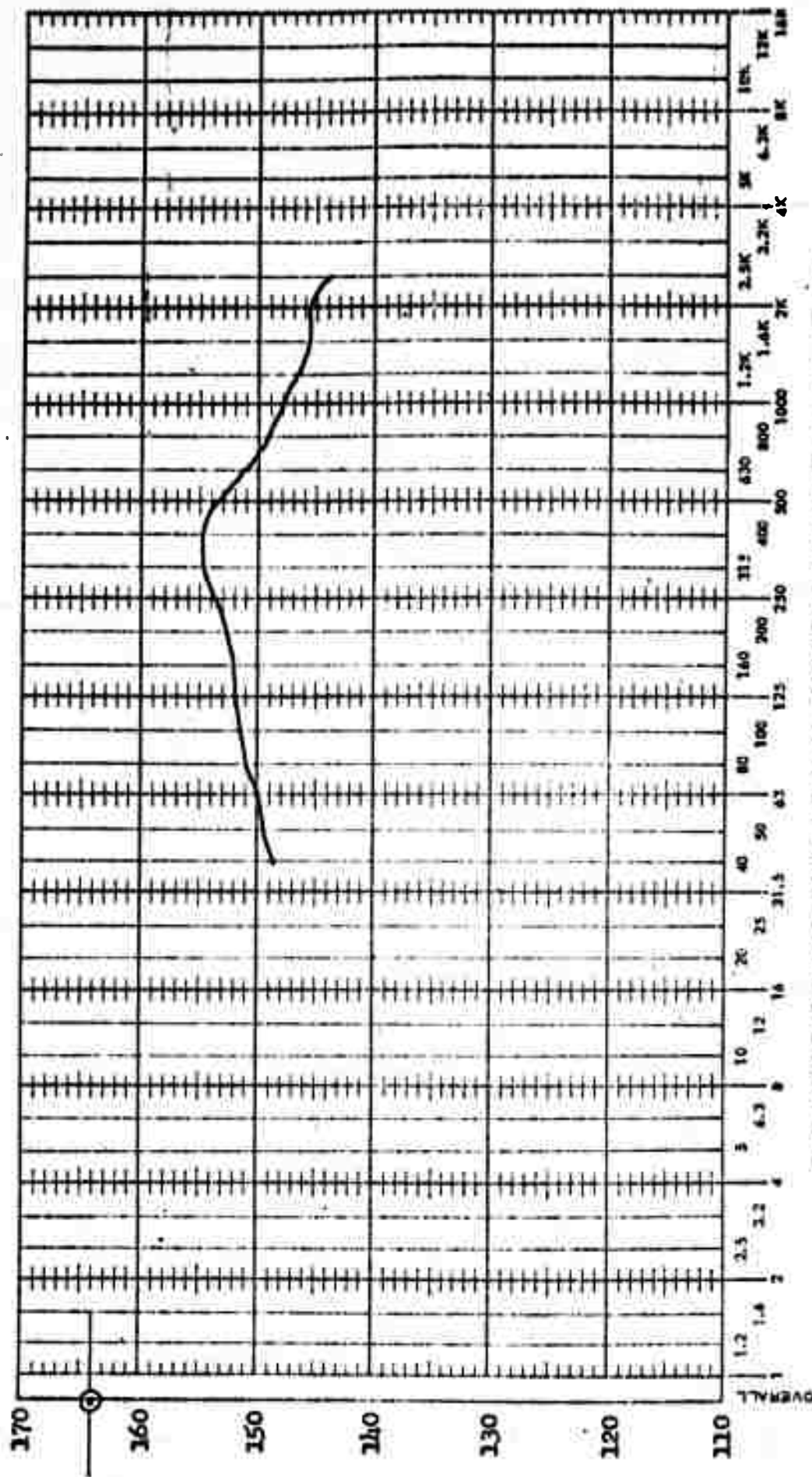
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

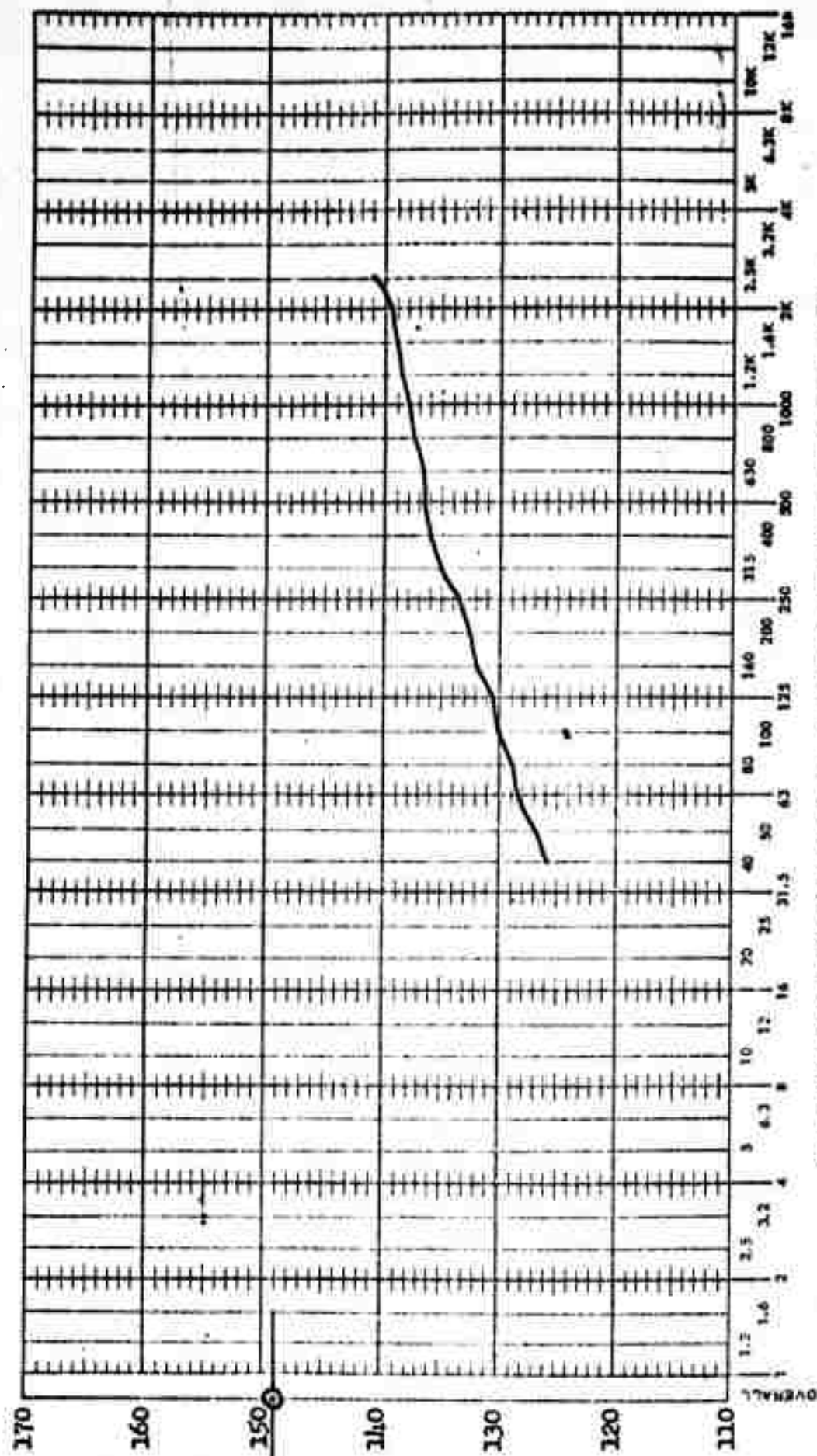
Test Point #27 Mach No. 0.9 Correlation No. 476

Figure 14 (Continued)

$\alpha' = -4^\circ$   $\beta = -4^\circ$



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #27 Mach No. 0.94 Correlation No. 482

$\alpha = -4^\circ$   $\beta = -4^\circ$

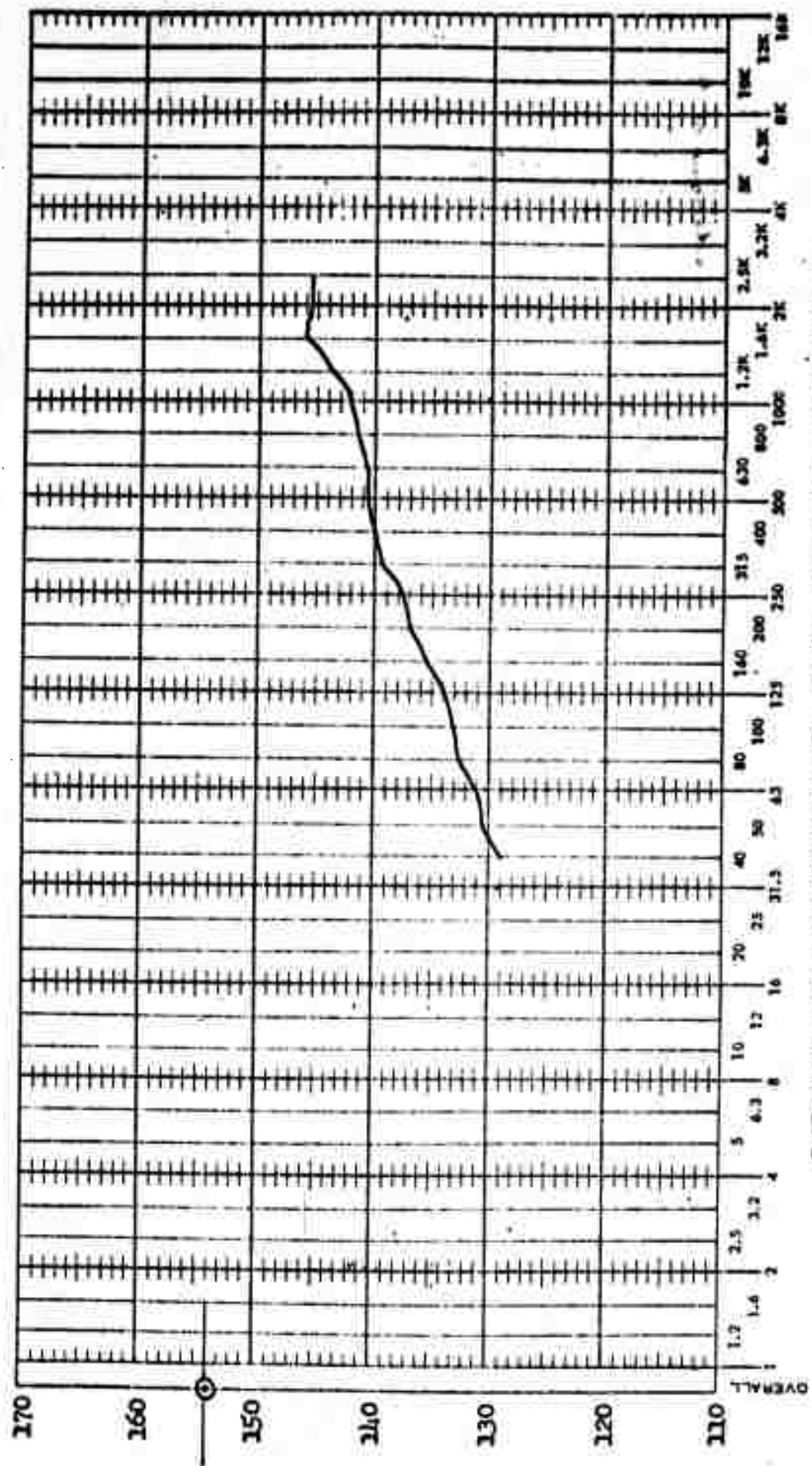
Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #28 Mach No. 0.75 Correlation No. 456

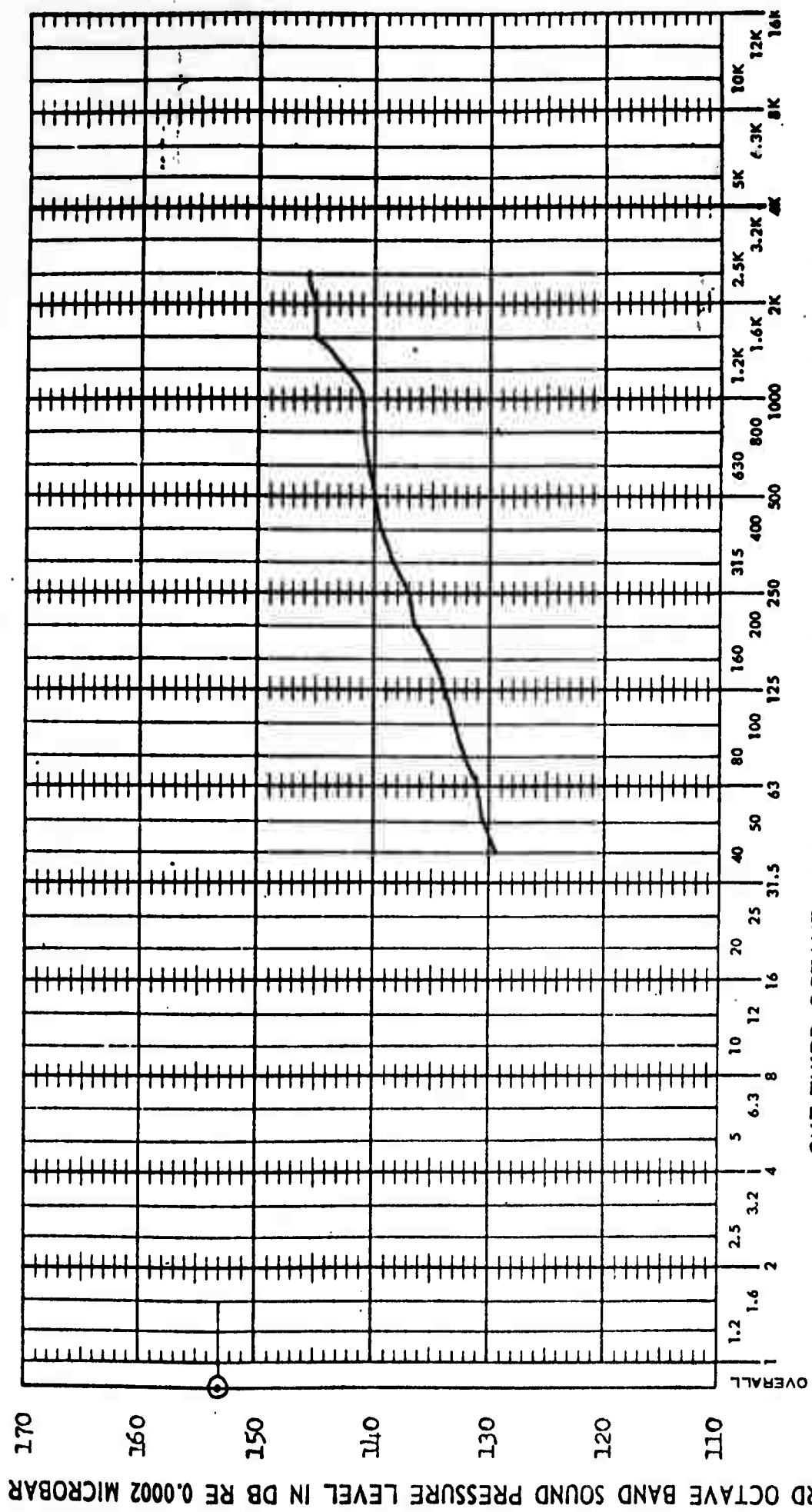
Figure 14 (Continued)

$\alpha = 0^\circ$   $\beta = -4^\circ$





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ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #28 Mach No. 0.75 Correlation No. 1457  $\alpha = -4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

BOEING

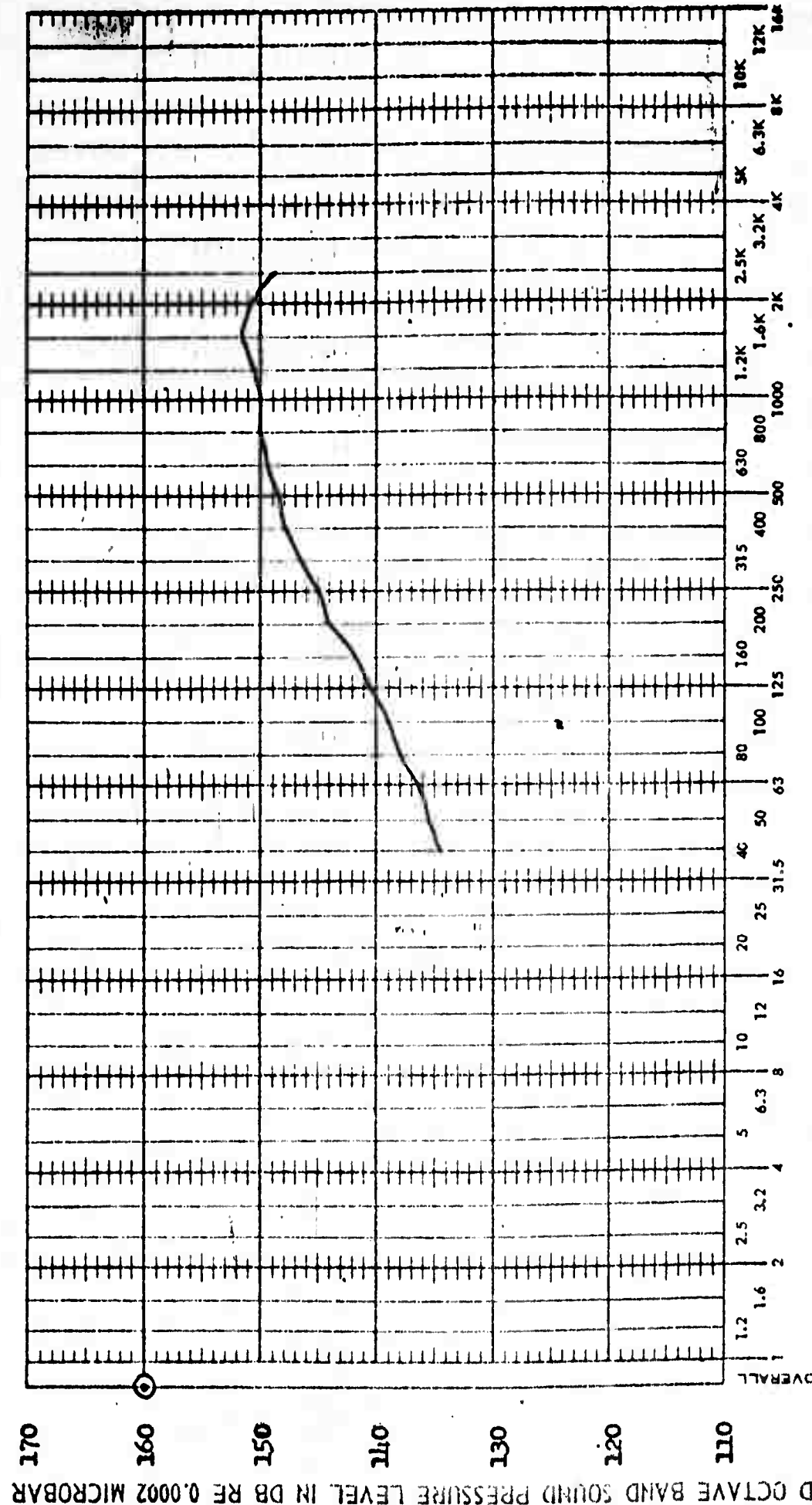
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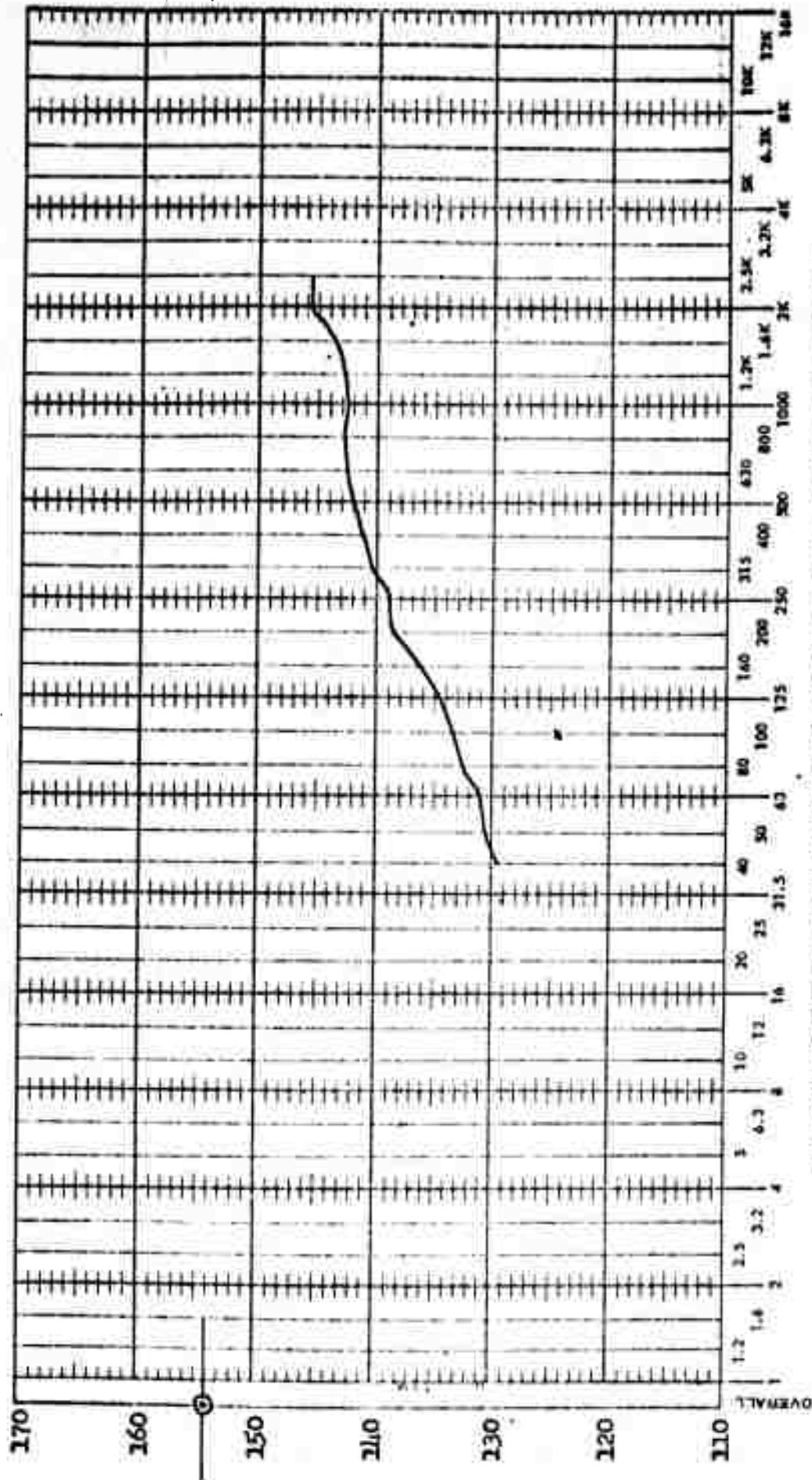
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #28 Mach No. 0.8 Correlation No. 458

$\alpha = -4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

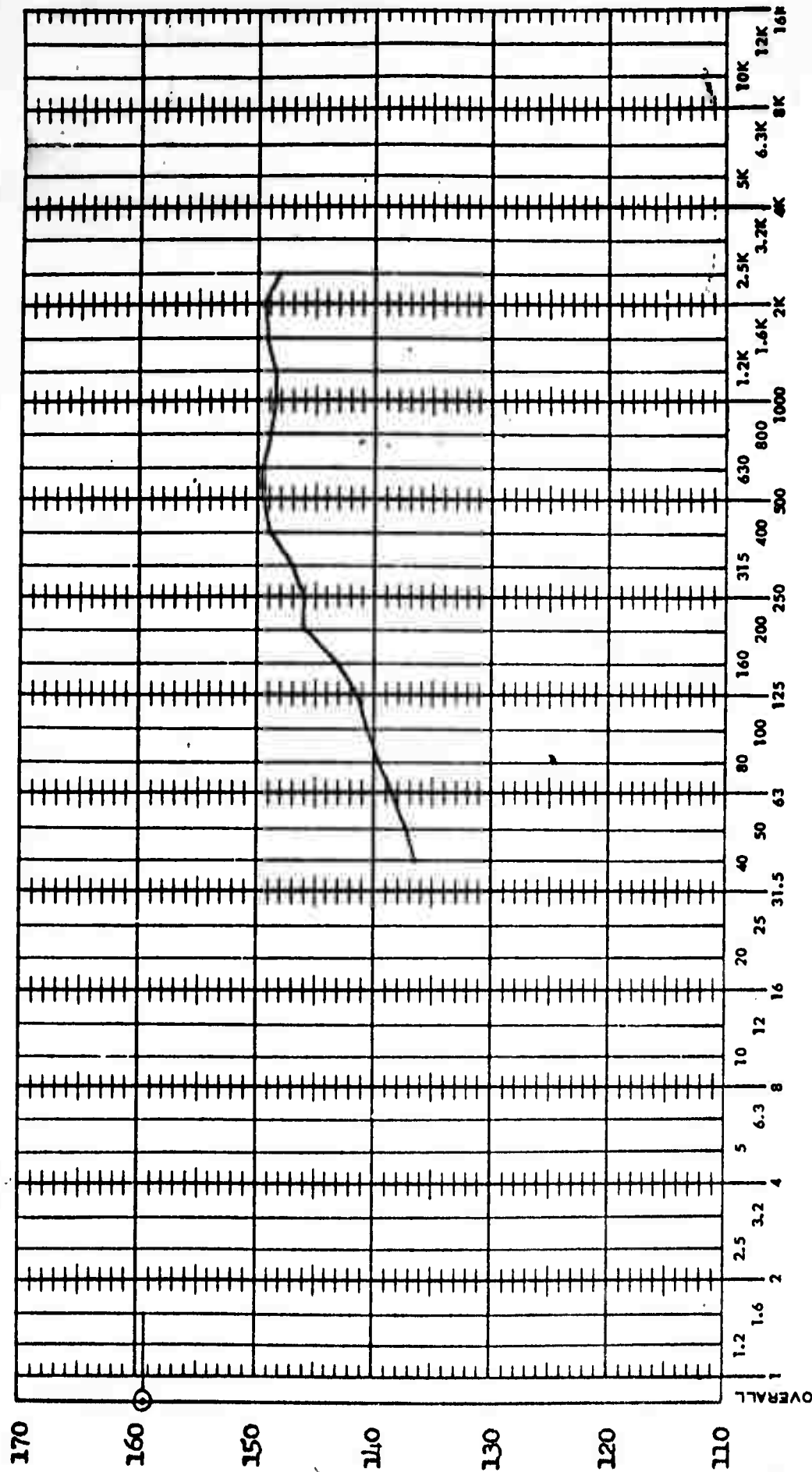
Test Point #28 Mach No. 0.6 Correlation No. 460

$\alpha = 0^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #28 Mach No. 0.82 Correlation No. 463

$\alpha = 0^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

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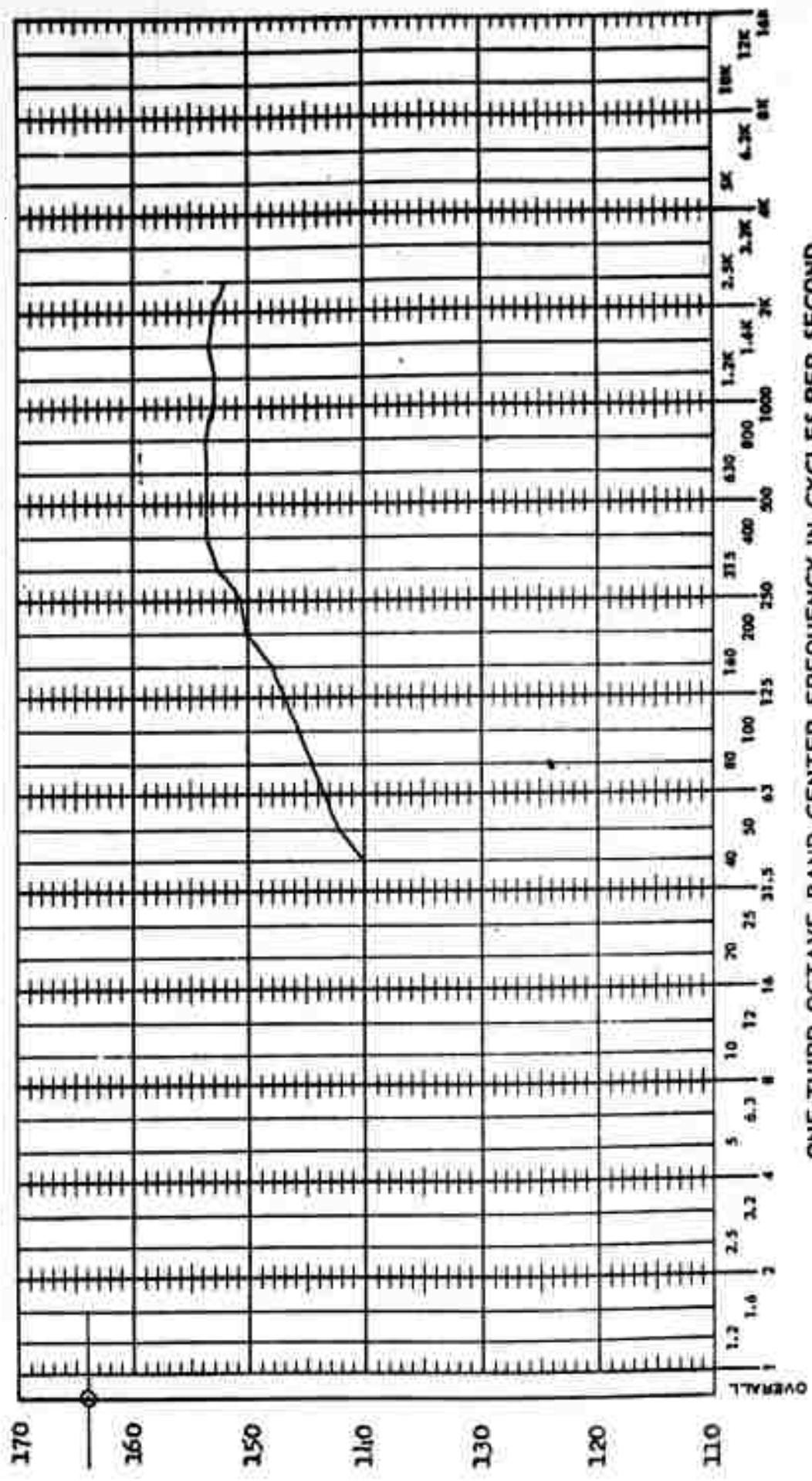
150





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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

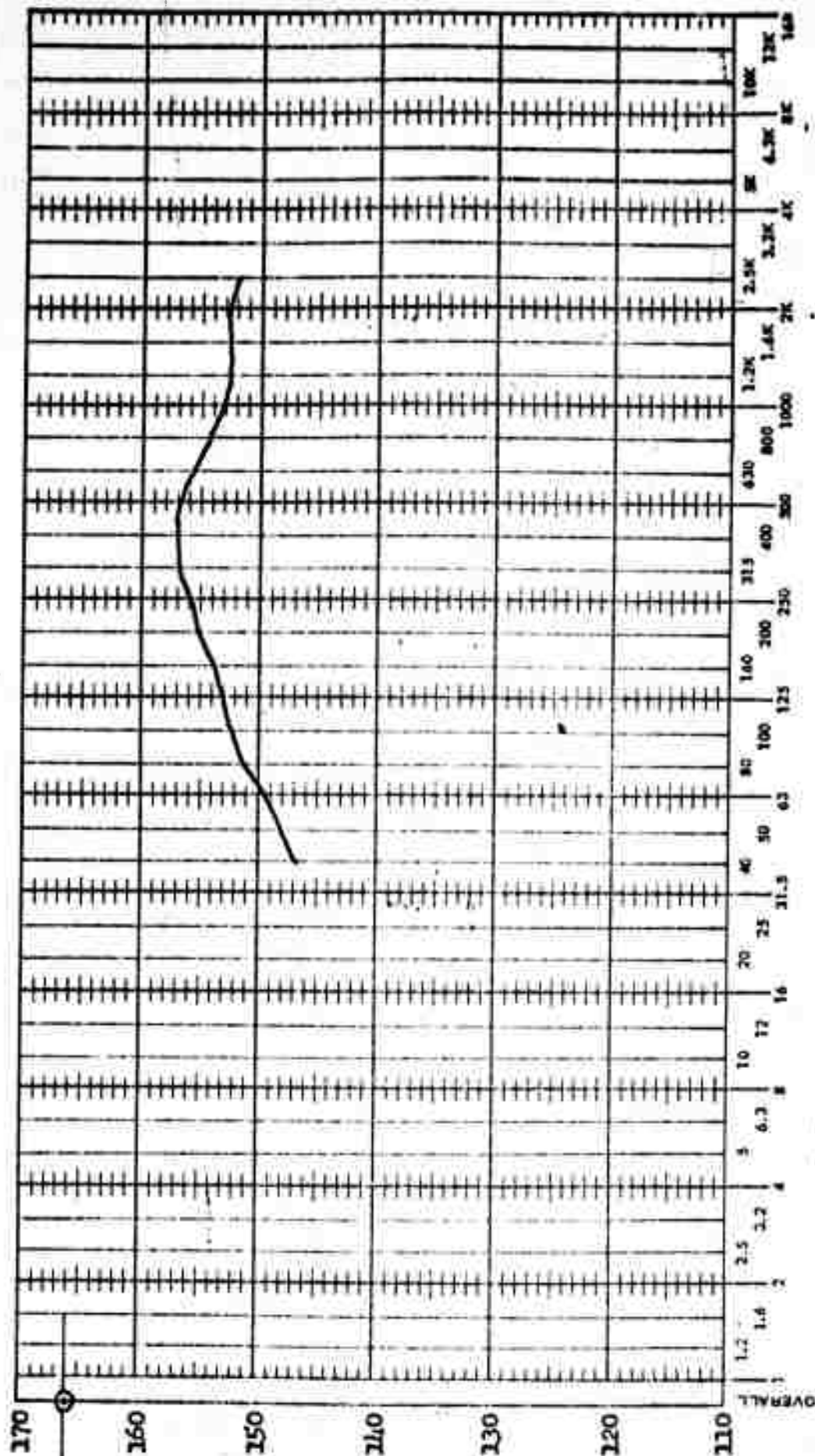
Test Point #28 Mach No. 0.82 Correlation No. 464

$\alpha = -7^\circ$   $\beta = -7^\circ$

Figure 14 (Continued)



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #28 Mach No. 0.84 Correlation No. 1466

Figure 14 (Continued)

$\alpha = 0^\circ$   $\beta = -2^\circ$

BOEING

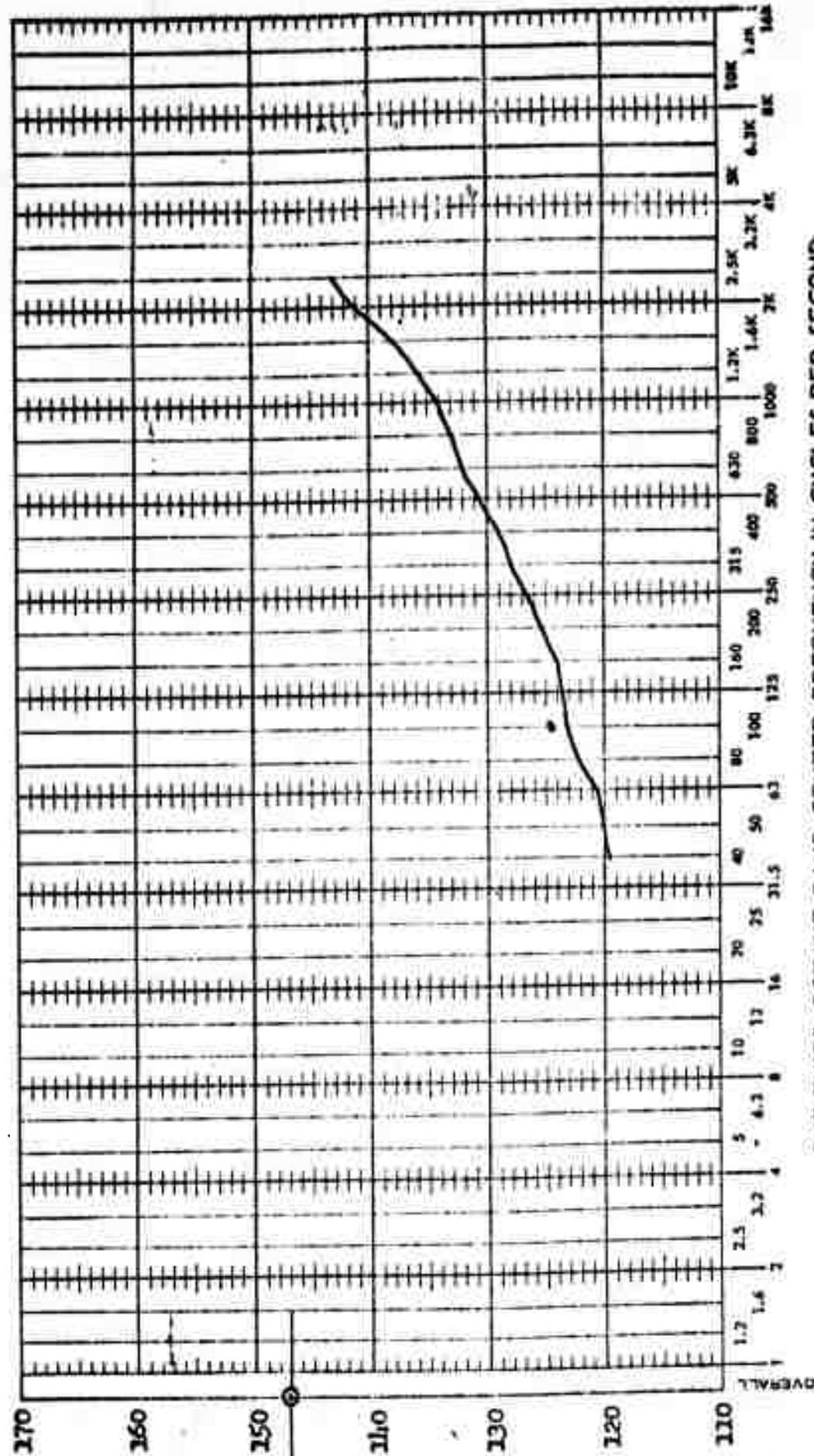
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



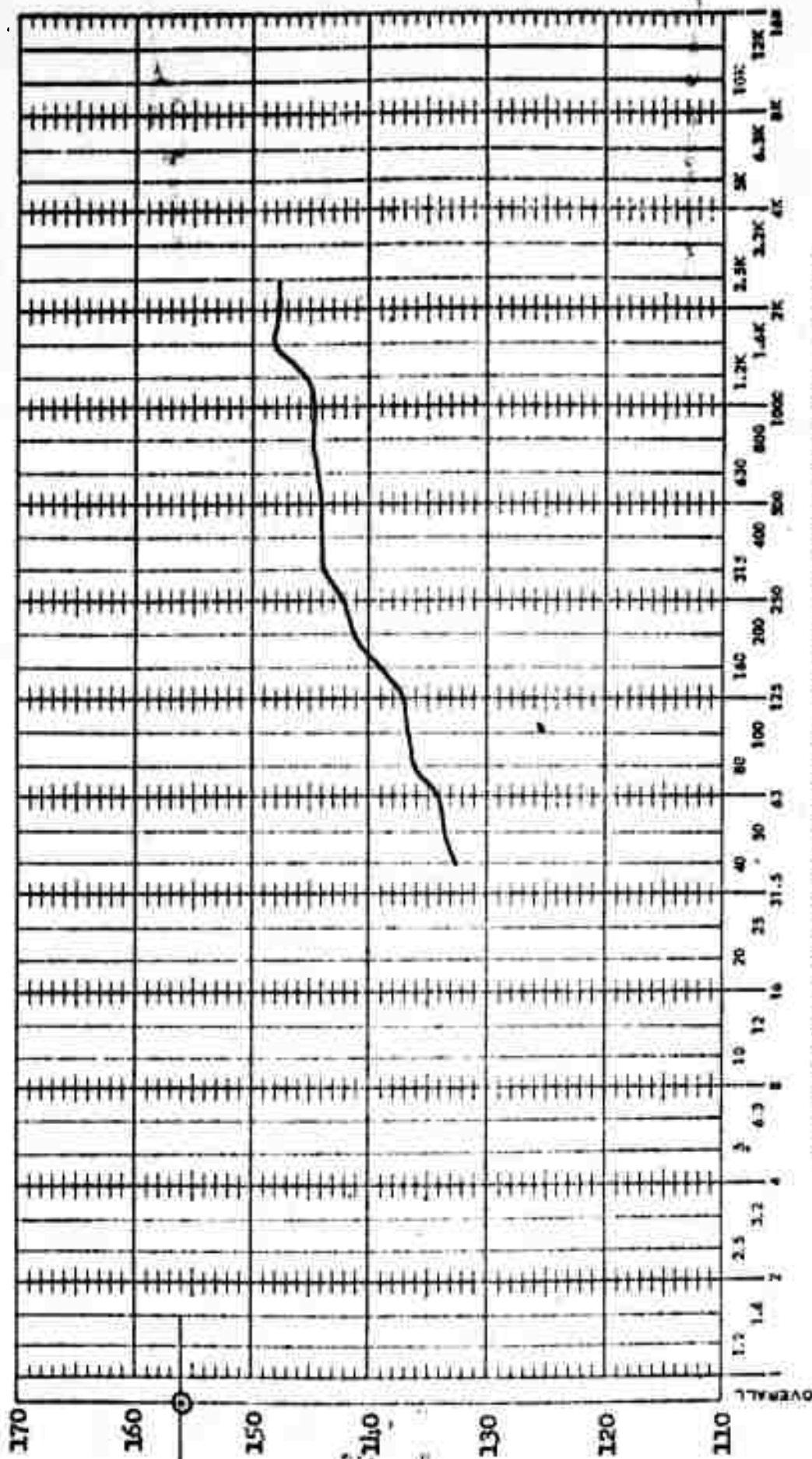
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #28 Mach No. 0.92 Correlation No. L77

$\delta = -4^\circ$   $\delta = -4^\circ$

Figure 14 (Continued)

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #29 Mach No. 0.75 Correlation No. 455

$\alpha = 4$

$\beta = -7$

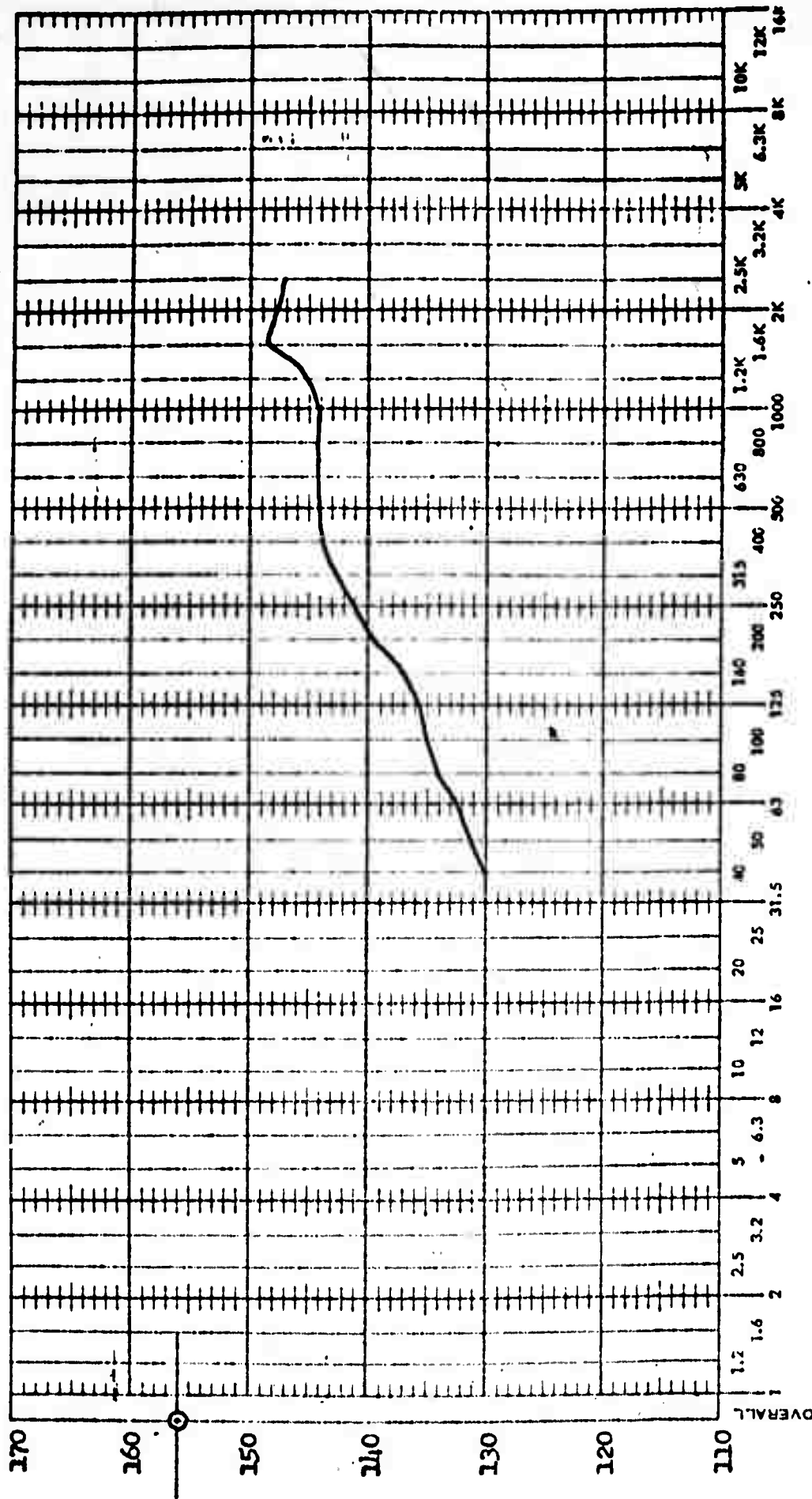
Figure 14 (Continued)





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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #29 Mach No. 0.75 Correlation No. 156  $\alpha = 0^\circ$   $\beta = -7^\circ$

Figure 14 (Continued)

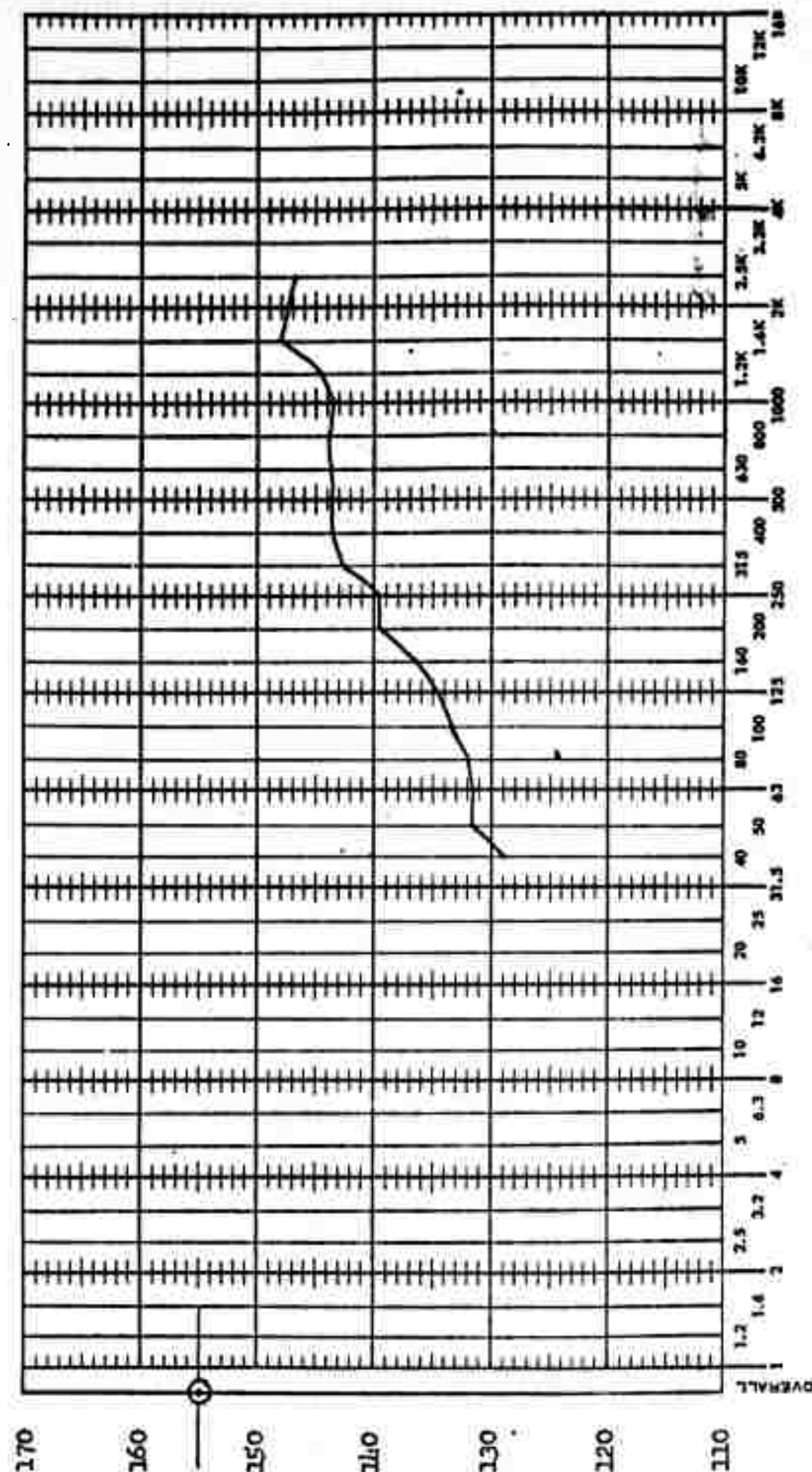
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR

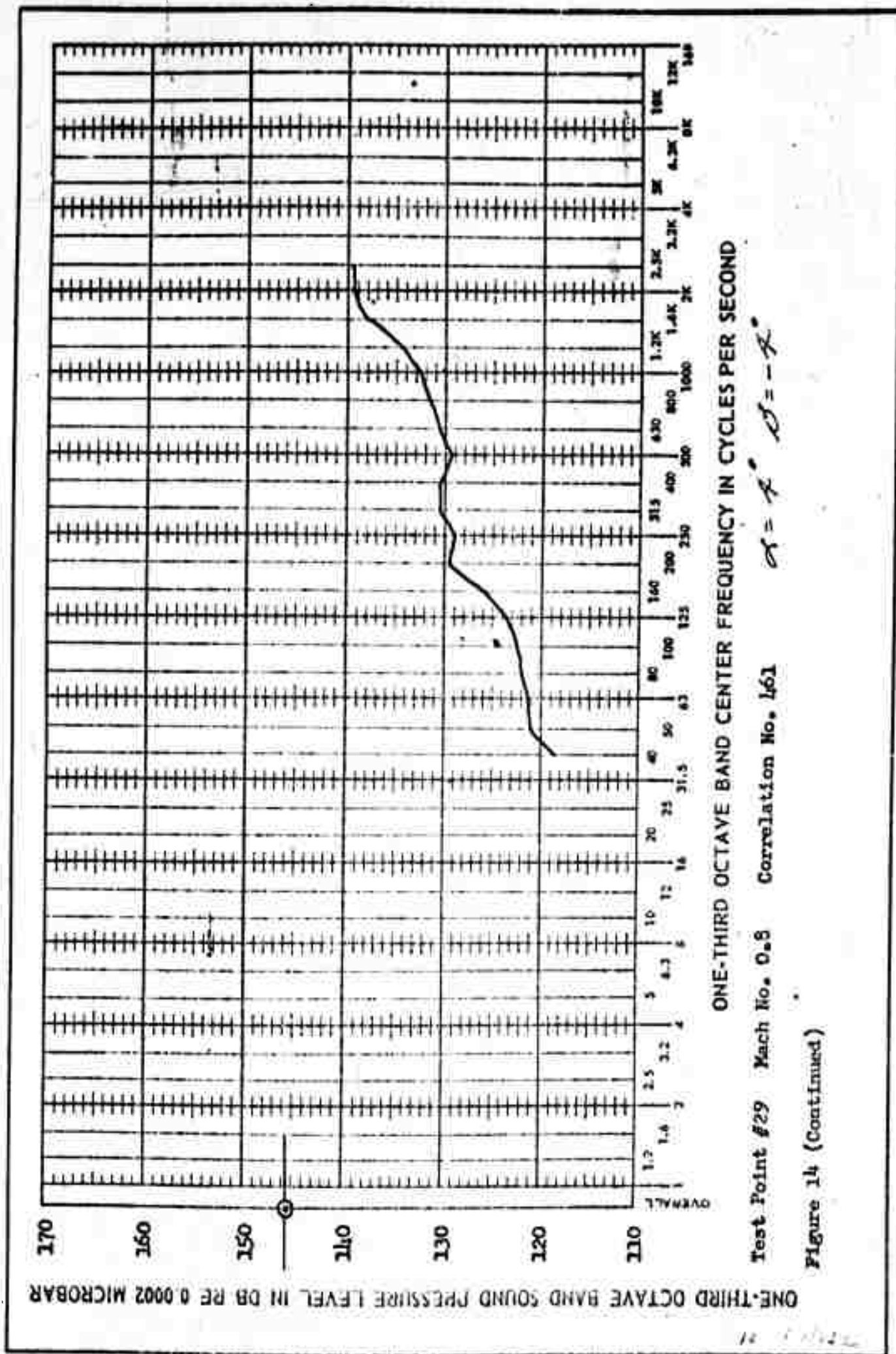


ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #29 Mach No. 0.75 Correlation No. 457

Figure 14 (Continued)

$\alpha = -4$   $\beta = -4$

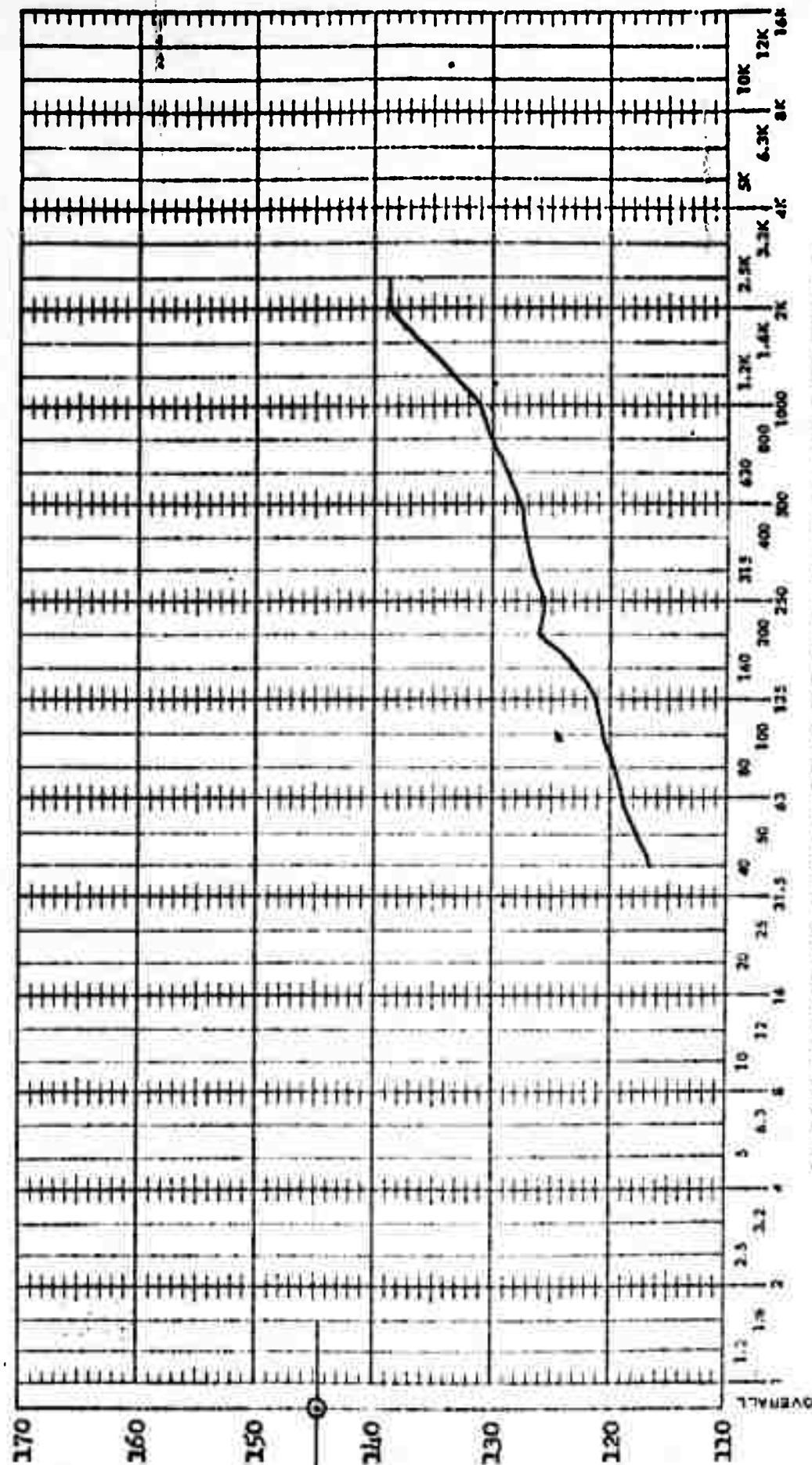


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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #29 Mach No. 0.82 Correlation No. 1462

$\alpha = 4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

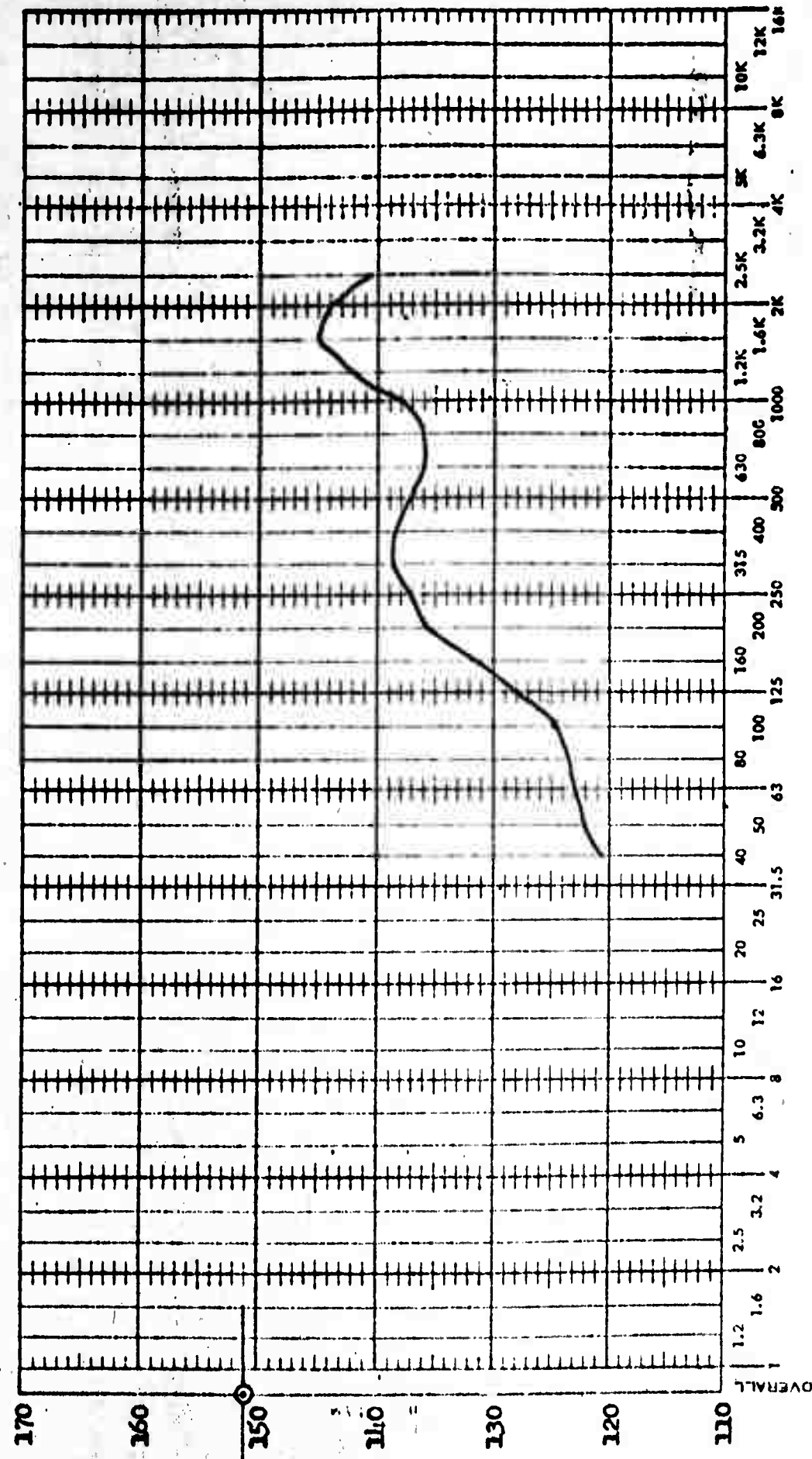
BOEING

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

$\alpha = 0^\circ$   $\beta = -4^\circ$

Test Point #30 Mach No. 0.8 Correlation No. 356

Figure 14 (Continued)

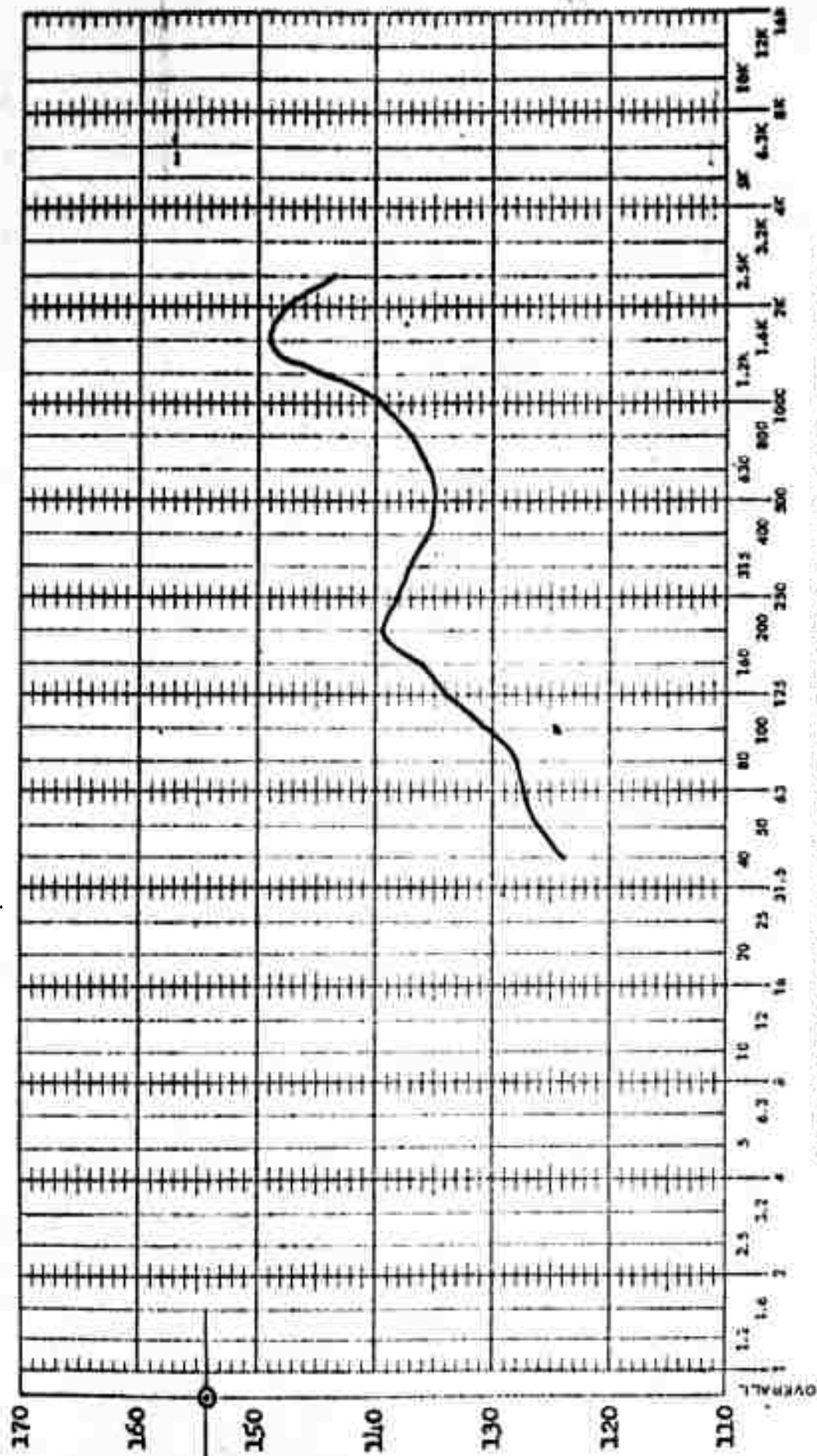
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

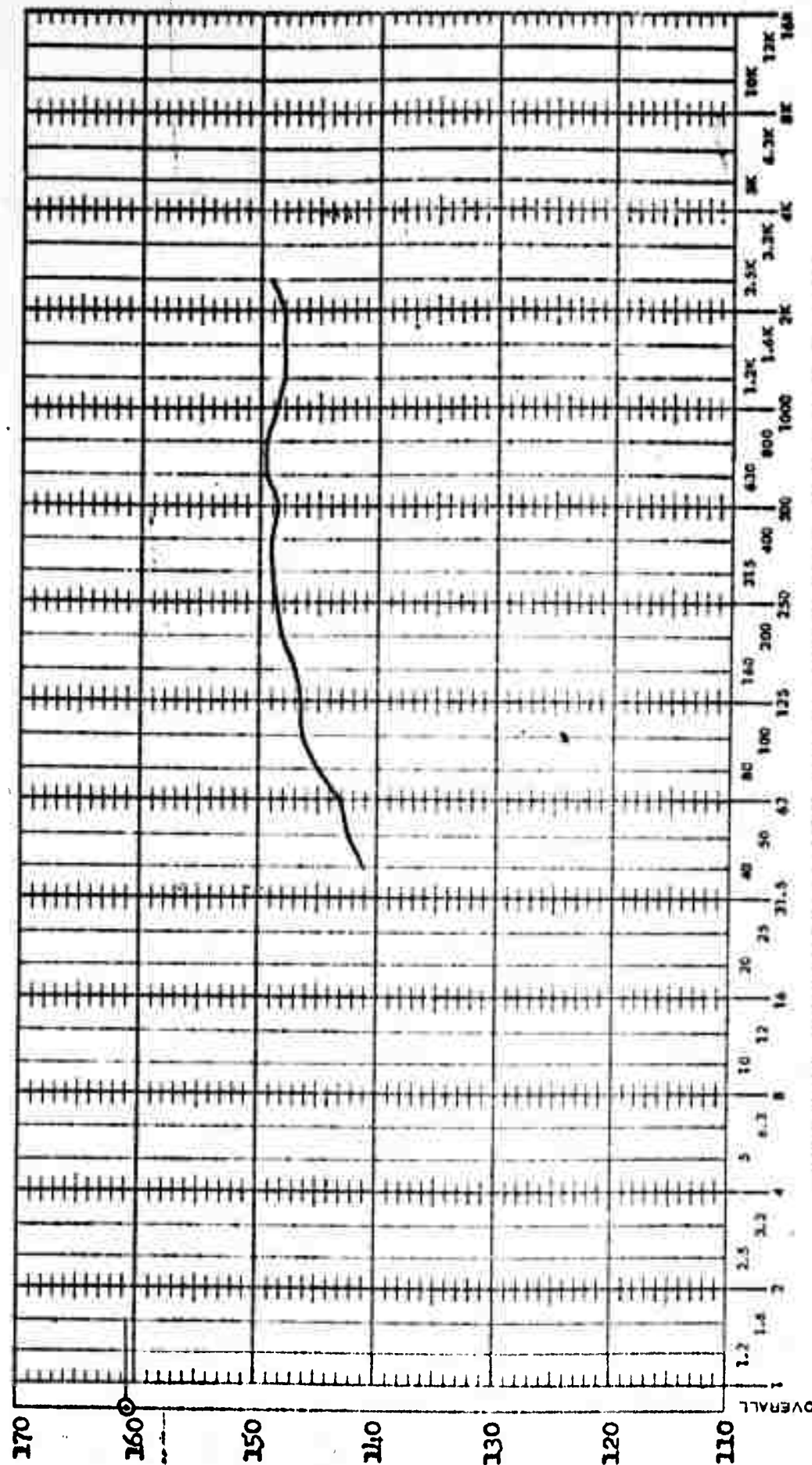
Test Point #30 Mach No. 0.85 Correlation No. 357

$\alpha = 0^\circ$   $\beta = -2^\circ$

Figure 14 (Continued)



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #30 Mach No. 0.95 Correlation No. 362

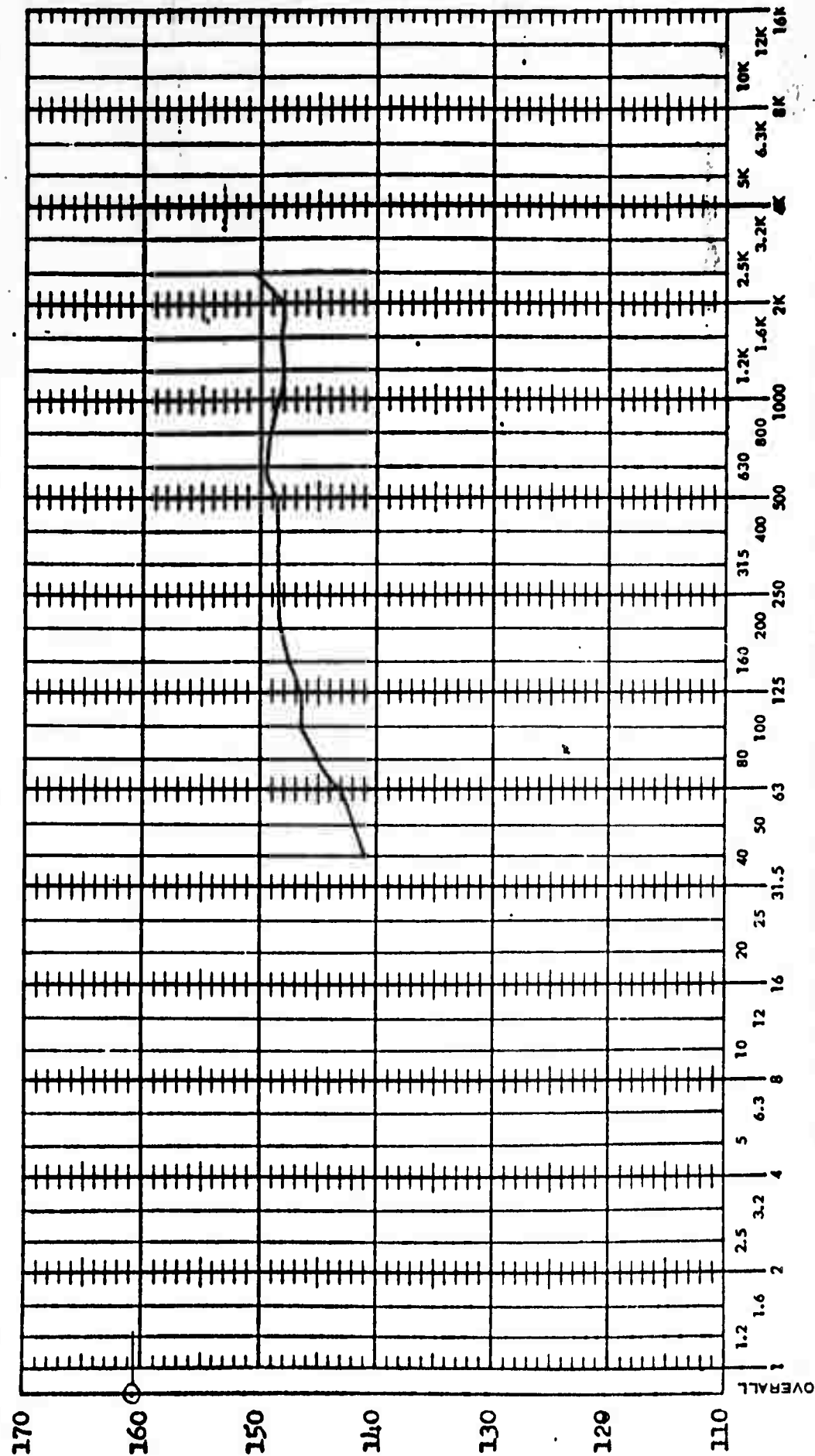
Figure 14 (Continued)

$\alpha = 0^\circ$   $\beta = 4^\circ$

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ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test point #30 Mach No. 1.0 Correlation No. 363

$\alpha = 0^\circ$   $\beta = 2^\circ$

Figure 14 (Continued)

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR

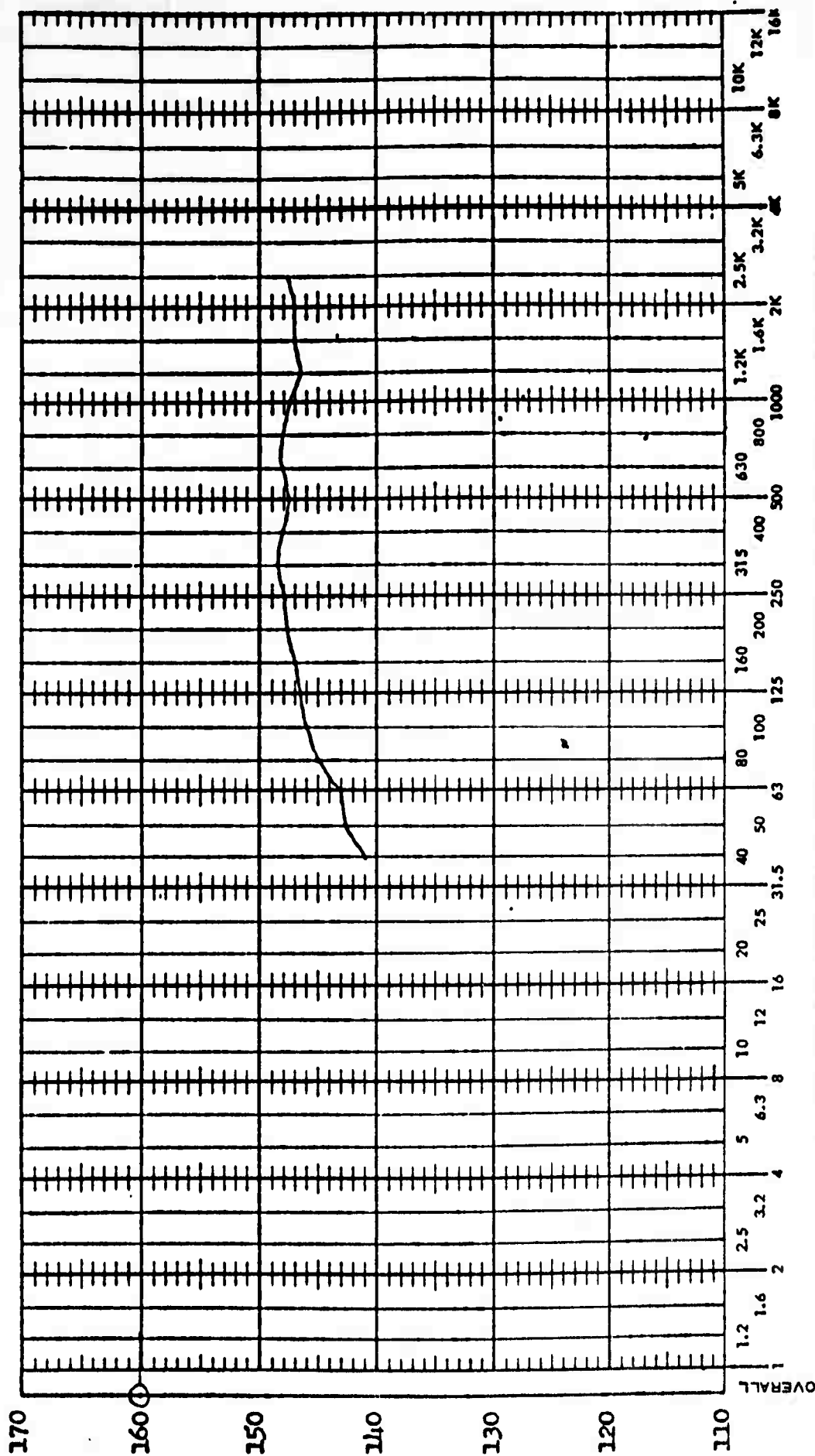
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test point #30 Mach No. 1.0 Correlation No. 364

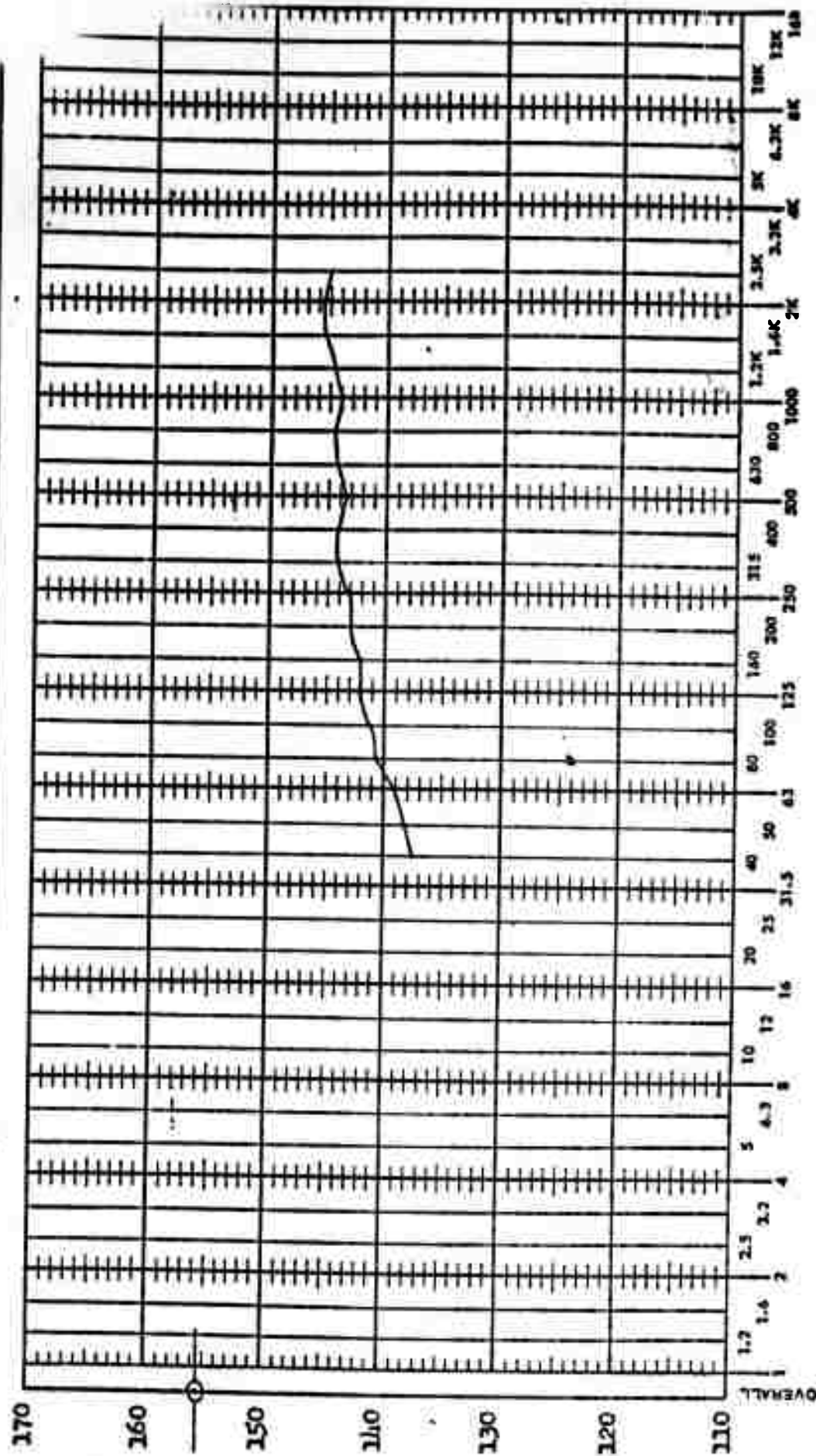
Figure 14 (Continued)

$\beta = 0^\circ$   $\beta = -2^\circ$



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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



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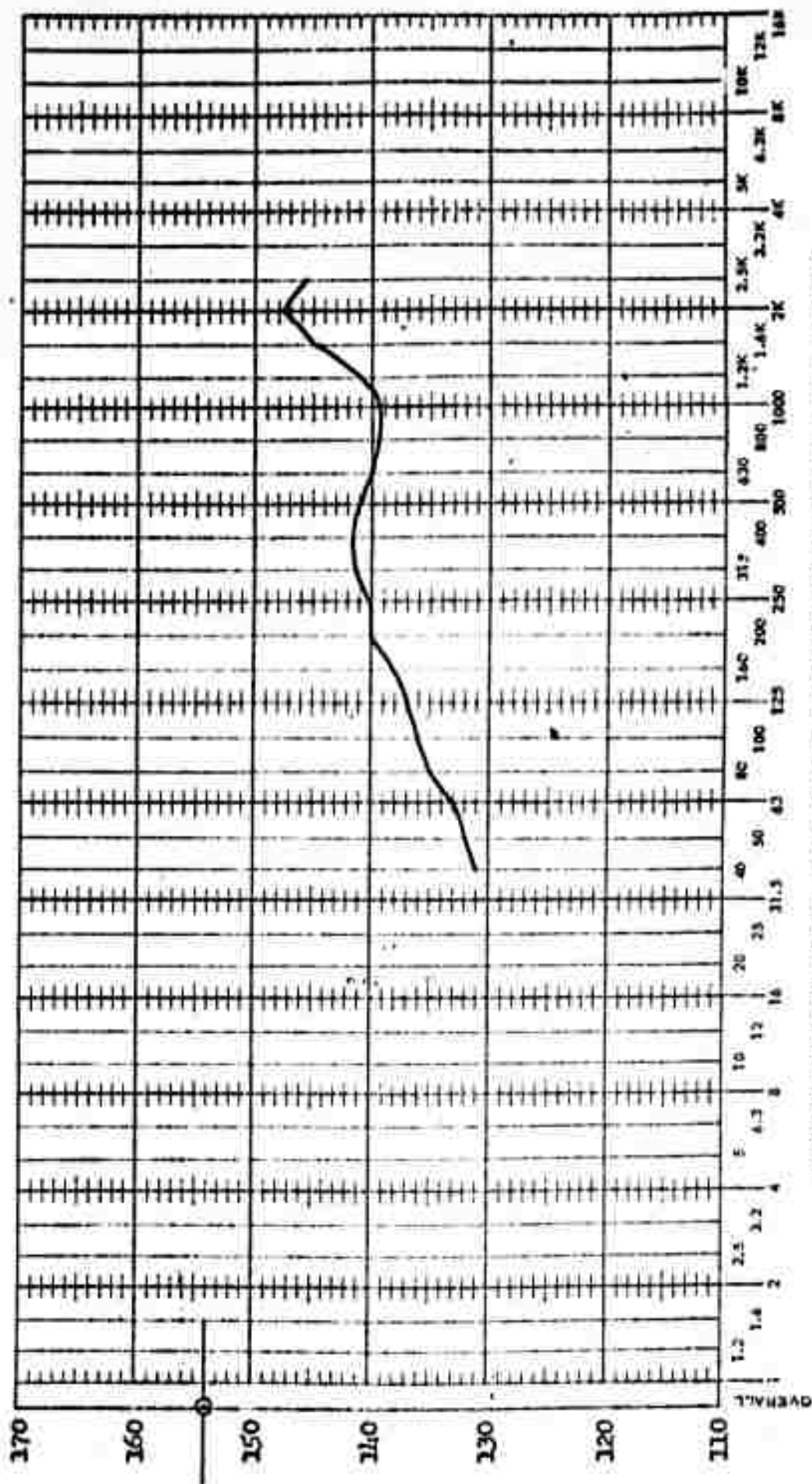
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #31 Mach No. 0.84 Correlation No. 467

$\alpha = 4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

BOEING

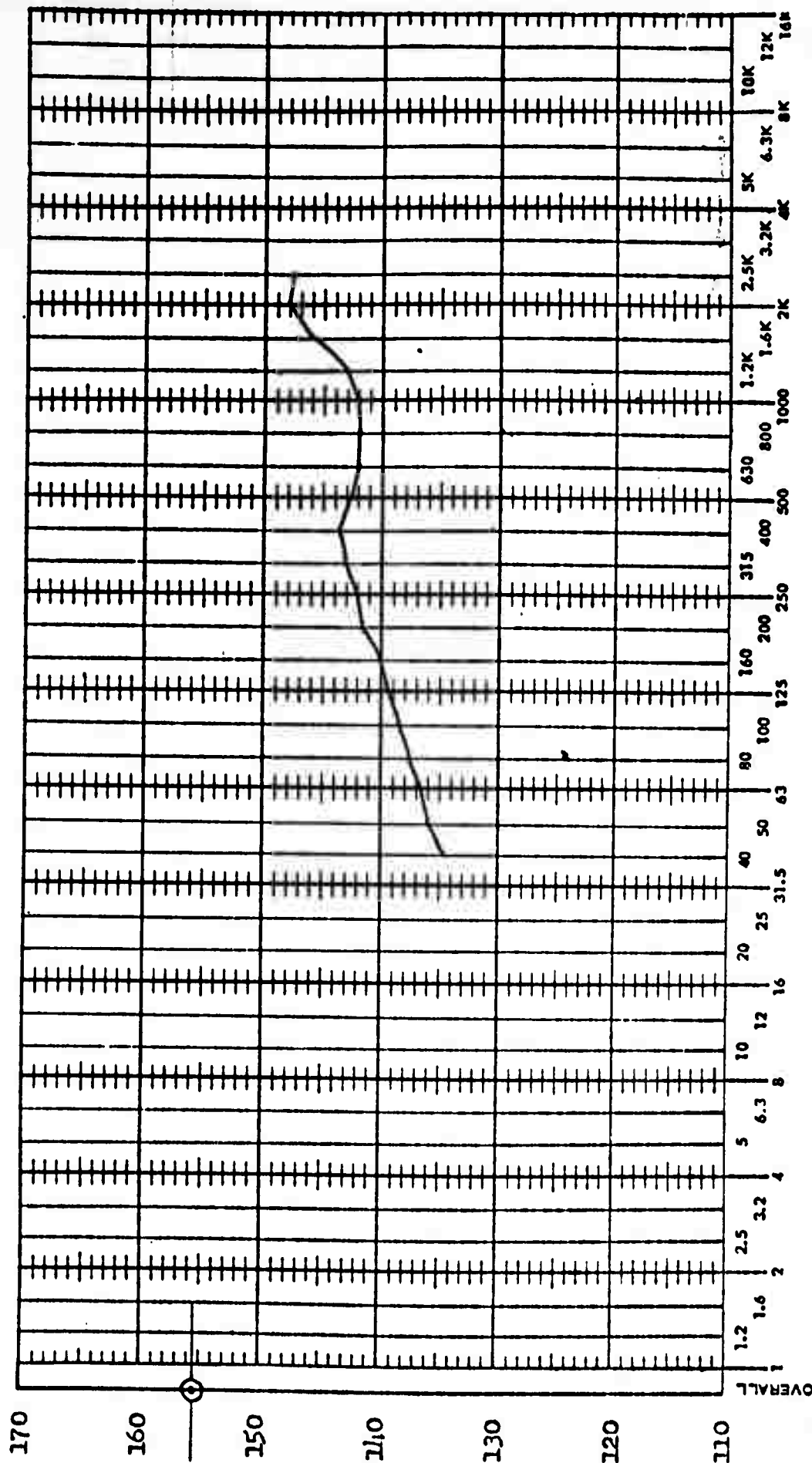
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #31 Mach No. 0.86 Correlation No. 1468

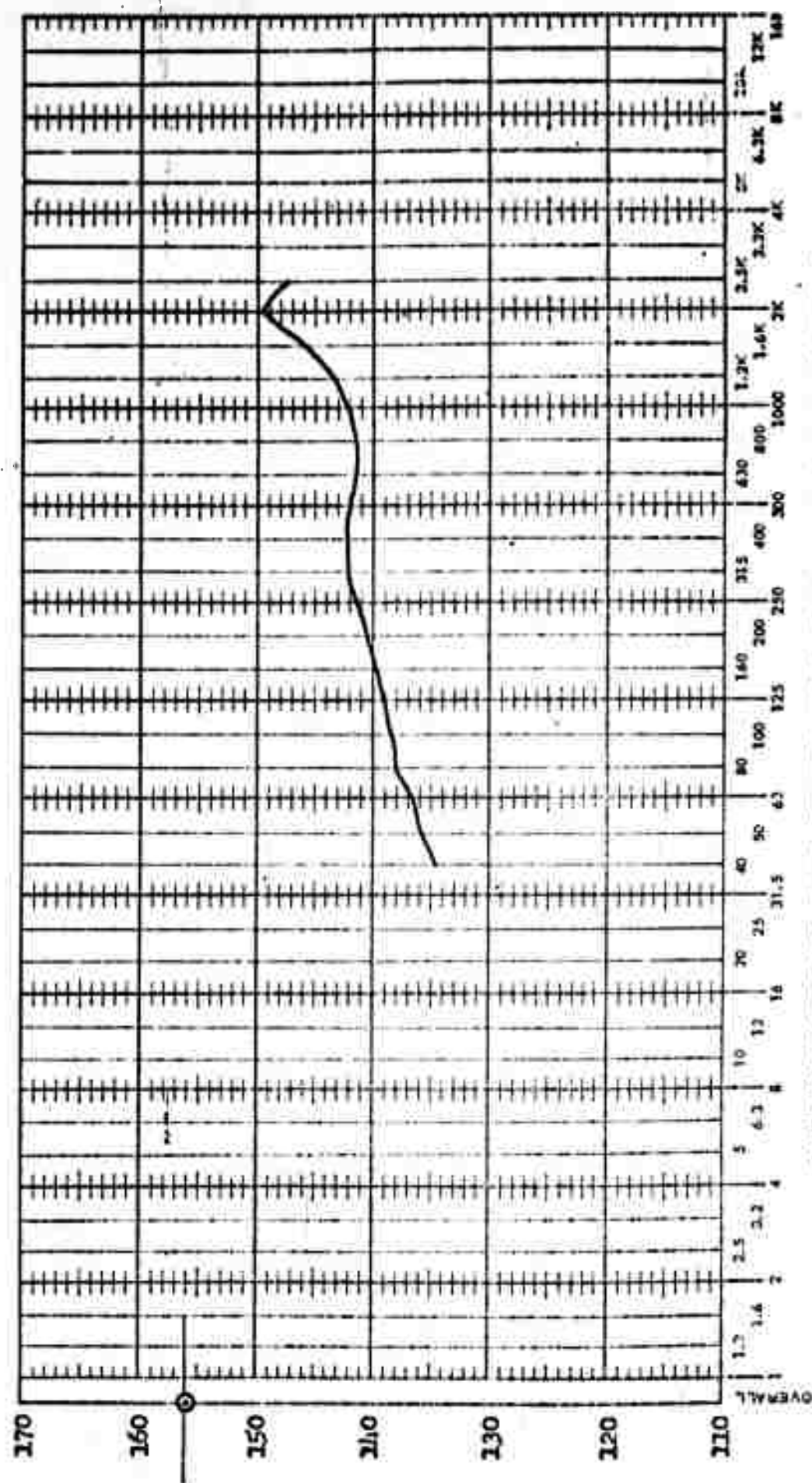
$\alpha = 4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)



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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #31 Mach No. 0.58 Correlation No. 473

$\alpha = 4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

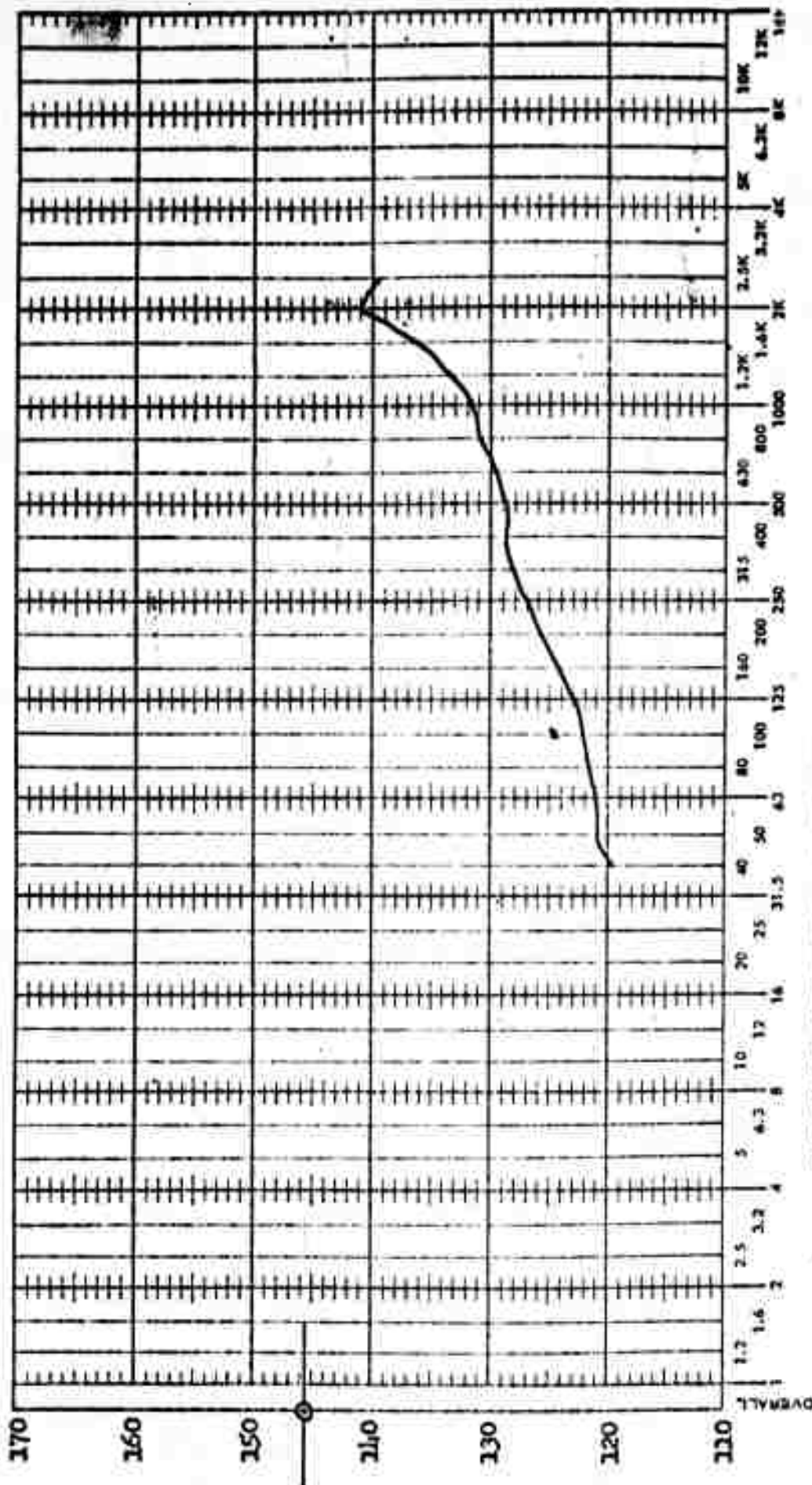
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



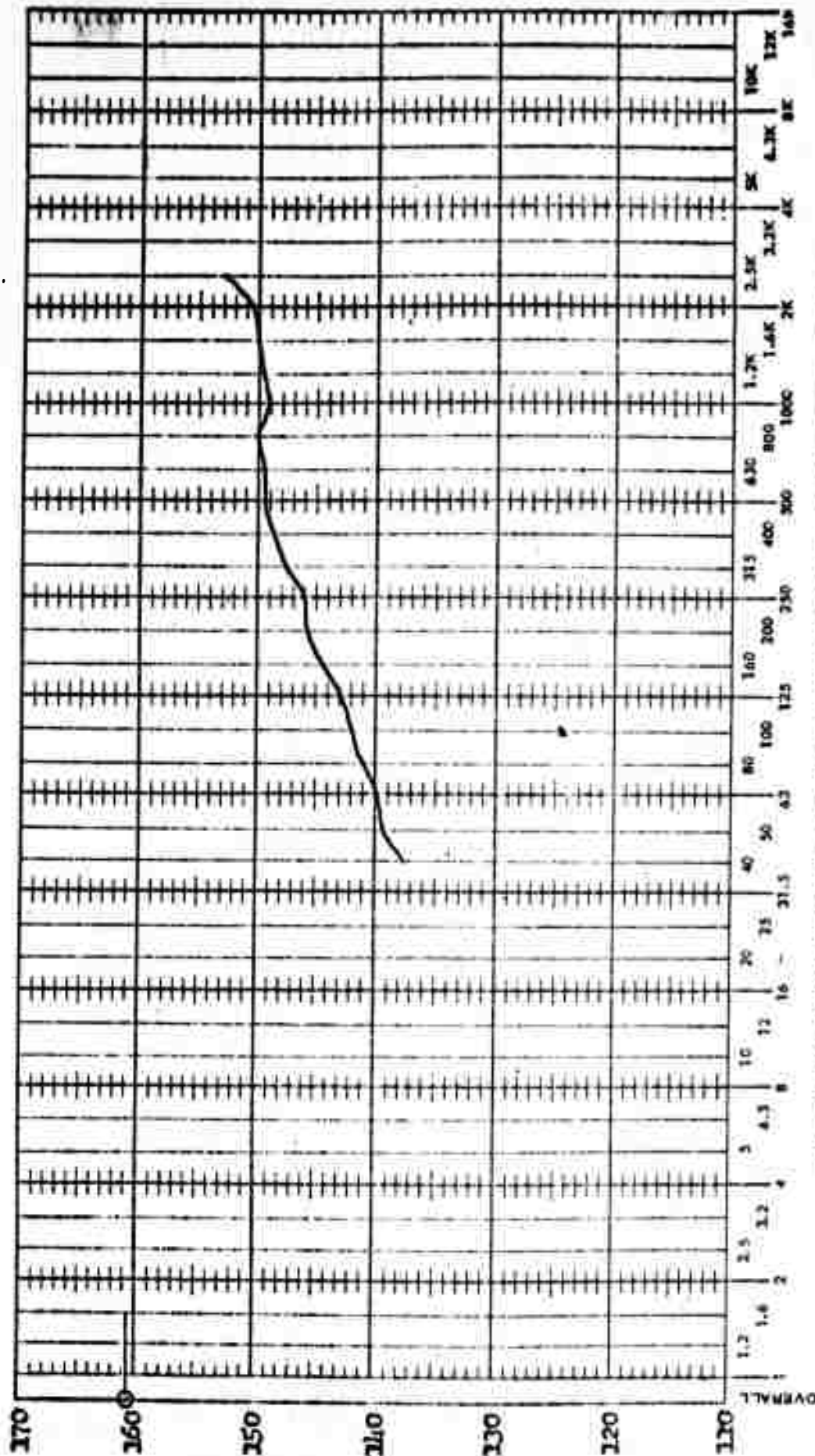
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #31 Mach No. 0.94 Correlation No. 480

Figure 14 (Continued)

$\alpha = 4^\circ$   $\beta = -8^\circ$

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

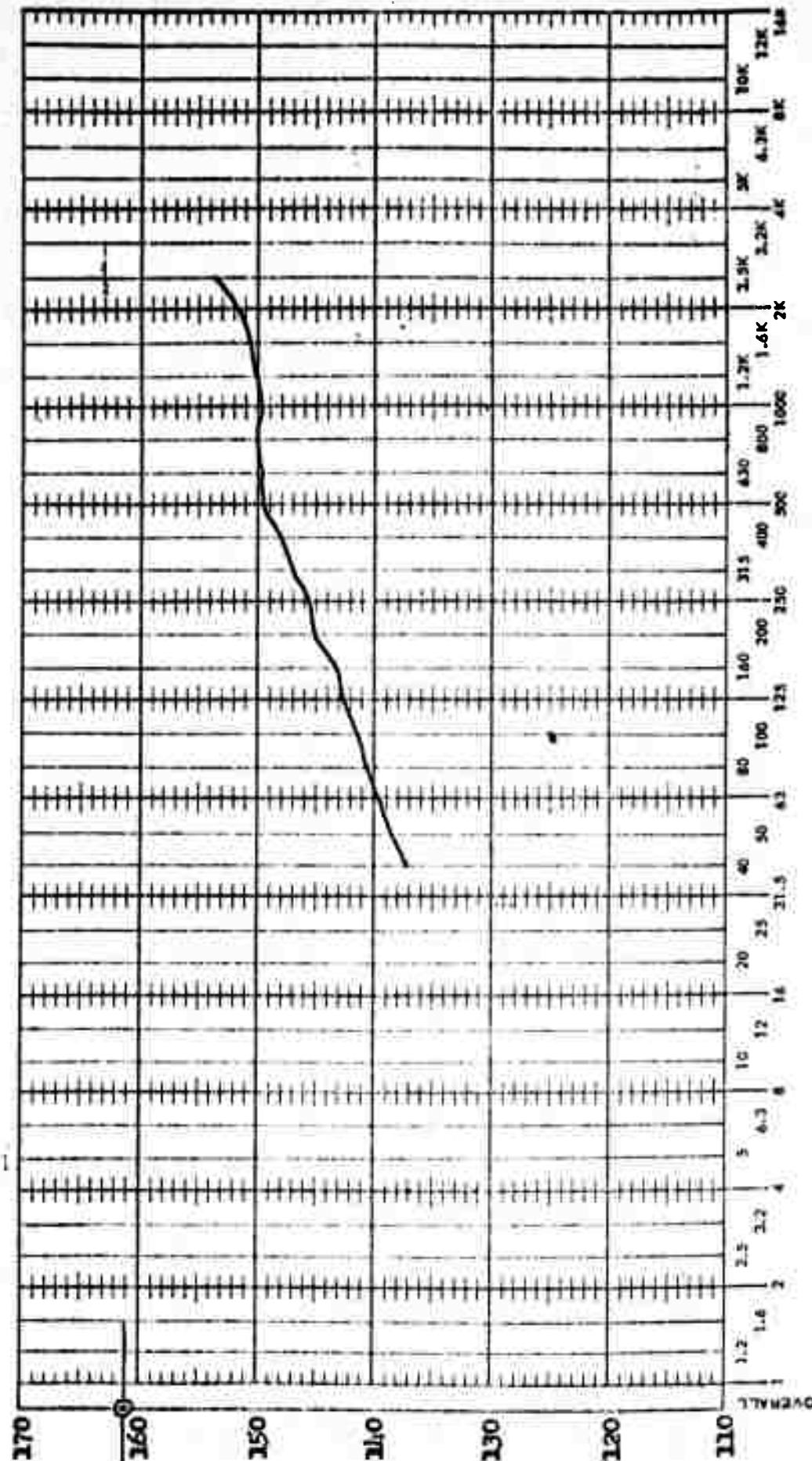
Test Point #32 Mach No. 0.8 Correlation No. 356

Figure 14 (Continued)

$\alpha = 0$   $\beta = -7$



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #32 Mach No. 0.85 Correlation No. 357

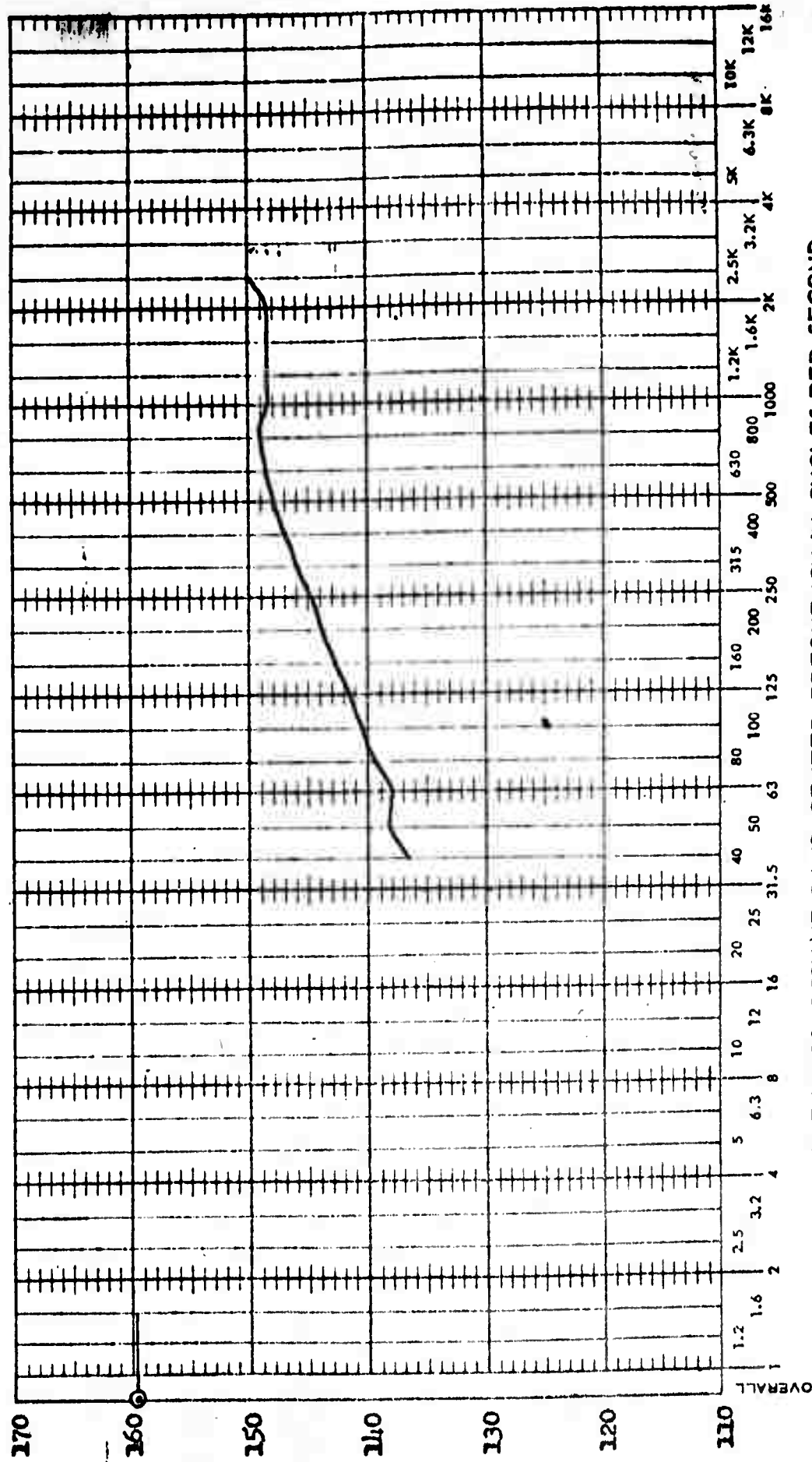
Figure 14 (Continued)

$\alpha = 0^\circ$   $\beta = -4^\circ$



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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #32 Mach No. 0.85 Correlation No. 358

$\alpha = 0^\circ$   $\beta = 4^\circ$

Figure 14 (Continued)

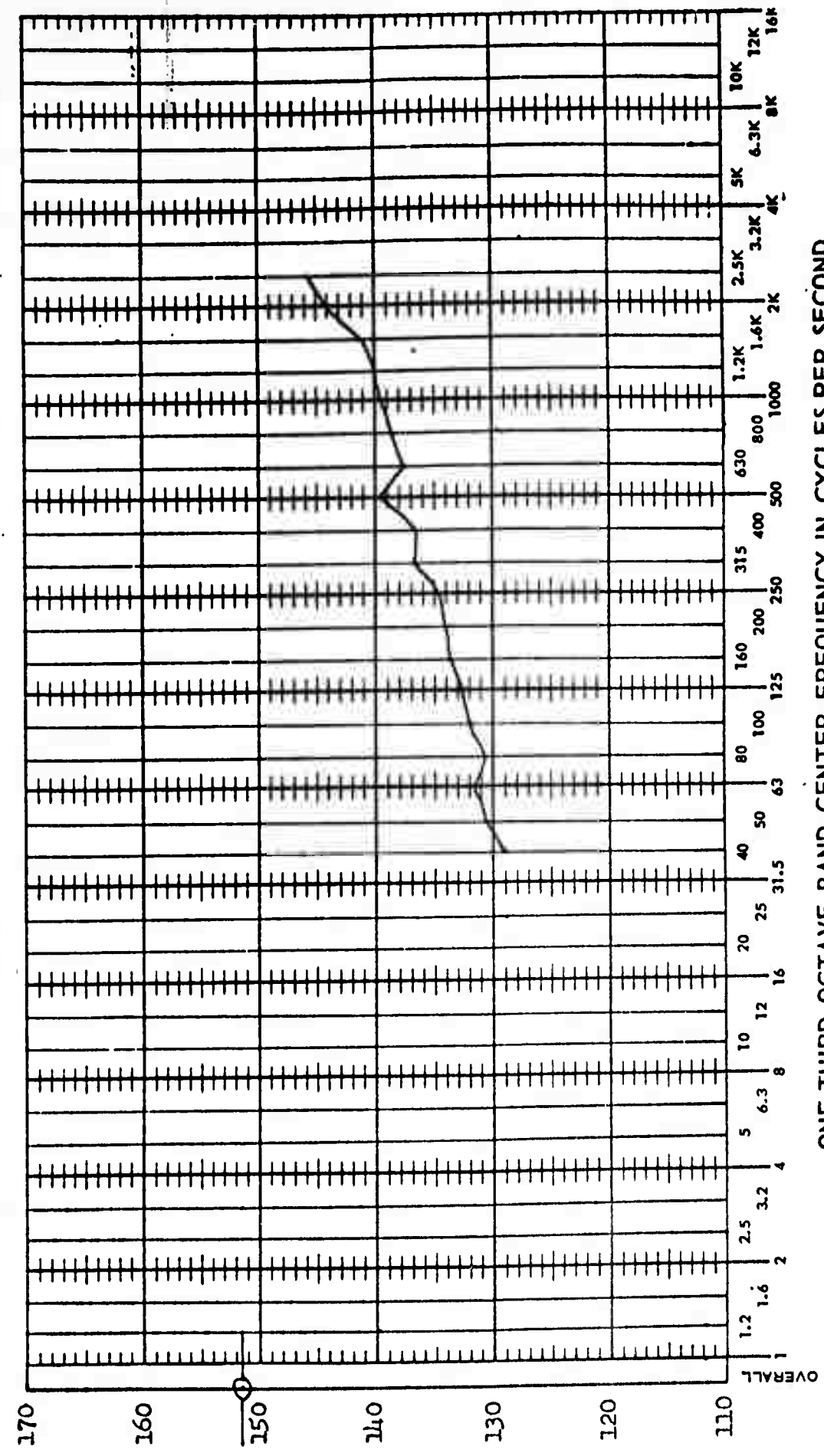
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test point #32 Mach No. 1.0 Correlation No. 406

$\alpha' = -4^\circ$   $\beta = -4$

Figure 14 (Continued)

BOEING

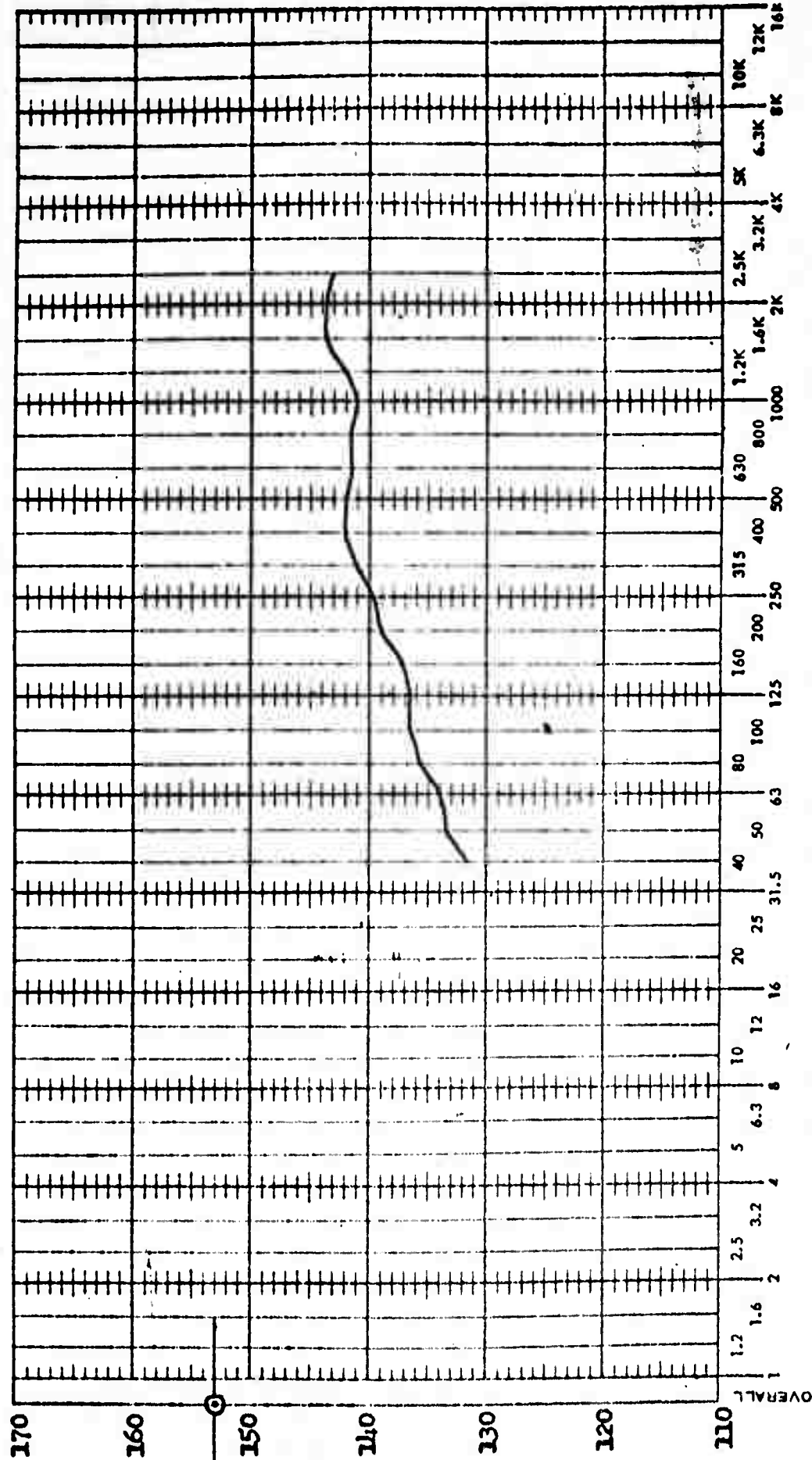
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #33 Mach No. 0.8 Correlation No. 461

Figure 14 (Continued)

$\alpha = 4^\circ$   $\beta = -4^\circ$

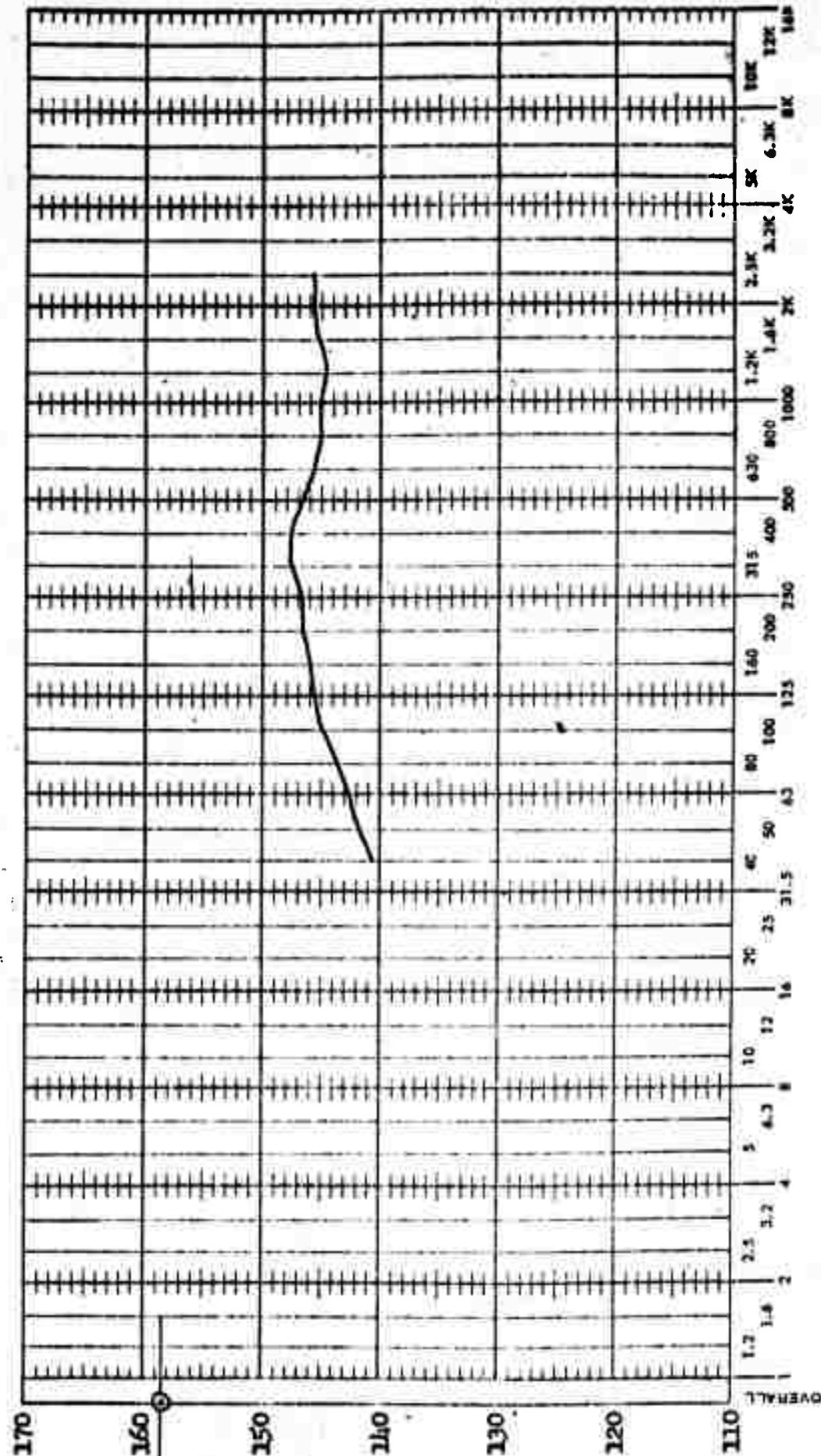
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #33 Mach No. 0.84 Correlation No. 467

Figure 14 (Continued)

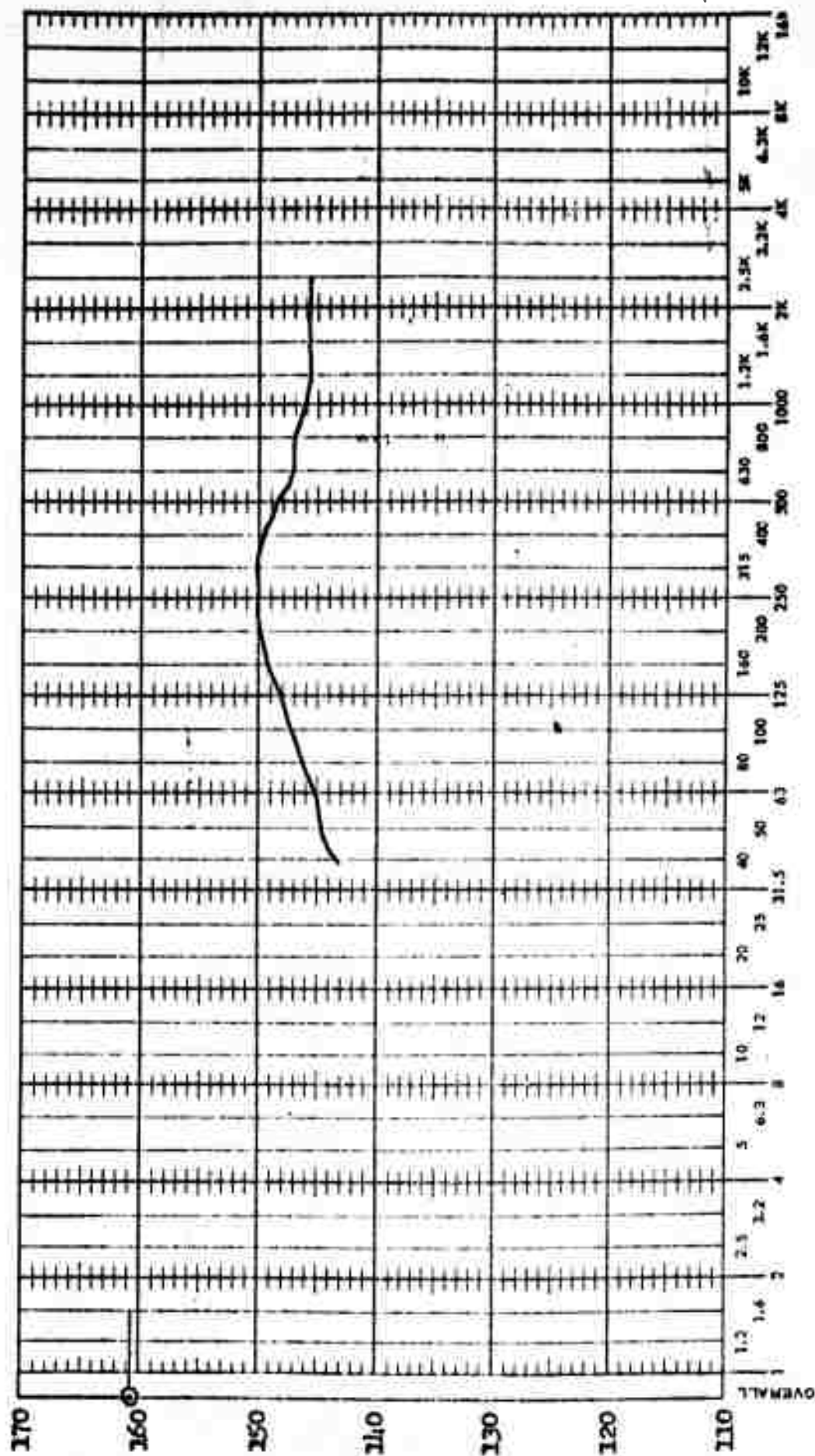
BOEING

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

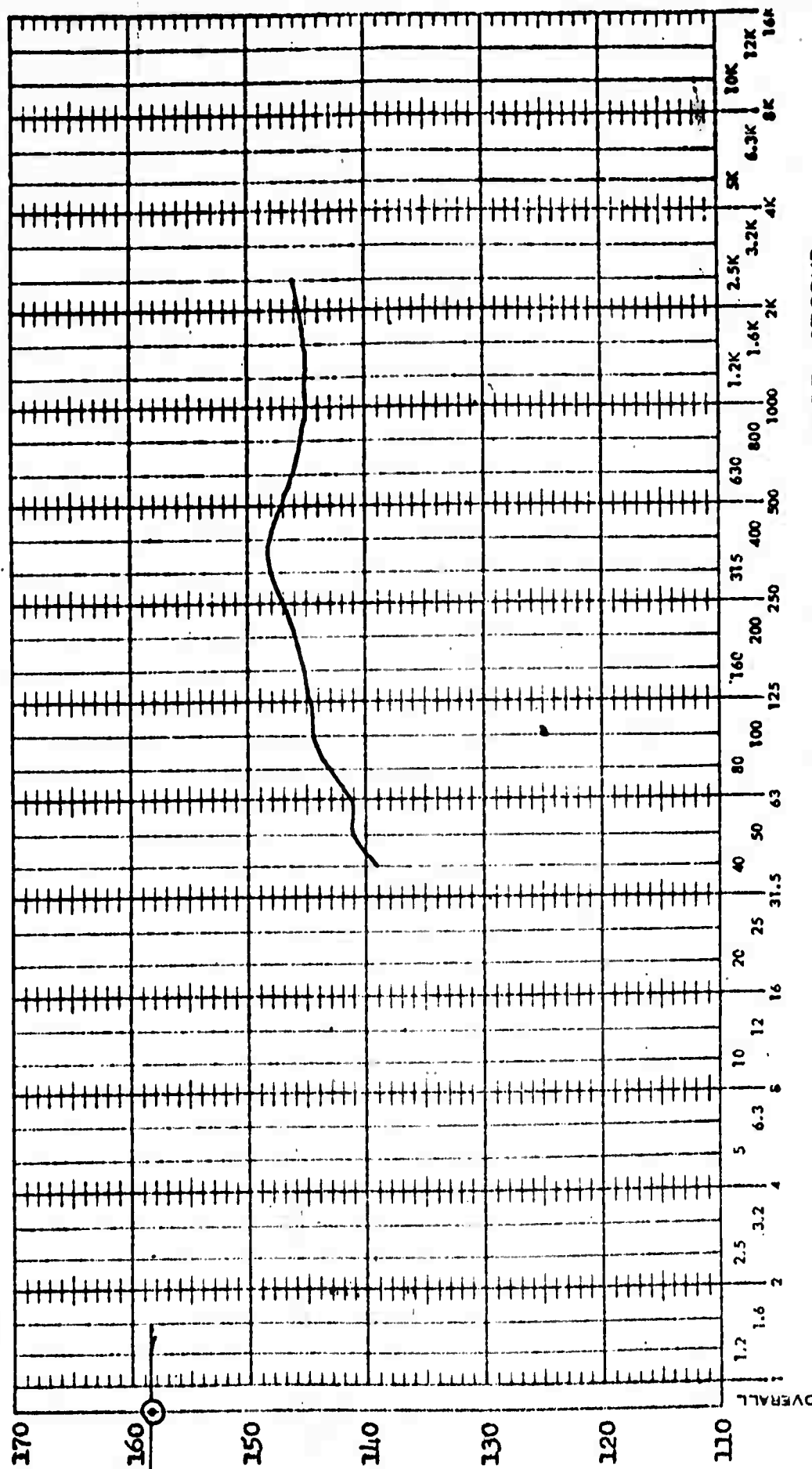
Test Point #33 Mach No. 0.86 Correlation No. 468

$\alpha = 2^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



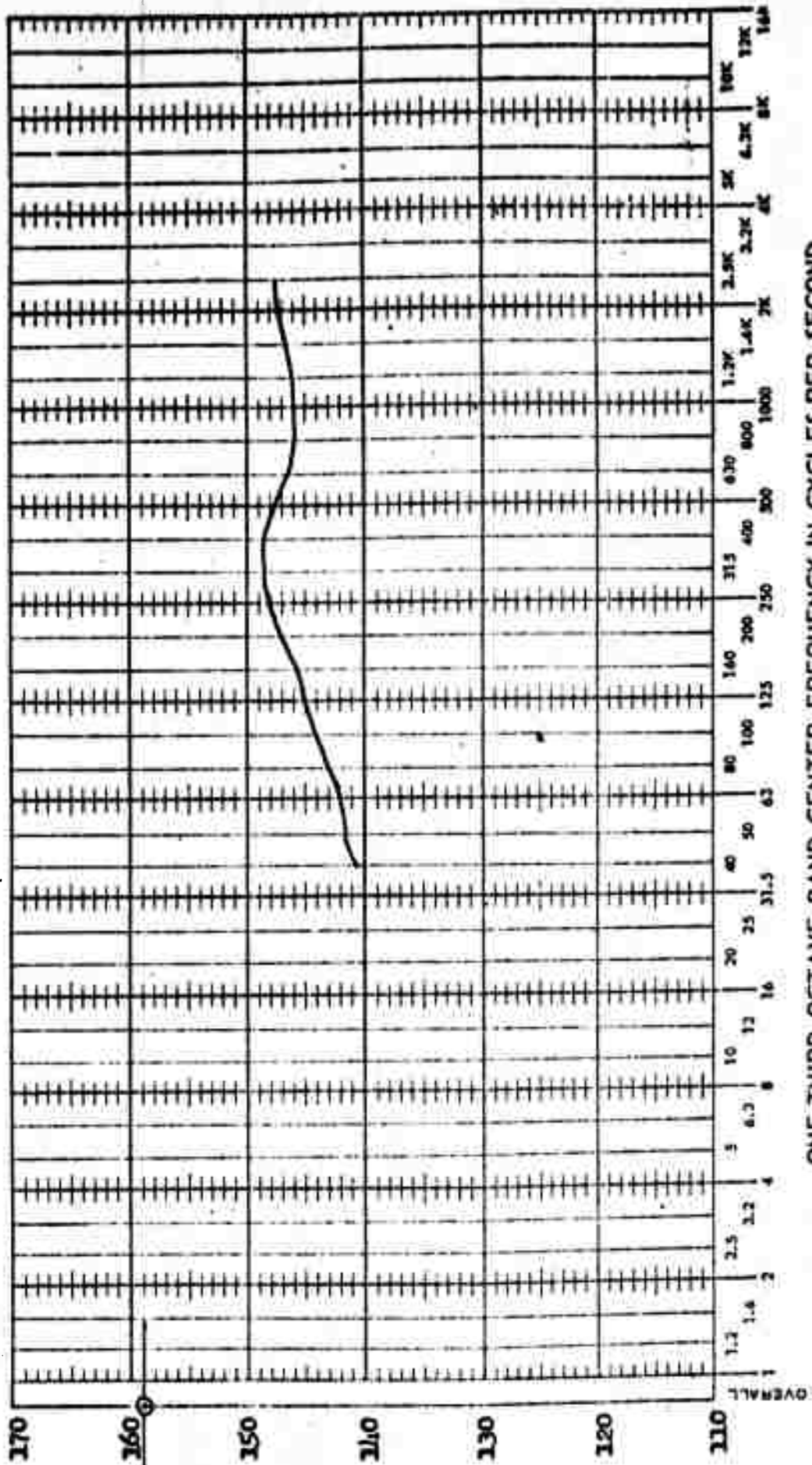
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #33 Mach No. 0.86 Correlation No. 469

$\alpha = 0$   $\beta = -4$

Figure 14 (Continued)

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



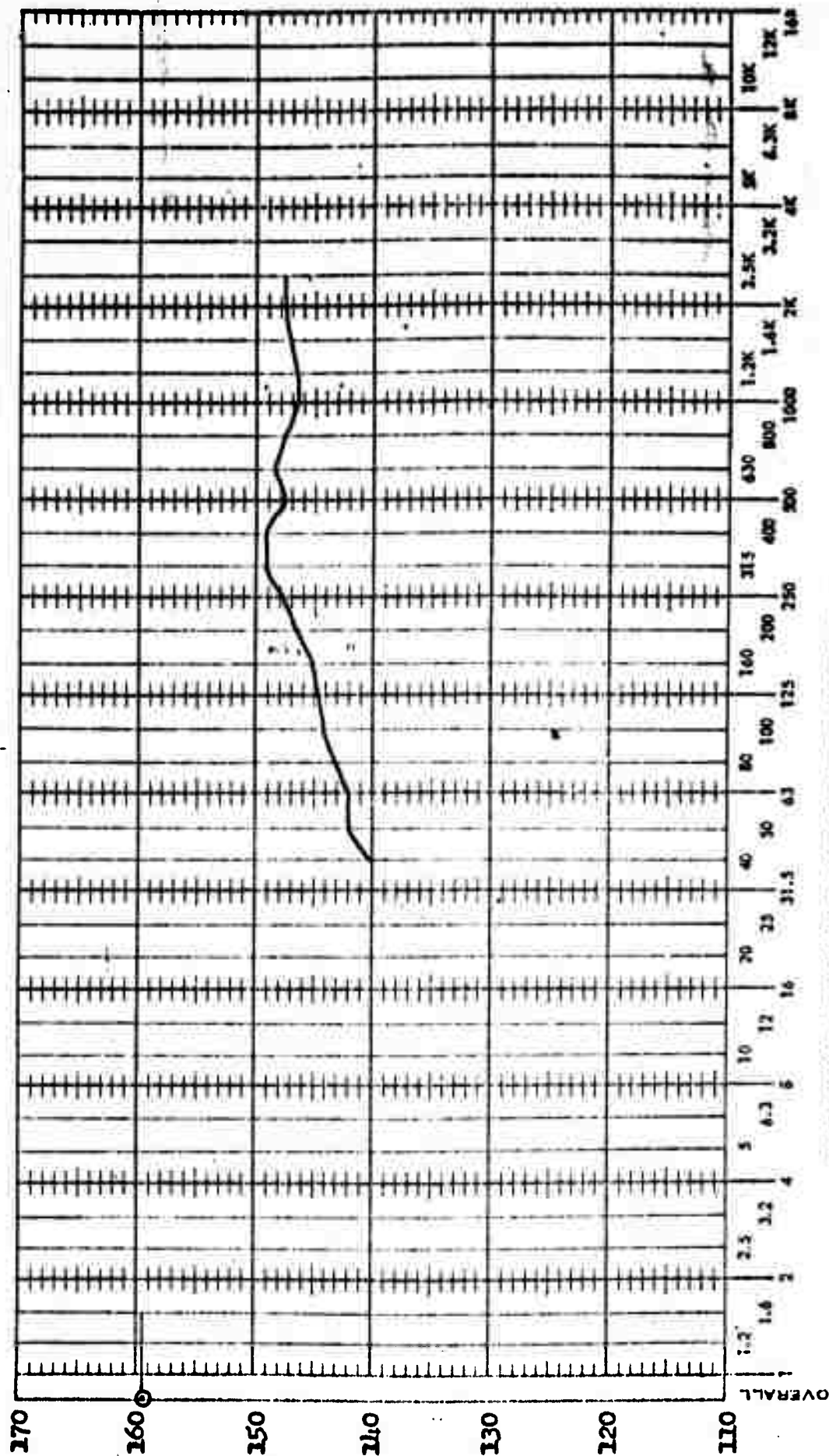
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #33 Mach No. 0.86 Correlation No. 470

Figure 14 (Continued)

$\alpha = -4^\circ$   $\beta = -4^\circ$

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

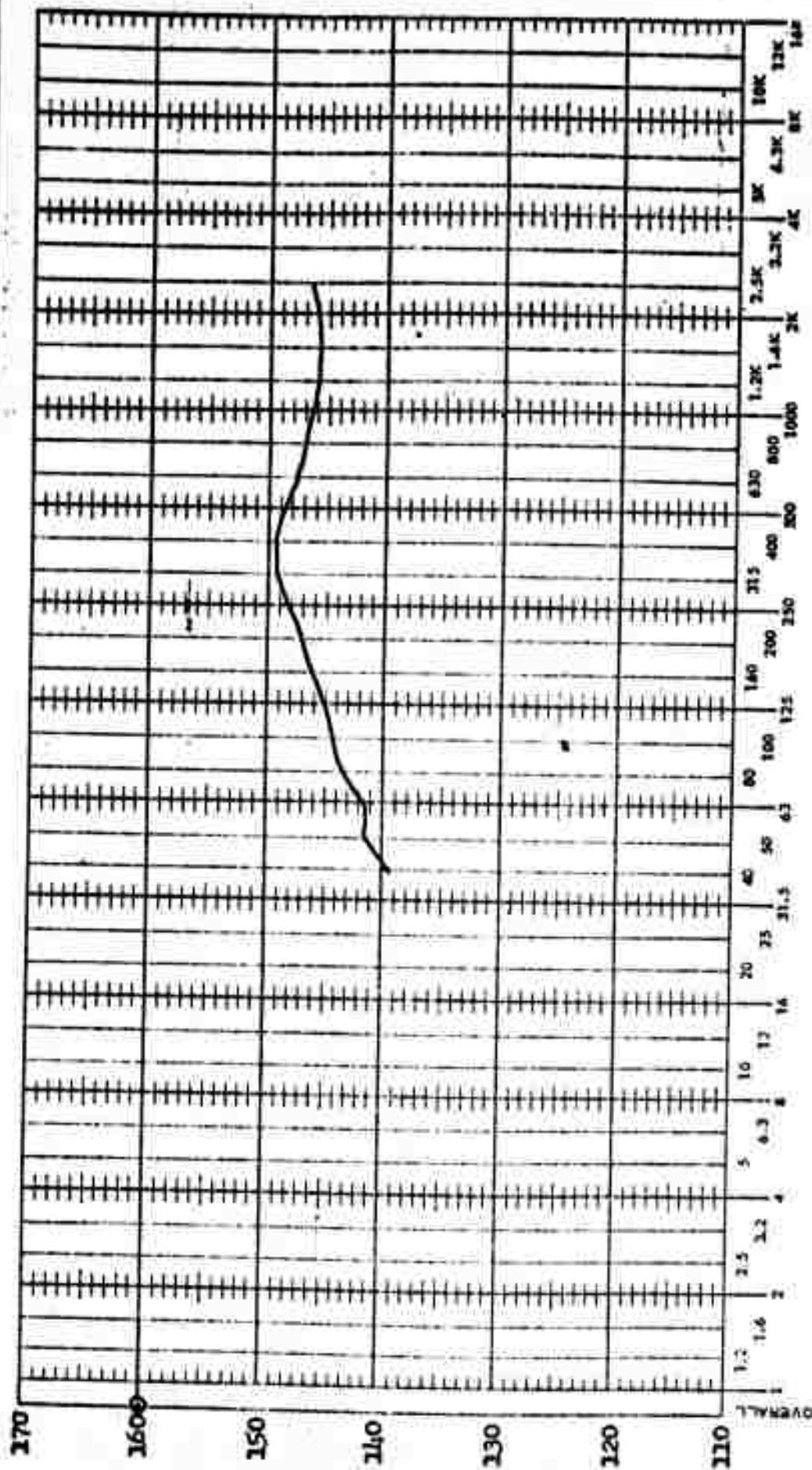
Test Point #33 Mach No. 0.88 Correlation No. 471

$\alpha = -4^\circ$   $\beta = -7^\circ$

Figure 14 (Continued)

182

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #33 Mach No. 0.88 Correlation No. 472

Figure 14 (Continued)

$\alpha = 0$   $\beta = -4$

BOEING

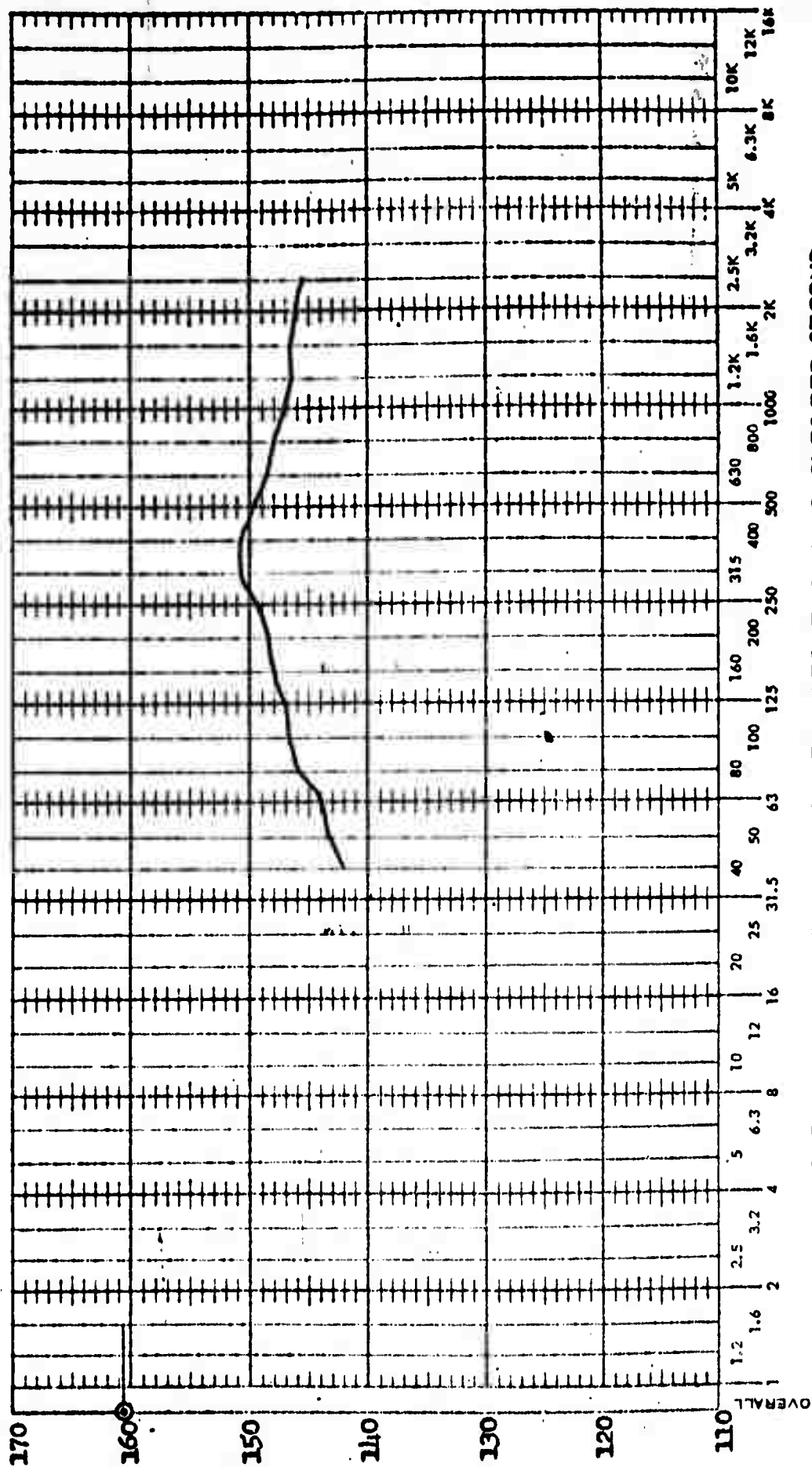
NO. 72-2648

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #33 Mach No. 0.88 Correlation No. 473

$\alpha = 4^\circ$   $\beta = -2^\circ$

Figure 14 (Continued)

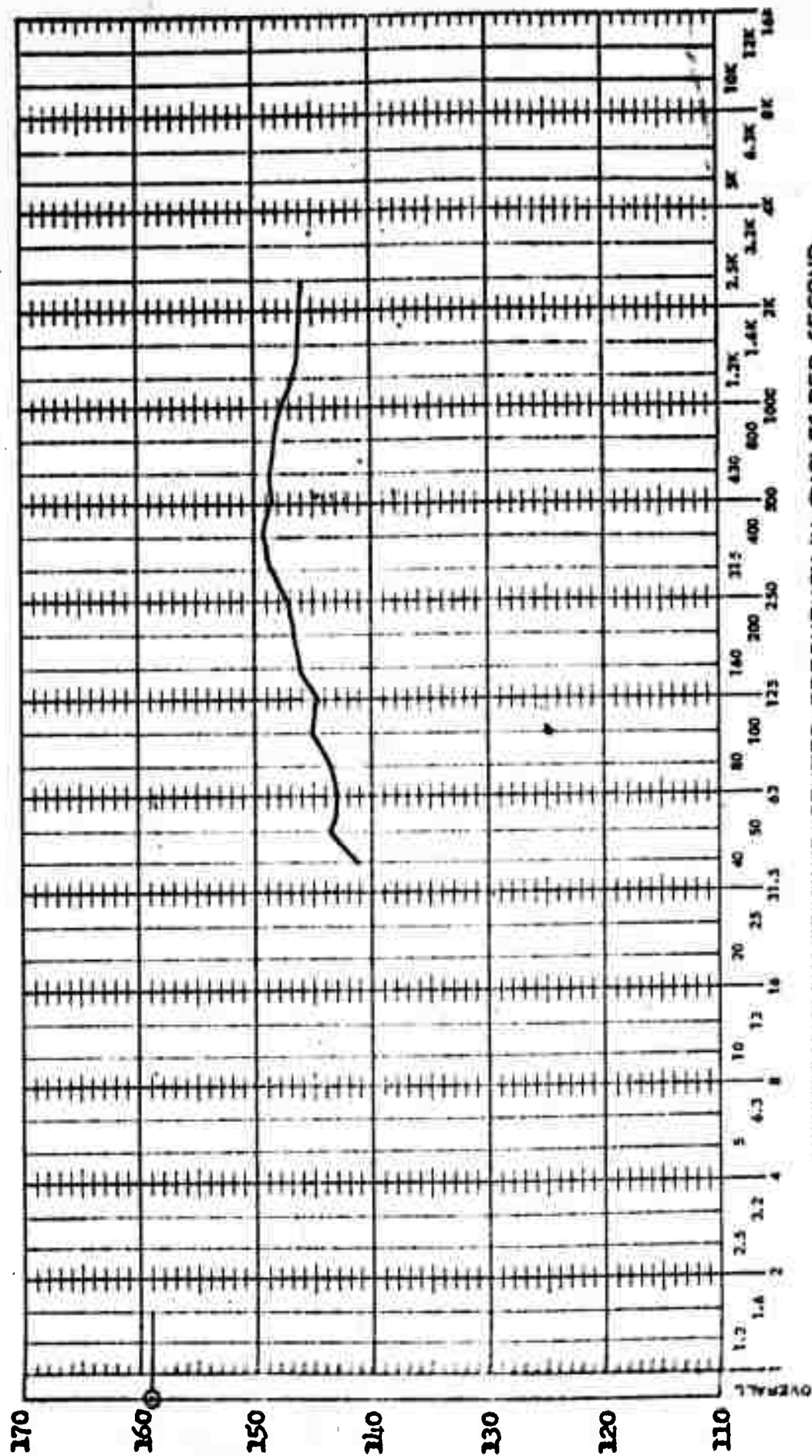
BOEING

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

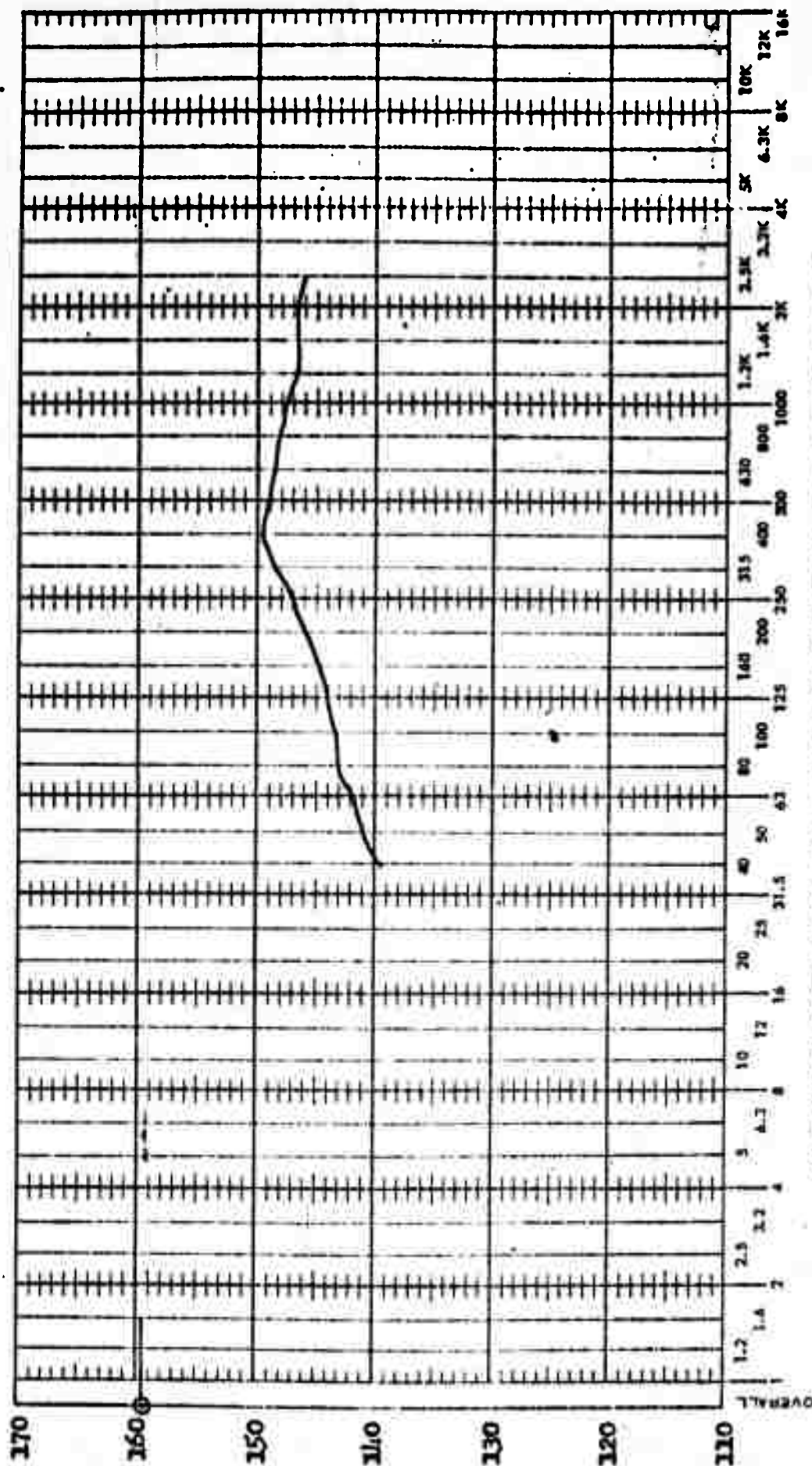
Test Point #33 Mach No. 0.9 Correlation No. 474

$\alpha = 4^\circ$   $\beta = -2^\circ$

Figure 14 (Continued)



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

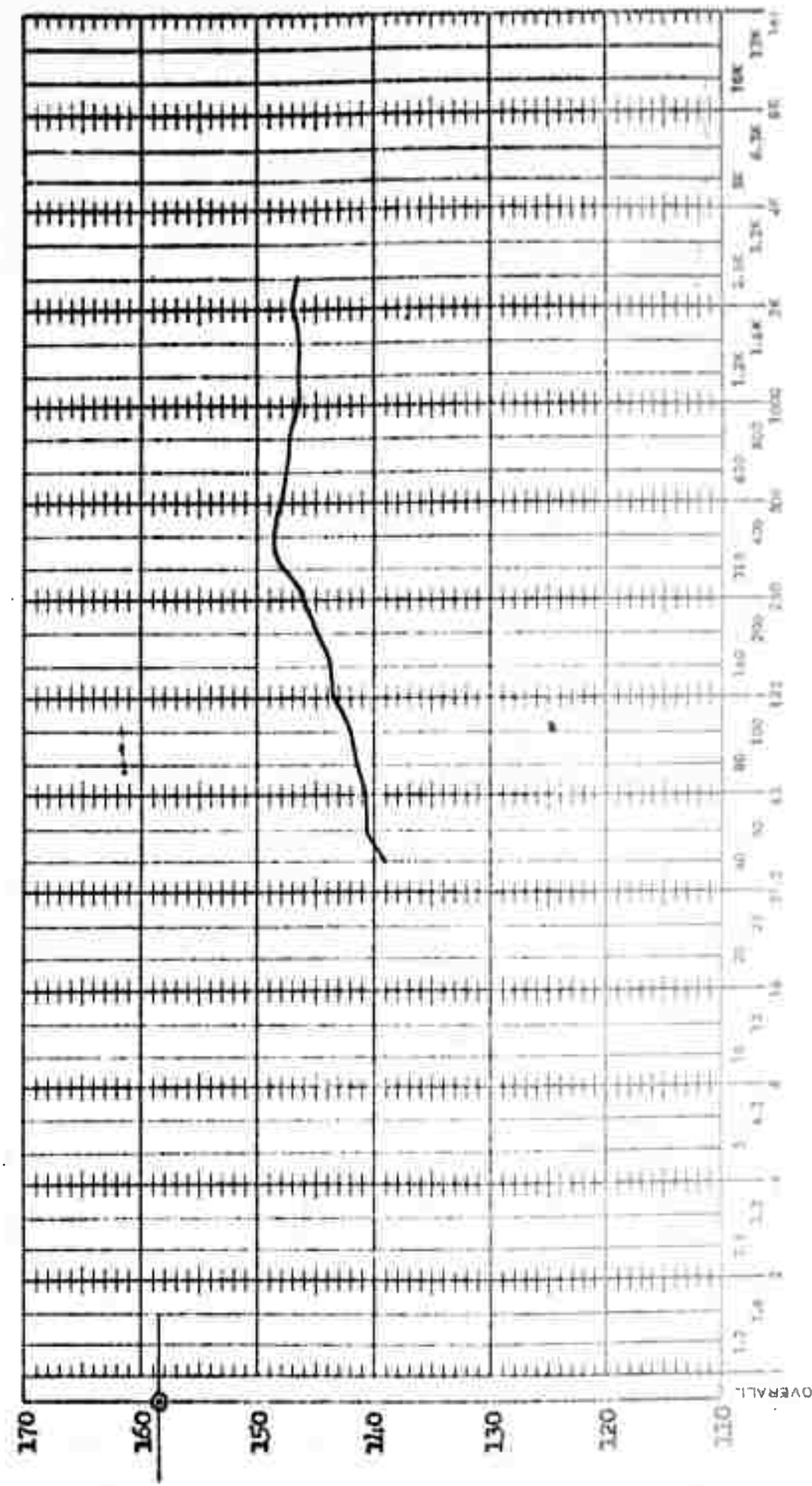
Test Point #33 Mach No. 0.9 Correlation No. 475

$\alpha = 0^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #33 Mach No. 0.9 Correlation No. 476

$\alpha = -4^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

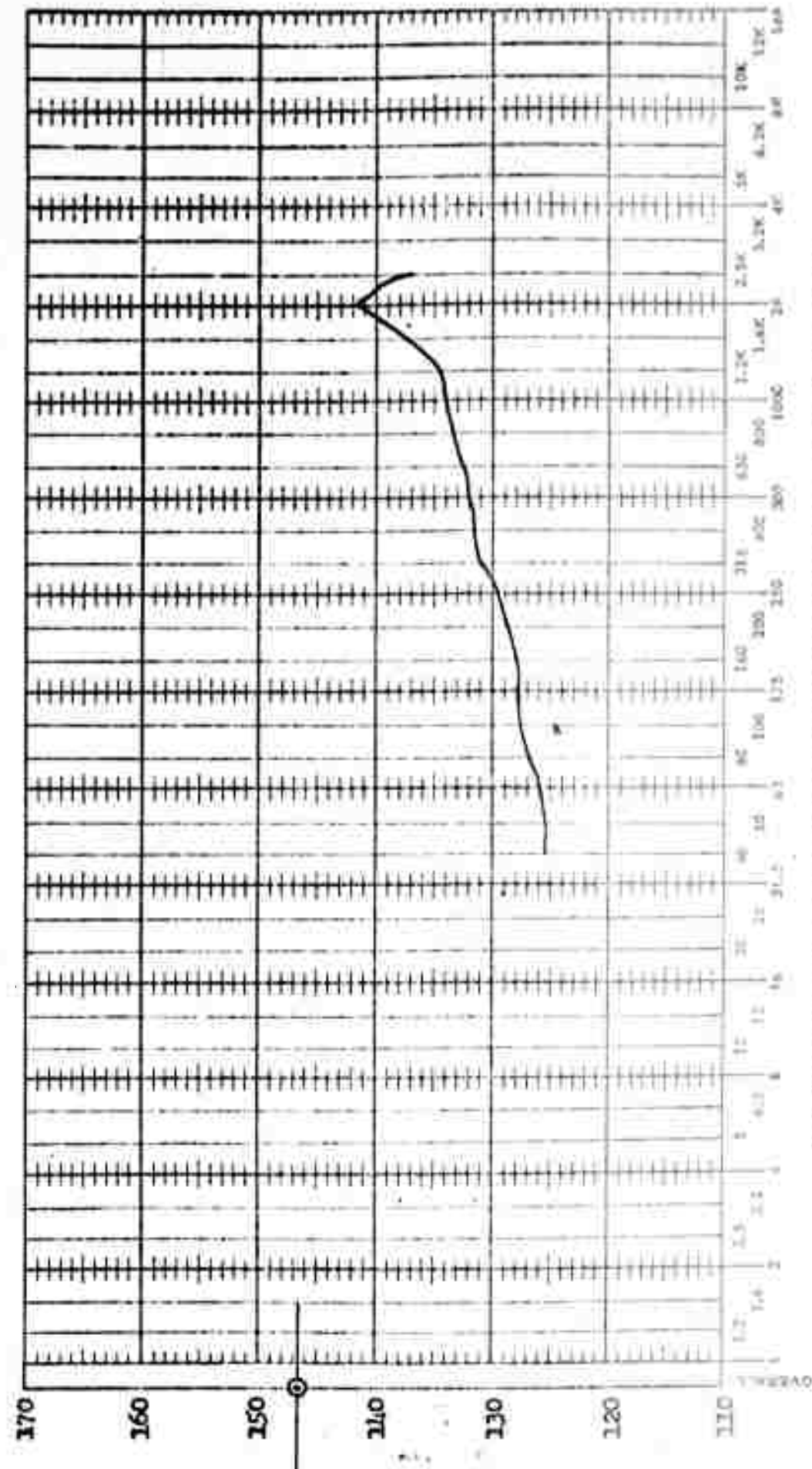
BOEING

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

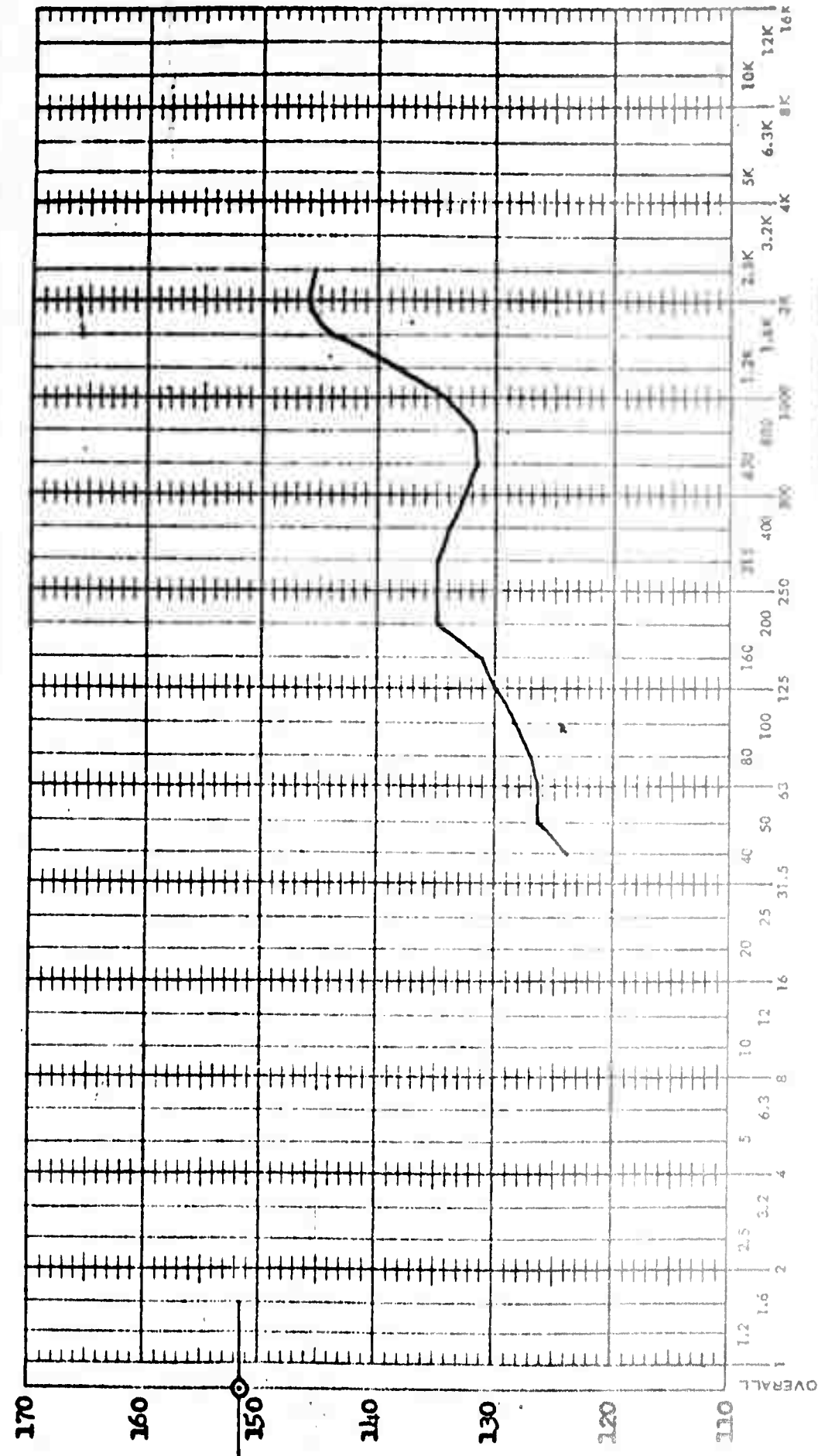
Test Point #33 Mach No. 0.74 Correlation No. 480

Figure 1A (Continued)



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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

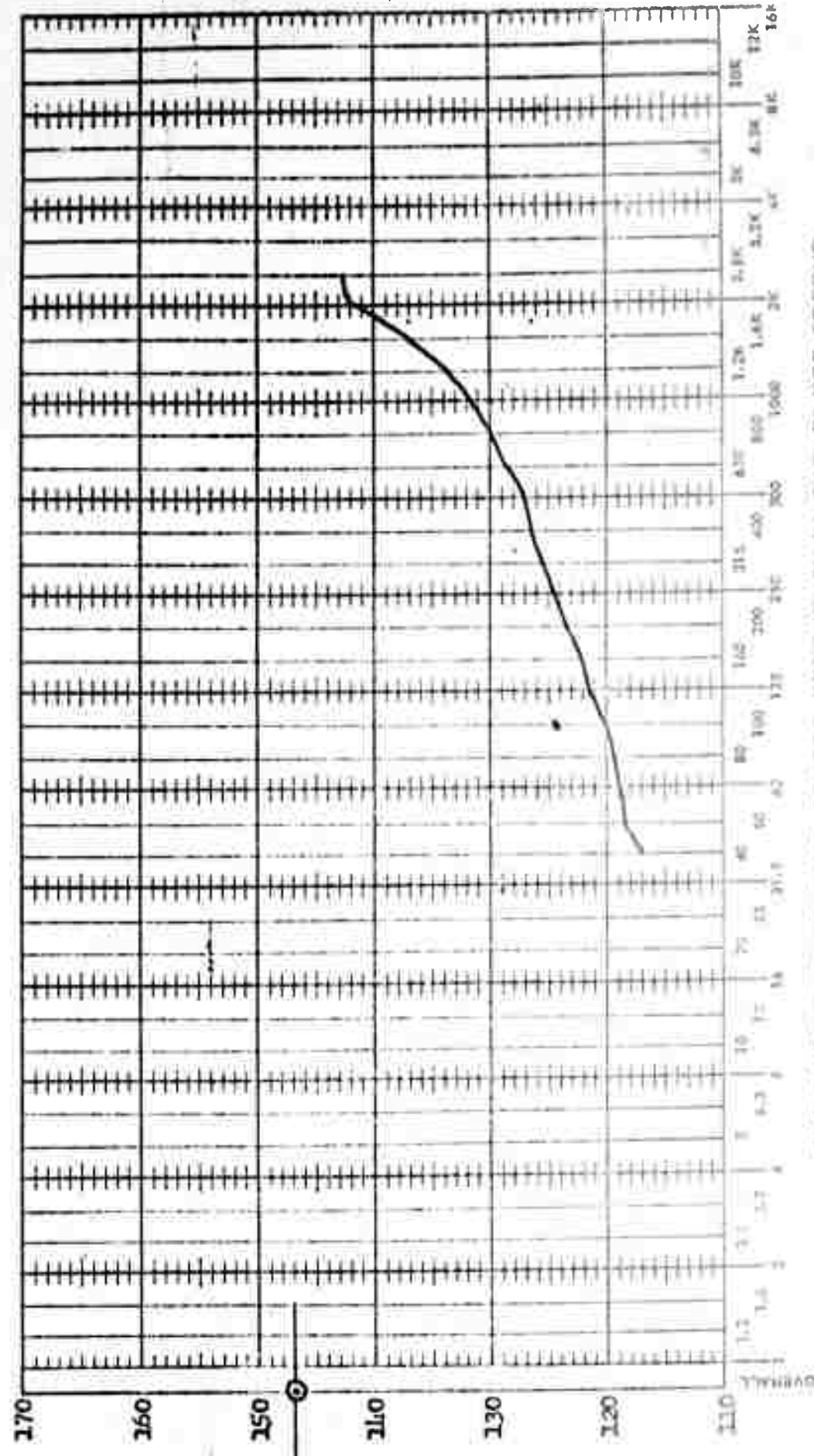
Test Point #34 Mach No. 0.85 Correlation No. 358

Figure 14 (Continued)

$\alpha = 0$



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

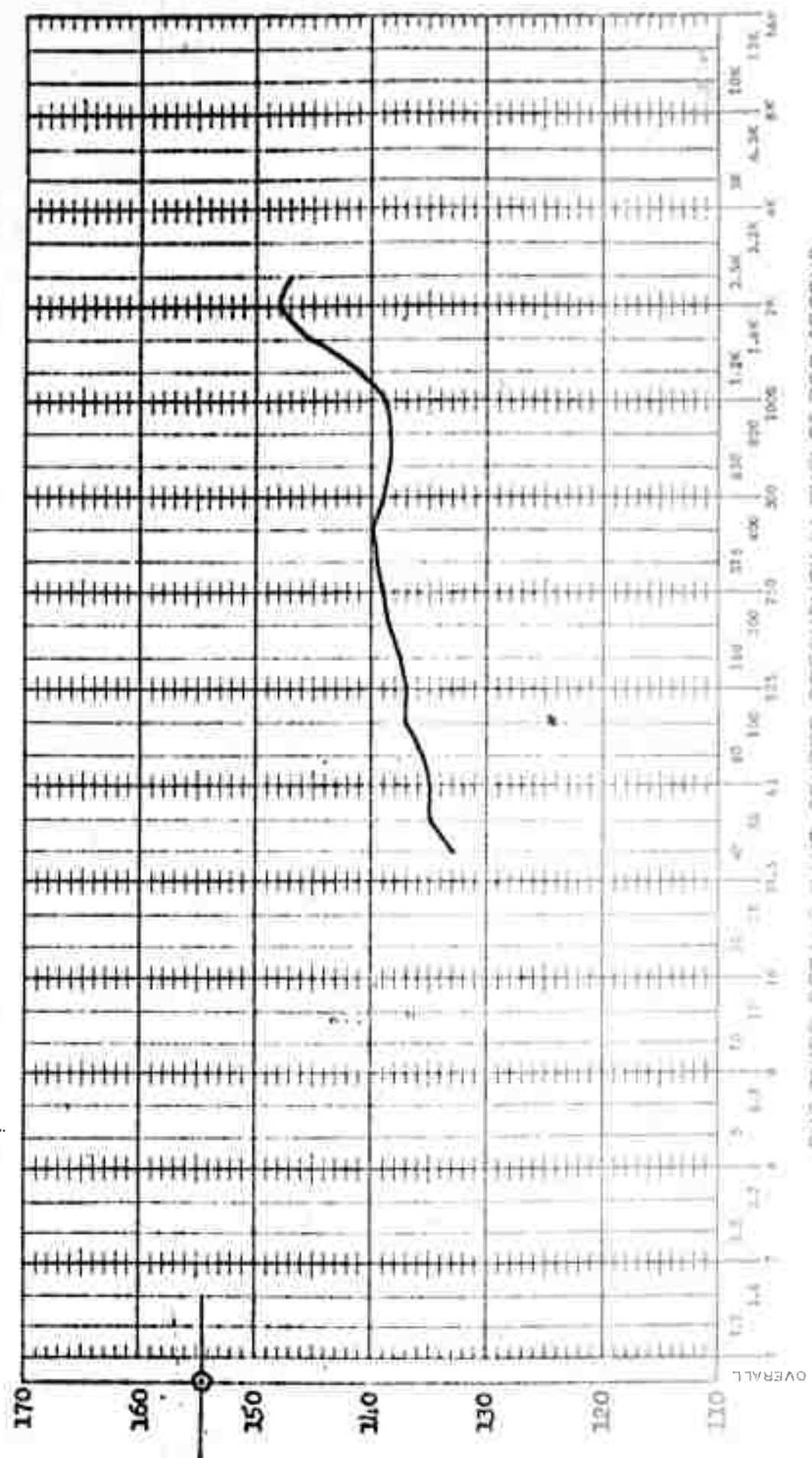
$\alpha = 0^\circ$   $\beta = 90^\circ$

Test Point #11 Mach No. 0.95 Correlation No. 362

Figure 1A (continued)

081

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #34 Mach No. 0.86 Correlation No. 469

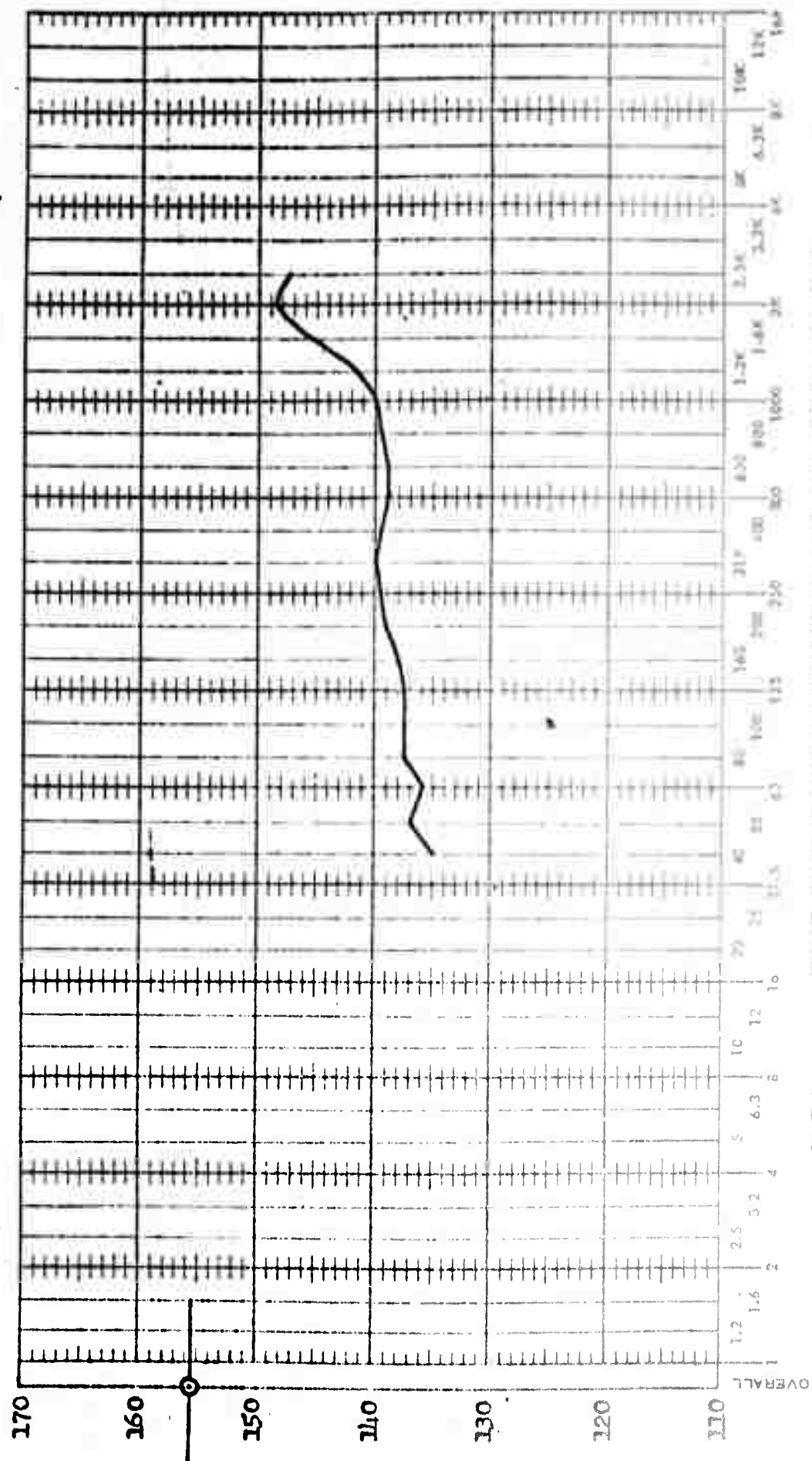
$\alpha = 0^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR

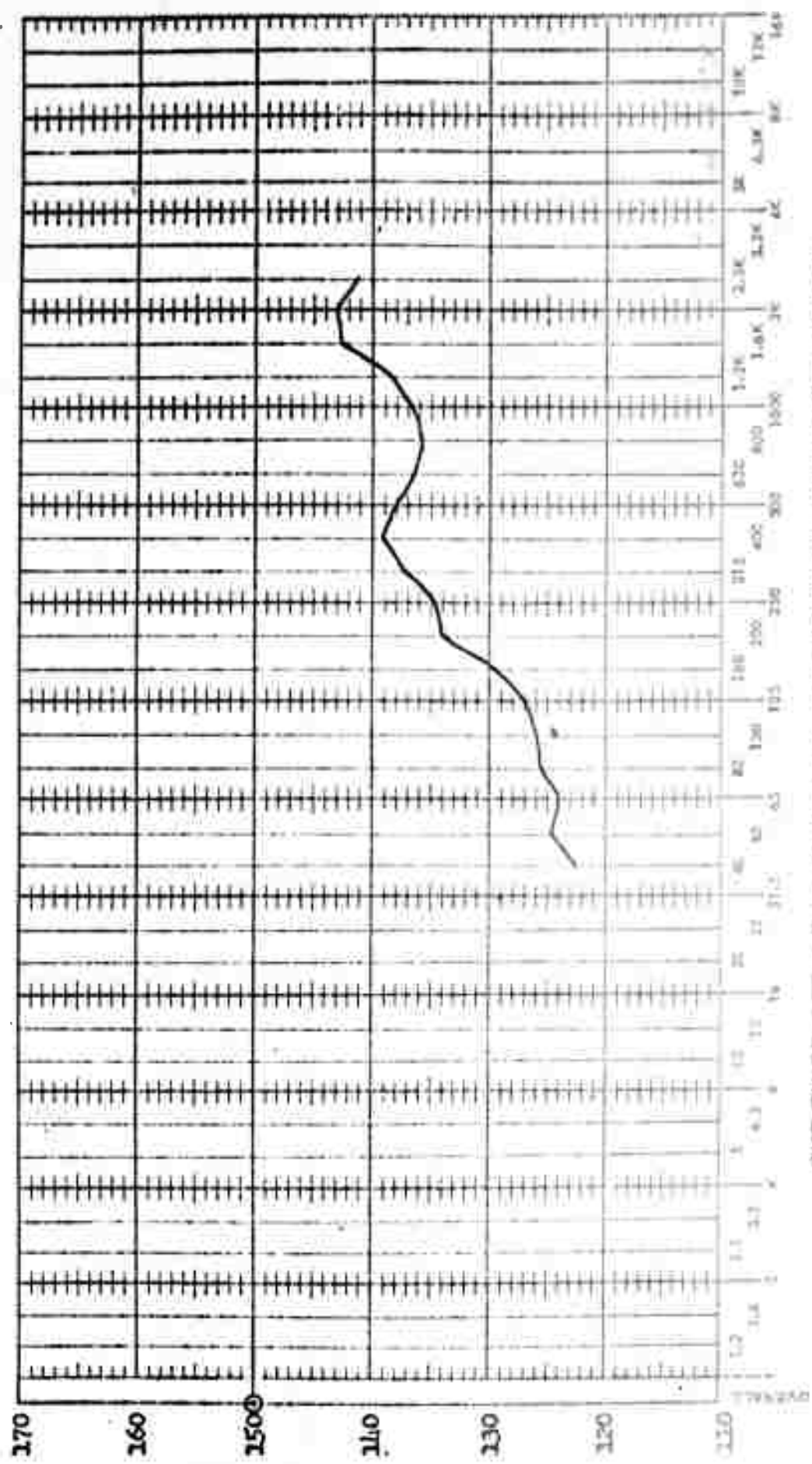


ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #34 Mach No. 0.88 Correlation May 1972

Figure 14 (Continued)

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #35 Mach No. 0.5 Correlation No. 460

Figure 14 (Continued)

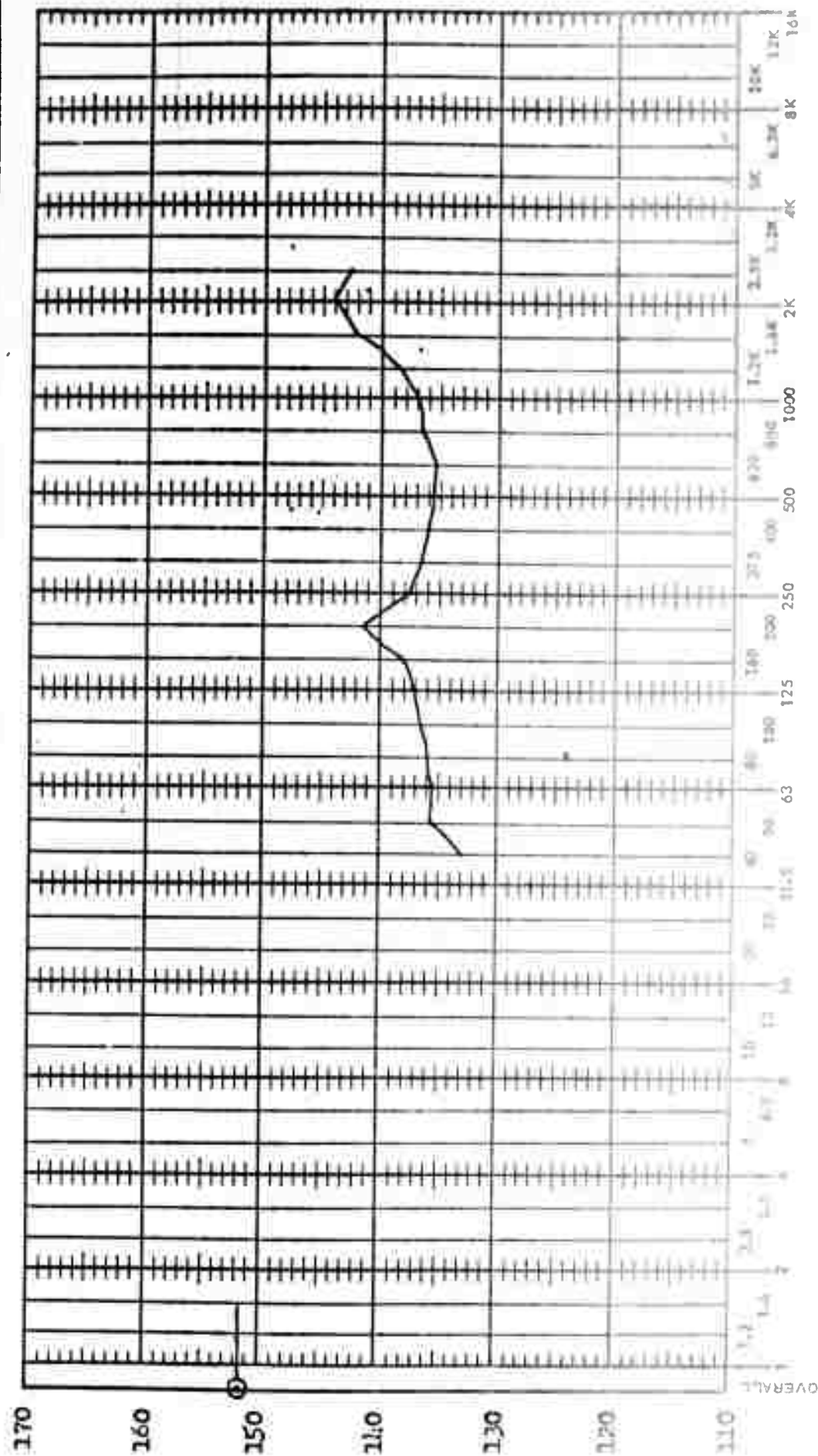
$\alpha = 0^\circ$   $\beta = -4^\circ$

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #35 Mach No. 0.84 Correlation No. 466

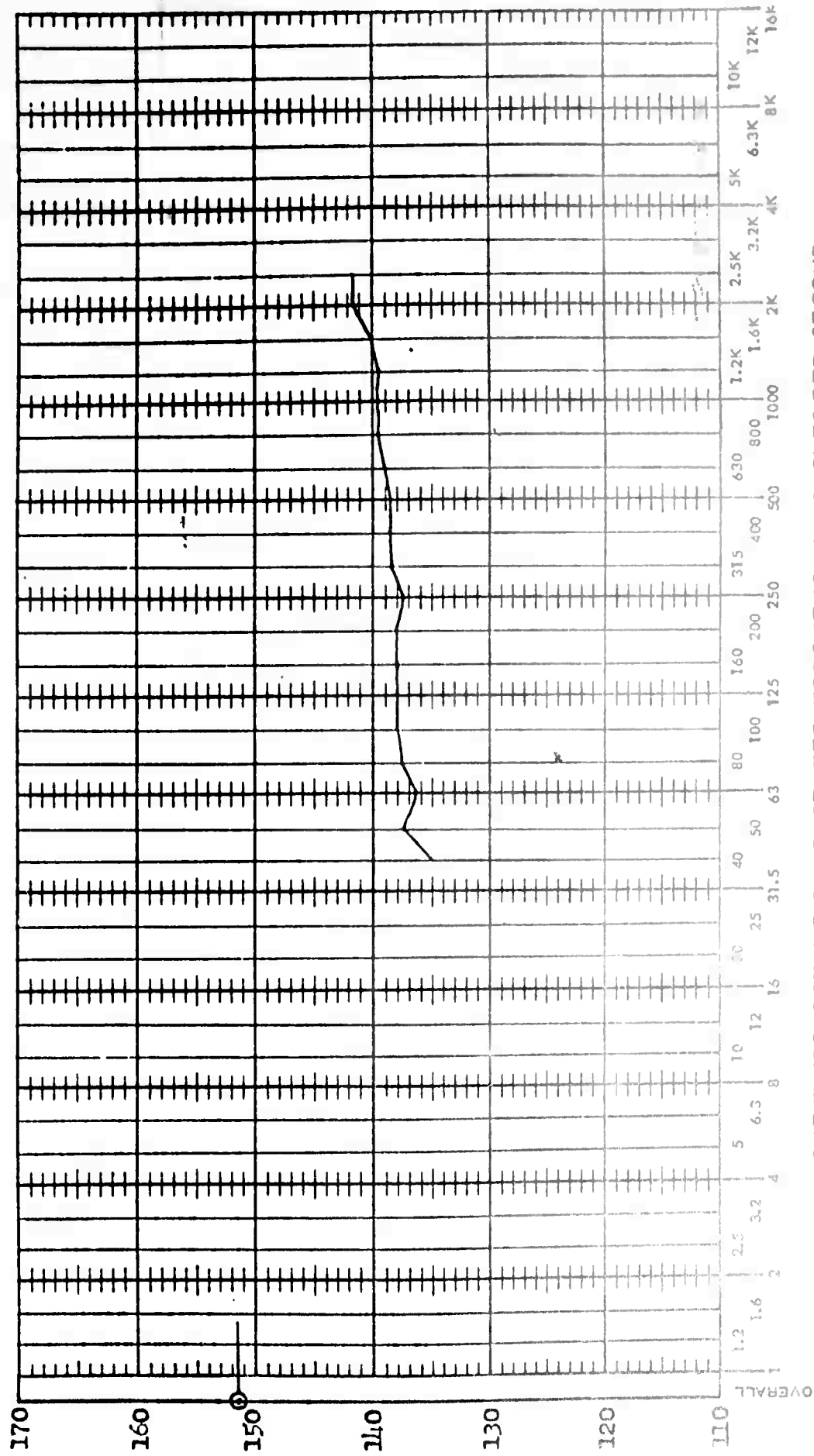
$\alpha = 0$   $\beta = -4$

Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #35 Mach No 0.88

Correlation No. 472

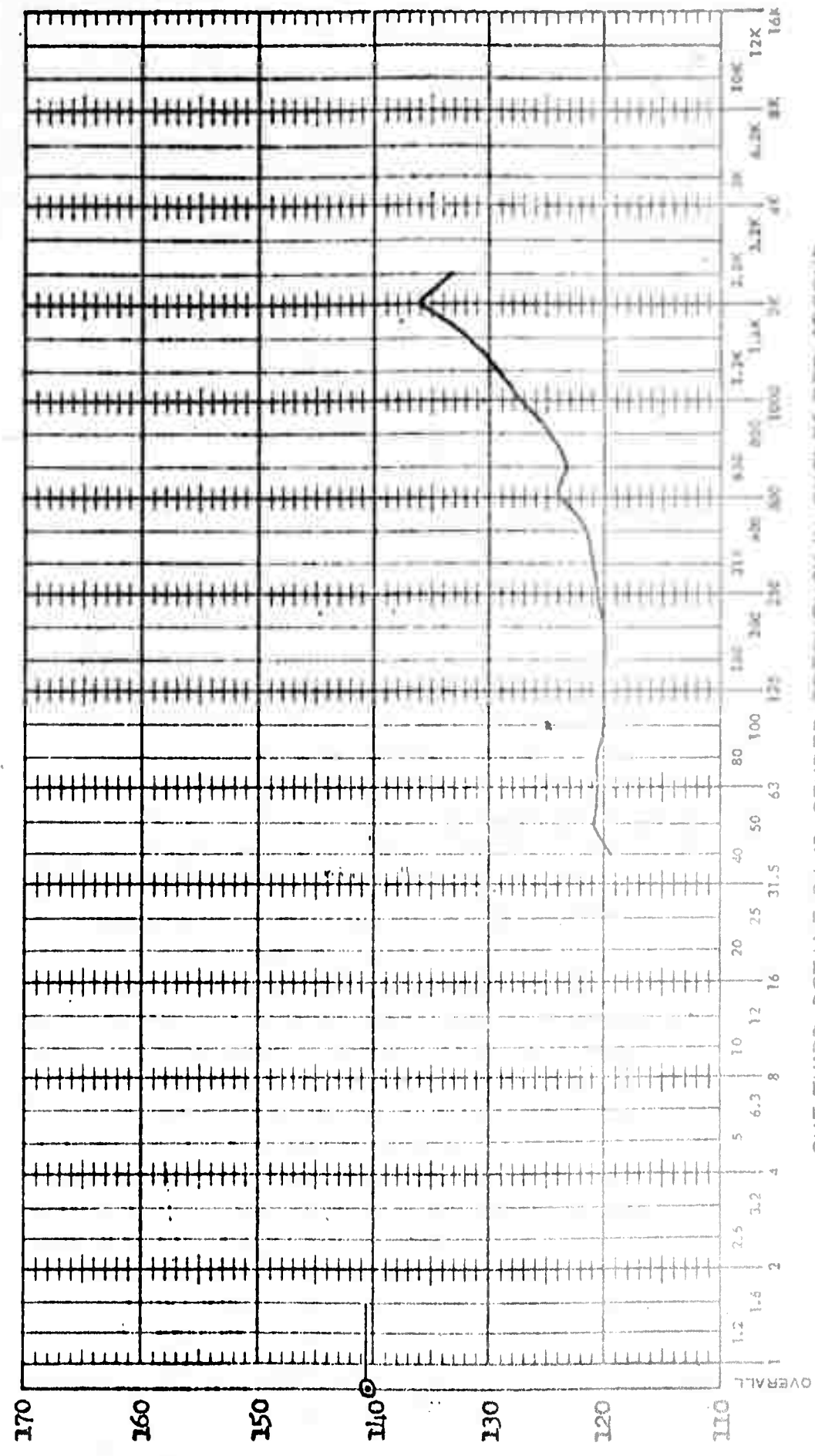
$\alpha = 0^\circ$

$\beta = -2^\circ$

Figure 14 (Continued)

195

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR

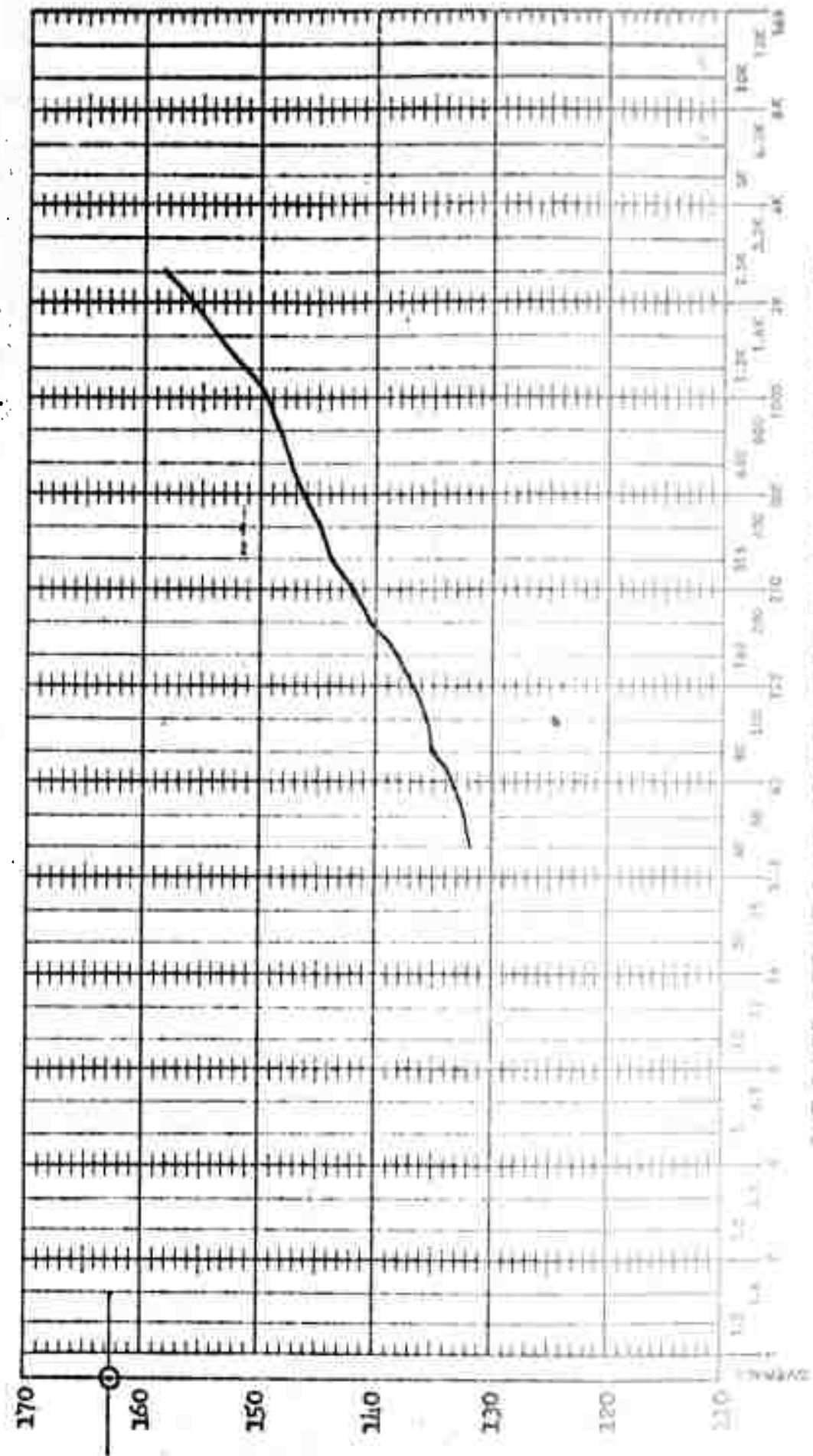


ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #35 Mach No. 0.92 Correlation No. 478

Figure 14 (Continued)

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



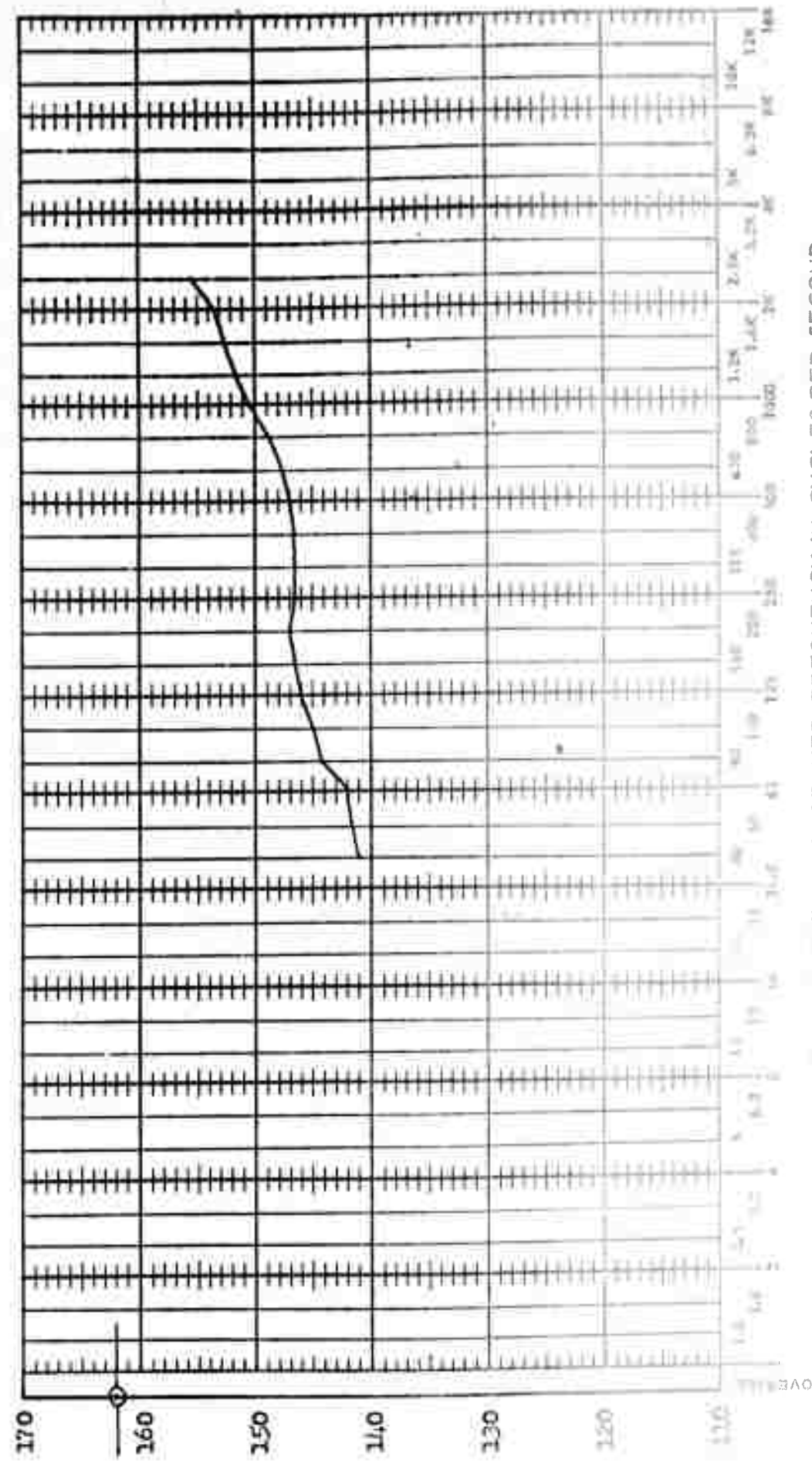
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #36 Mach No. 0.9 Correlation No. 360

$\alpha = 0^\circ$   $\beta = -5^\circ$

Figure 14 (Continued)

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #36 Mach No. 0.95 Correlation No. 361

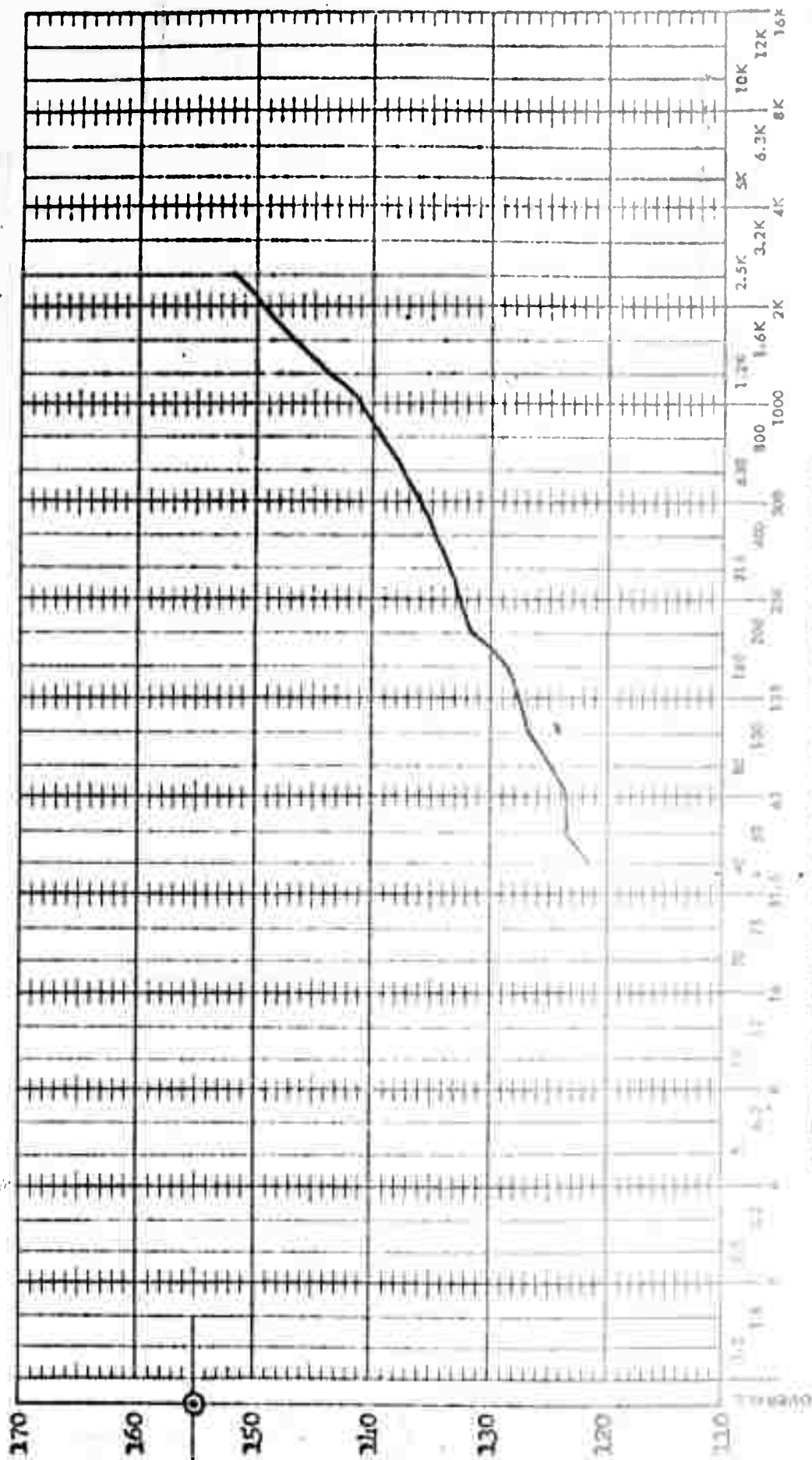
$\alpha = 0^\circ$   $\beta = -4^\circ$

Figure 14 (Continued)

12660  
100

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #56; Mach. No. 0.75; Correlation No. 1.55

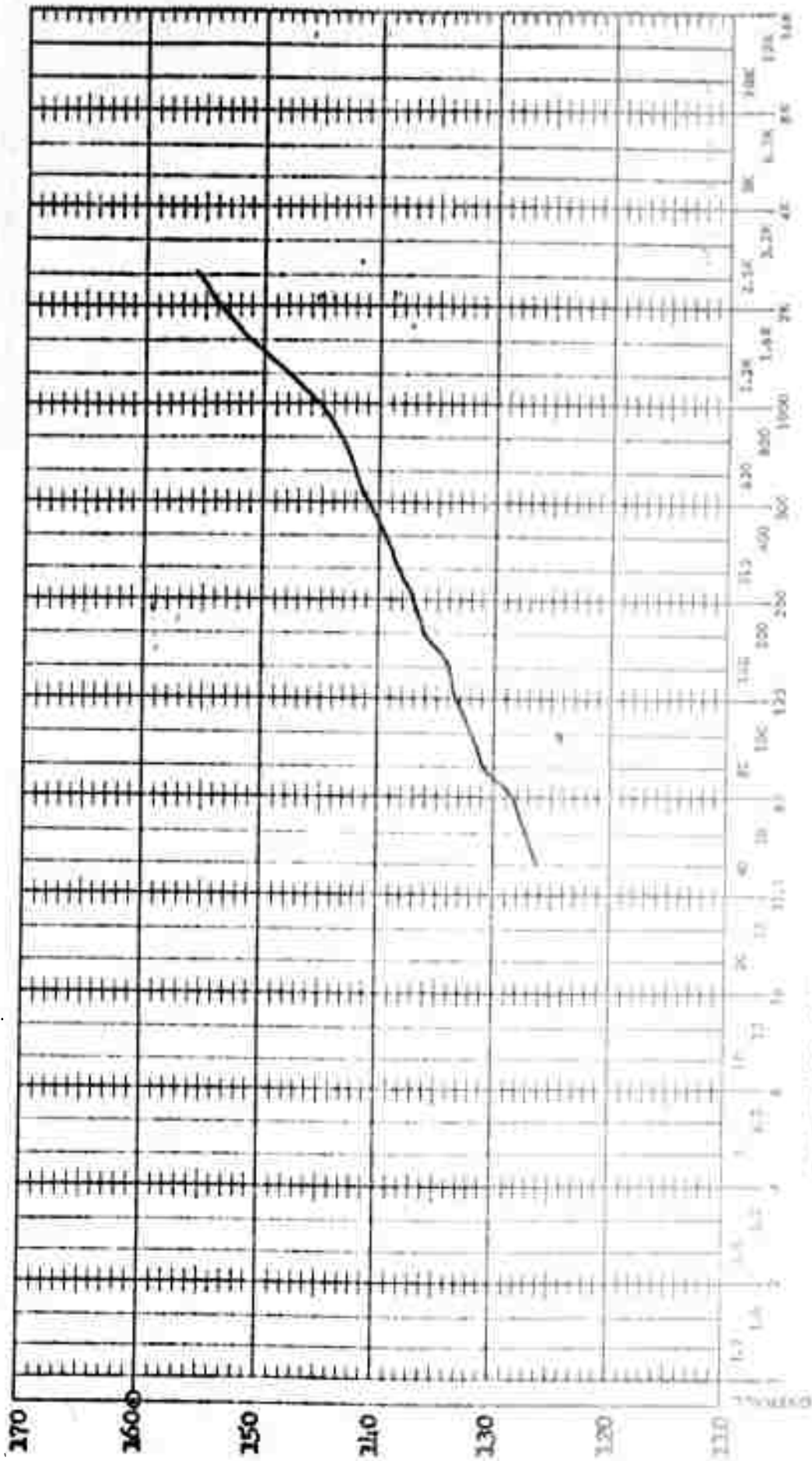
Figure 14 (Continued)

$\beta = -4$

$\alpha = 4$



ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



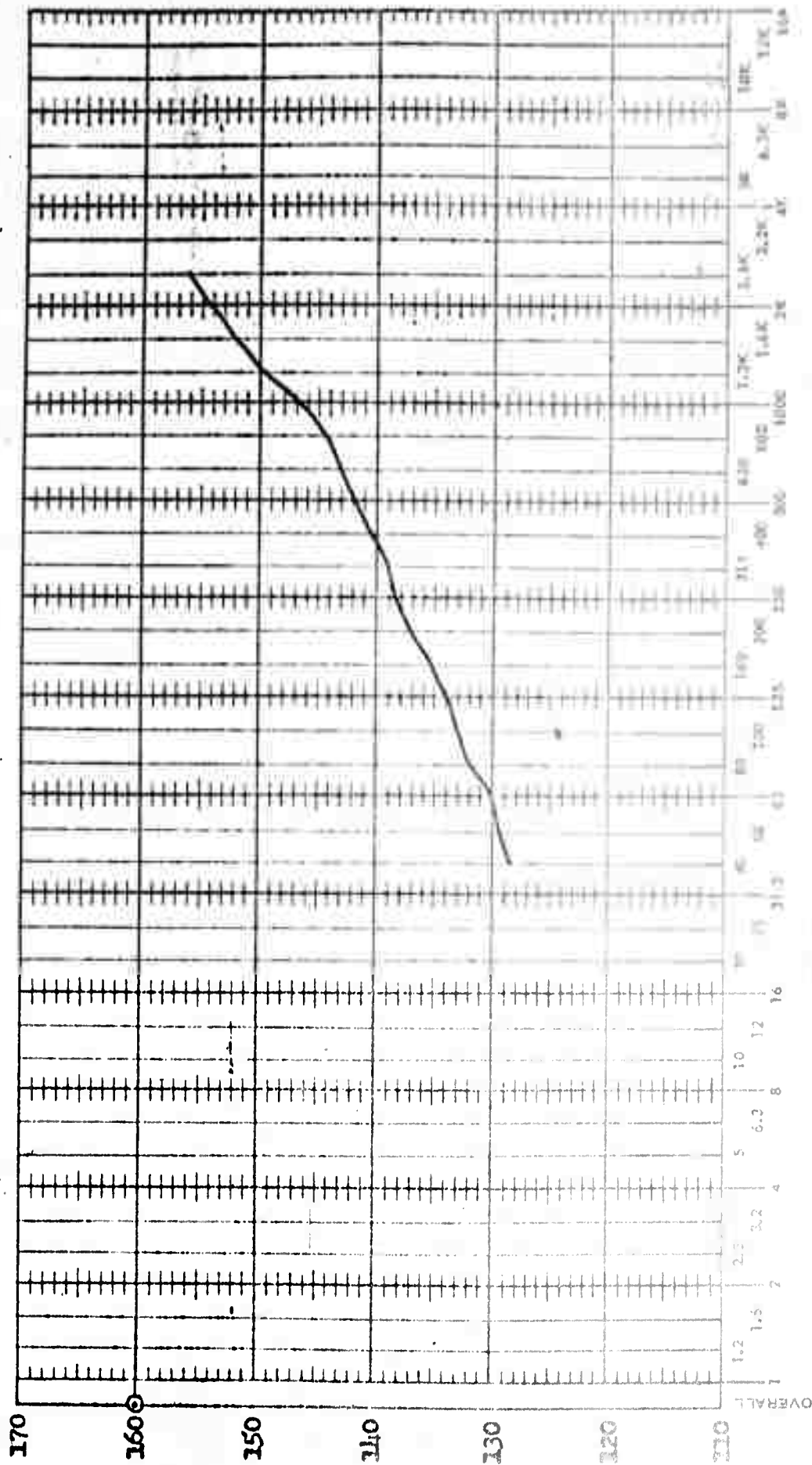
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #16 Mach No. 0.92 Correlation No. 462

Figure 14 (Continued)

$\alpha = 7.5^\circ$   $\beta = -7.5^\circ$

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

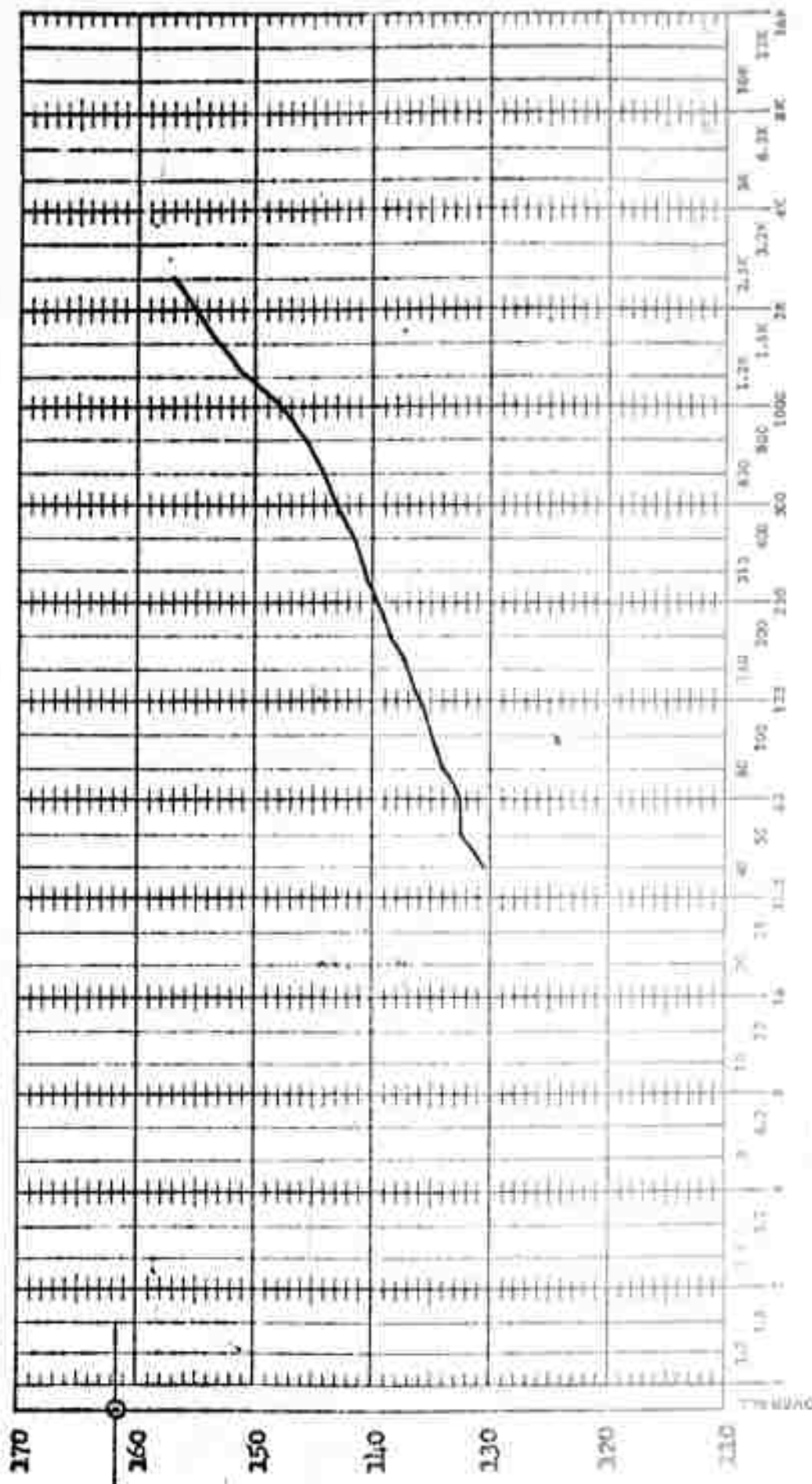
Test Point #36 Mach No. 0.84 Correlation No. 467

Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #35 Mach No. 0.85 Correlation No. 468

Figure 14 (Continued)

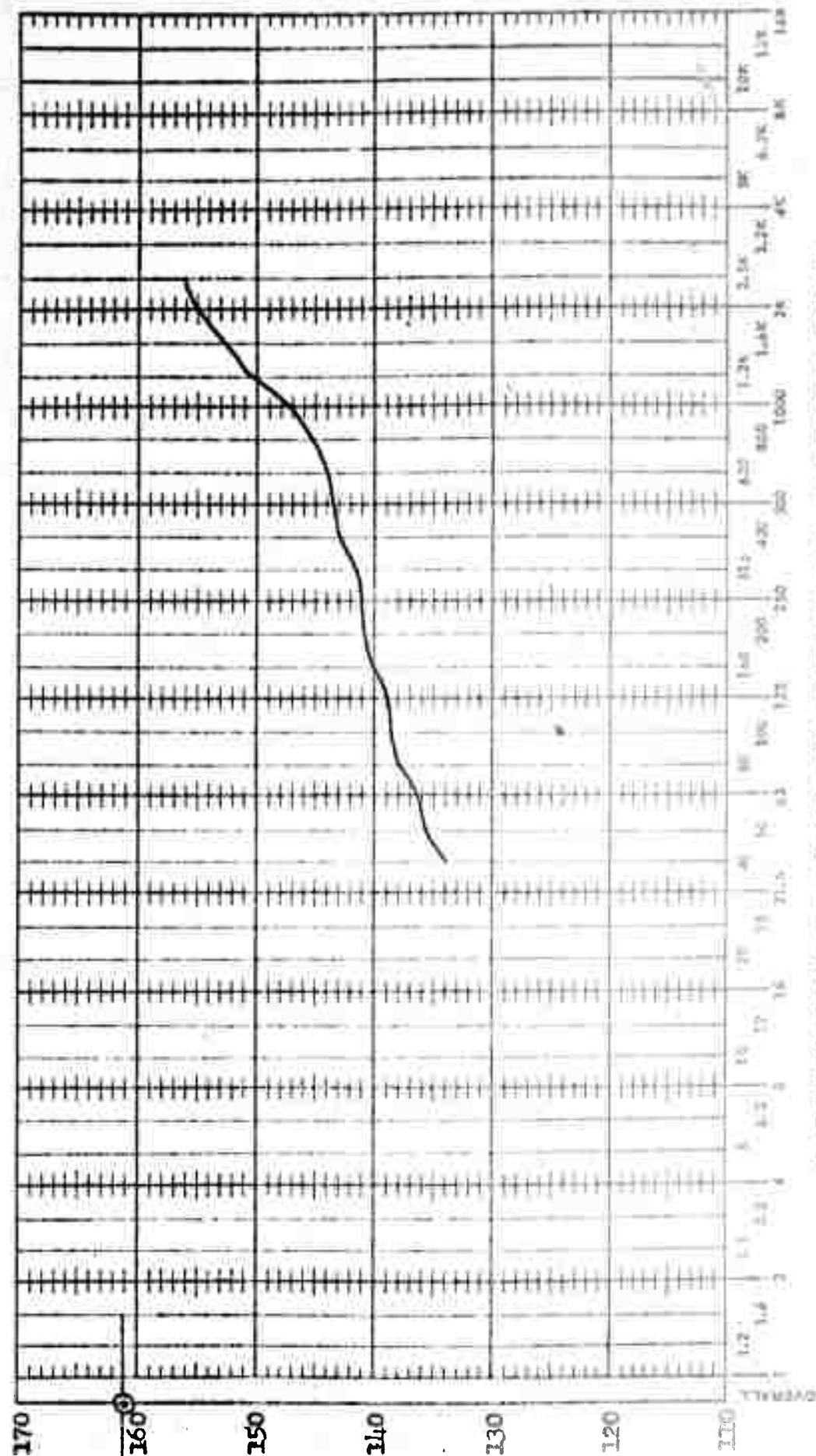
$\alpha = 4^\circ$   $\beta = -7^\circ$

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



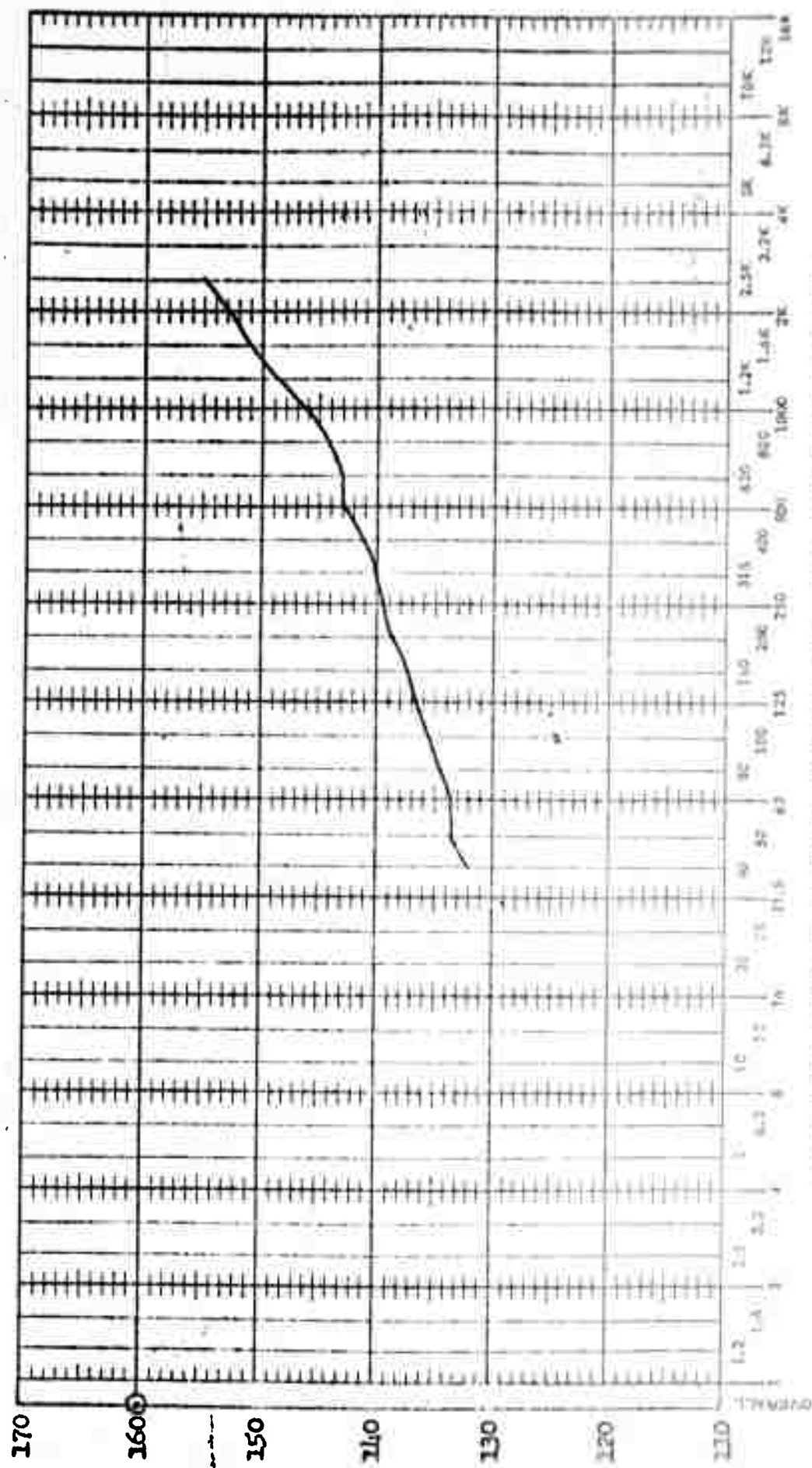
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #35 Mach No. 0.06 Correlation No. 1457

 $\alpha = 0$   $\beta = -2$ 

Figure 14 (Continued)

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #36 Mach No. 0.56 Correlation No. 470

$\alpha = -9$   $\beta = -4$

Figure 14 (Continued)

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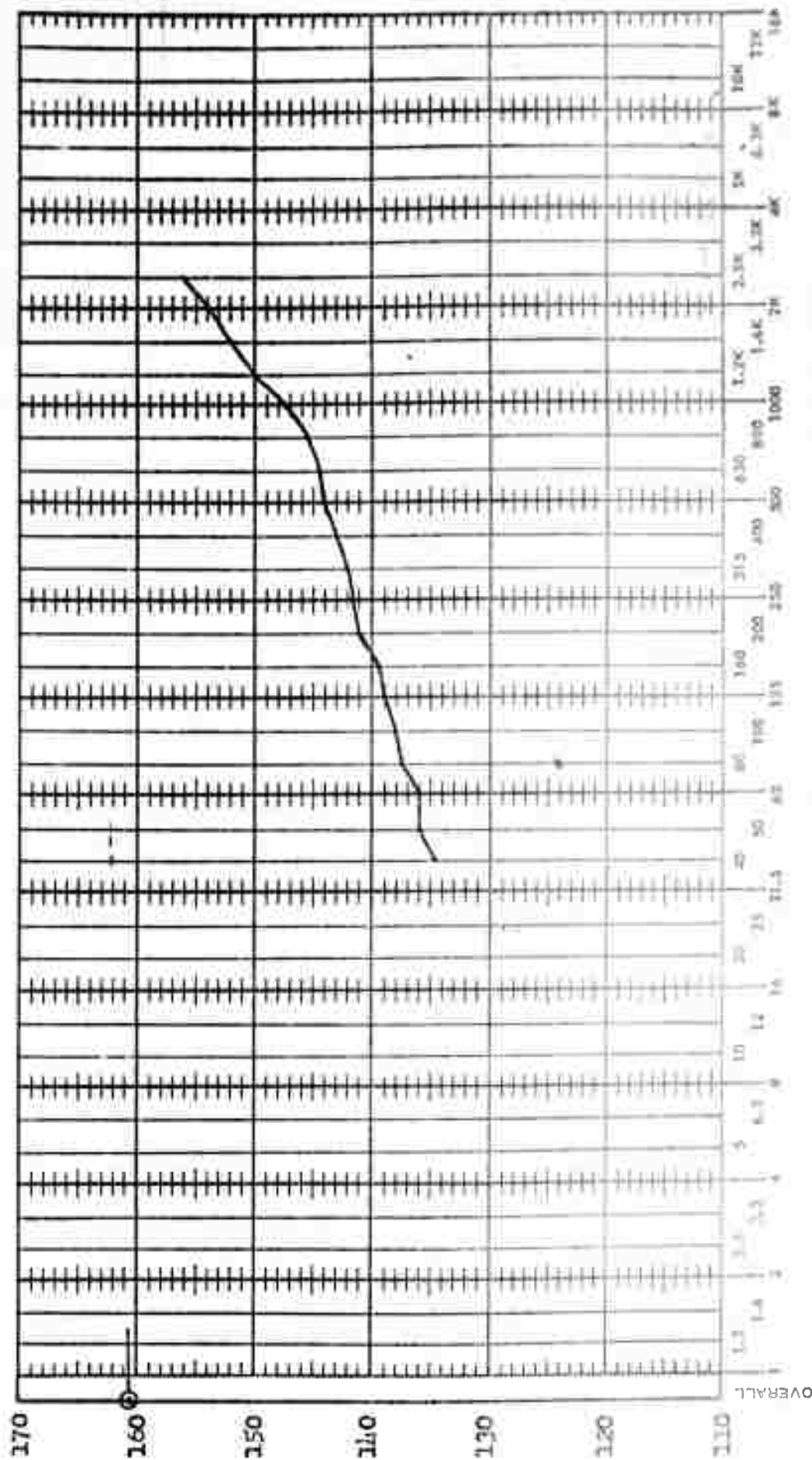
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #36 Mach No. 0.88 Correlation No. 471

$\alpha = -4^\circ$   $\beta = 4^\circ$

Figure 14 (Continued)

BOEING

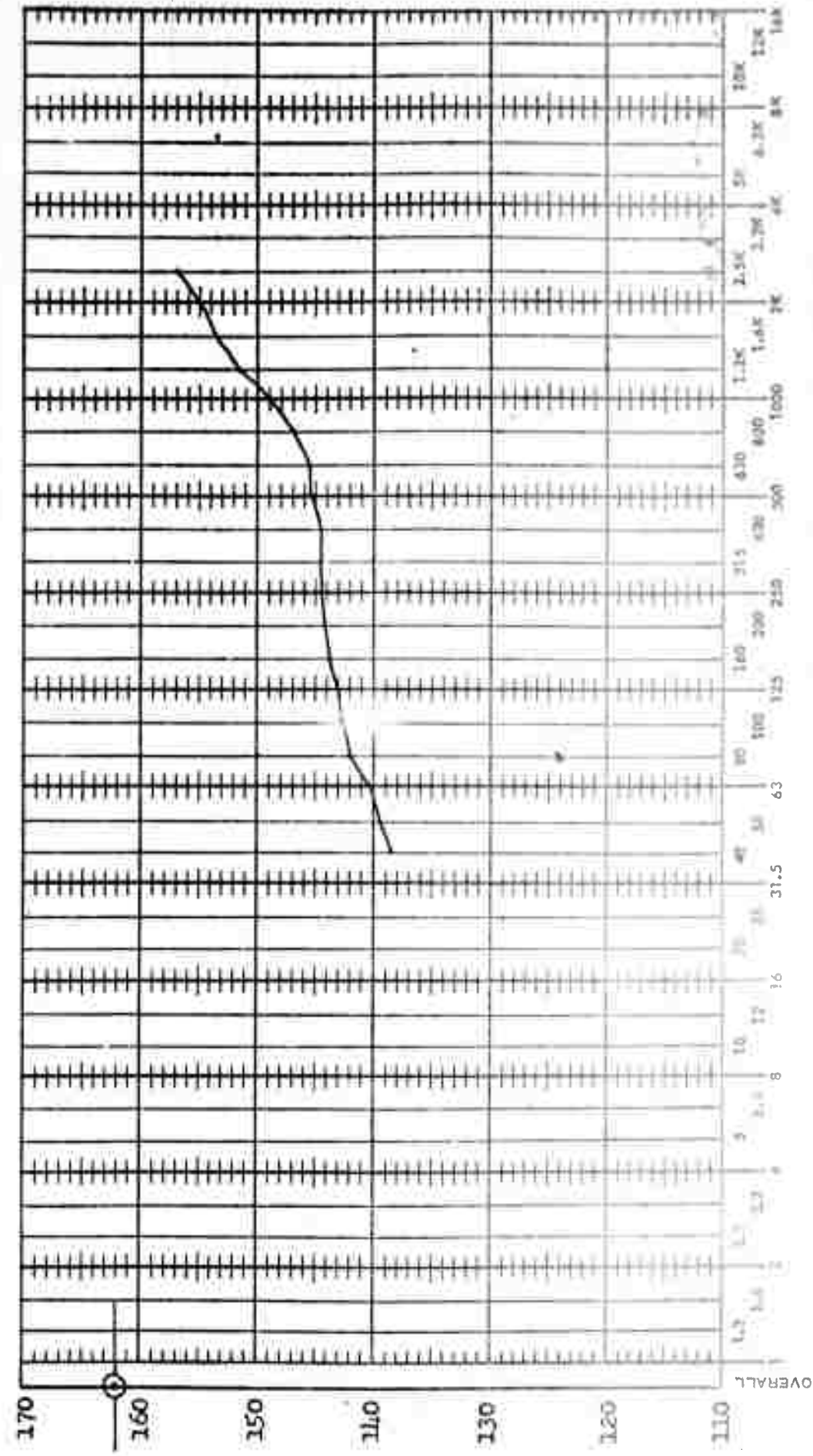
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

$\alpha = 0^\circ$

$\beta = -4^\circ$

Test Point #36 Mach No. 0.88

Correlation No. 472

Figure 14 (Continued)

BOEING

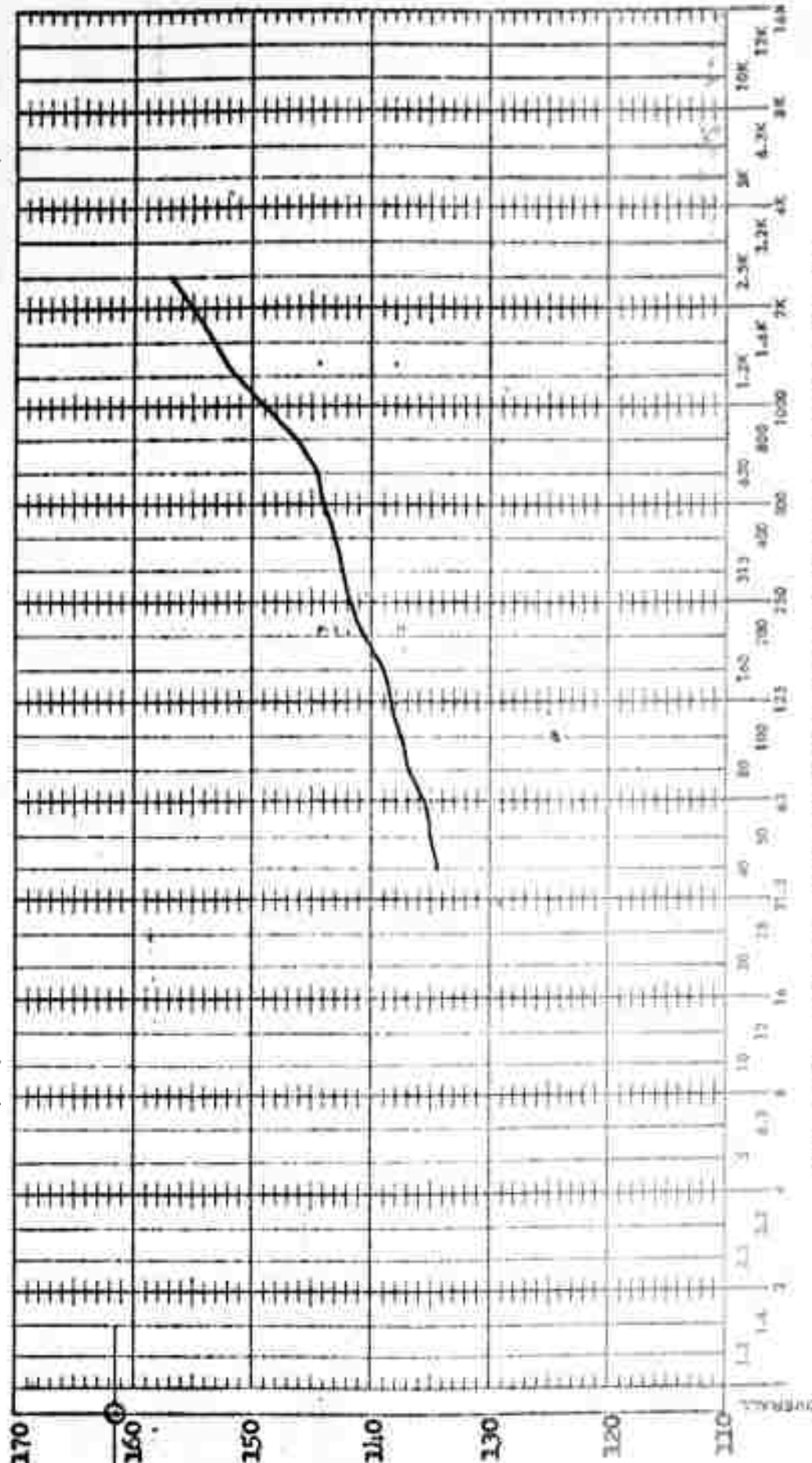
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #36 Mach No. 0.88 Correlation No. 473

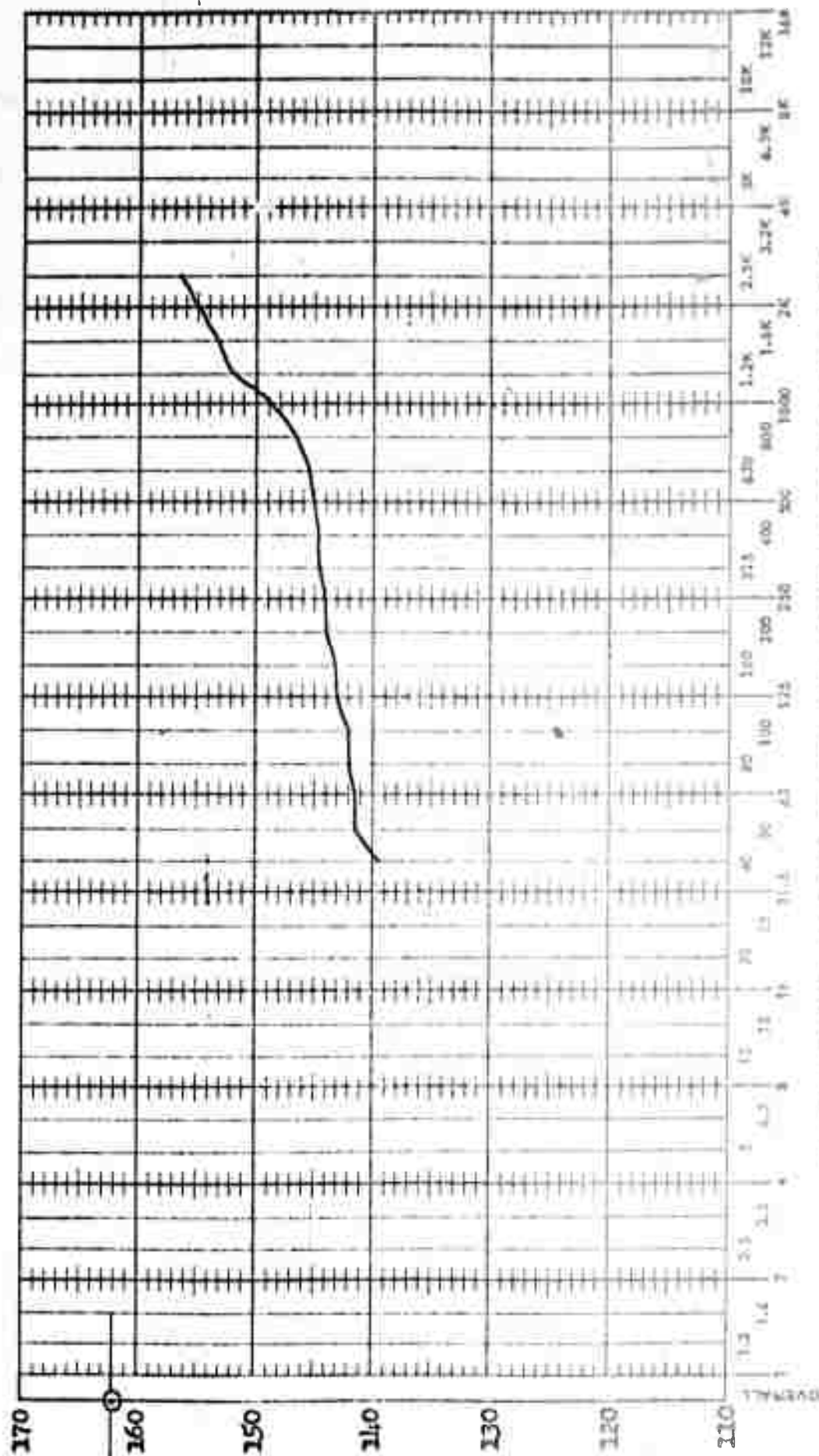
Figure 14 (Continued)

$\delta = 15^\circ$   $\beta = 4^\circ$

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #36 Xch. No. 0.9 Correlation No. 1171

 $\phi = 180^\circ$   $\phi = -180^\circ$ 

Figure 14 (Continued)

REFERENCE

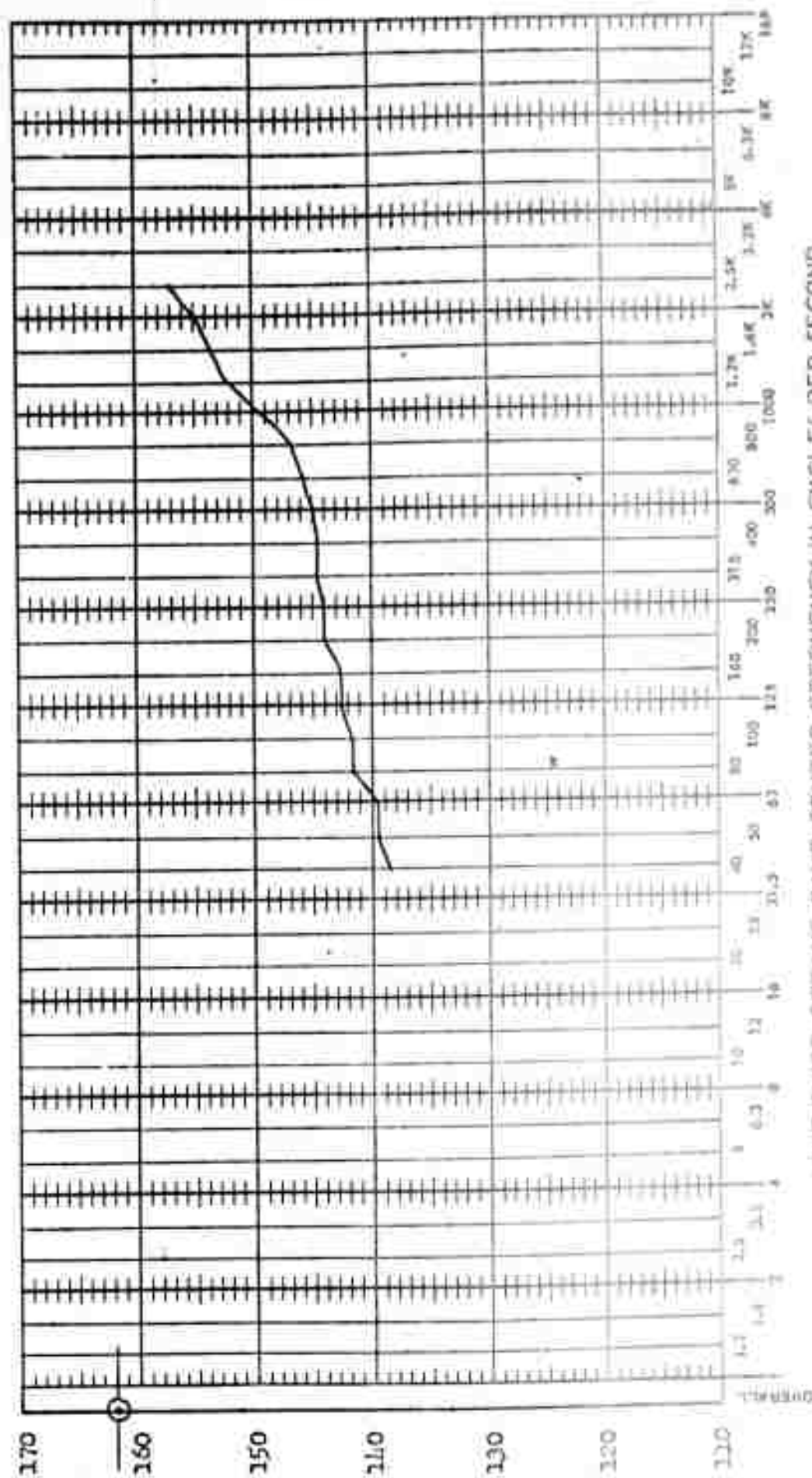
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #36 Mach No. 0.9 Correlation No. 175

Figure 14 (Continued)

BOEING

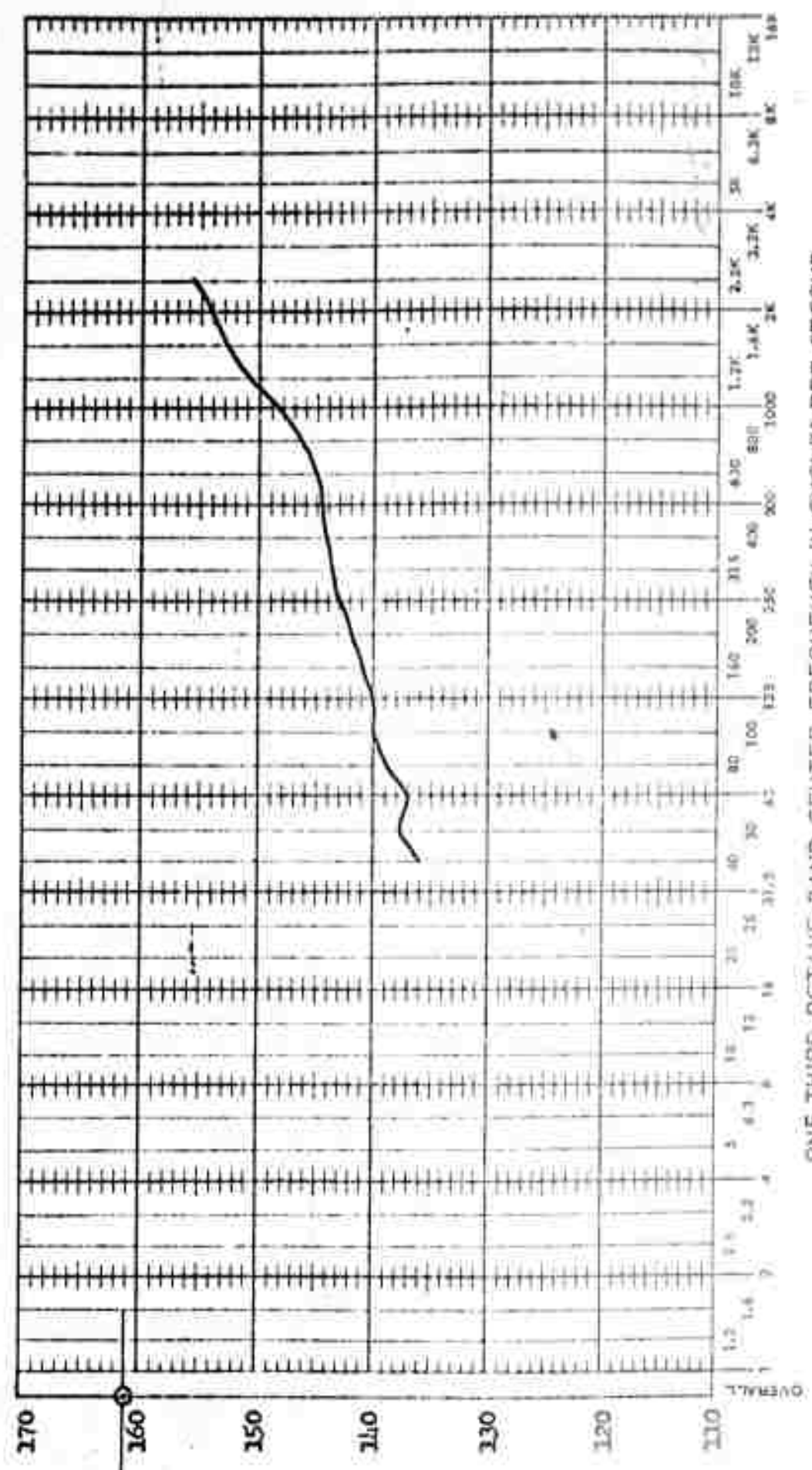
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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



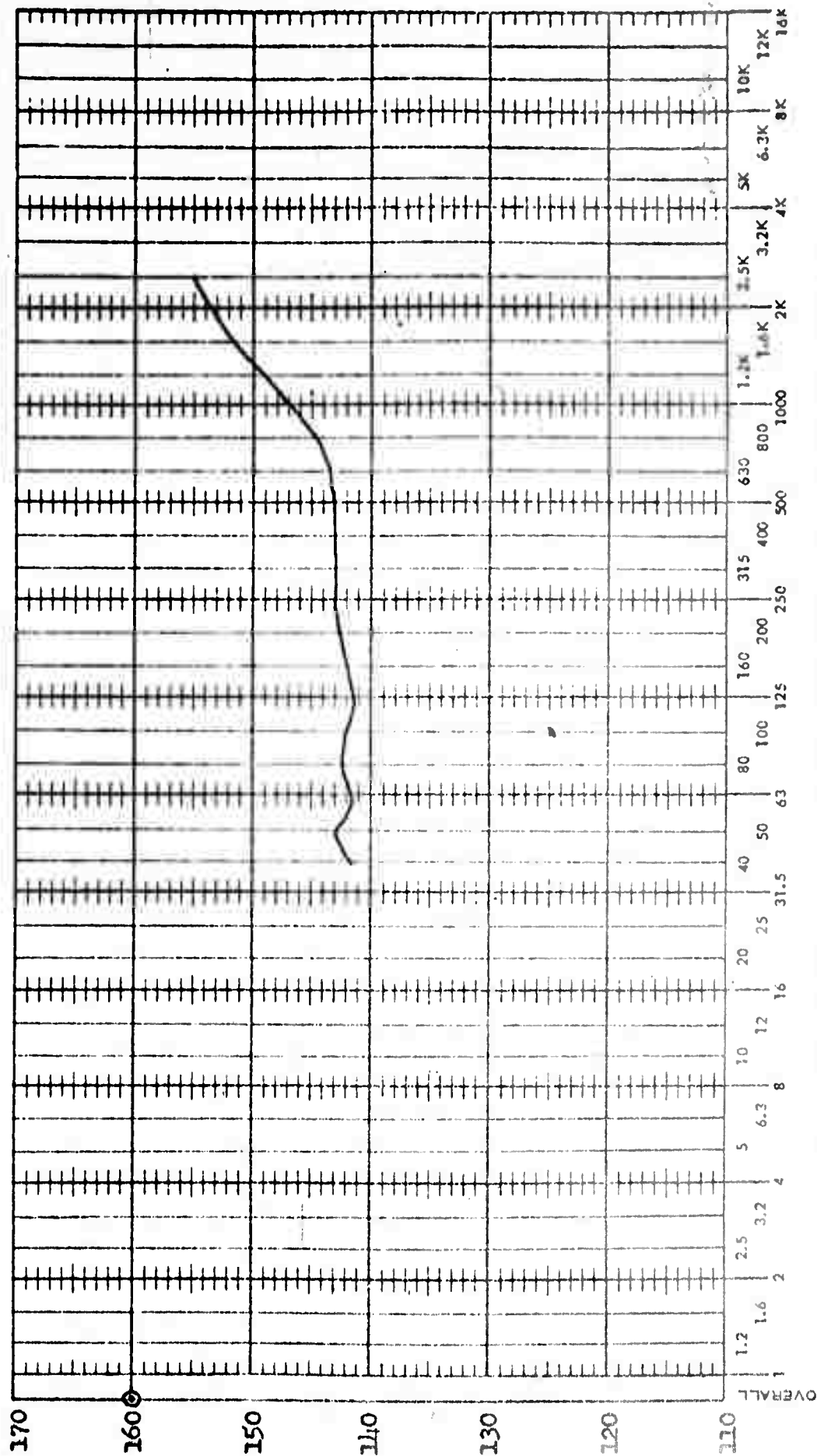
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #36 Mach No. 0.9 Correlation No. 476

$\alpha = -2^\circ$   $\beta = -2^\circ$

Figure 14 (Continued)

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #36 Mach No. 0.92 Correlation No. 477

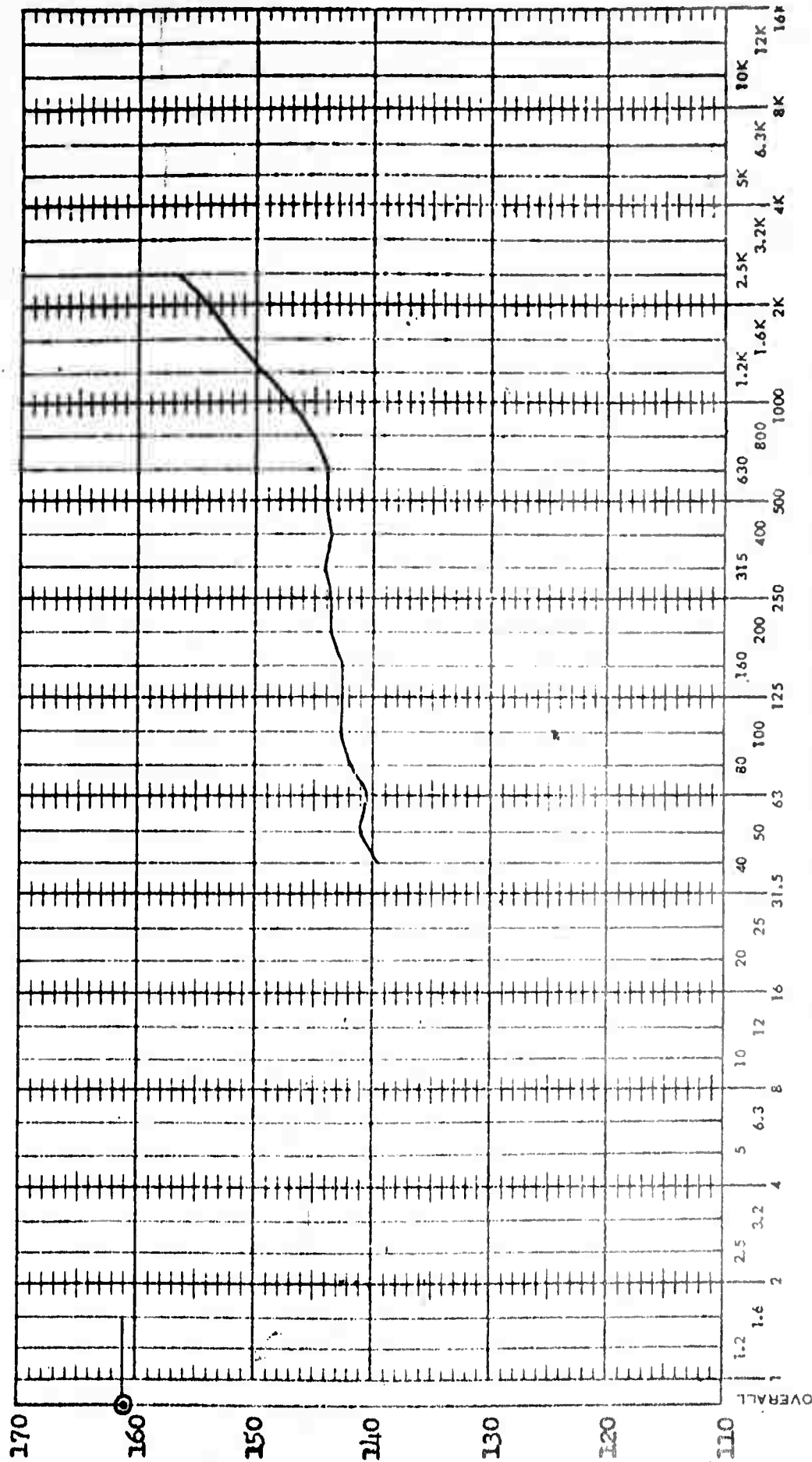
 $\alpha = -4^\circ$   $\beta = -4^\circ$ 

Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #36 Mach No. 0.92 Correlation No. 478

$\alpha = 0^\circ$

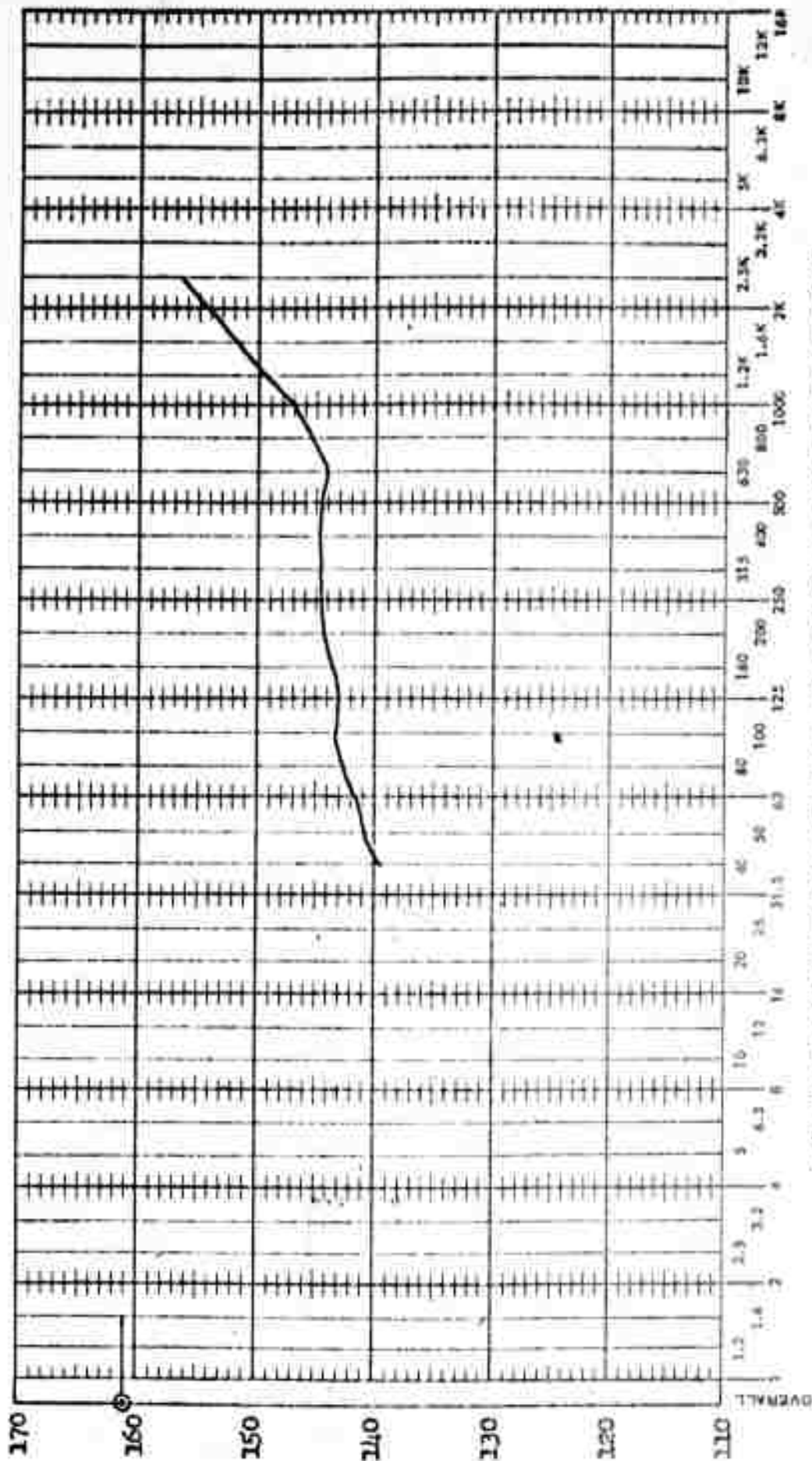
$\beta = -2^\circ$

Figure 14 (Continued)

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ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



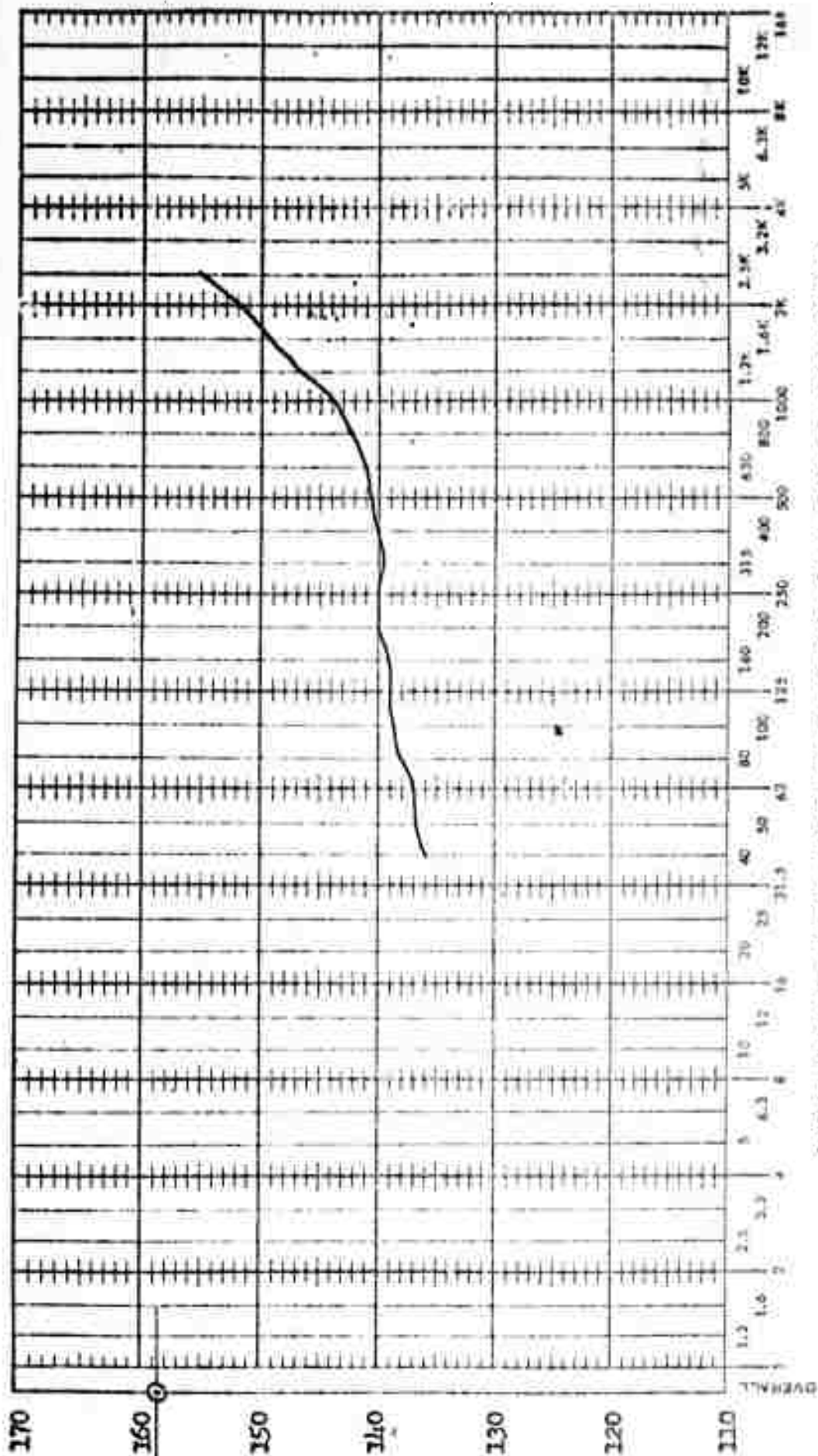
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #36 Mach No. 0.92 Correlation No. 479

$\sigma = 7.5$   $\beta = -2.5$

Figure 14 (Continued)

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #36 Mach No. 0.91 Correlation No. 481

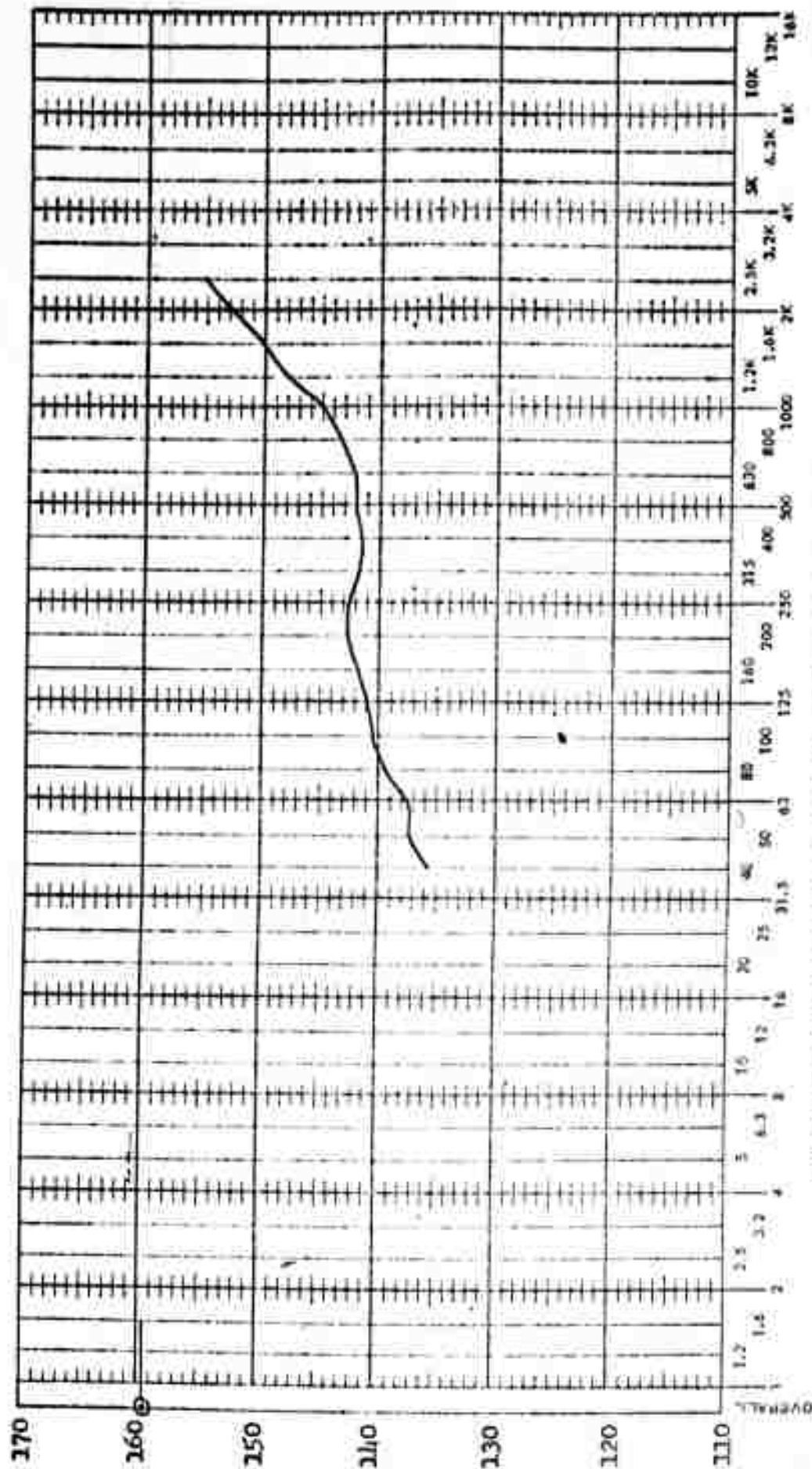
$\alpha = 0^\circ$   $\beta = -5^\circ$

Figure 14 (Continued)

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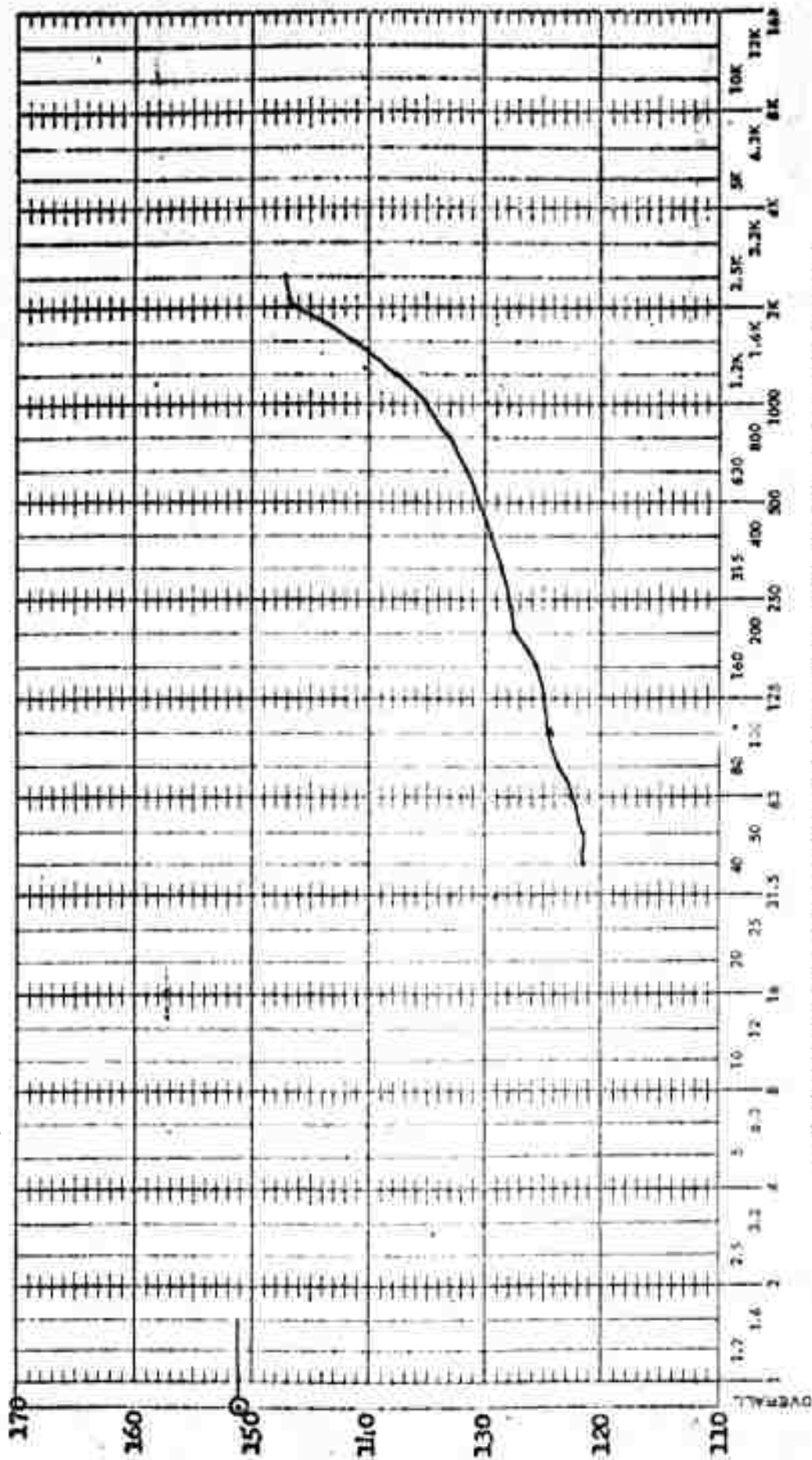


Test Point #36 Mach No. 0.94 Correlation No. 432

Figure 14 (Continued)

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Test Point #36 Mach No. 0.96 Correlation No. 485

$\alpha = 4$   $\beta = -2$

Figure 14. One-third octave band sound pressure levels. (Concluded)

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Table VII. OVERALL SOUND PRESSURE LEVEL (continued) \*See Text

db RE: .0002 Microbars

Correlation No.	310	311	312	313	314	315	316	317	318	319	320	321	322	323	324
Each No.	0.9	0.9	0.9	0.95	0.95	0.95	0.95	1.0	1.0	1.08	1.08	1.08		0.825	0.825
$\alpha$	0	+4	-4	-4	0	+4	+4	0	-4	-4	0	+4		+4	+2
$\beta$	0	0	0	0	0	0	0	0	0	0	0	0		0	0
Transducer Location															
1															
2		155.5*												154.0*	
3															
4															
5															
6															
7															
8															
9															
12															
14															
15															
16															
18															
20				146.0	146.5	149.0*	148.0*	145.0	145.0	148.0	147.0	148.5*			
21															
23															
26															
27															
28															
29															
30															
31															
32	148.5	146.5	148.0												
33															
34															
35															
36															

**db RE: .0002 Microbars**

[illegible]



dB RE: .0002 Microbars

[illegible]



Table VII. OVERALL SOUND PRESSURE LEVEL (cont'd)\*See Text

db RE: .0002 Microbars

Correlation No.	355	356	357	358	359	360	361	362	363	364	365	366	367	368	369
Mach No.	0.8	0.8	0.85	0.85	0.9	0.9	0.95	0.95	1.0	1.0	1.08	1.08	0.825	0.825	0.825
$\alpha$	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
$\beta$	+4	-4	-4	+4	+4	-4	-4	+4	+4	-4	-4	+4	+4	+2	0
Transducer Location															
1															
2															
3															
4															
5															
6															
7															
8															
9															
12															
14															
15															
16															
18															
20	140.0	144.0*	147.0*	140.5	140.0	149.7	145.0*	141.0*	142.0	144.0	144.0	144.0			
21	143.0	150.5*	149.0*	143.0	142.5	151.0*	148.0*	142.5	144.5	145.0	145.5	145.5			
23	144.0	144.0	145.0	145.0	145.0	144.5	145.5	147.5*	148.5*	147.0	149.5	150.5*			
26	158.5	158.5*	144.5*			142.5	142.0			142.5	142.5				
27															
28	160.0	151.0													
29															
30		151.0*	154.0*												
31															
32	160.5*	161.0*	159.5*												
33															
34			148.5	151.5*											
35						151.5									
36					157.0	162.5*	162.0*	149.0	147.5						

Table VII. OVERALL SOUND PRESSURE LEVEL (continued)

db RE: .0002 Microbars

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49 38 0002 Microbana

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db RE: .0002 Microbars

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Table VII. OVERALL SOUND PRESSURE LEVEL (cont'd)

db RE: .0002 Microbars

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225-

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Table VII. OVERALL SOUND PRESSURE LEVEL (continued) \*See Text

db RE: .0002 Microbars

Correlation No.	445	446	447	448	449	450	451	452	453	454	455	456	457	458	459
Mach No.	0.85	0.85	0.85	0.85	0.875	0.875	0.875	0.875	0.875		0.75	0.75	0.75	0.8	
$\alpha$	+2	0	-2	-4	-4	-2	0	+2	+4		+4	0	-4	-4	
$\beta$	-2	-2	-2	-2	-2	-2	-2	-2	-2		-4	-4	-4	-4	
Transducer Location															
1															
2											149.5				
3	148.0	147.0	146.5	146.5	147.0	147.0	147.0	147.5	148.0		146.0	147.5	147.5		
4											154.0		148.5		
5											BAD				
6											143.5	143.5	149.0	150.0	
7											148.5		151.0		
8											147.0				
9											143.5	144.0	146.0	146.5	
12											141.5	140.5	142.0	141.5	
14															
15															
16											142.0	142.5	142.5	142.0	
18															
20															
21															
23															
26															
27															
28											160.0	159.0	159.0	161.0	
29											151.0	154.0	153.0	160.0	
30											156.0	156.0	155.0		
31											148.5	148.5	148.5	151.0	
32											149.0	148.5	148.5		
33															
34											149.0	149.0	148.5	151.5	
35											148.0	148.0	148.0	150.0	
36											155.0	151.5	151.0	153.5	



Table VII. OVERALL SOUND PRESSURE LEVEL (cont'd) \*See Text  
db RE: .0002 Microbars

Correlation No. Mach No.	460	461	462	463	464	465	466	467	468	469	470	471	472	473	474
$\alpha$	0.8	0.8	0.82	0.82	0.82	0.84	0.84	0.84	0.86	0.86	0.86	0.88	0.88	0.88	0.9
$\beta$	0	+4	+4	0	-4	-4	0	+4	+4	0	-4	-4	0	+4	+4
Transducer Location	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4
1															
2		153.5	152.0					156.0*	157.0*					156.5*	156.5*
3															
4		146.0	146.0		147.5*	147.0*		145.5	145.0		146.0	147.5*		144.0	143.5
5		155.0	154.5		149.0*	148.5*		153.5	153.0					152.0	151.5
6															
7	143.5	143.5	143.5	143.5	149.0	150.0	143.0	143.0	143.0	142.5	151.5	152.0	142.5	143.0	143.5
8		150.5	151.0		154.0	158.5*		153.0	156.5			149.5*		158.0	143.5
9		148.5	146.0					148.0	145.0					148.5	148.0
12	144.5	144.5	144.5	144.0	146.5	147.0	144.0	144.0	143.5	143.5	147.0	147.0	143.0	143.0	143.0
14	141.0	141.5	142.0	141.0	141.5	142.0	141.0	142.0	142.0	141.5	142.5	143.5	142.0	142.0	142.5
15															
16		143.0	143.0	142.5	142.0	142.0	143.0	143.0	142.5	142.0	141.5	142.0	142.0	142.0	142.0
18															
20															
21															
23															
26															
27	161.0*	161.0*	161.0*	161.0*	161.0*	162.0*	162.0*	162.0*	162.0*	162.0*	162.5*	163.0*	162.5*	163.0*	
28	154.0*	154.0			159.5*	164.0*	166.0*								
29		145.5*	144.5*	142.5	144.0	144.0	142.5	142.0	142.5	143.0	145.0	145.0	143.0	142.5	142.5
30	150.5	150.5	151.0	151.0	151.5	152.0	151.5	151.0	151.0	151.5	151.5	153.0	155.0	155.0	153.0
31					151.0	152.0	151.5	154.0*	155.5*	152.5	152.0	151.0	153.0	156.0*	
32															
33	151.5	153.0*	155.5	154.0	153.5	156.0	156.5	158.5*	160.5*	158.0*	159.0*	159.5*	160.0*	160.5*	159.0*
34					149.0	150.0	150.0	150.0	150.0	154.5*	151.5	151.0	155.5*	150.5	152.0
35	150.0*	150.0	149.5	150.0	152.0	151.5	151.5*	150.5	150.0	150.5	151.0	151.0	151.5*	153.5	
36	155.0		160.0*	156.5	154.5	156.0	157.5	160.0*	161.5*	161.0*	160.0*	160.5*	162.0*	161.5*	162.0*

Table VII. OVERALL SOUND PRESSURE LEVEL (cont)\*See Text

db RE: .0002 Microbars

Correlation No.	475	476	477	478	479	480	481	482	483	484	485	486	487	488
Mach No.	0.9	0.9	0.92	0.92	0.92	0.94	0.94	0.94	0.96	0.96	0.96	0.98	0.98	0.98
$\alpha$	0	-4	-4	0	+4	+4	0	-4	-4	0	+4	+4	0	-4
$\beta$	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4
Transducer Location														
1														
2					155.0						147.0*	146.5*		
3														
4		146.0	146.0		143.0			146.5	147.5					148.5
5					151.5	151.5					152.0	152.0		
6														
7	142.5	152.5	152.5	143.0	143.5	144.0	143.5	153.0	153.5	143.5	144.5	145.0	144.0	153.5
8		146.0	145.0		142.0			145.0	145.5					146.0
9					148.0	148.0					148.0	148.5		
12	142.5	146.5	146.5	143.0	143.0	143.5	143.0	147.0	147.0	143.5	144.0	145.0	143.5	147.0
14	142.5	144.5	144.5	141.5	143.0	143.0	142.0	142.5	143.0	143.0	143.5	144.0	143.5	144.0
15														
16	142.0	142.0	142.0	141.5	141.5	142.0	143.0	142.5	142.5	143.0	143.5	143.5	142.5	142.5
18														
20														
21														
23														
26														
27		164.0*												
28			147.0*	143.5	146.0	145.5	143.5	149.0	146.5	148.0	147.0	146.0	147.5	143.5
29														
30	143.0	145.0	145.0	143.0	142.0	142.5	142.5	144.5	145.5	143.0	142.5	143.5	144.0	146.0
31	145.0		144.5	144.0	145.0	143.5	144.0	144.5	144.5	144.0	144.0	144.0	144.5	143.0
32	153.0	154.5				145.5*	143.5	142.0	142.0	142.0	143.0	142.5	142.0	142.0
33														
34	159.5*	158.5*	162.0*			146.5*	149.0		145.5	147.0	144.5	145.0	147.5	145.0
35	152.0	151.5				142.5	143.0	143.5	143.5	143.0	142.0	142.5	143.5	144.5
36	162.0*	161.0*	160.0*	161.0*	140.5	140.5	140.0	140.0	140.5	140.0	141.0	141.0	141.0	140.5
							158.5*	159.5*	148.0	148.0	151.0*	146.0	147.0	147.0

Table VII. OVERALL SOUND PRESSURE LEVEL (cont'd) \*See Text

db RE: .0002 Microbars

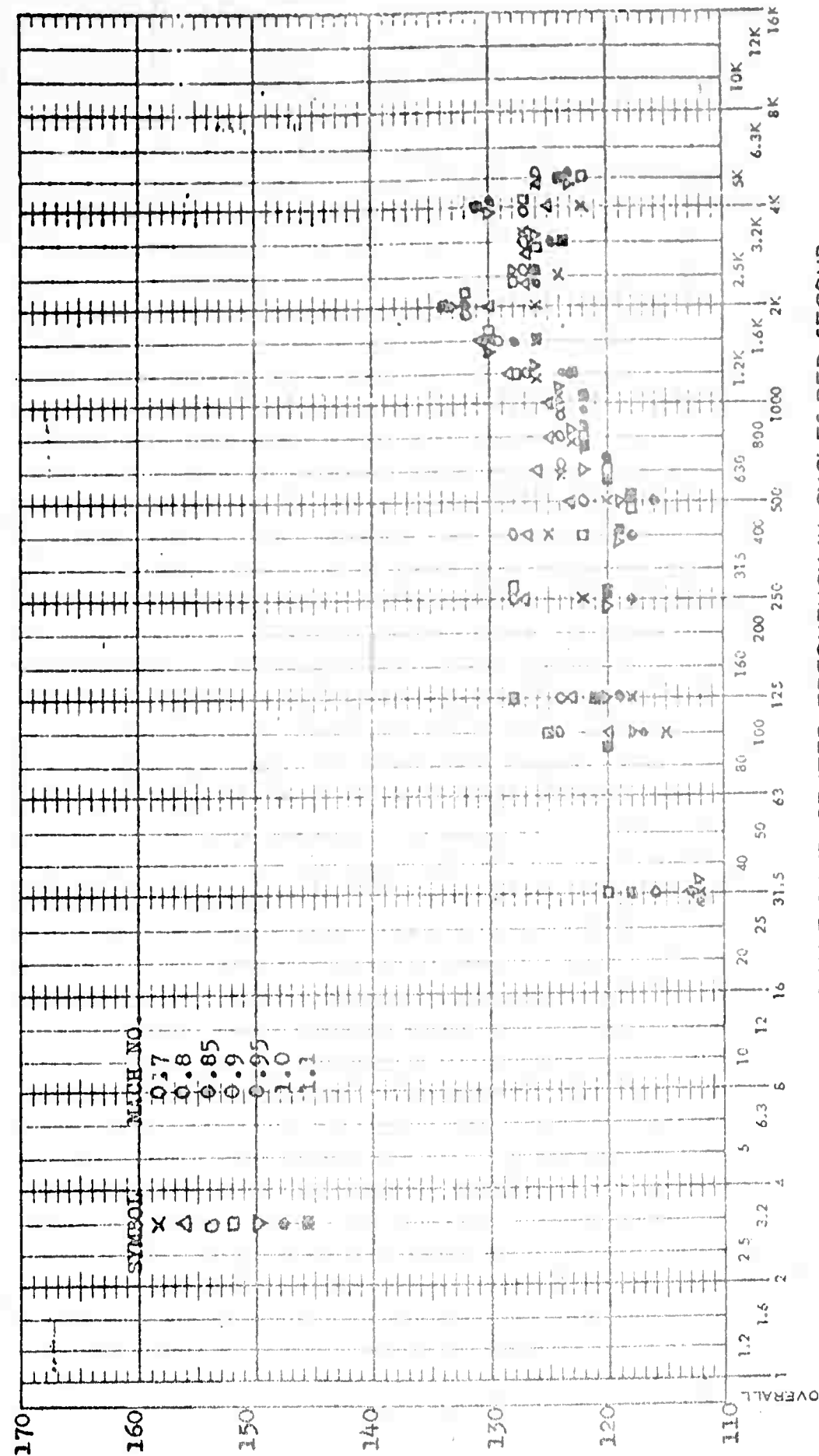
Correlation No.	489	490	491	492	493	494	495	496	497	498	499	500	501	502	503
Mach No.	1.0	1.0	1.0	1.02	1.02	1.02	1.04	1.04	1.04	1.06	1.06	1.06	1.08	1.08	1.08
$\alpha$	-4	0	+4	+4	0	-4	-4	0	+4	+4	0	-4	-4	0	-4
$\beta$	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4	-4
Transducer Location															
1															
2		142.5	143.0					143.5	143.5					144.0	143.0
3															
4					145.5	153.0					146.5	153.5			
5			151.0	150.5					151.0	152.0					153.0
6															
7	152.5	146.5	146.0	146.0	145.0	152.0	155.0*	148.0*	147.0	147.5	145.5	154.5	160.5*	149.0*	
8					143.5	147.0					144.5	148.5			
9			149.0	149.0	146.0*				149.5	152.5*	147.0*				153.0
12	151.0*	143.5	145.0	144.5	144.0	147.0	150.5*	144.5	145.0	145.5	144.5	151.5*	153.0*	145.0	
14	144.5	144.0	149.0*	149.0*	144.5	145.0	145.0	145.0	150.0*	148.0	145.5	146.0	146.0	146.0	
15															
16	142.5	143.0	147.5*	147.5*	143.0	142.5	142.5	143.0	147.5*	144.0	143.5	143.5	143.5	143.5	
18															
20															
21															
23															
26															
27	143.5	148.0	145.5	145.0	147.0	144.0	143.5	147.0	144.0	144.5	146.0	144.5	145.0	146.0	
28	145.5	145.0	147.0	146.0	144.5	146.0	146.5	145.5	147.5	148.0	146.0	147.0	147.5	146.5	
29	147.0	145.0	144.0	144.5	145.0	147.0	147.5	145.5	145.0	145.0	145.5	148.0	148.0	145.5	
30	144.5	145.0	145.0	144.0	143.0	143.0	144.0	144.0	144.5	144.0	145.0	144.0	144.0	144.0	
31	142.5	143.0	143.5	143.0	143.0	143.0	143.5	143.5	143.0	144.0	144.5	143.5	144.0	144.0	
32															
33	145.5	147.5	144.5	145.5	148.0	146.5	147.0	148.5	145.0	146.0	148.5	148.0	148.5	148.0	
34	144.5	144.0	143.5	143.5	144.0	144.5	145.0	144.5	144.0	144.5	145.0	145.5	145.5	145.0	
35	140.5	142.0	141.0	141.5	142.5	141.5	141.5	142.0	142.0	142.5	143.0	143.0	143.0	143.0	143.0
36	145.5	145.0	145.0	145.0	145.5	146.0	146.0	146.0	146.0	146.5	147.0	147.0	147.5	146.5	



Table VII.

Correlation No.	504	505	506
Each No.	0.725	0.7	0.65
α	+4	+4	+4
β	-4	-4	-4
Transducer Location			
1			
2			
3			
4			
5			
6			
7			
8			
9	146.0*		
12			
14			
15			
16	140.5	140.0	139.0
18			
20			
21			
23			
26			
27			
28			
29			
30			
31			
32			
33			
34			
35	147.0	145.0	141.5
36			

ONE-THIRD OCTAVE BAND SOUND PRESSURE LEVEL IN DB RE 0.0002 MICROBAR



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

Figure 15. Wind tunnel background noise.

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Table VIII. ACOUSTIC TEST RESULTS - SPACE-CORRELATION COEFFICIENTS

Transducer Location:	Reference at Test Point 4 Test at Test Point 5
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[illegible]

ONE THIRD OCTAVE BAND CENTER FREQUENCY

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BOEING

2-5535-2-17

Table VIII. ACOUSTIC TEST RESULTS - SPACE-CORRELATION COEFFICIENTS (continued)

Reference at Test Point 4  
Transducer Location: Test at Test Point 8

Correlation No.	464	465	470	471	504
Mach No.	0.82	0.84	0.86	0.88	0.725
$\infty$	-4	-4	-4	-4	+4
$\emptyset$	-4	-4	-4	-4	-4

CORRELATION COEFFICIENT					
40	+0.52	+0.75	-0.08	0	+0.83
50	+0.64	+0.56	+0.61	+0.22	+0.77
63	+0.67	+0.85	+0.49	+0.43	+0.54
80	+0.54	+0.69	+0.46	+0.44	+0.72
100	+0.72	+0.60	+0.31	+0.33	+0.64
125	+0.67	+0.57	+0.13	+0.38	+0.56
160	+0.60	+0.47	+0.28	+0.23	+0.47
200	+0.56	+0.52	+0.06	+0.12	+0.43
250	+0.47	+0.28	-0.06	+0.12	+0.52
320	+0.22	0	-0.22	+0.22	+0.49
400	-0.12	-0.17	-0.22	+0.06	+0.43
500	-0.17	-0.13	-0.13	0	+0.33
630	-0.17	-0.16	-0.20	+0.06	+0.17
800	-0.14	-0.12	-0.16	+0.12	+0.06
1000	-0.20	-0.10	-0.09	+0.23	0
1250	0	-0.10	+0.10	+0.25	+0.06
1600	-0.39	-0.10	+0.09	+0.32	+0.06
2000	-0.20	0	+0.20	+0.34	+0.33
2500					

ONE THIRD OCTAVE BAND CENTER FREQUENCY

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Table VIII. ACOUSTIC TEST RESULTS - SPACE-CORRELATION COEFFICIENTS (continued)

Transducer Location: Reference at Test Point 4  
Test at Test Point 9

Correlation No.	164	165	170	171
Mach No.	0.82	0.84	0.86	0.88
$\infty$	-4	-4	-4	-4
$\theta$	-4	-4	-4	-4

	CORRELATION COEFFICIENT			
40	+0.33	+0.35	+0.51	+0.58
50	+0.60	+0.43	+0.19	+0.22
63	+0.58	+0.52	+0.45	+0.38
80	+0.66	+0.52	+0.58	+0.53
100	+0.73	+0.56	+0.28	+0.48
125	+0.56	+0.56	+0.48	+0.43
160	+0.60	+0.60	+0.52	+0.34
200	+0.60	+0.47	+0.38	+0.33
250	+0.28	+0.17	+0.22	+0.33
320	+0.12	-0.12	0	+0.12
400	-0.06	-0.06	+0.06	+0.13
500	-0.18	+0.06	+0.06	+0.13
630	-0.06	0	0	+0.06
800	0	0	0	-0.07
1000	+0.12	+0.12	+0.06	+0.13
1250				
1600				
2000				
2500				

ONE THIRD OCTAVE BAND CENTER FREQUENCY

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Transducer Location:	Reference at Test Point 14	Test at Test Point 12

Correlation No.	307	351	407	408	502
Mach No.	0.85	1.08	1.08	1.08	1.08
$\infty$	-4	+4	-4	+4	0
$\beta$	0	0	-4	-4	-4

Correlation Coefficient					
40	+0.77	0	+0.17	0	+0.22
50	+0.82	+0.12	+0.33	-0.06	+0.22
63	+0.67	-0.06	+0.38	-0.12	+0.43
80	+0.70	0	+0.38	-0.12	+0.22
100	+0.60	0	+0.28	-0.12	+0.12
125	+0.70	+0.17	+0.28	+0.12	+0.33
160	+0.67	+0.06	+0.17	-0.06	+0.17
200	+0.67	+0.06	+0.17	+0.17	+0.12
250	+0.52	+0.12	+0.33	+0.12	+0.33
320	+0.43	+0.06	+0.12	+0.17	+0.12
400	+0.38	+0.06	+0.12	+0.12	+0.17
500	+0.33	+0.12	+0.12	+0.06	+0.22
630	+0.12	0	+0.12	-0.06	+0.23
800	+0.13	0	+0.12	-0.12	+0.12
1000	0	+0.06	+0.06	-0.06	0
1250					
1600					
2000					
2500					

ONE THIRD OCTAVE BAND CENTER FREQUENCY

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Reference at Test Point 14  
Transducer Location: Test at Test Point 15

Correlation No.	321	351	407	408																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																														</
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VIII 5/15

Reference at Test Point 27  
Transducer Location: Test at Test Point 26

Correlation No.	307	360	397
Mach No.	0.85	0.9	0.8
$\alpha$	-4	0	+4
$\beta$	0	-4	-4

CORRELATION COEFFICIENT			
40	+0.33	0	+0.17
50	+0.12	-0.23	0
63	0	-0.06	+0.12
80	0	+0.12	0
100	-0.17	-0.12	0
125	-0.06	+0.06	+0.13
160	0	0	0
200	+0.06	0	0
250	-0.12	0	+0.06
320	0	+0.12	0
400	+0.17	+0.06	-0.06
500	0	0	-0.16
630	0	-0.07	+0.08
800	0	+0.09	0
1000	0	0	0
1250			
1600			
2000			
2500			

ONE THIRD OCTAVE BAND CENTER FREQUENCY

1/6

1000

ONE THIRD OCTAVE BAND CENTER FREQUENCY

VIII 6/15

Transducer Location: Reference at Test Point 14  
Test at Test Point 16

Correlation No.	307	351	407	502
Mach No.	0.85	1.08	1.08	1.08
$\alpha$	-4	+4	-4	0
$\beta$	0	0	-4	-4

	CORRELATION COEFFICIENT				
40	+.22	-.13	+.14	-.31	
50	-.23	-.06	-.39	-.39	
63	0	-.23	0	-.07	
80	-.17	-.06	+.23	-.23	
100	-.12	-.06	0	-.23	
125	-.12	-.12	+.33	-.06	
160	0	-.06	0	-.22	
200	+.13	-.22	+.24	-.28	
250	+.06	-.06	0	-.38	
320	0	-.06	+.06	-.17	
400	+.12	0	0	-.12	
500	+.06	-.06	+.16	0	
630					
800					
1000					
1250					
1600					
2000					
2500					
$\Sigma$ SPL					

ONE THIRD OCTAVE BAND CENTER FREQUENCY

ONE THIRD OCTAVE BAND CENTER FREQUENCY

Transducer Location:	Reference at Test Point 27	Test at Test Point 28

[illegible]

ONE THIRD OCTAVE BAND CENTER FREQUENCY

VIII 8/15



Reference at Test Point 27  
Transducer Location: Test at Test Point 29

Correlation No.	307
Mach No.	0.85
$\alpha$	-4
$\beta$	0

CORRELATION COEFFICIENT	
40	+0.21
50	-0.24
63	0
80	+0.06
100	0
125	-0.12
160	+0.06
200	0
250	0
320	0
400	0
500	-0.06
630	
800	
1000	
1250	
1600	
2000	
2500	

ONE THIRD OCTAVE BAND CENTER FREQUENCY

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ONE THIRD OCTAVE BAND CENTER FREQUENCY

VIII 9/15

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**BOEING**

2-5535-2.

[illegible]

ONE THIRD OCTAVE BAND CENTER FREQUENCY

VII 10/15

Table VIII.

ACOUSTIC TEST RESULTS - SPACE-CORRELATION COEFFICIENTS (continued)

Reference at Test Point 32

[illegible]

ONE THIRD OCTAVE BAND CENTER FREQUENCY

VIII 11/15

Transducer Location: Reference at Test Point 32  
Test at Test Point 31

Correlation No.	307	357
Mach No.	0.85	0.85
$\infty$	-4	0
$\beta$	0	-4

CORRELATION COEFFICIENT	
40	+0.22
50	+0.06
63	+0.22
80	0
100	+0.17
125	+0.12
160	+0.06
200	+0.17
250	+0.12
320	+0.06
400	+0.06
500	0
630	0
800	0
1000	
1250	
1600	
2000	
2500	

ONE THIRD OCTAVE BAND CENTER FREQUENCY

XIII 12/15





Transducer Location:	Reference at Test Point 32 Test at Test Point 35
----------------------	---

Correlation No.	307	397	415
Mach No.	0.85	0.8	0.9
$\infty$	-1	+4	-4
$\delta$	0	-4	+4

CORRELATION COEFFICIENT	
40	+0.34 0
50	+0.34 0
63	+0.17 +0.31 +0.06
80	+0.17 +0.27 -0.28
100	+0.06 +0.20 0
125	+0.22 -0.09 -0.06
160	+0.06 0 -0.06
200	+0.06 0 -0.06
250	0 -0.12 -0.06
320	0 0 +0.06
400	0 +0.07 +0.06
500	+0.06 +0.07 +0.06
630	0 -0.07 -0.06
800	0 -0.06 -0.06
1000	0 0 0
1250	
1600	
2000	
2500	

ONE THIRD OCTAVE BAND CENTER FREQUENCY
100
125
160
200
250
320
400
500
630
800
1000
1250
1600
2000
2500

ONE THIRD OCTAVE BAND CENTER FREQUENCY

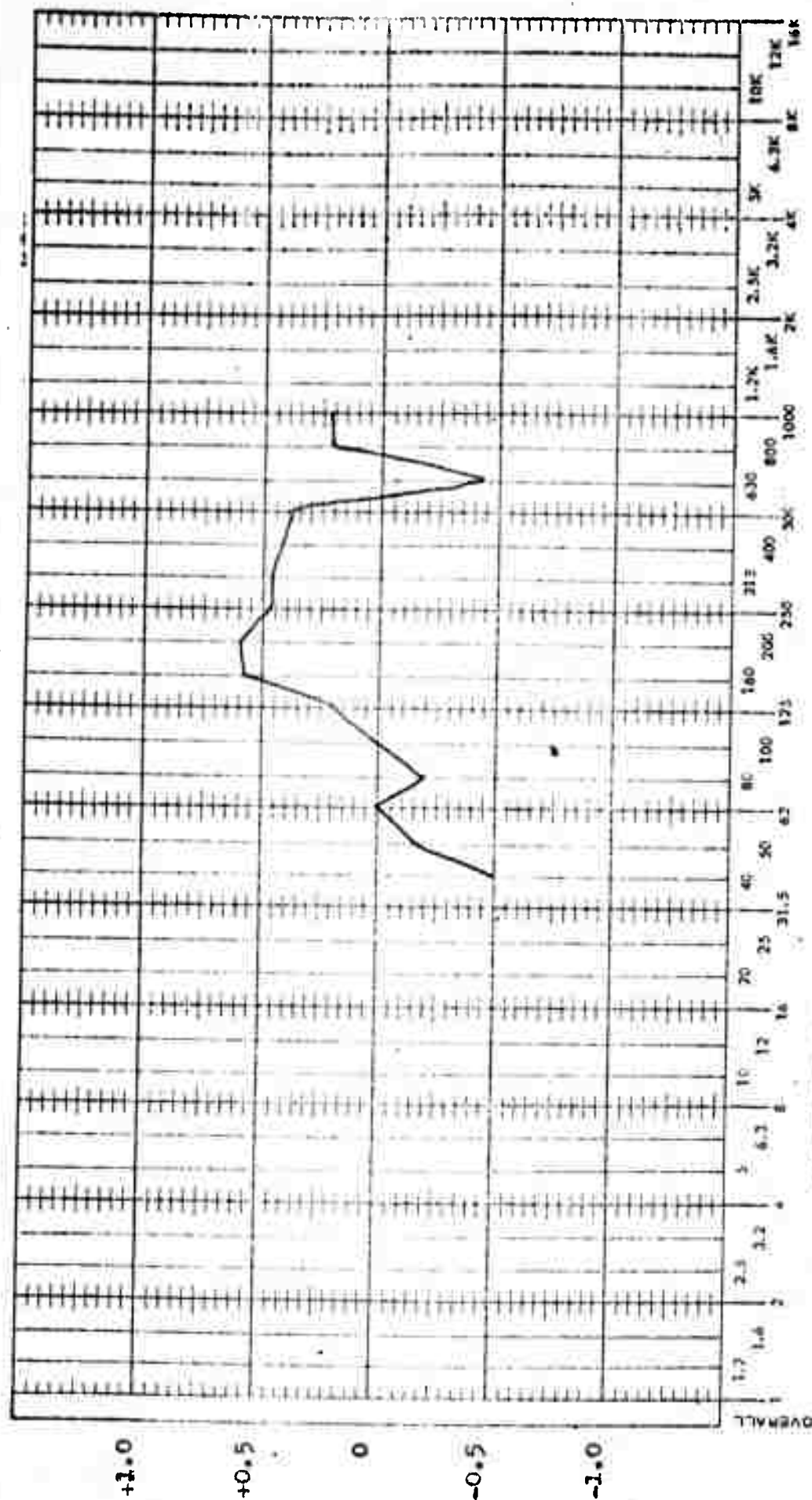
VIII 14/15

Transducer Location: Reference at Test Point 32  
Test at Test Point 36

[illegible]

ONE THIRD OCTAVE BAND CENTER FREQUENCY

VIII 15/15



REFERENCE TRANSDUCER 372, TEST POINT 4  
 TEST TRANSDUCER 407, TEST POINT 5

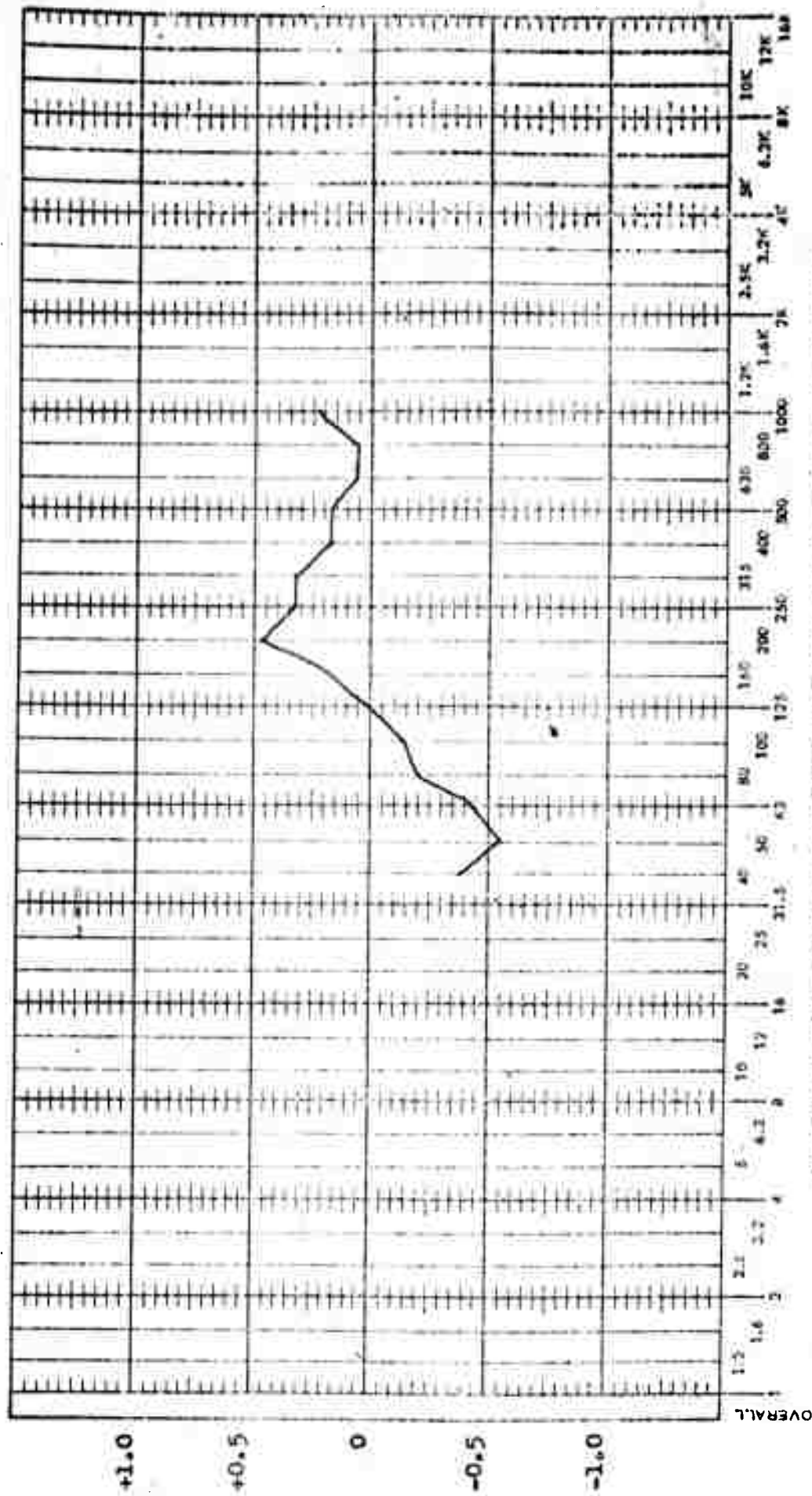
CORRELATION NO. 464  
 MACH NO. 0.82

Figure 16. Space correlation coefficients.



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SPACE CORRELATION COEFFICIENT (R)



ONE-THIRD-OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

REFERENCE TRANSDUCER 372, TEST POINT 4  
 TEST TRANSDUCER 407, TEST POINT 5  
 CORRELATION NO. L70  
 MACH NO. 0.86  
2.5  
4.0

Figure 16. Continued.



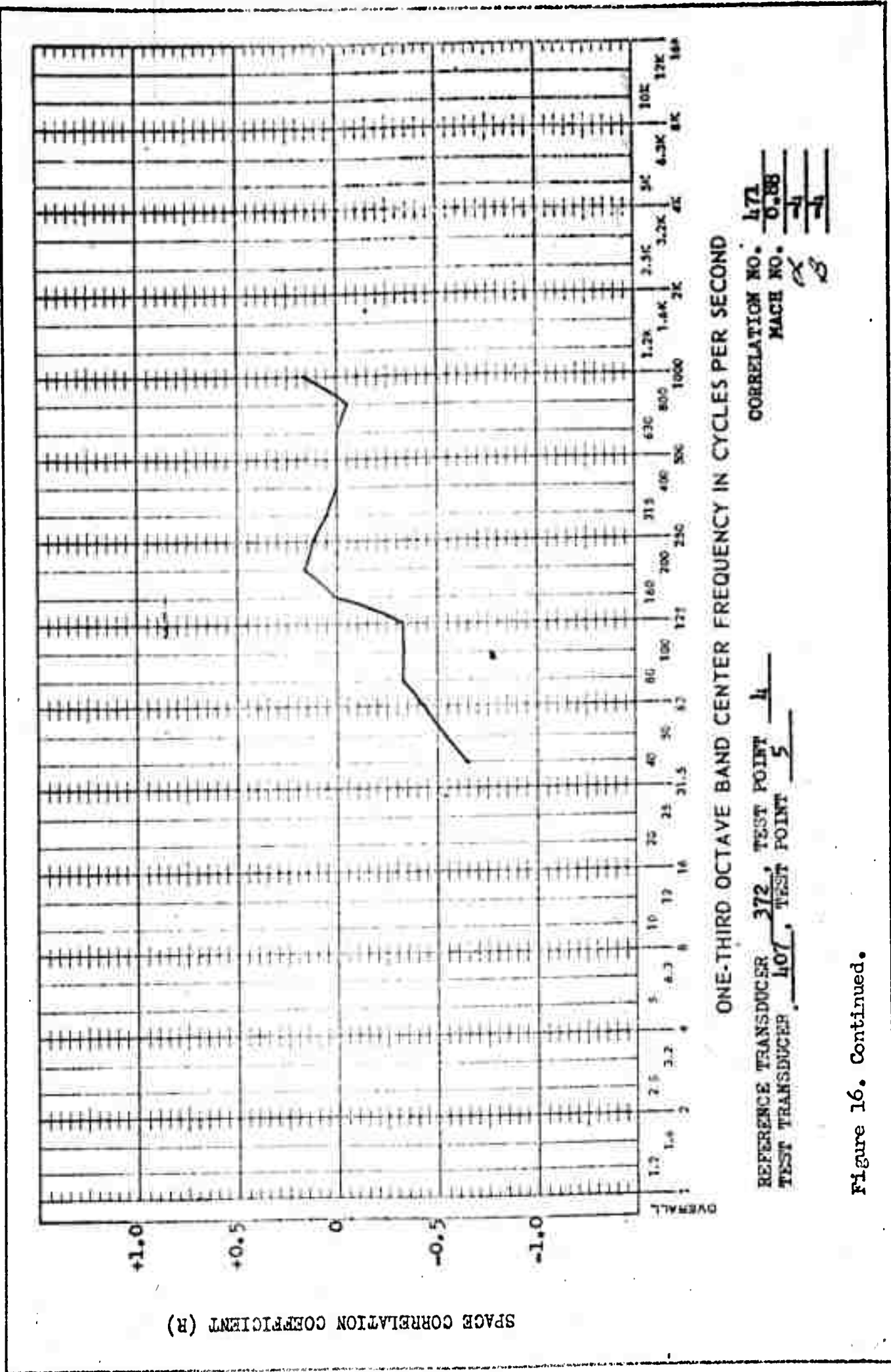
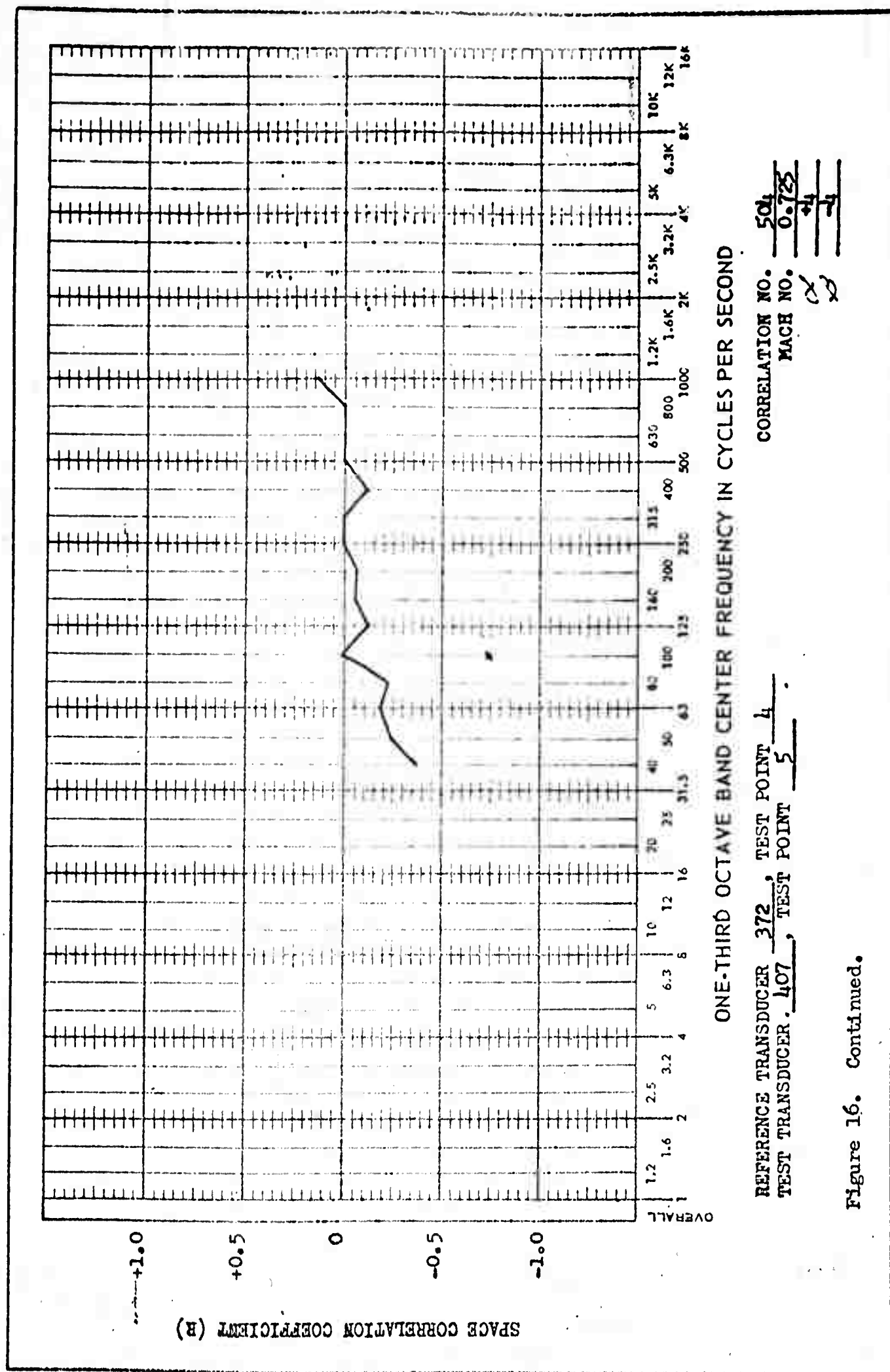
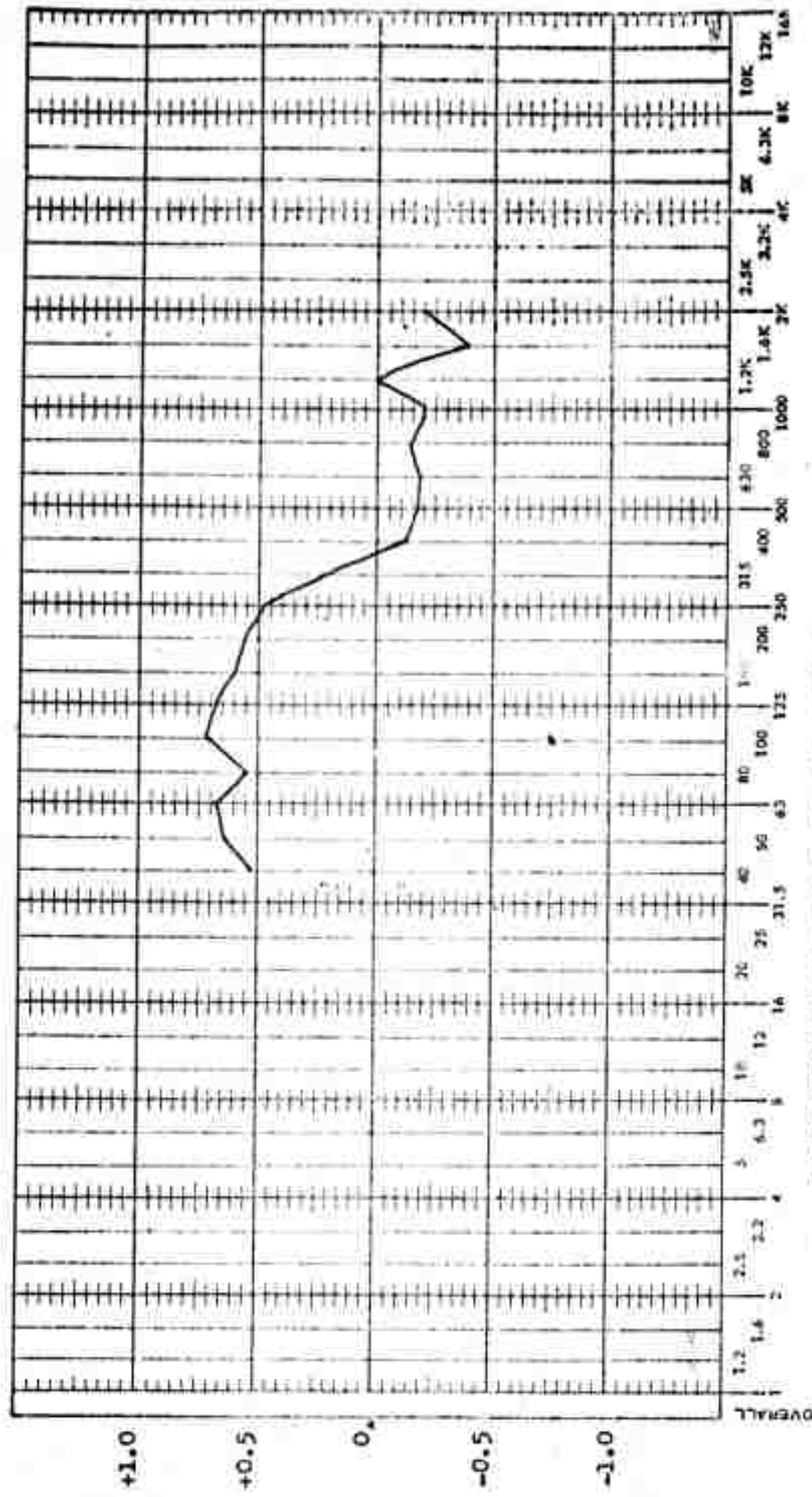


Figure 16. Continued.





ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

CORRELATION NO. 161  
MACH NO. 0.62

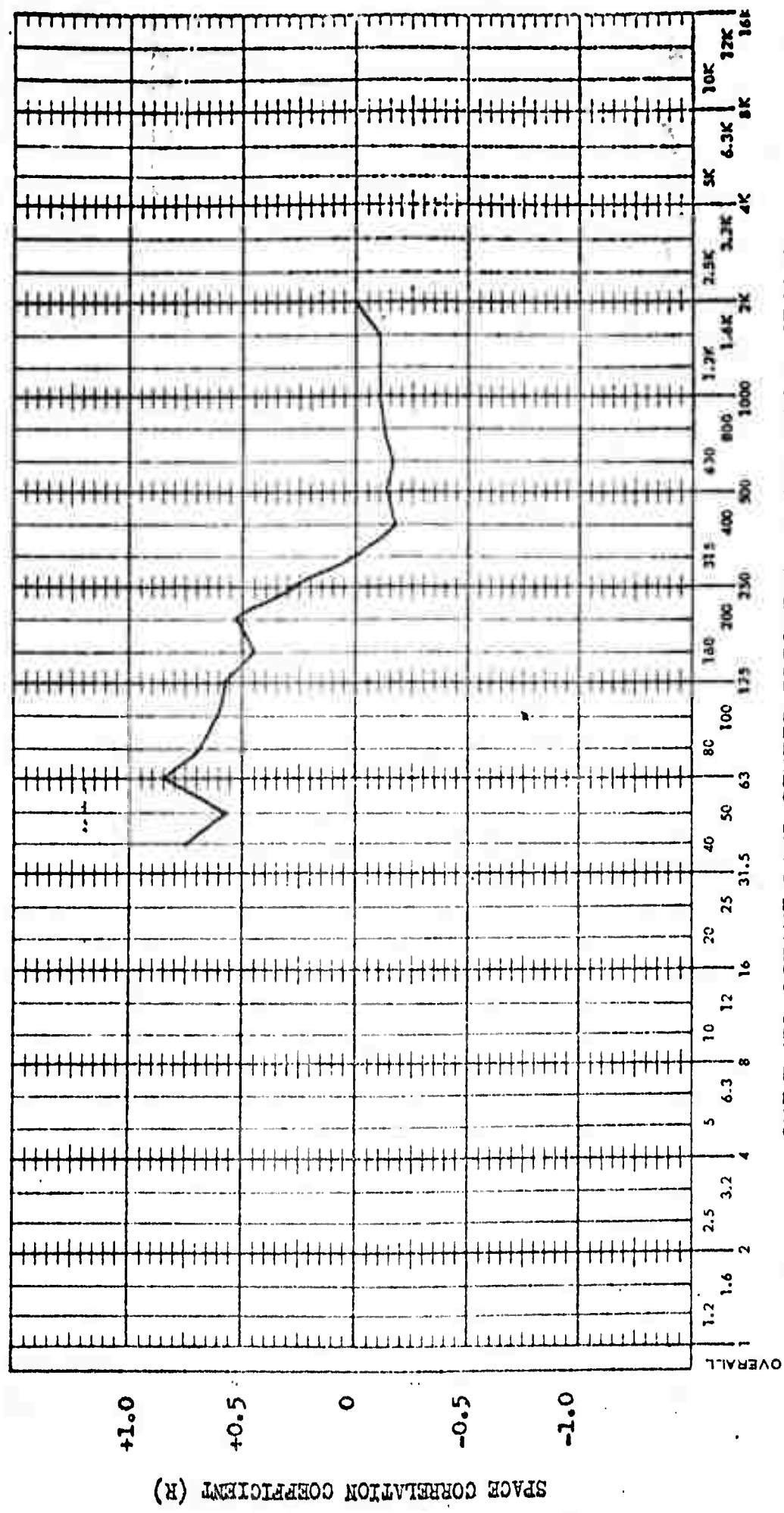
REFERENCE TRANSDUCER 372, TEST POINT 4  
TEST TRANSDUCER 416, TEST POINT 8

Figure 16. Continued.

SPACE CORRELATION COEFFICIENT (R)



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ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

CORRELATION NO. 165  
MACH NO. 0.84

REFERENCE TRANSDUCER 372, TEST POINT 4  
TEST TRANSDUCER 116, TEST POINT 8

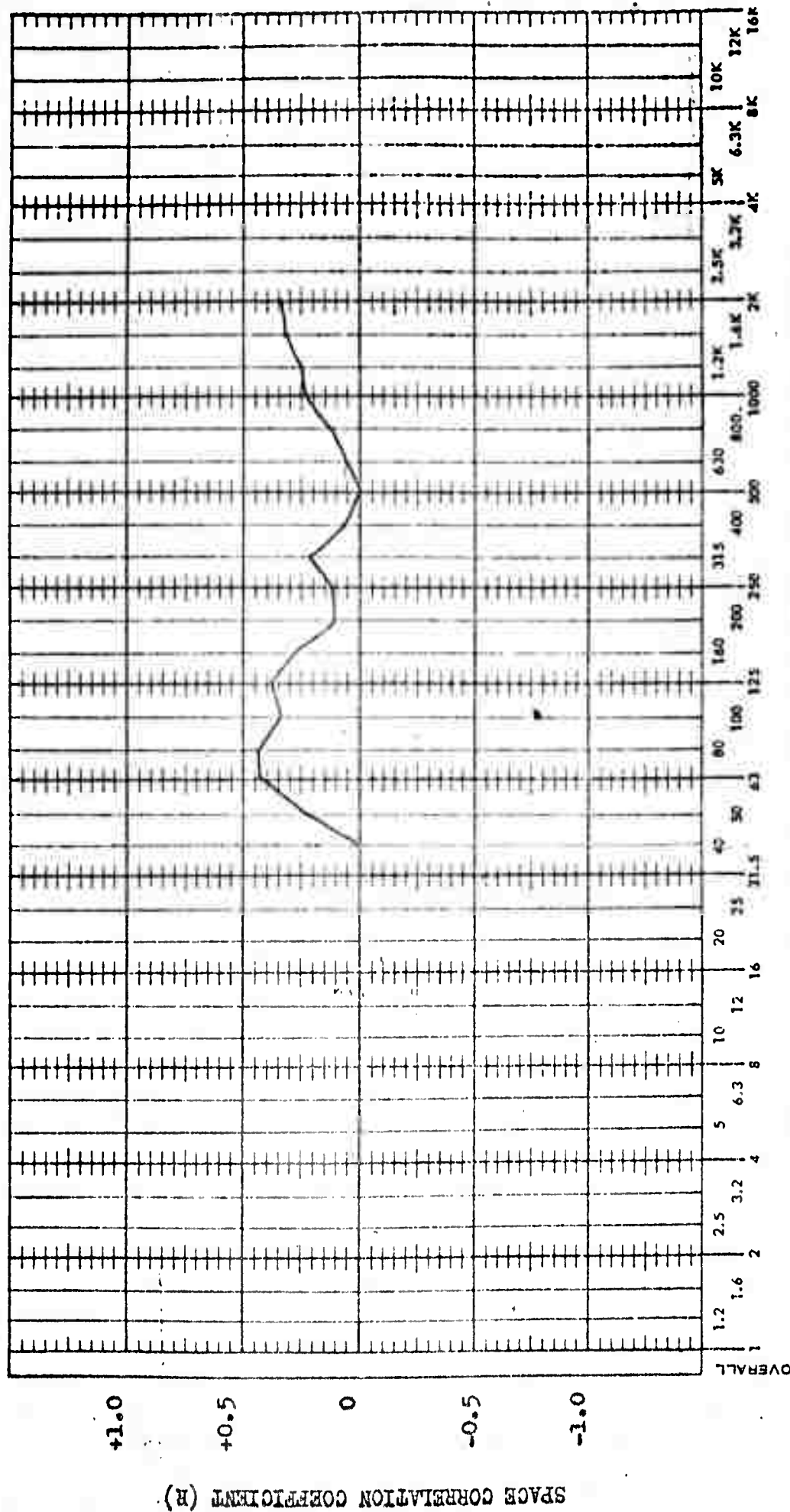
Figure 16. Continued.







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ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

CORRELATION NO. 472  
MACH NO. 0.88  
47

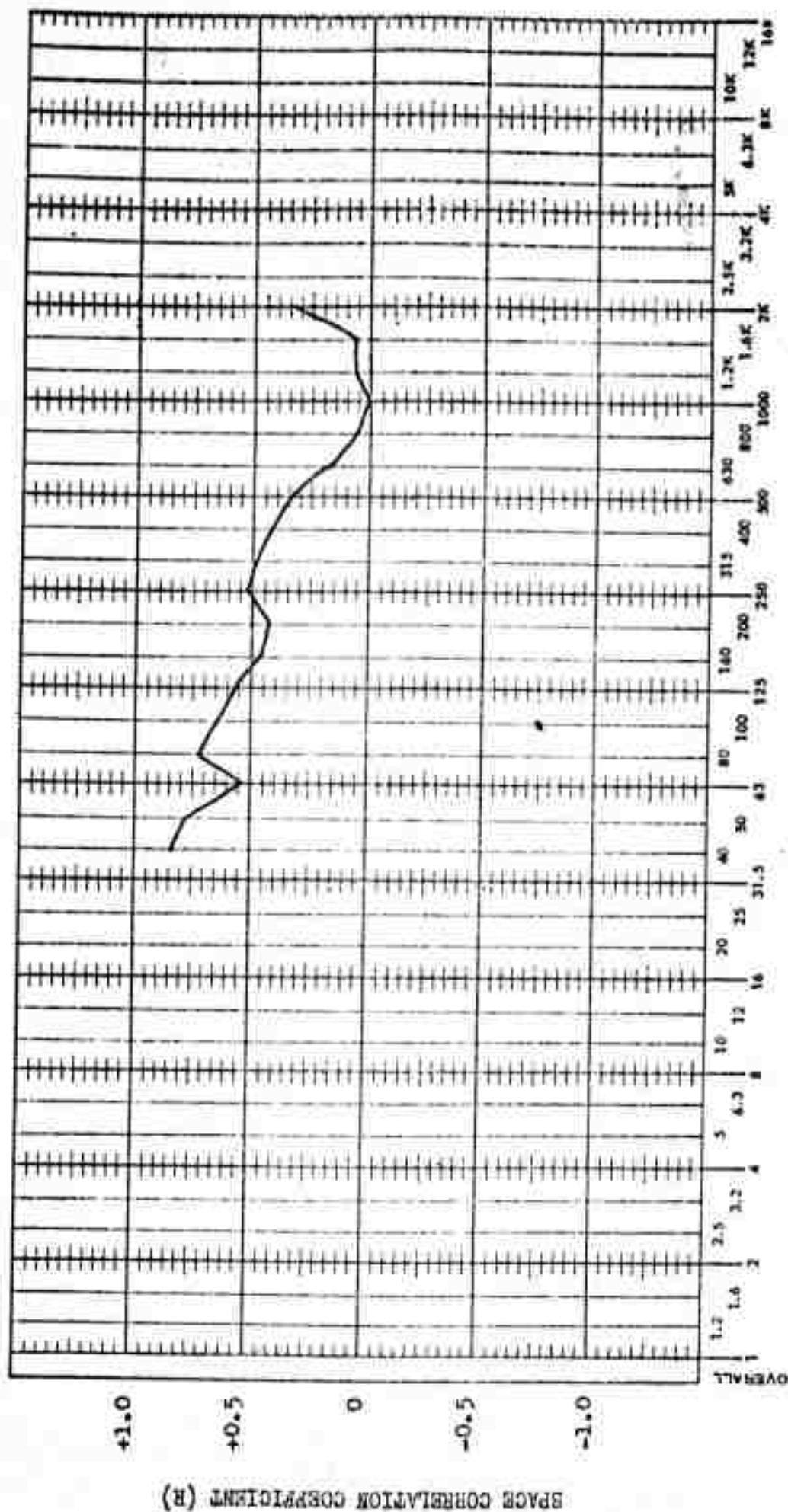
REFERENCE TRANSDUCER 372, TEST POINT 4  
TEST TRANSDUCER 416, TEST POINT 8

Figure 16. Continued.

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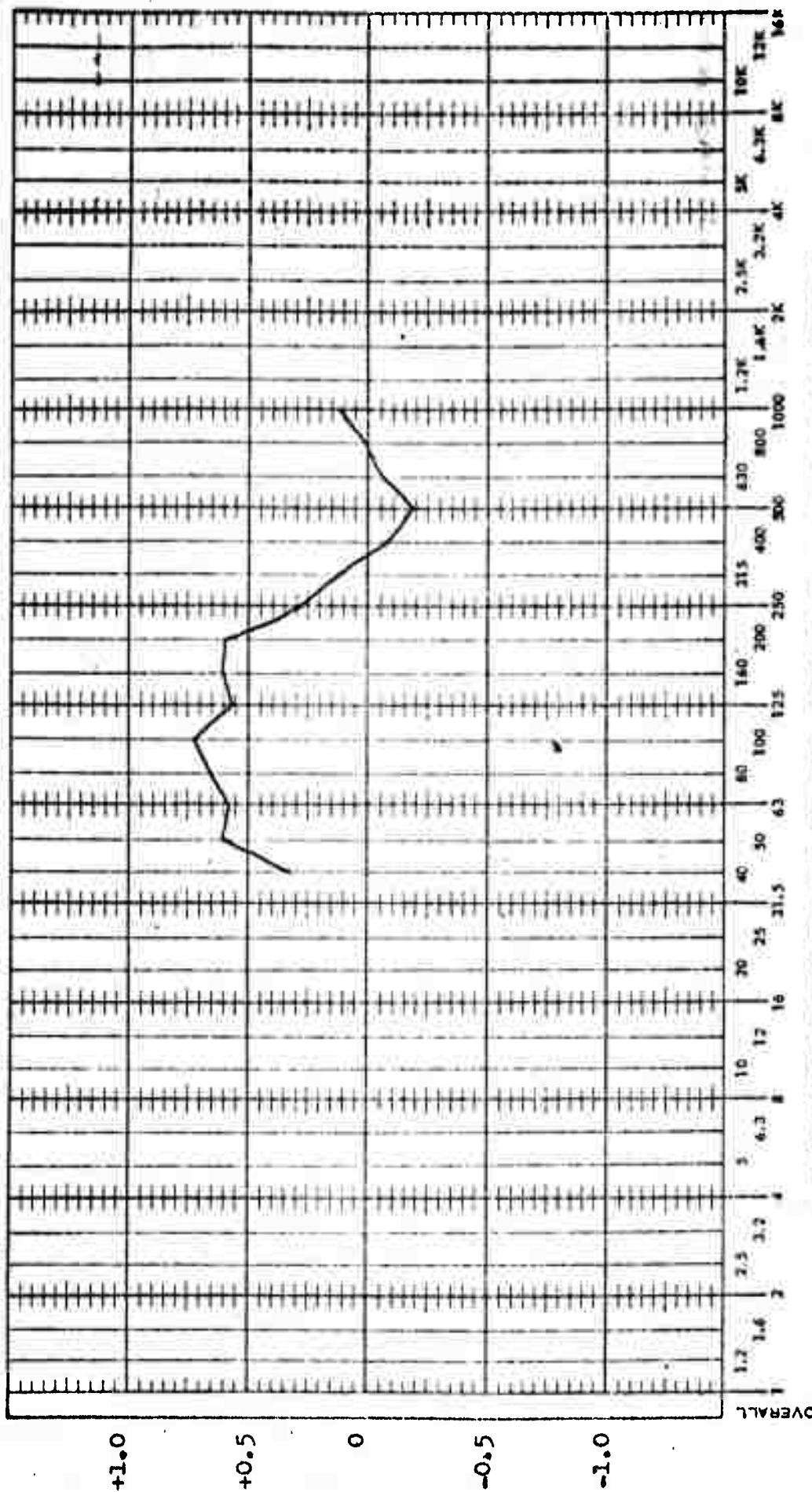
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ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

REFERENCE TRANSDUCER 372, TEST POINT 4  
 TEST TRANSDUCER 416, TEST POINT 8  
 CORRELATION NO. 504  
 MACH NO. 0.725

Figure 16. Continued.



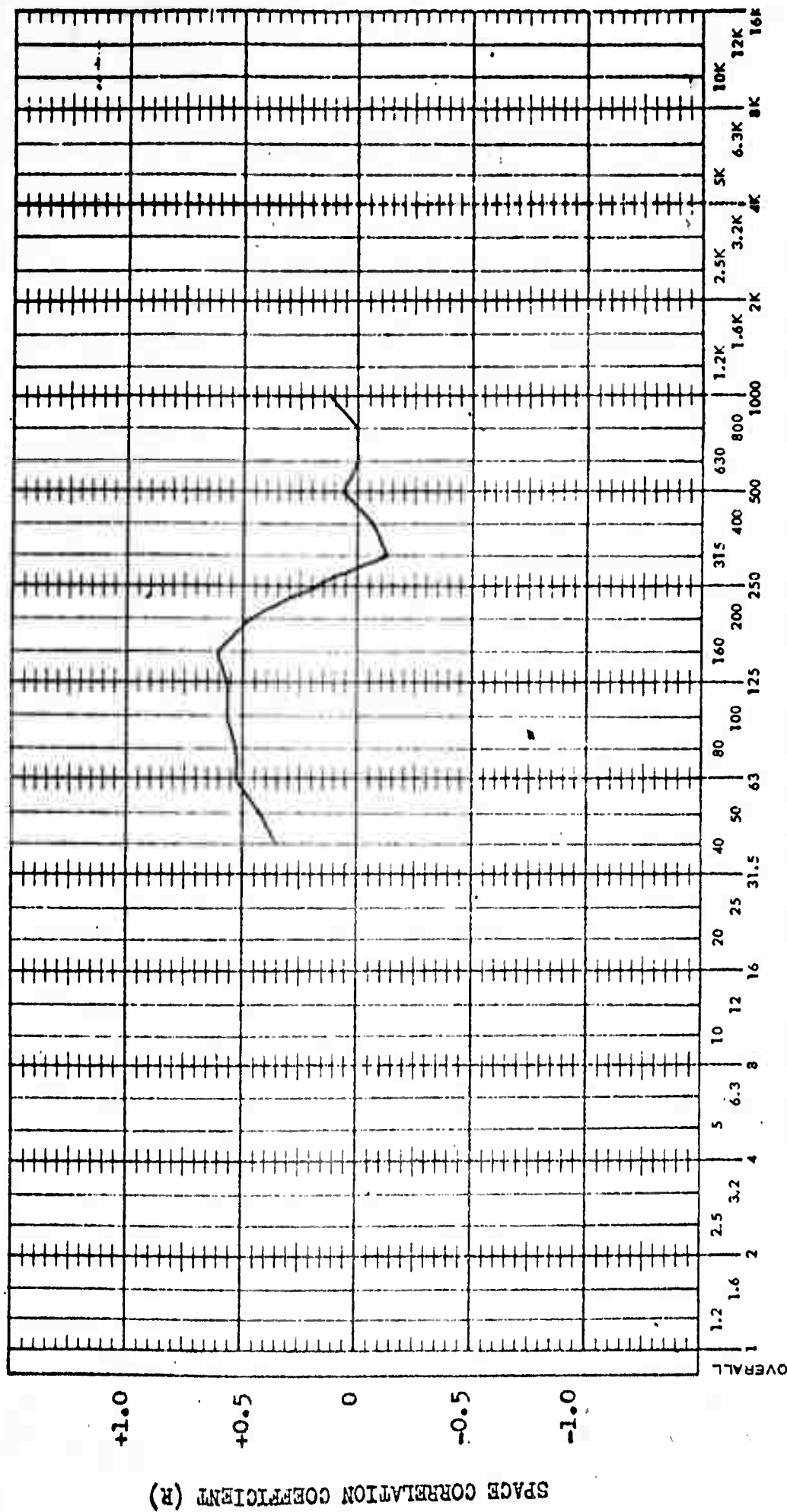
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

CORRELATION NO. 166  
MACH NO. 0.82

REFERENCE TRANSDUCER 372, TEST POINT 4  
TEST TRANSDUCER 355, TEST POINT 9

Figure 16. Continued.

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ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

REFERENCE TRANSDUCER 372, TEST POINT 4  
 TEST TRANSDUCER 355, TEST POINT 9  
 CORRELATION NO. 165  
 MACH NO. 0.84  
4  
4

Figure 16. Continued.

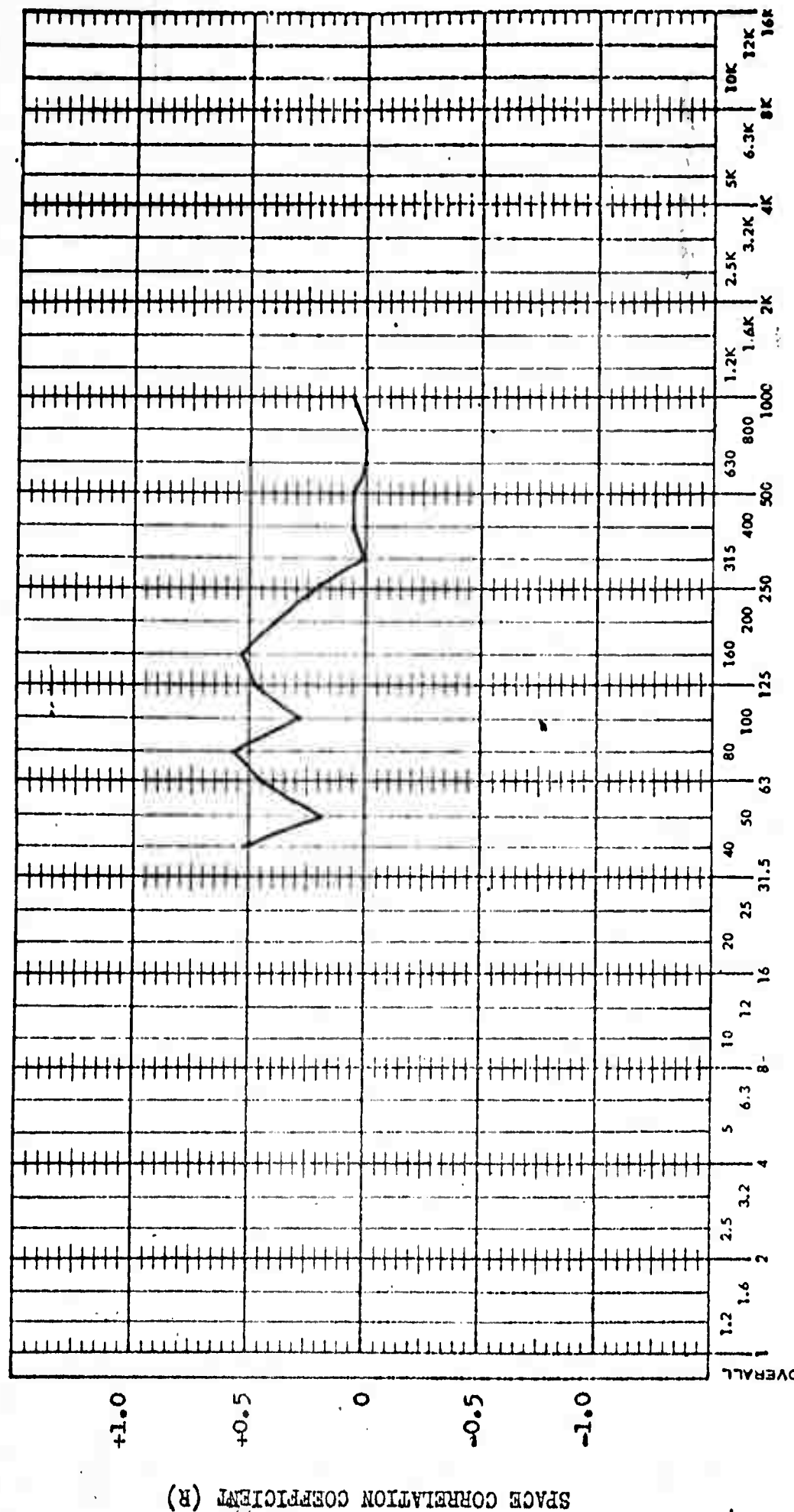
BOEING

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ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

CORRELATION NO. 470  
 MACH NO. 0.86

REFERENCE TRANSDUCER 372, TEST POINT 4  
 TEST TRANSDUCER 355, TEST POINT 9

Figure 16. Continued.



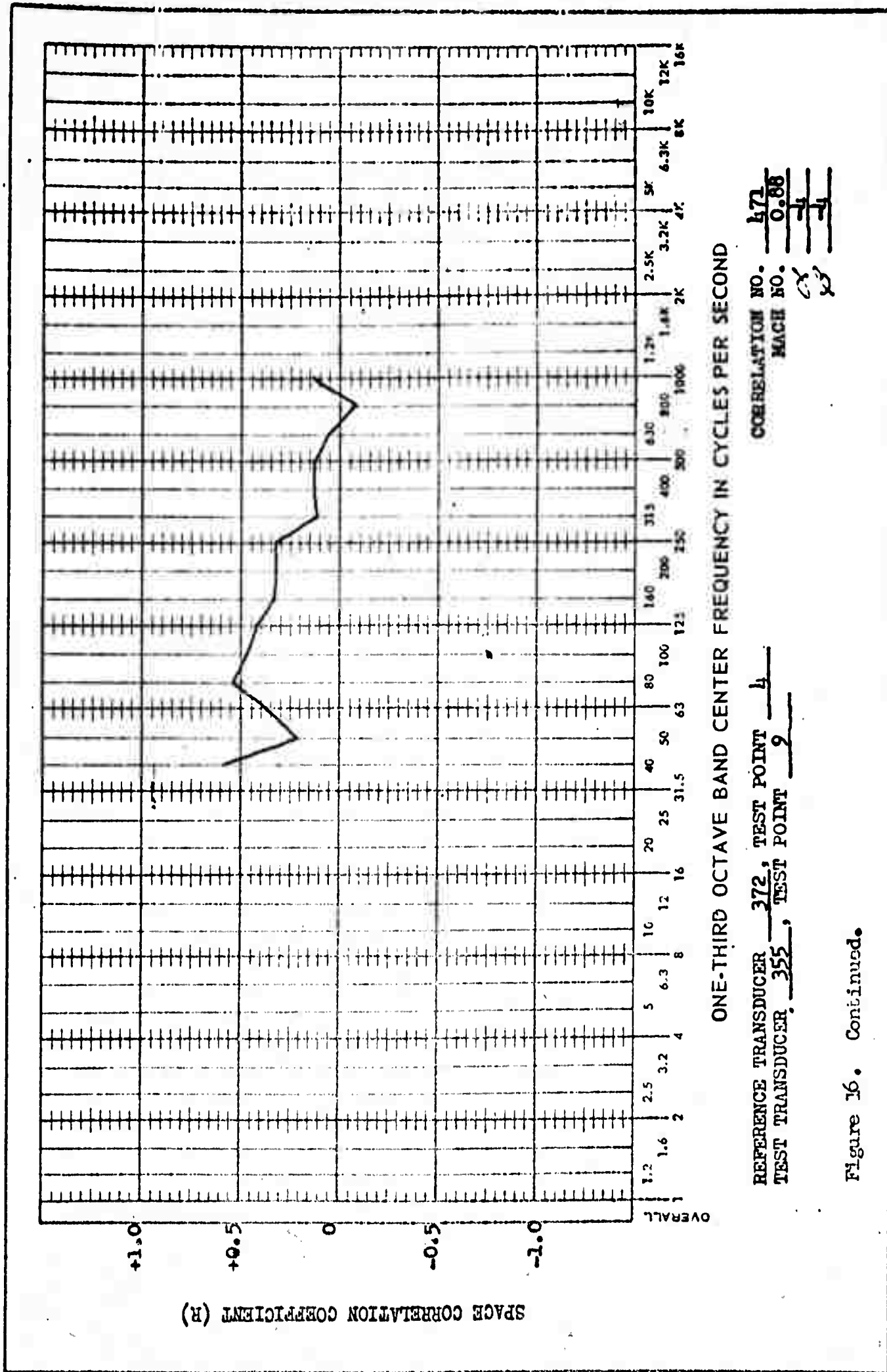


Figure 16. Continued.

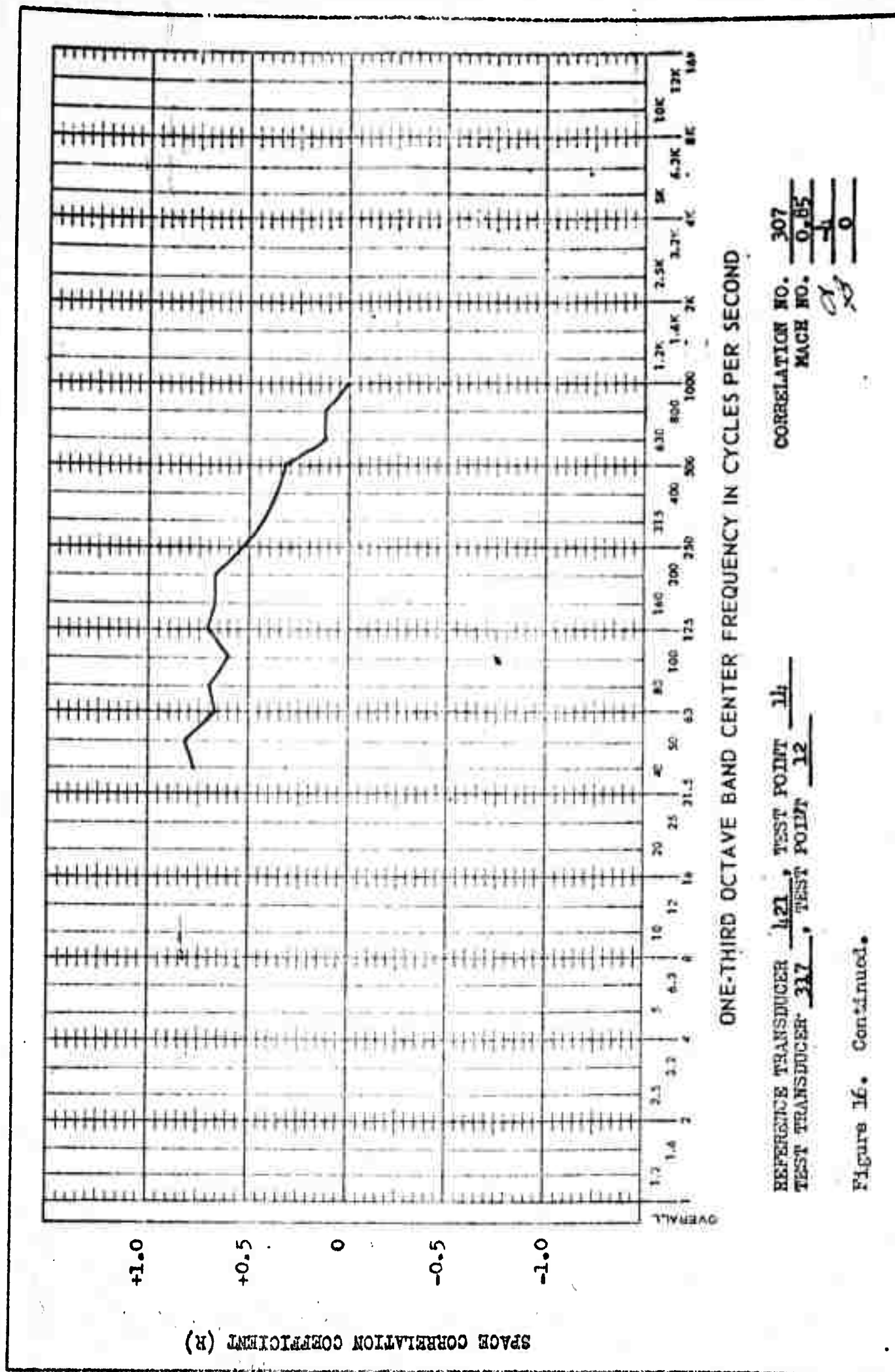
BUEING

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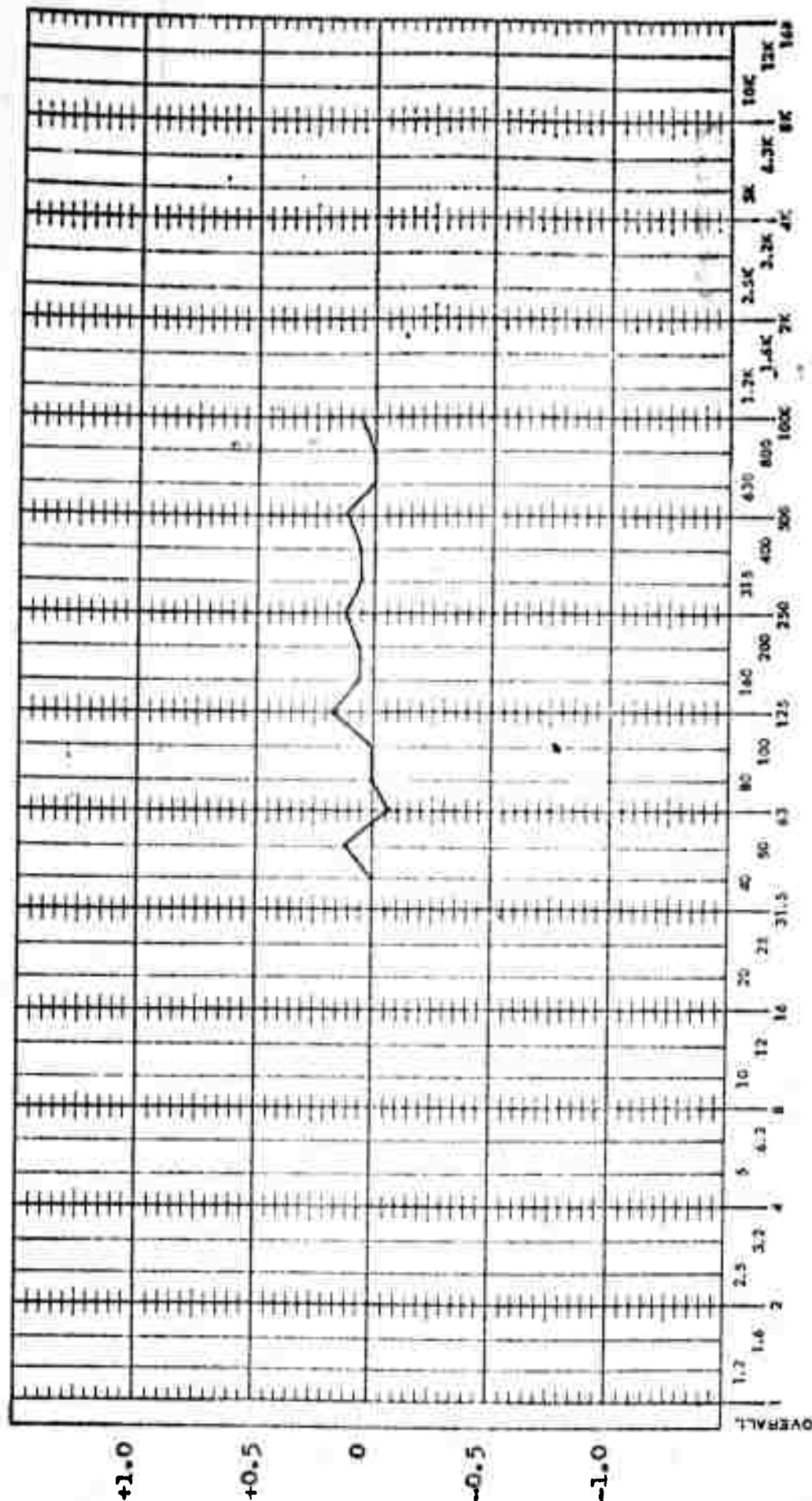


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NO. T2-2618

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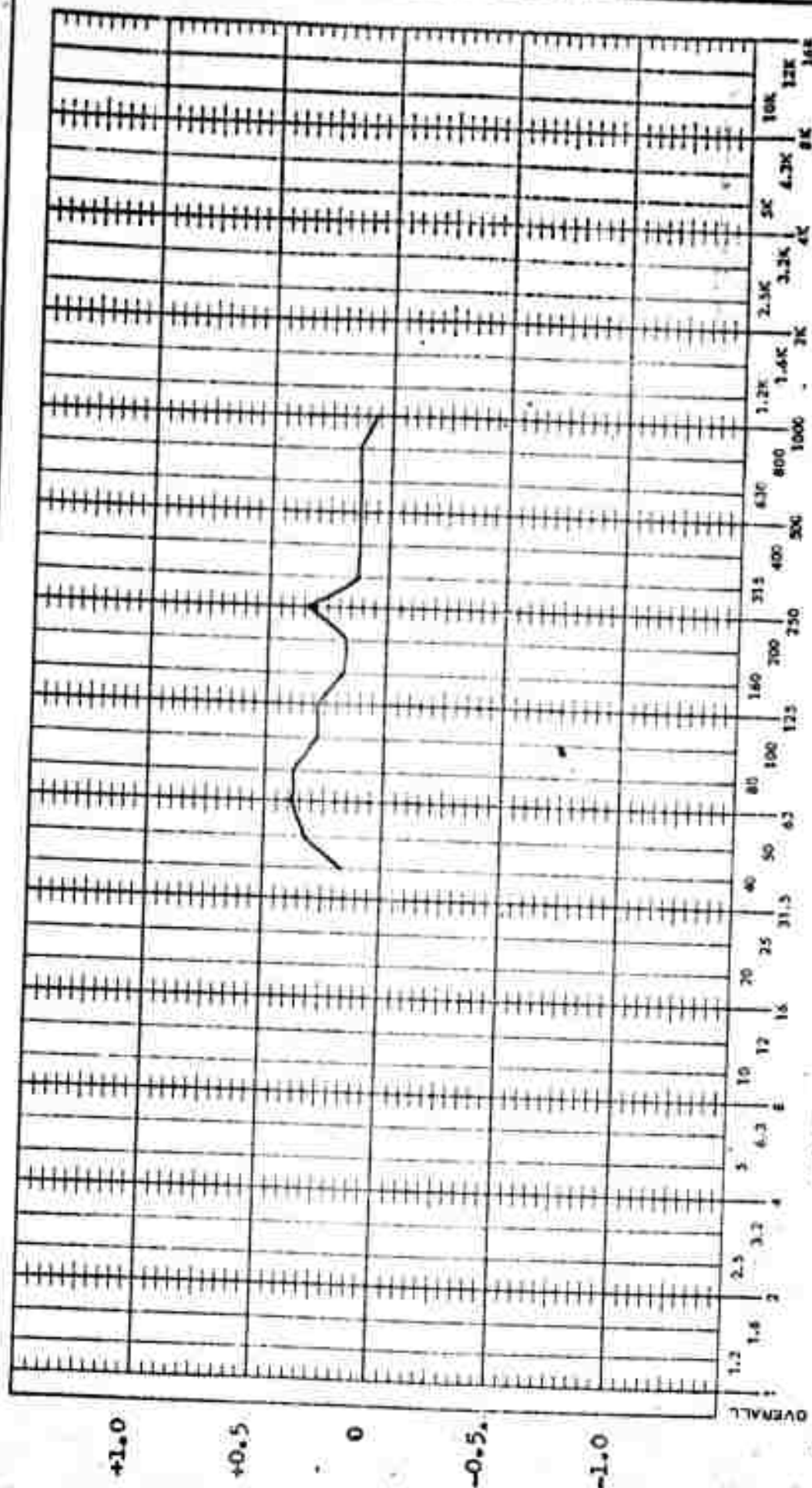
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ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

REFERENCE TRANSDUCER 421, TEST POINT 24  
 TEST TRANSDUCER 317, TEST POINT 12  
 CORRELATION NO. 351  
 MACH NO. 1.06  
0

Figure 16. Continued.



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

REFERENCE TRANSDUCER 421 TEST POINT 14  
TEST TRANSDUCER 317 TEST POINT 12

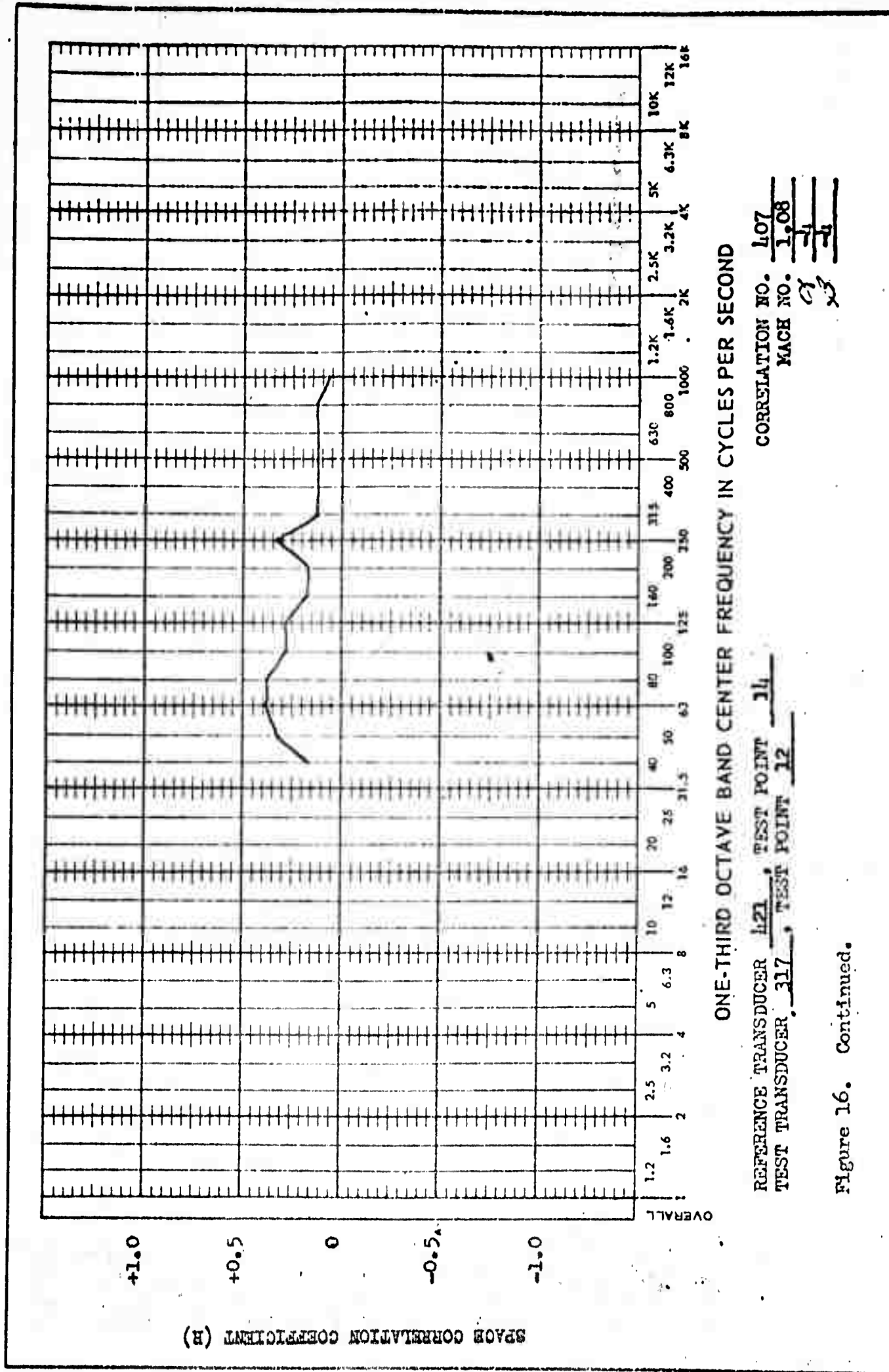
CORRELATION NO. 407  
MACH NO. 1.08  
0.2

Figure 16. Continued.

(H) INITIALISED NOISE NOISE NOISE









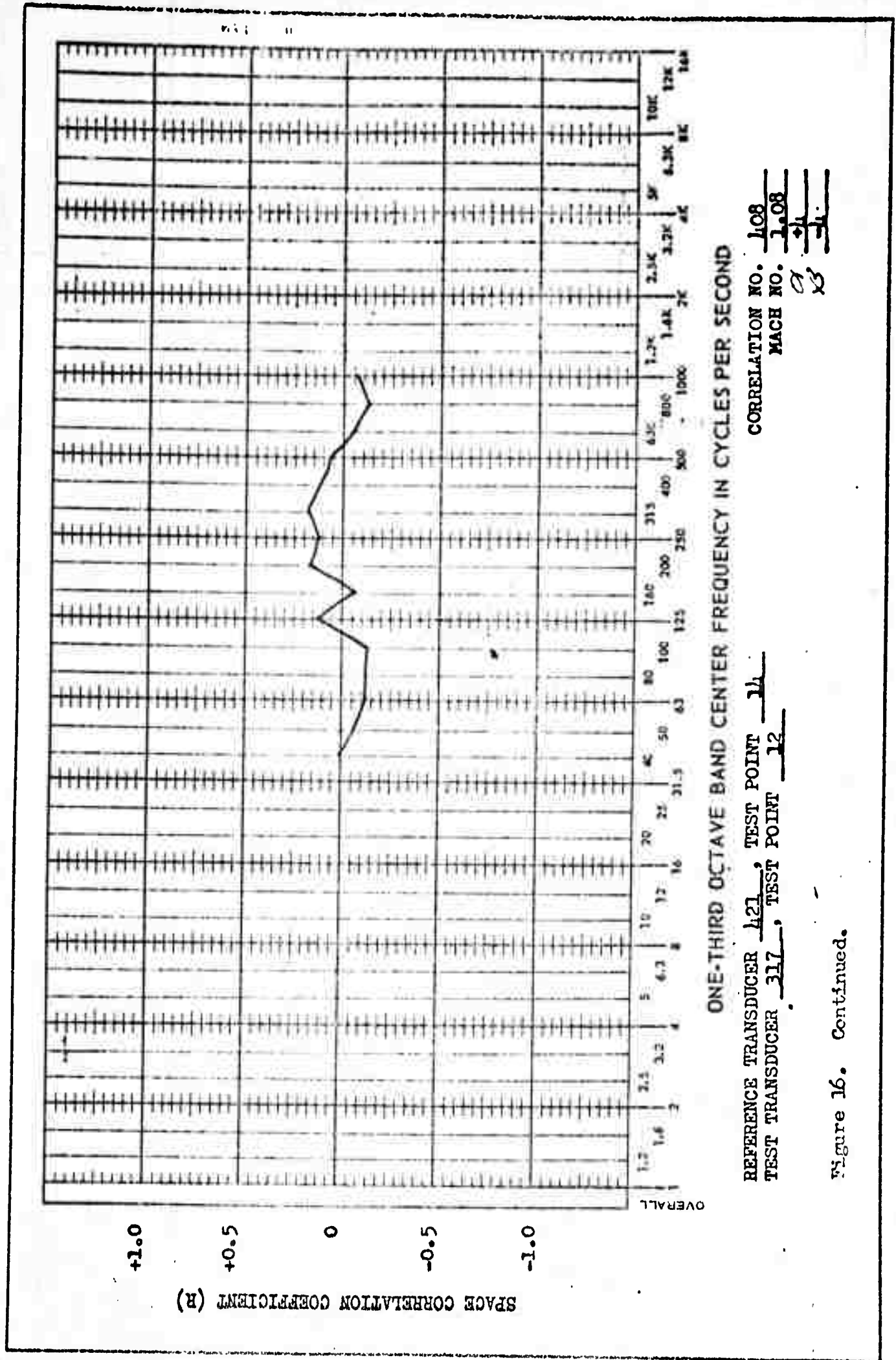


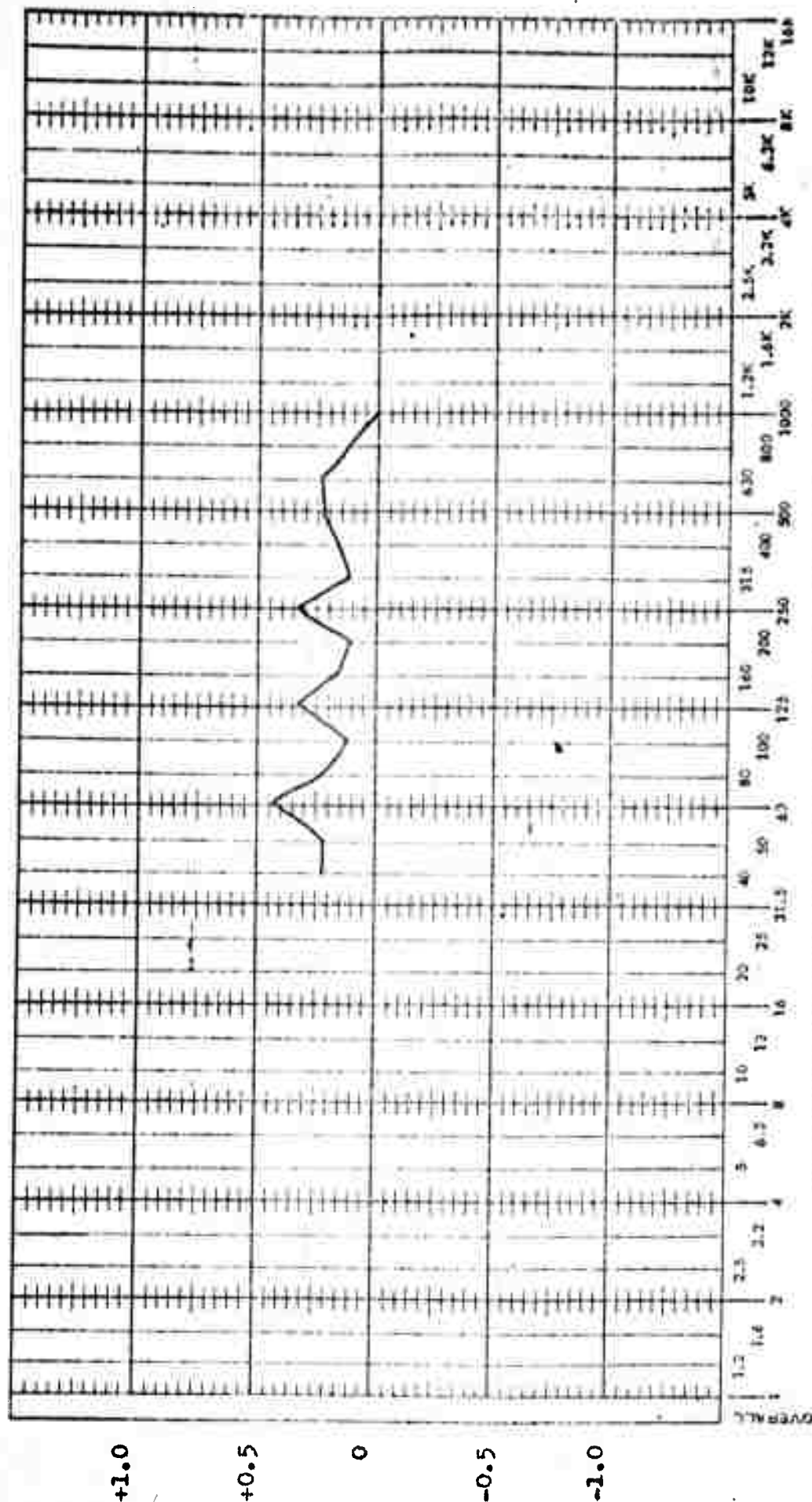
Figure 16. Continued.

ENGINE

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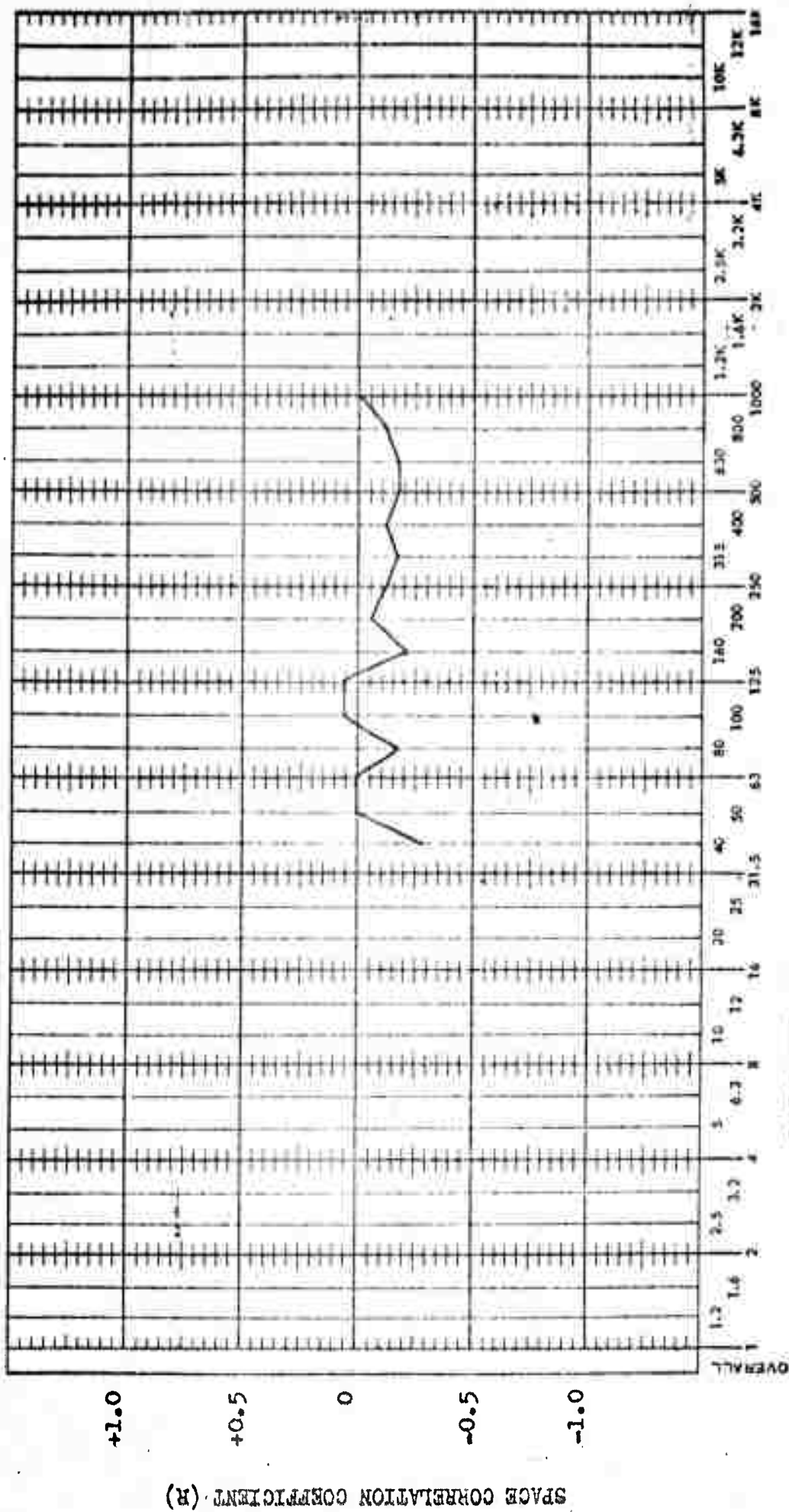


ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

REFERENCE TRANSDUCER 421, TEST POINT 14  
 TEST TRANSDUCER 317, TEST POINT 12  
 CORRELATION NO. 502  
 MACH NO. 1.08  
0  
4

Figure 16. Continued.

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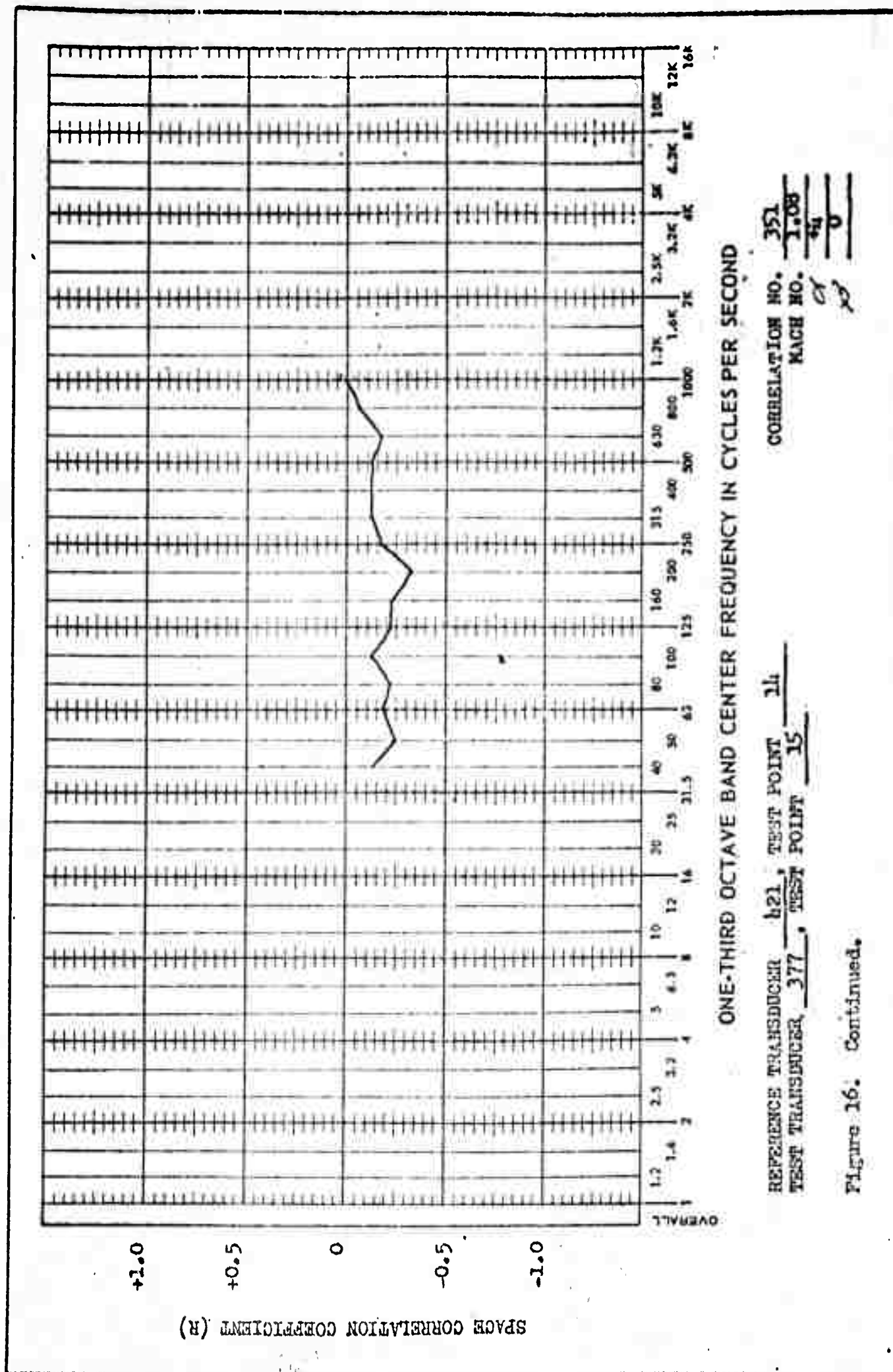


ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

REFERENCE TRANSDUCER 421, TEST POINT 21  
 TEST TRANSDUCER 377, TEST POINT 15

CORRELATION NO. 321  
 MACH NO. 1.08  
4  
0

Figure 16. Continued.

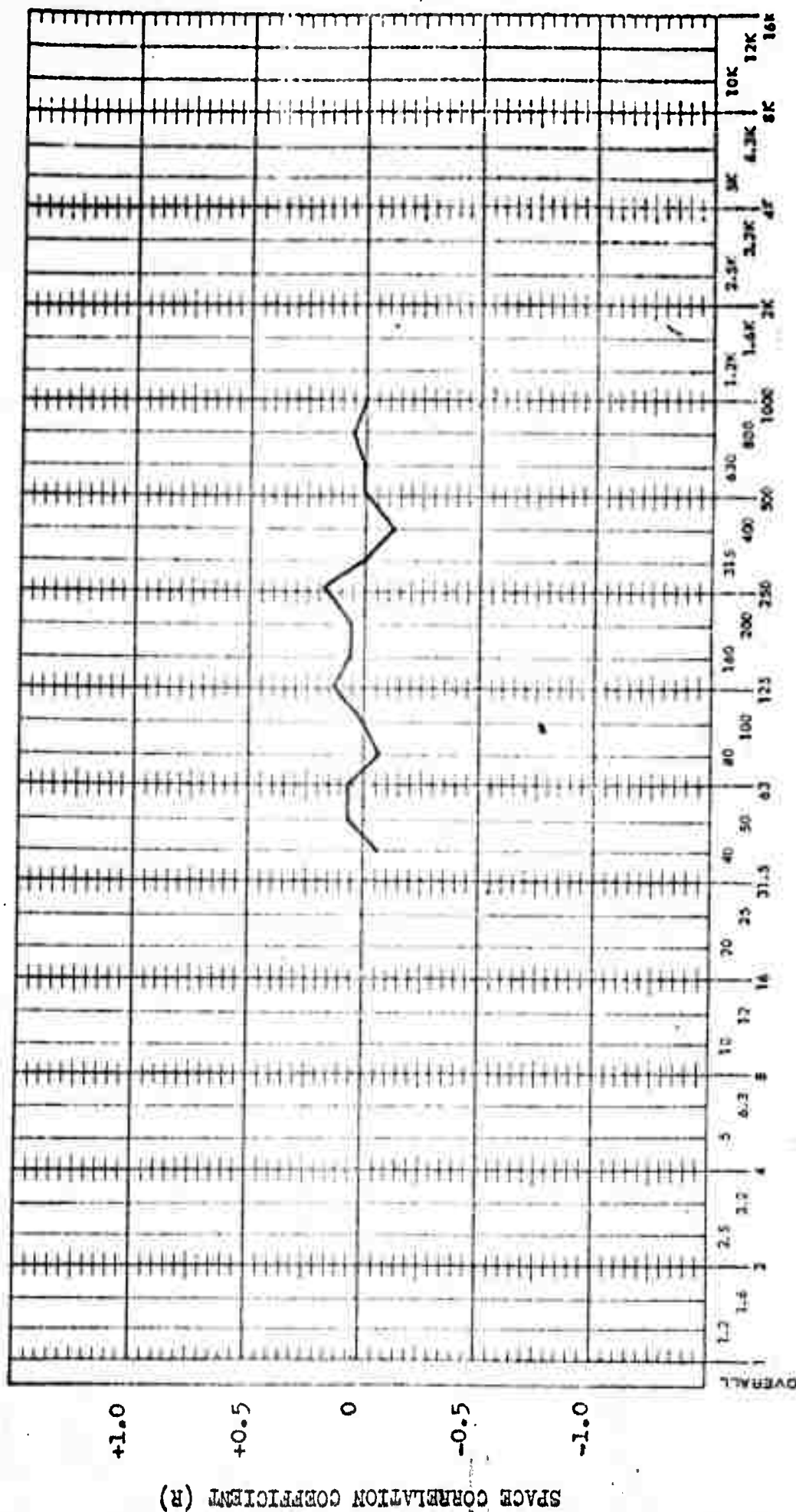


BOEING

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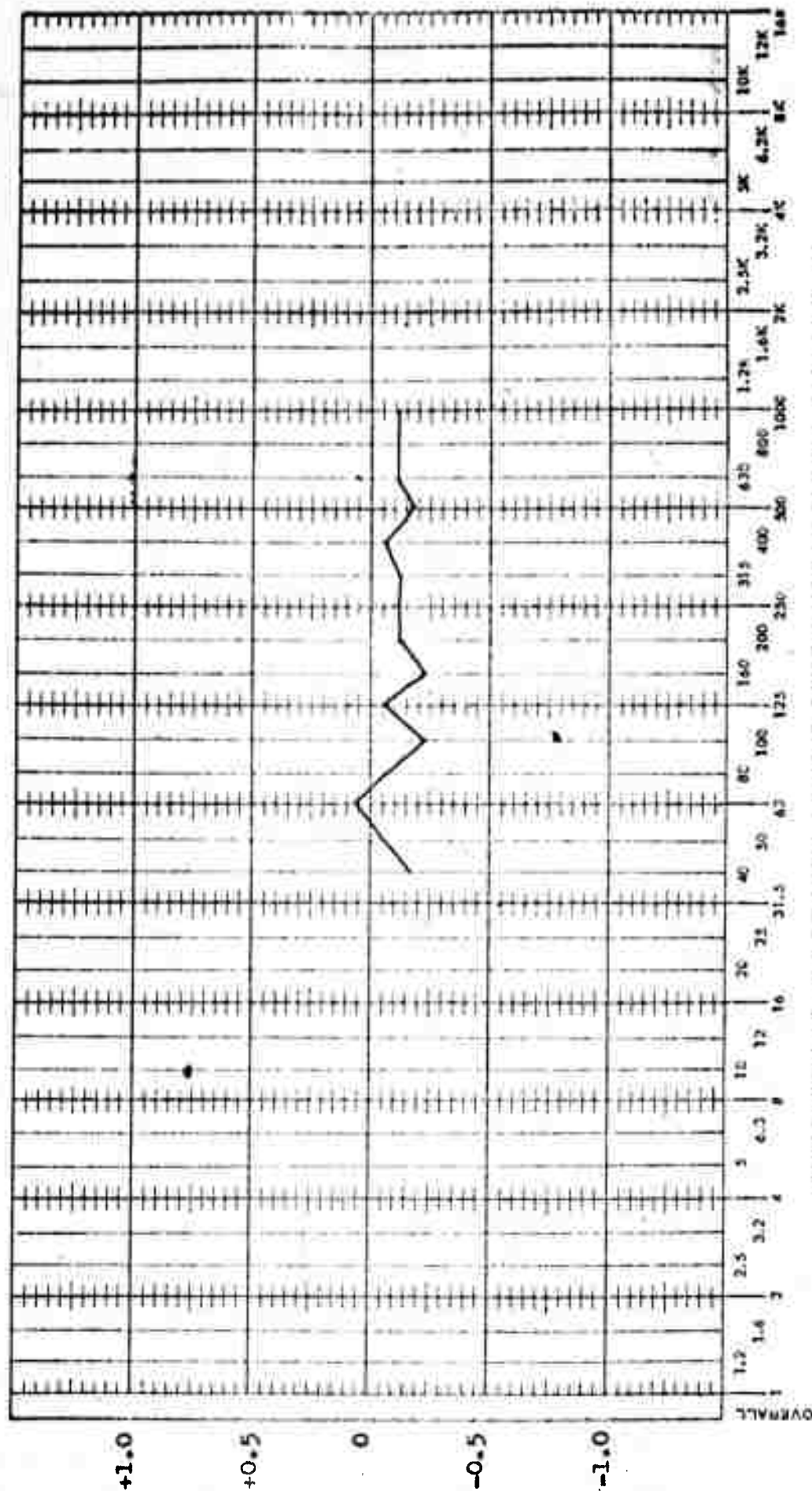


REFERENCE TRANSDUCER 1421, TEST POINT 11;  
TEST TRANSDUCER 177, TEST POINT 15

CORRELATION NO. 107  
MACH NO. 1.08

Figure 16. Continued.





ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

REFERENCE TRANSDUCER 421, TEST POINT 14  
 TEST TRANSDUCER 377, TEST POINT 15  
 CORRELATION NO. 108  
 MACH NO. 1.08

Figure 16. Continued.

SPACE CORRELATION COEFFICIENT (R)

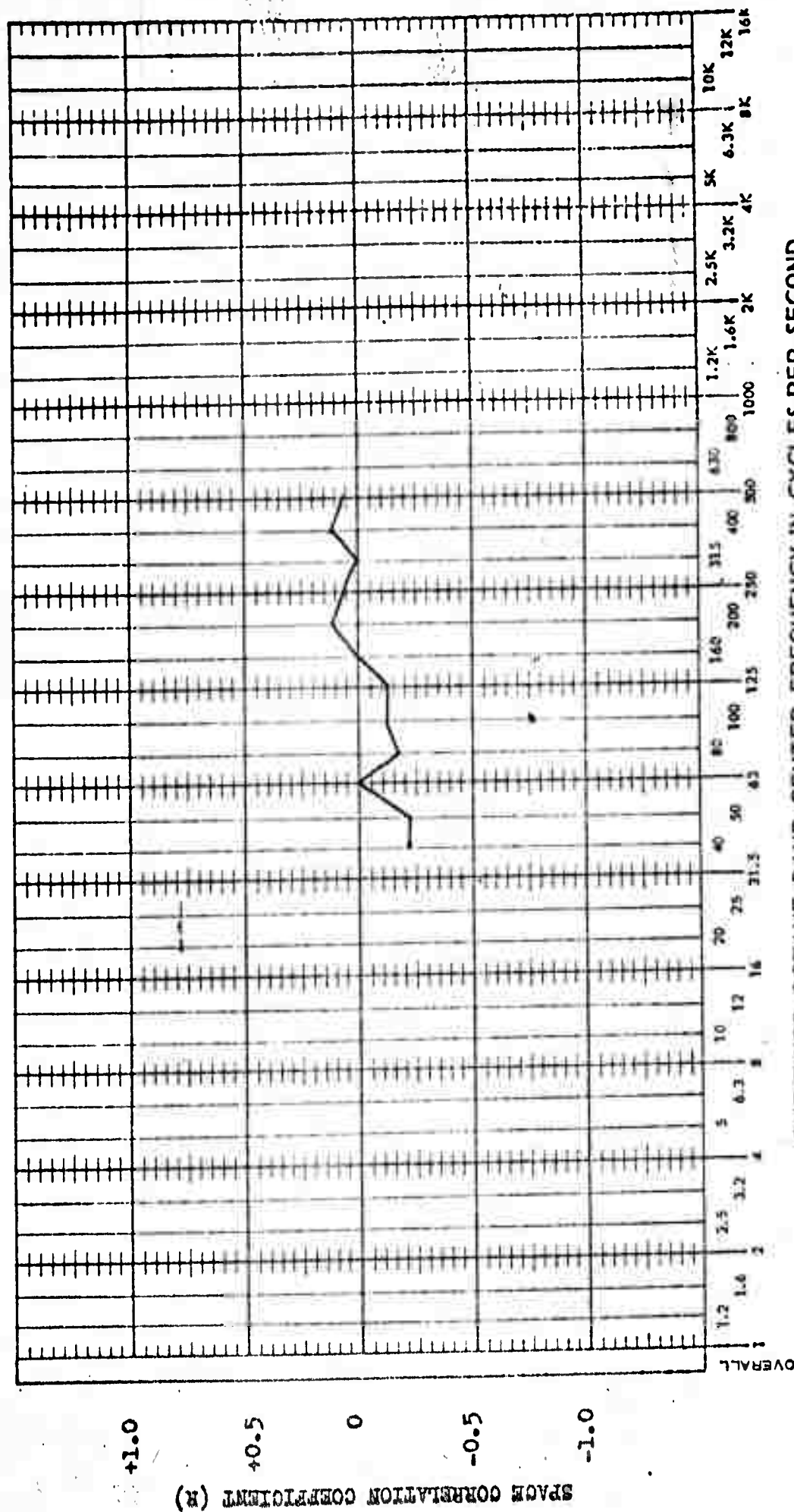
BOEING

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CORRELATION NO. 307  
MACH NO. 0.85  
0.9  
0

REFERENCE TRANSDUCER 121, TEST POINT 21  
TEST TRANSDUCER 117, TEST POINT 16

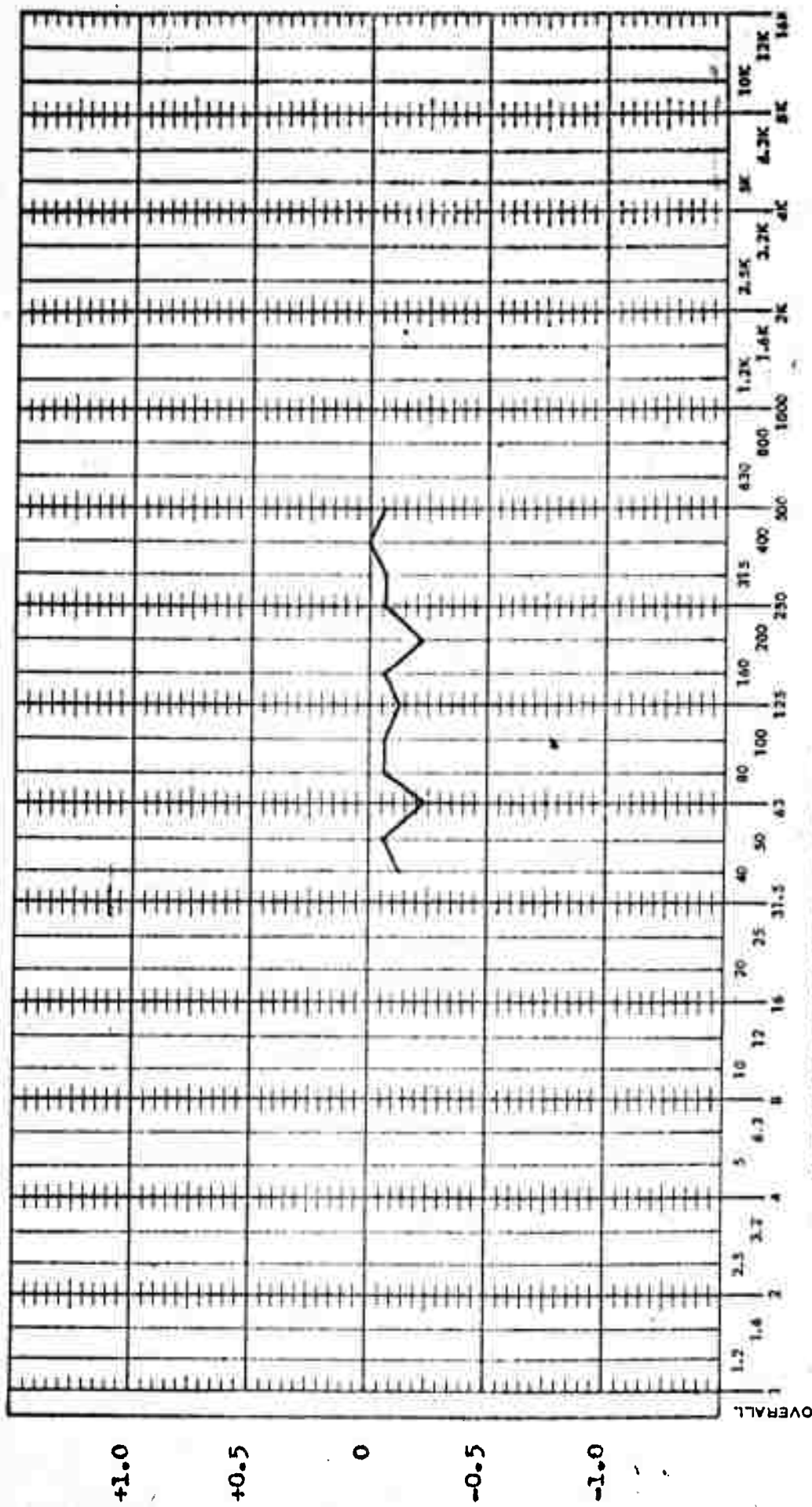
Figure 16. Continued.

BOEING

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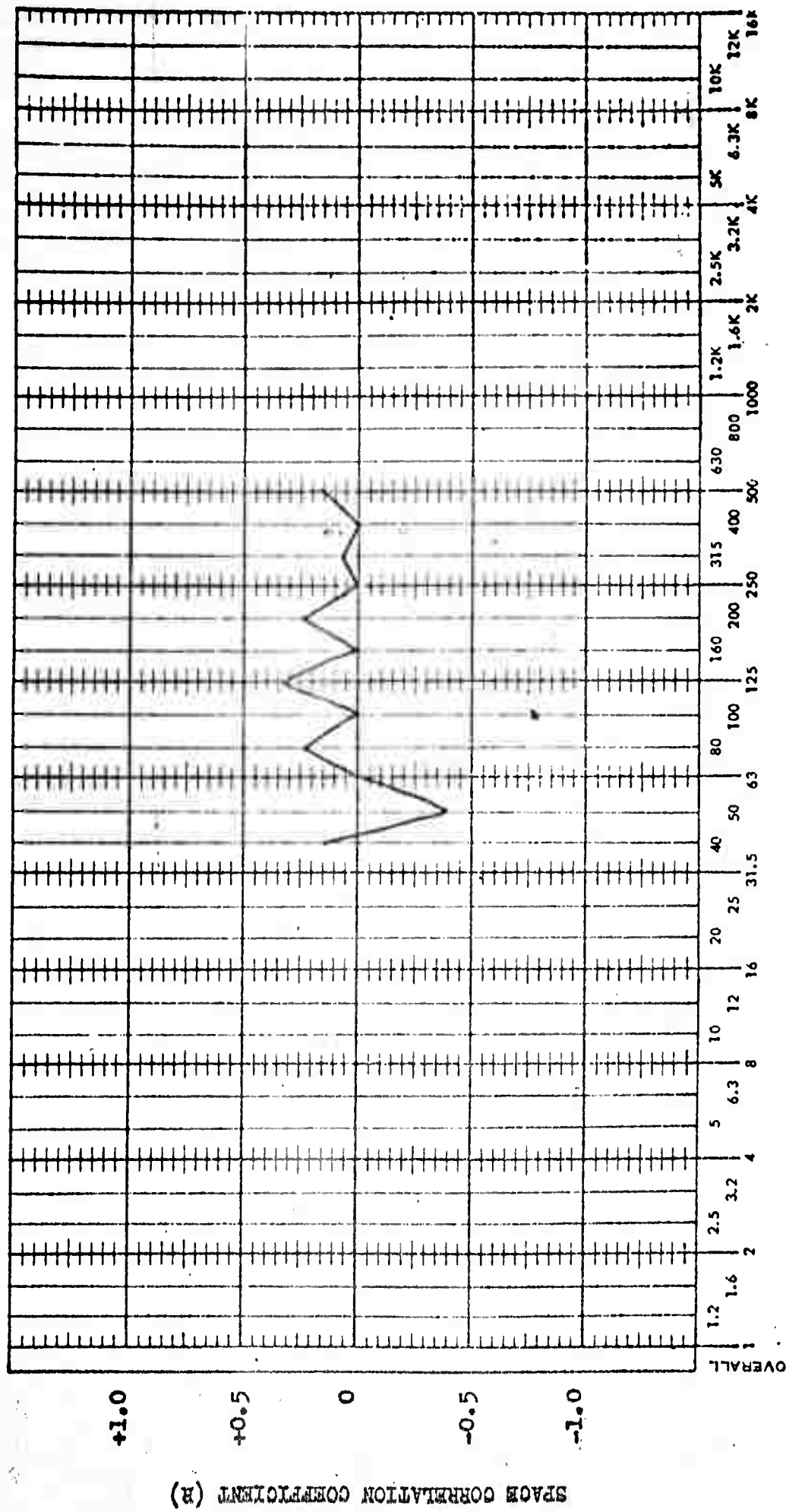


ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

REFERENCE TRANSDUCER 421, TEST POINT 24  
 TEST TRANSDUCER 417, TEST POINT 16  
 CORRELATION NO. 357  
 MACH NO. 1.08  
83

Figure 16. Continued.

270



REFERENCE TRANSDUCER 421, TEST POINT 14  
 TEST TRANSDUCER, 417, TEST POINT 16

CORRELATION NO. 407  
 MACH NO. 1.08

Figure 16. Continued.

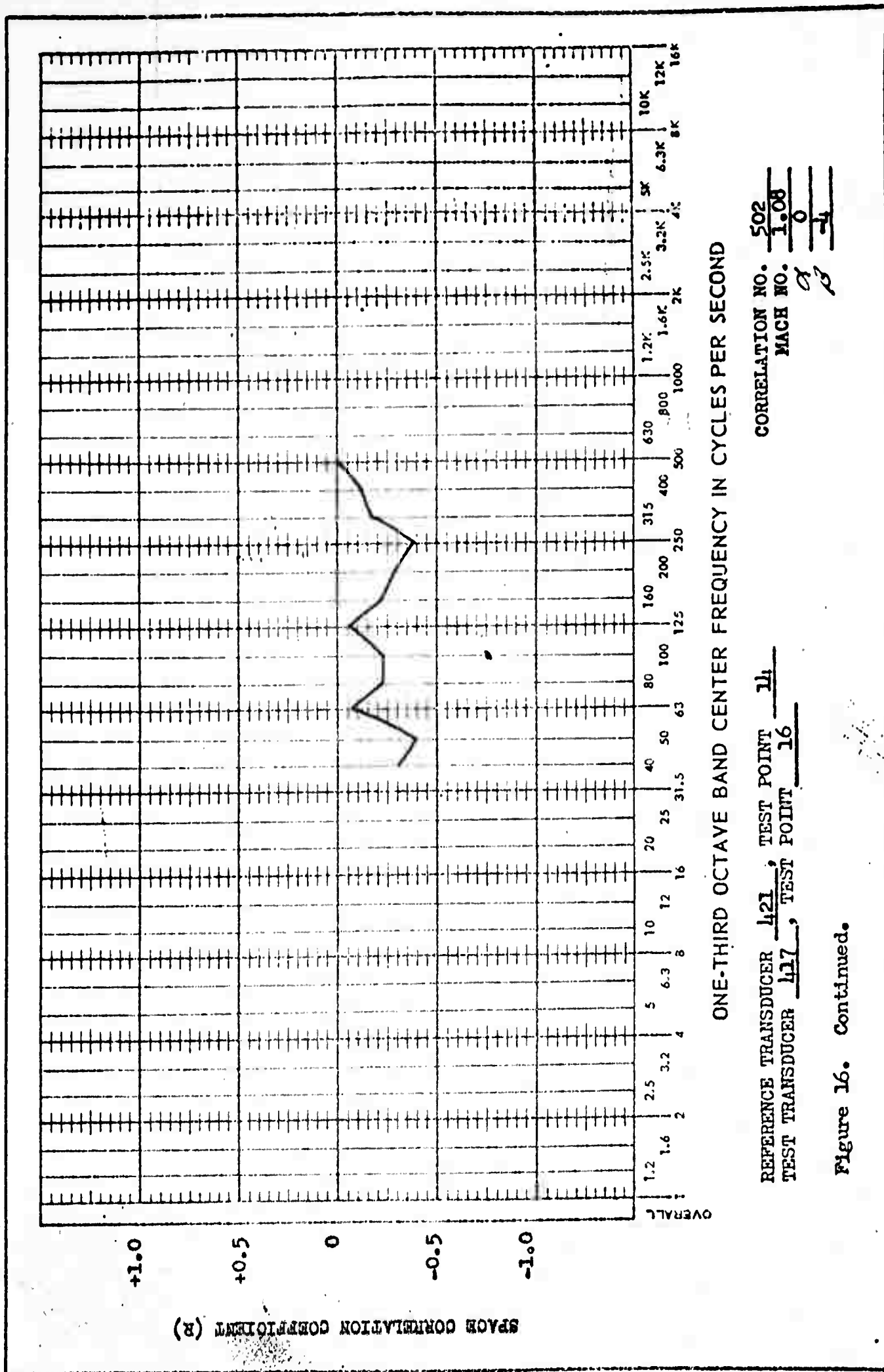
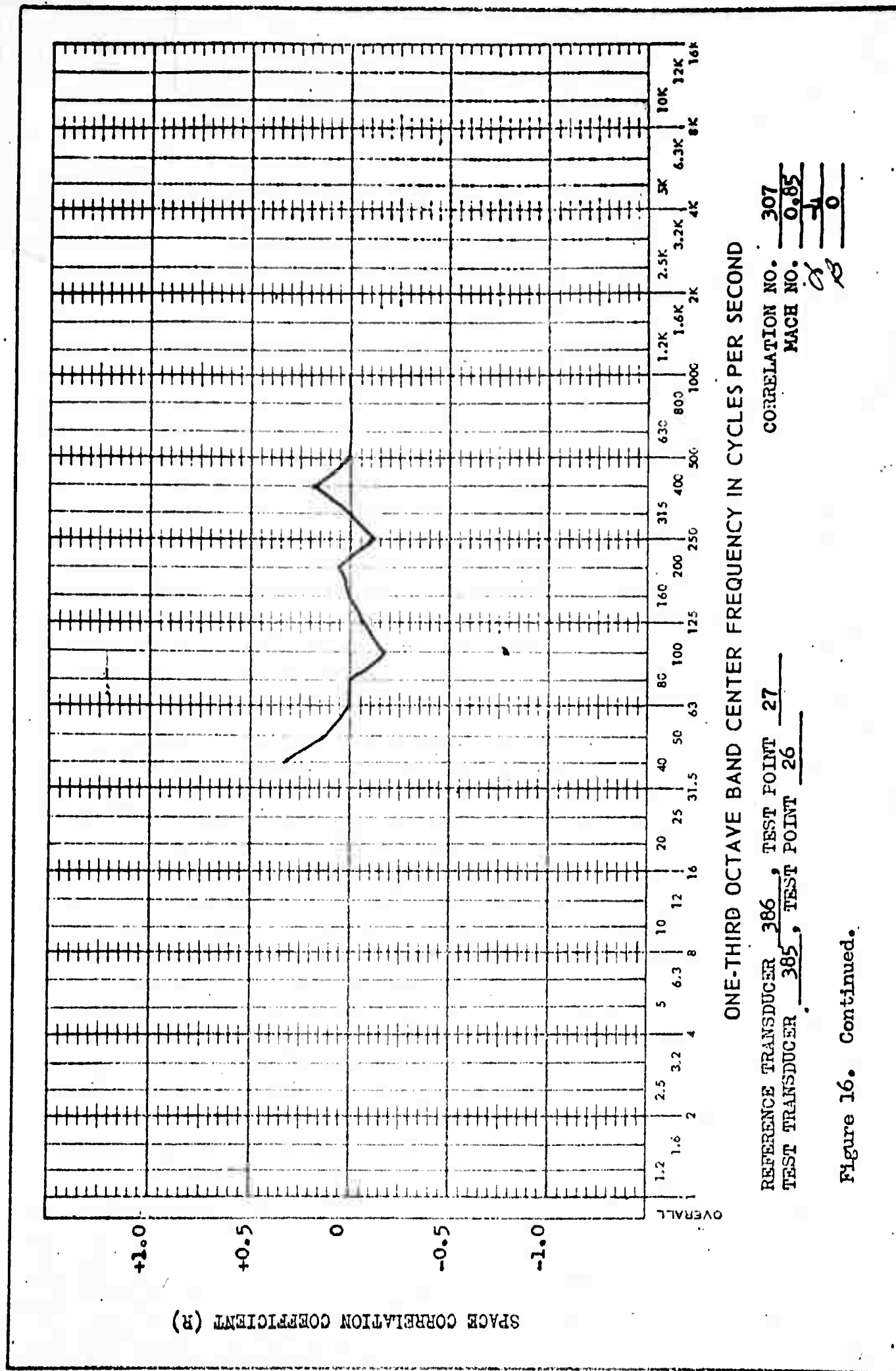


Figure 16. Continued.







REFERENCE TRANSDUCER 386, TEST POINT 27  
 TEST TRANSDUCER 385, TEST POINT 26

CORRELATION NO. 307  
 MACH NO. 0.85

0.85  
 0.85  
 0

BOEING

NO. T2-2618

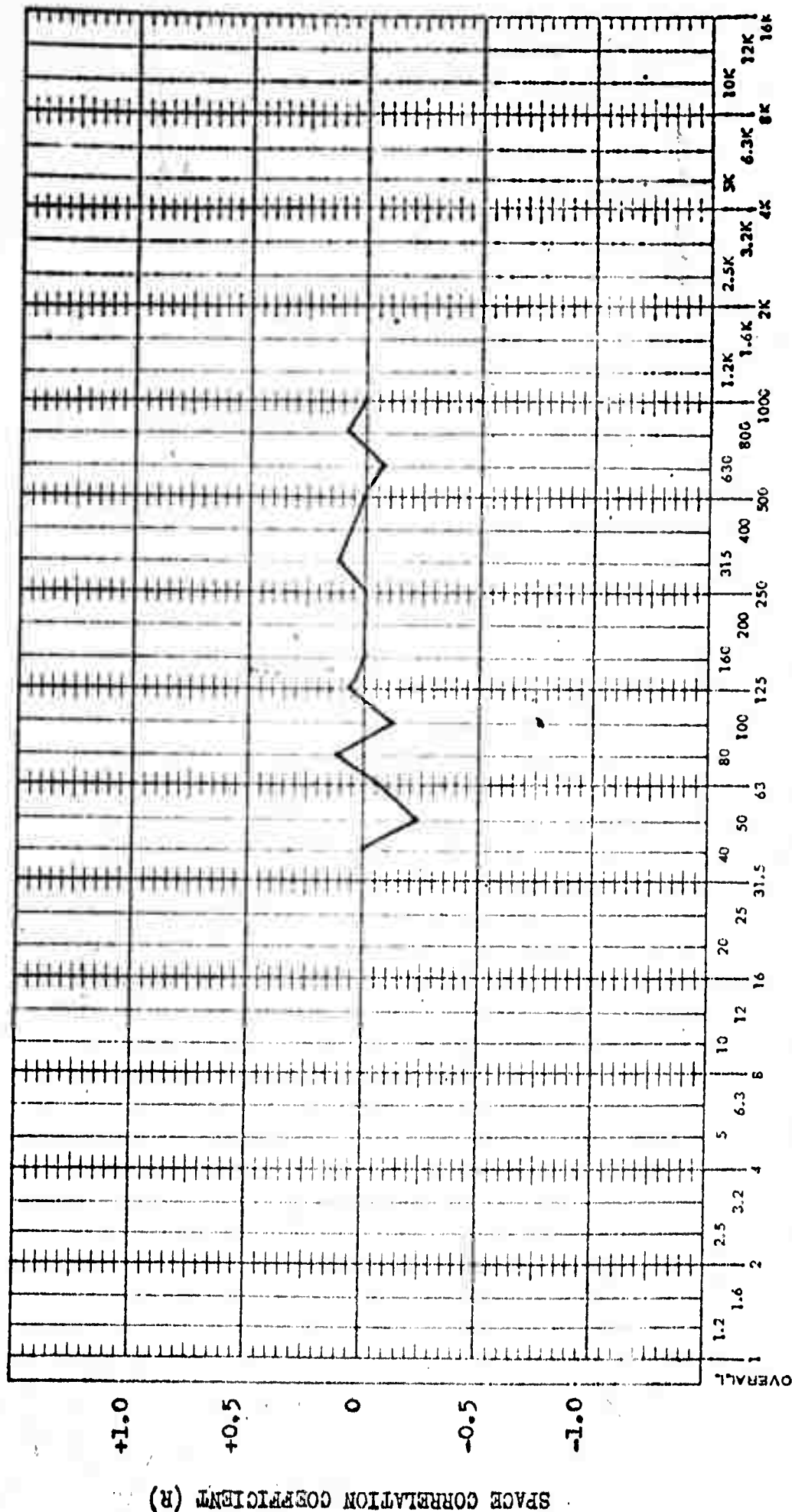
PAGE

270



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CORRELATION NO. 360  
MACH NO. 0.9  
2/0

REFERENCE TRANSDUCER 386, TEST POINT 27  
TEST TRANSDUCER 385, TEST POINT 26

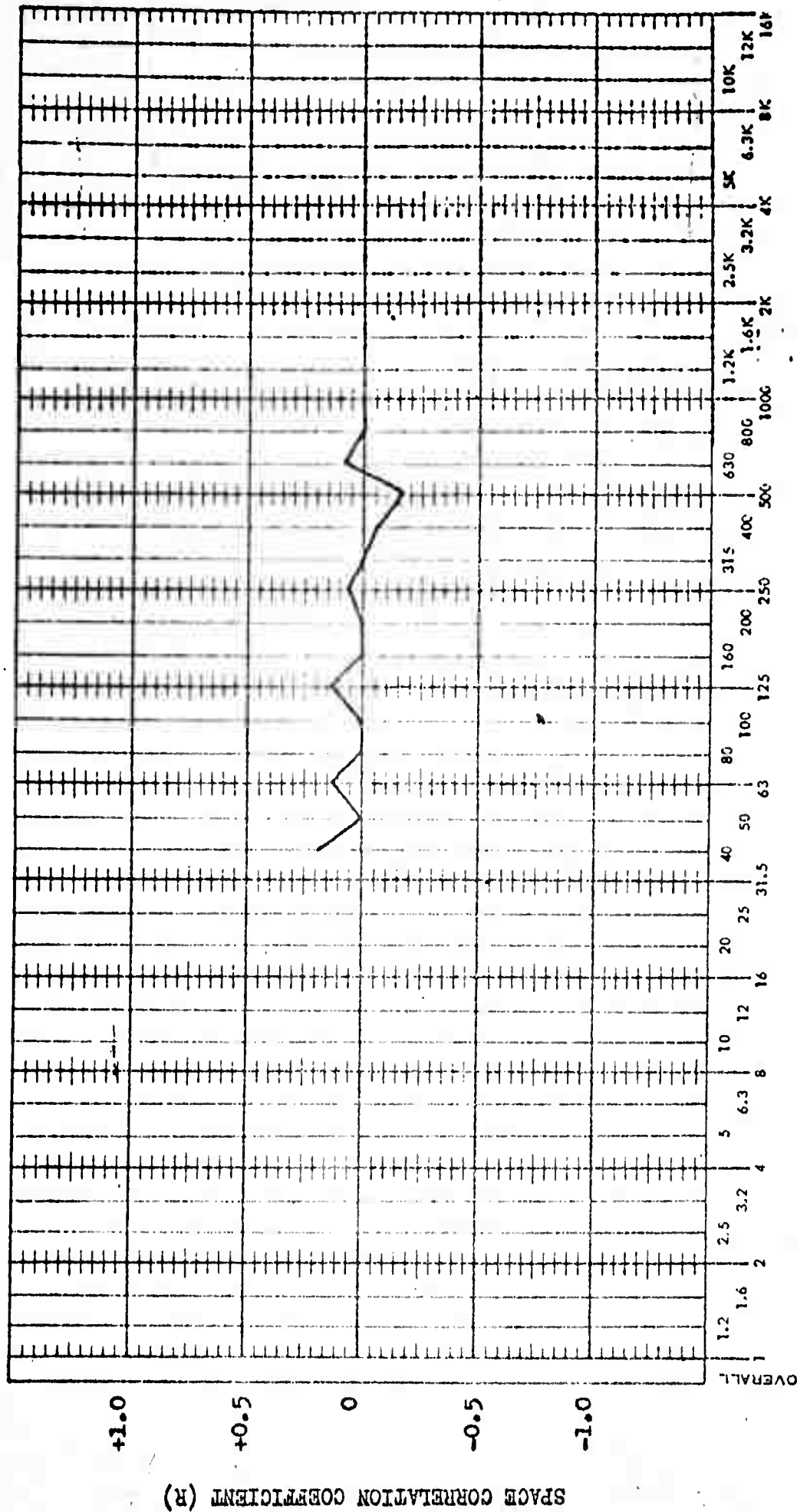
Figure 16. Continued.

BUENING

NO. T2-2618

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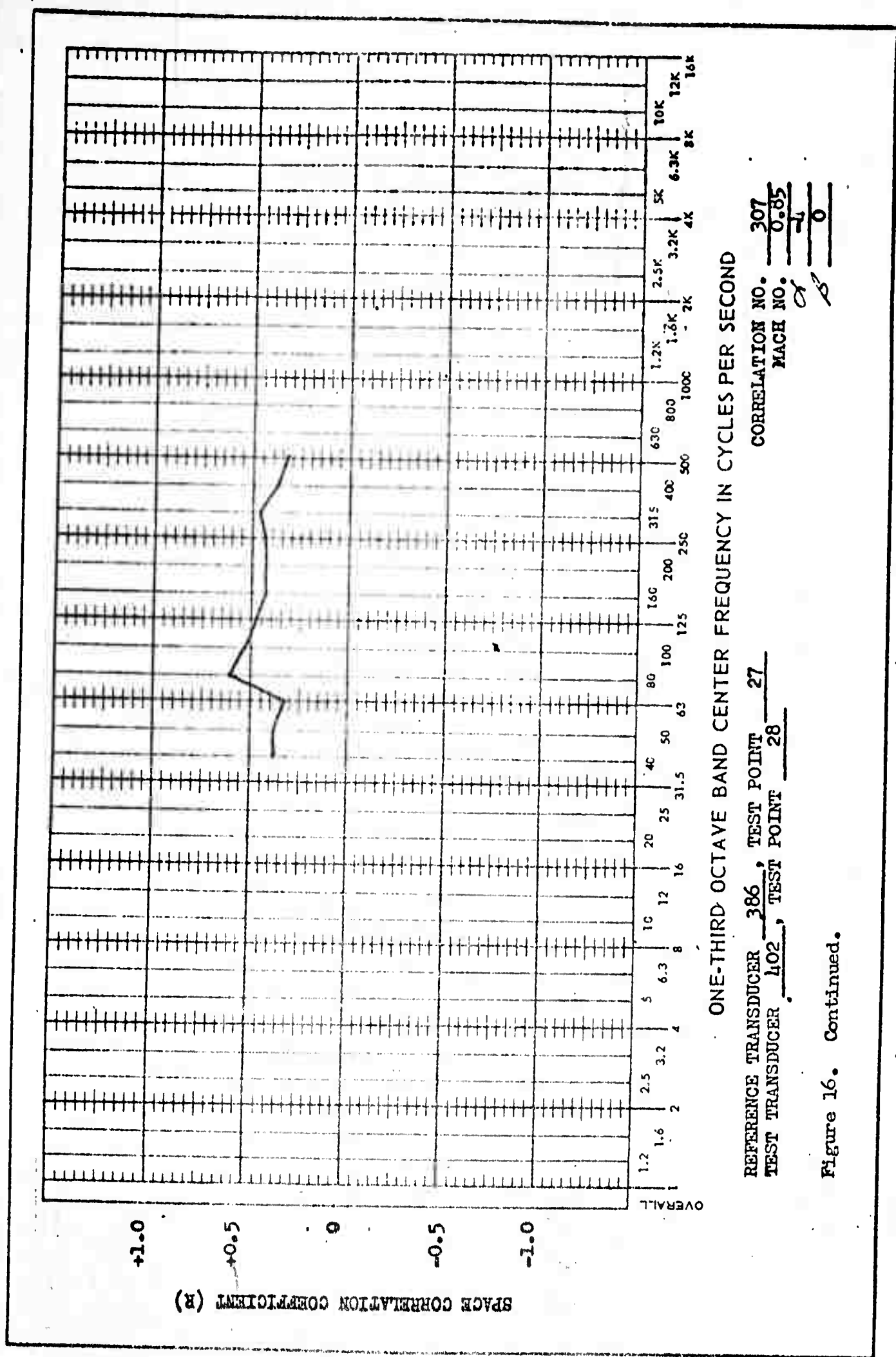


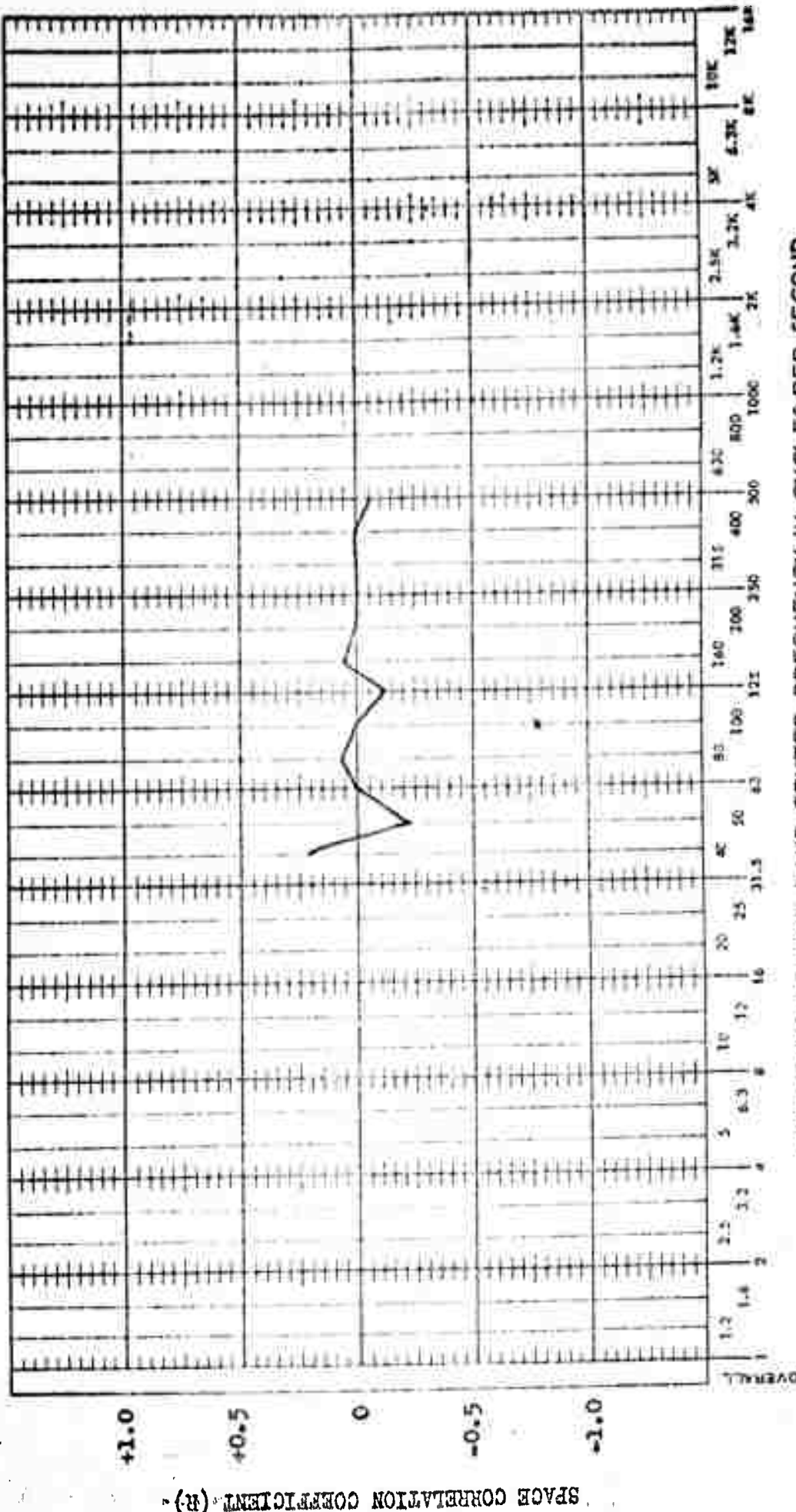
REFERENCE TRANSDUCER 386, TEST POINT 27  
 TEST TRANSDUCER 385, TEST POINT 26

CORRELATION NO. 397  
 MACH NO. 0.8  
2  
3

Figure 16. Continued.

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REFERENCE TRANSDUCER 386 TEST POINT 27  
TEST TRANSDUCER 345 TEST POINT 29

CORRELATION NO. 307  
MACH NO. 0.85

Figure 16. Continued.



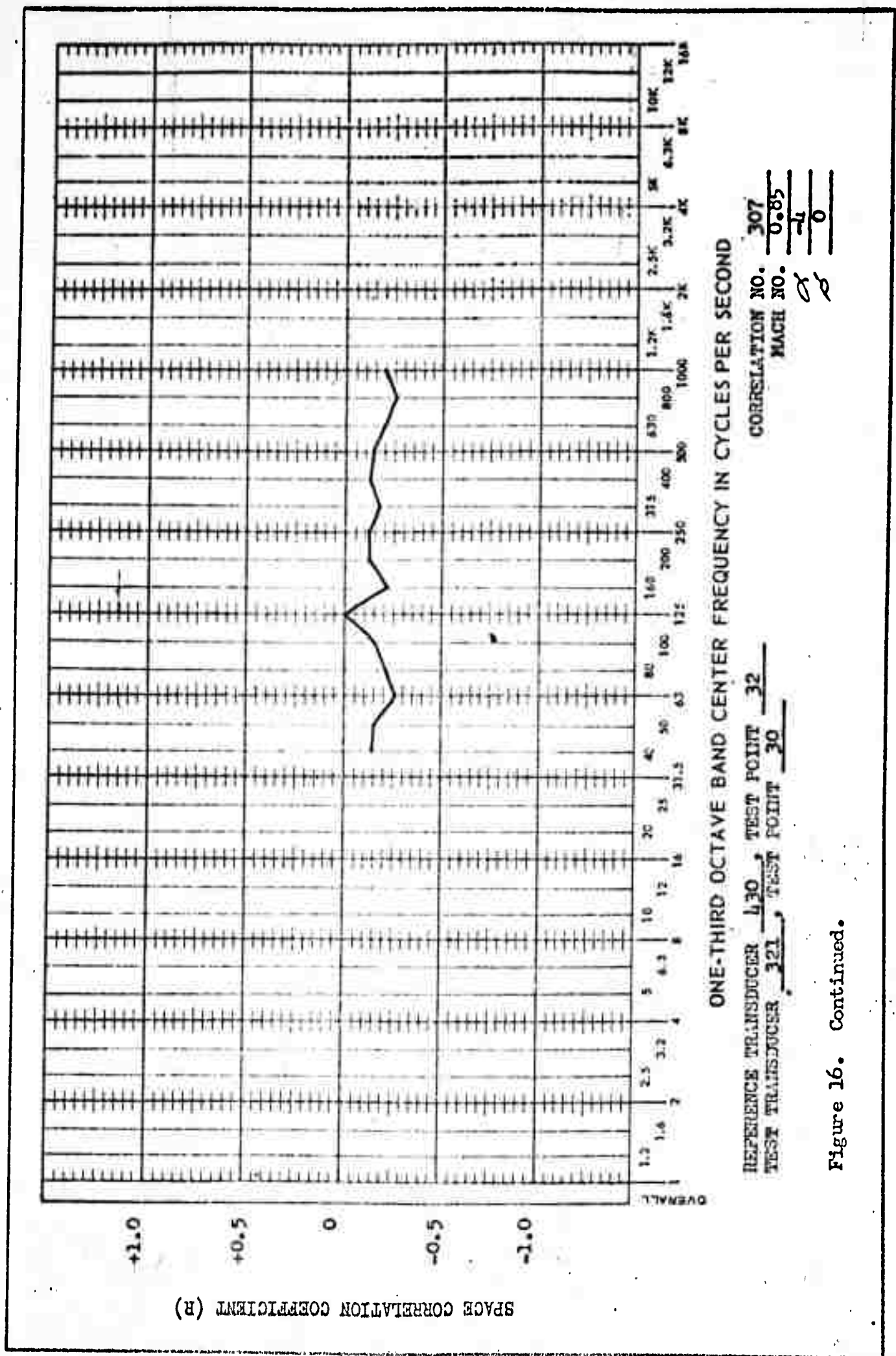
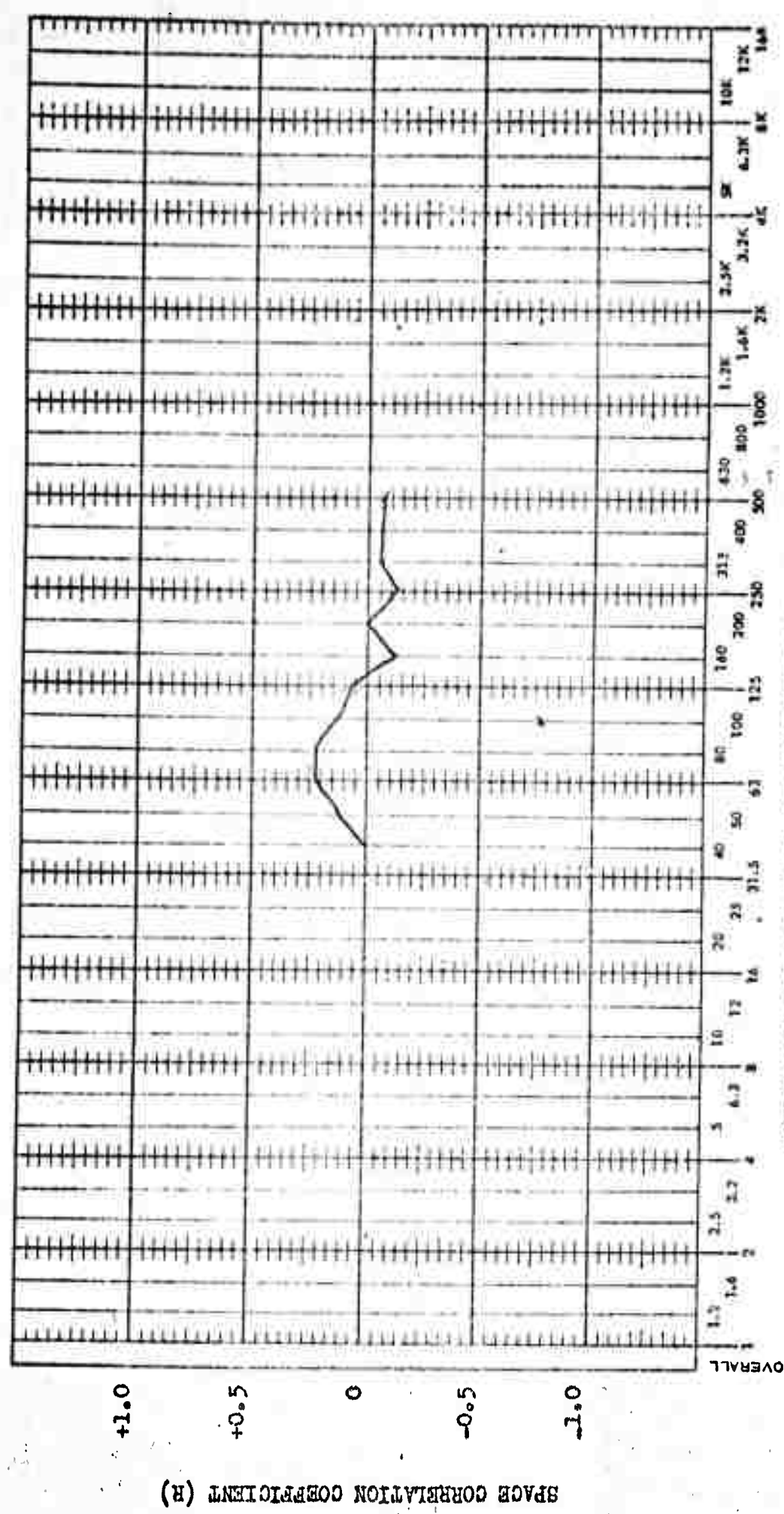


Figure 16. Continued.



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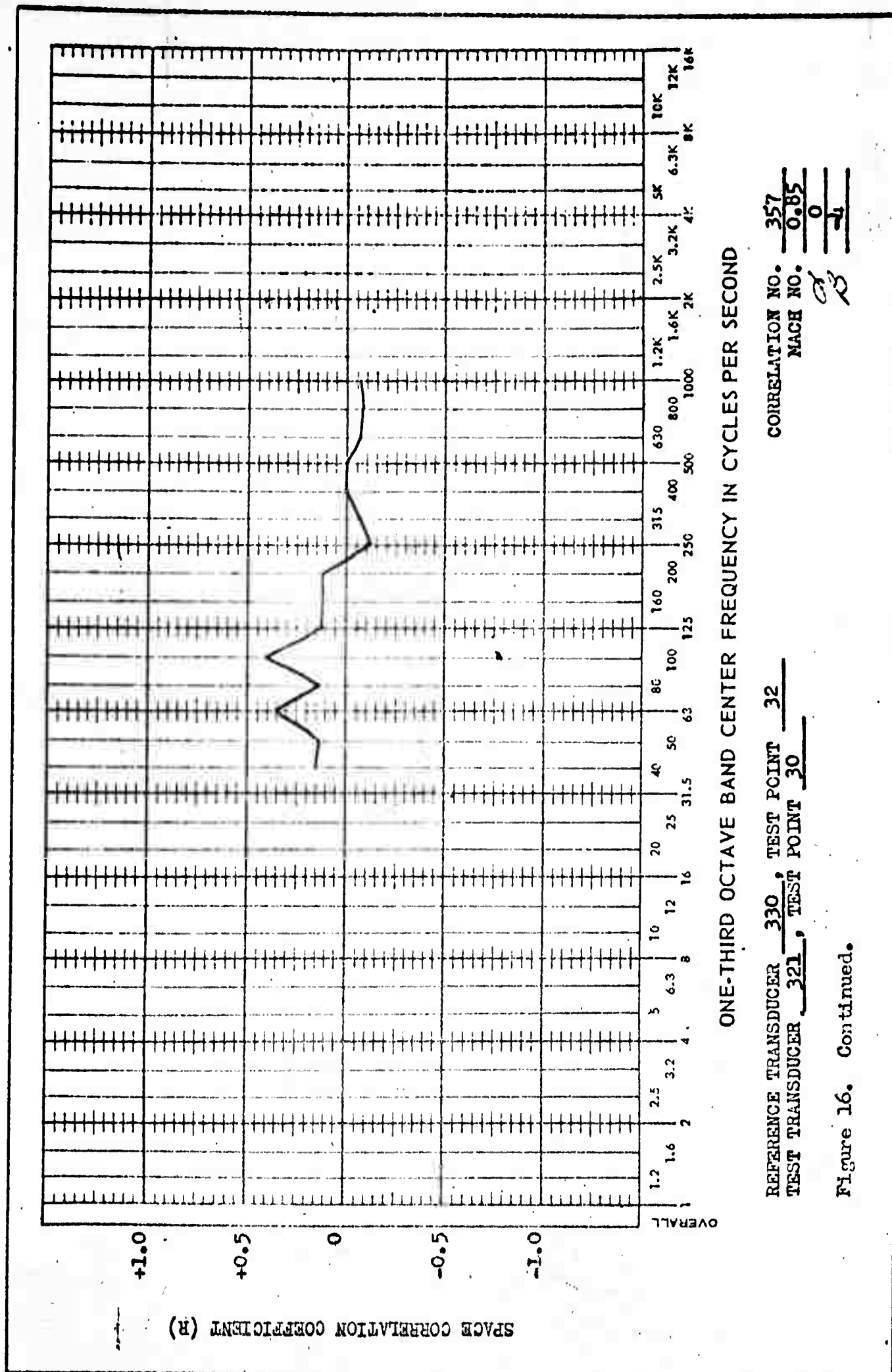
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

REFERENCE TRANSDUCER 430 TEST POINT 32  
 TEST TRANSDUCER 338 TEST POINT 21  
 CORRELATION NO. 307  
 MACH NO. 0.85  
20

Figure 16. Continued.



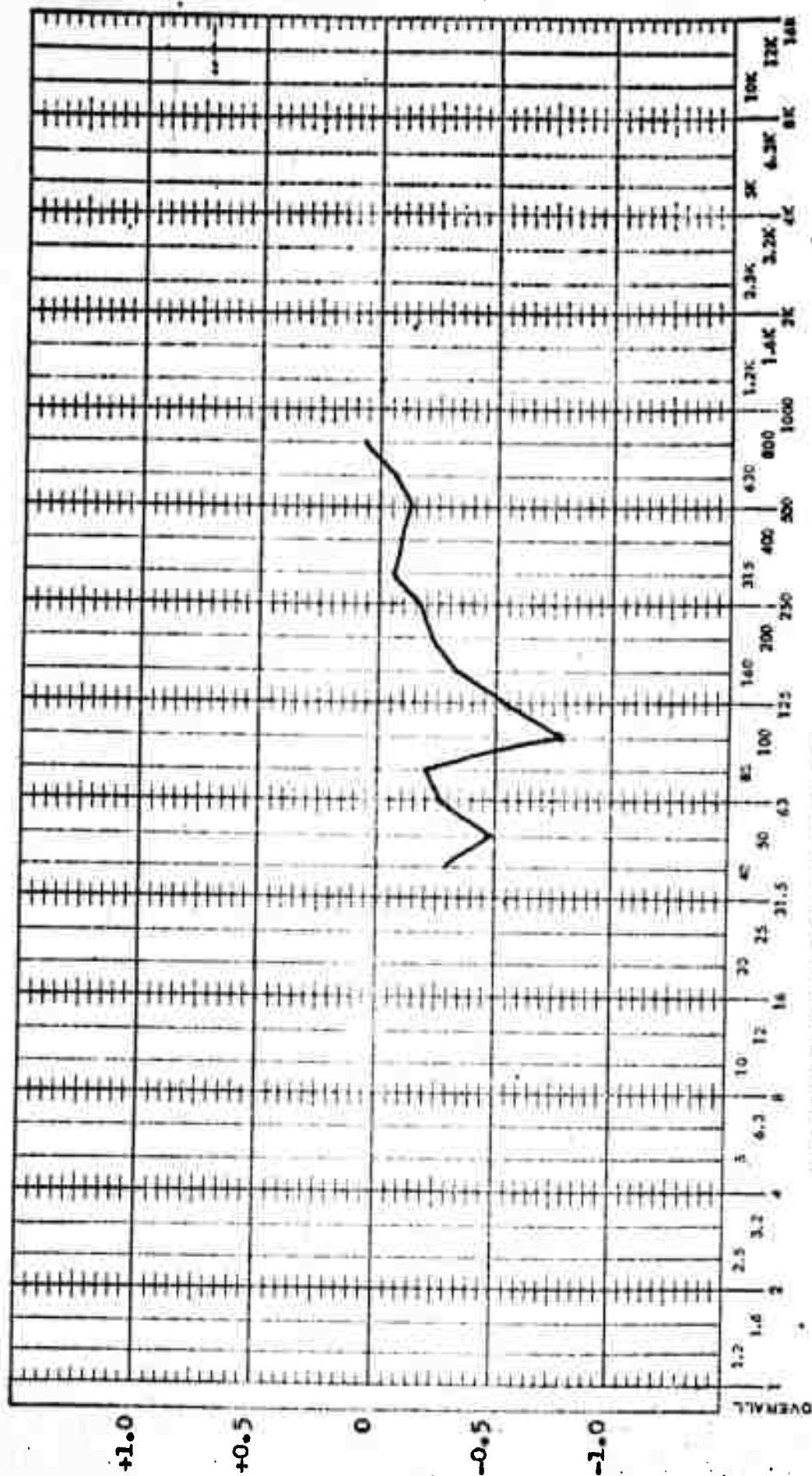
280



CORRELATION NO. 357  
MACH NO. 0.85  
0  
4

REFERENCE TRANSDUCER 330, TEST POINT 32  
TEST TRANSDUCER 321, TEST POINT 30

Figure 16. Continued.



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

REFERENCE TRANSDUCER 105, TEST POINT 32  
TEST TRANSDUCER 321, TEST POINT 30

CORRELATION NO. 397  
MACH NO. 0.8  
0.8

Figure 16. Continued.

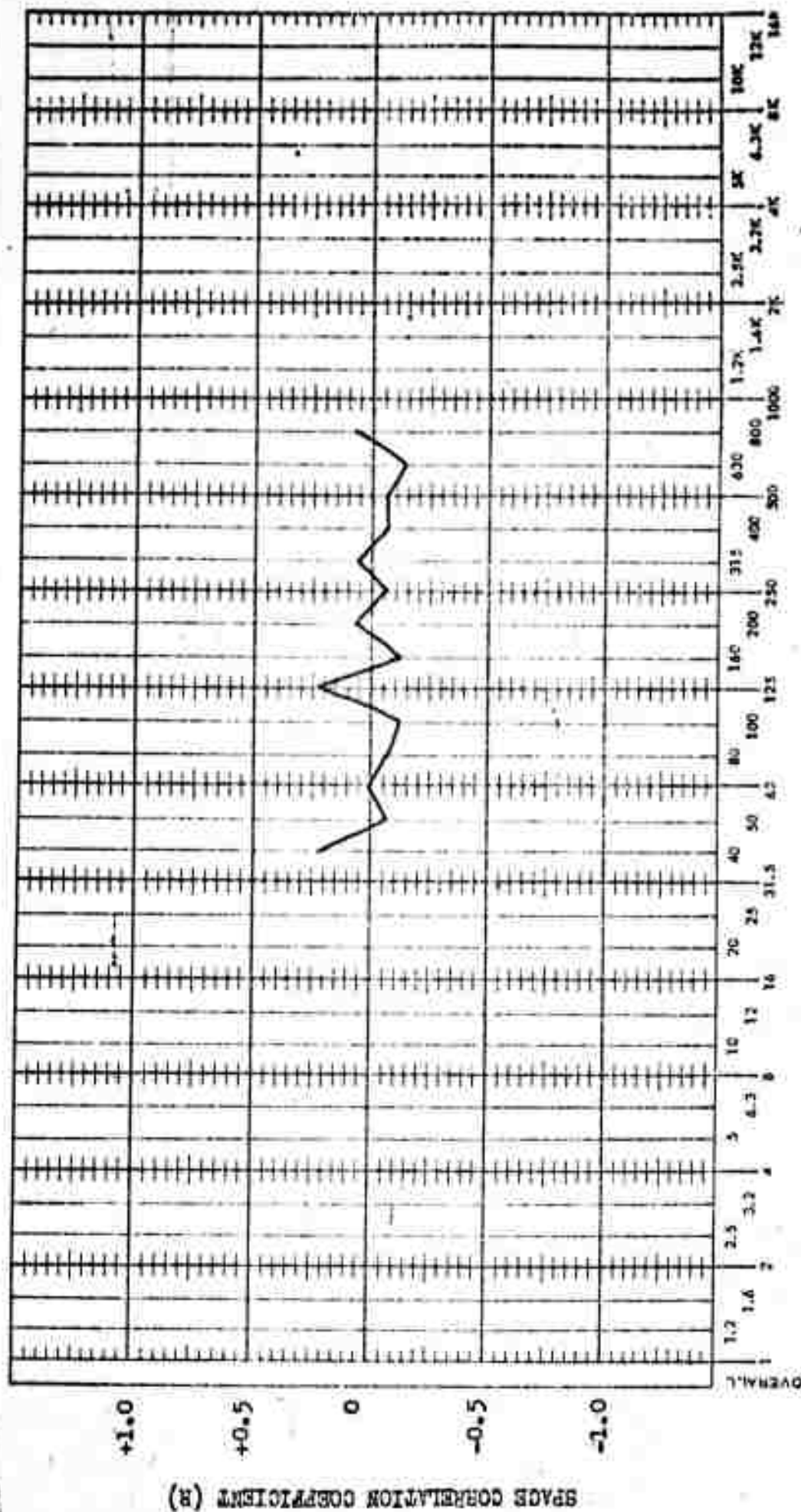
BOEING

NO. T2-2 618

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REFERENCE TRANSDUCER 405, TEST POINT 32  
 TEST TRANSDUCER 321, TEST POINT 30

CORRELATION NO. 415  
 MACH NO. 0.9

Figure 16. Continued.

BOEING

NO. T2-2648  
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BOEING

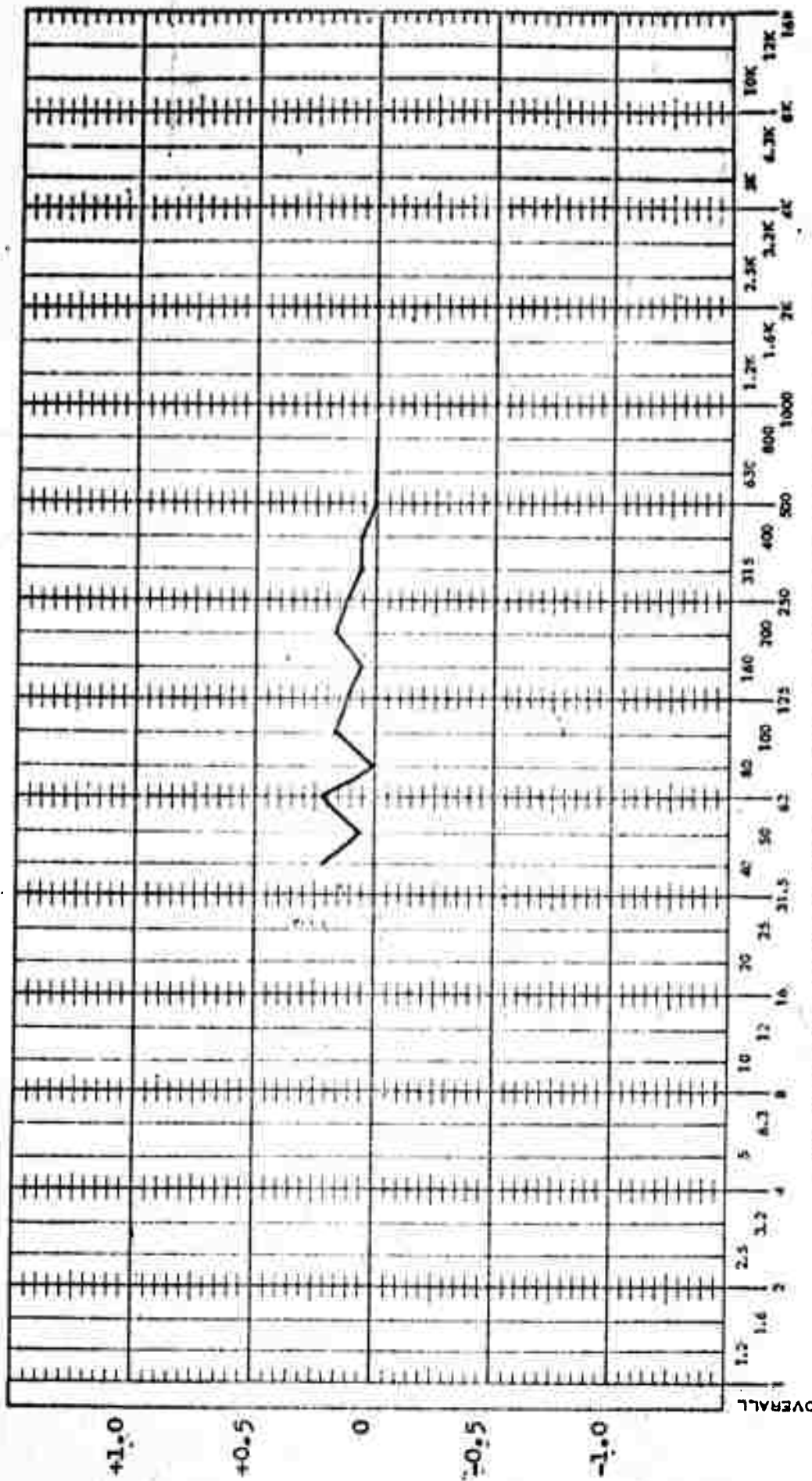
NO. T2-2648

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SPACE CORRELATION COEFFICIENT (R)

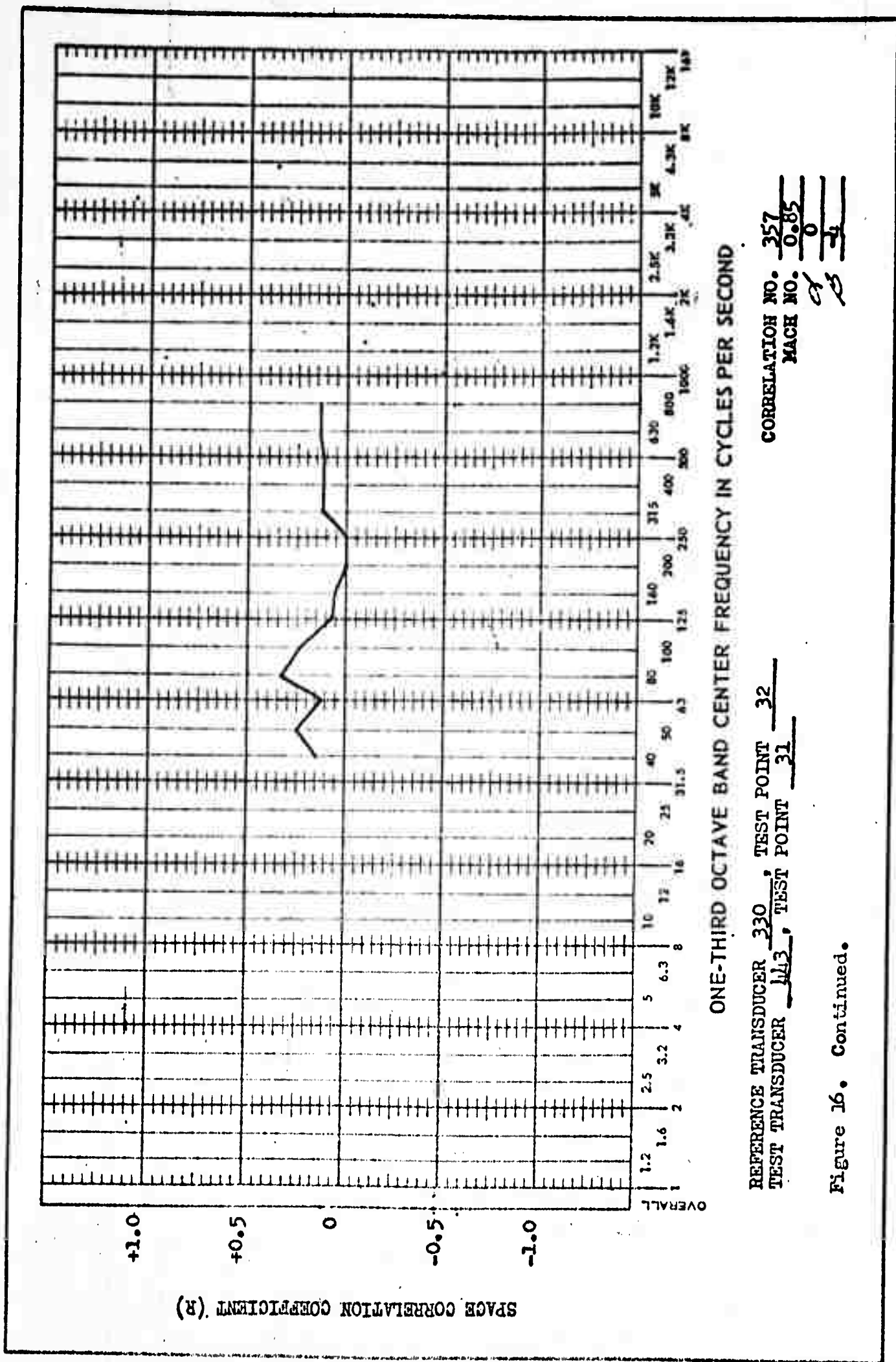


ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

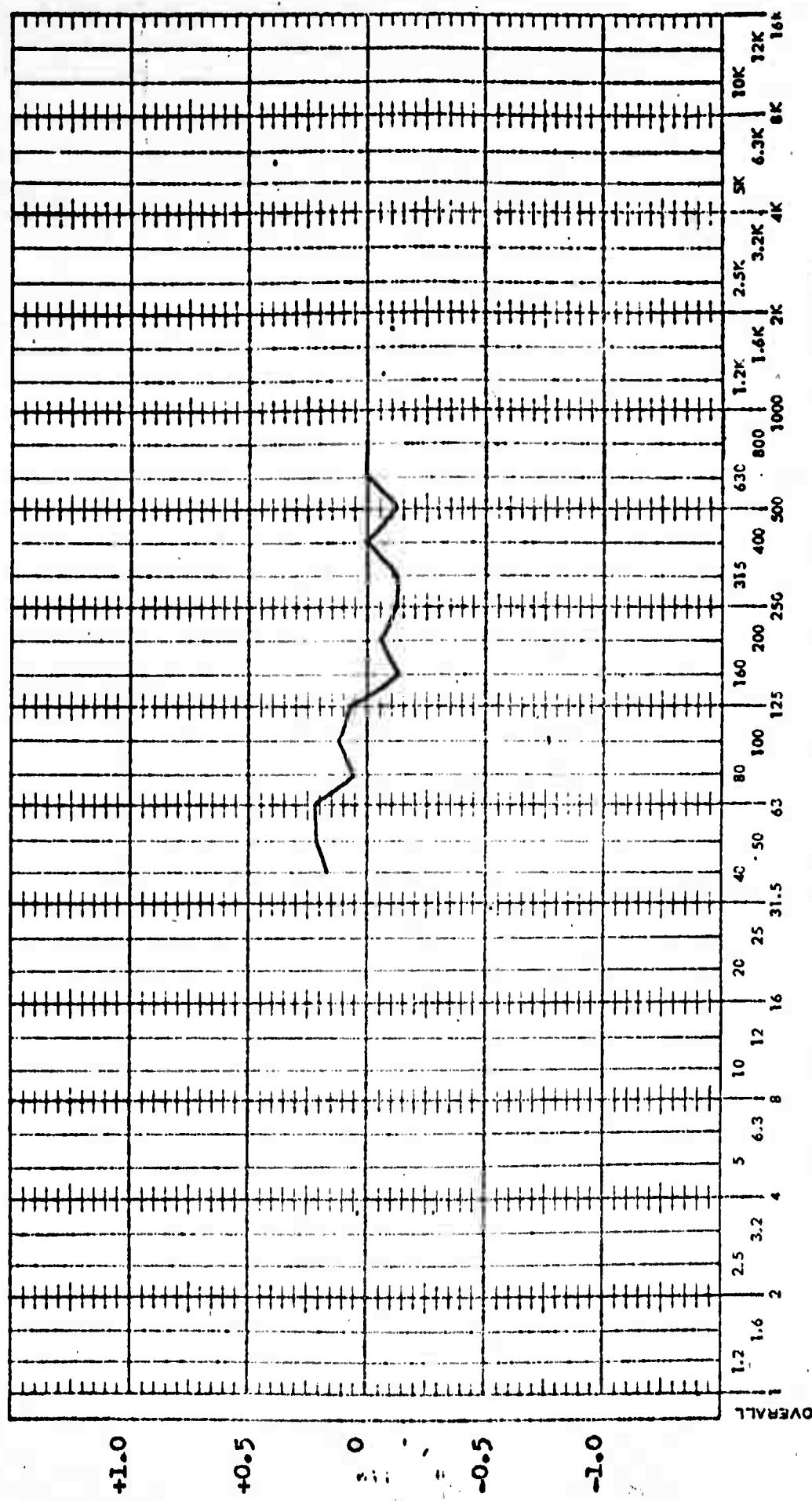
REFERENCE TRANSDUCER L30, TEST POINT 32  
TEST TRANSDUCER L43, TEST POINT 31

CORRELATION NO. 307  
MACH NO. 0.85  
23

Figure 16. Continued.



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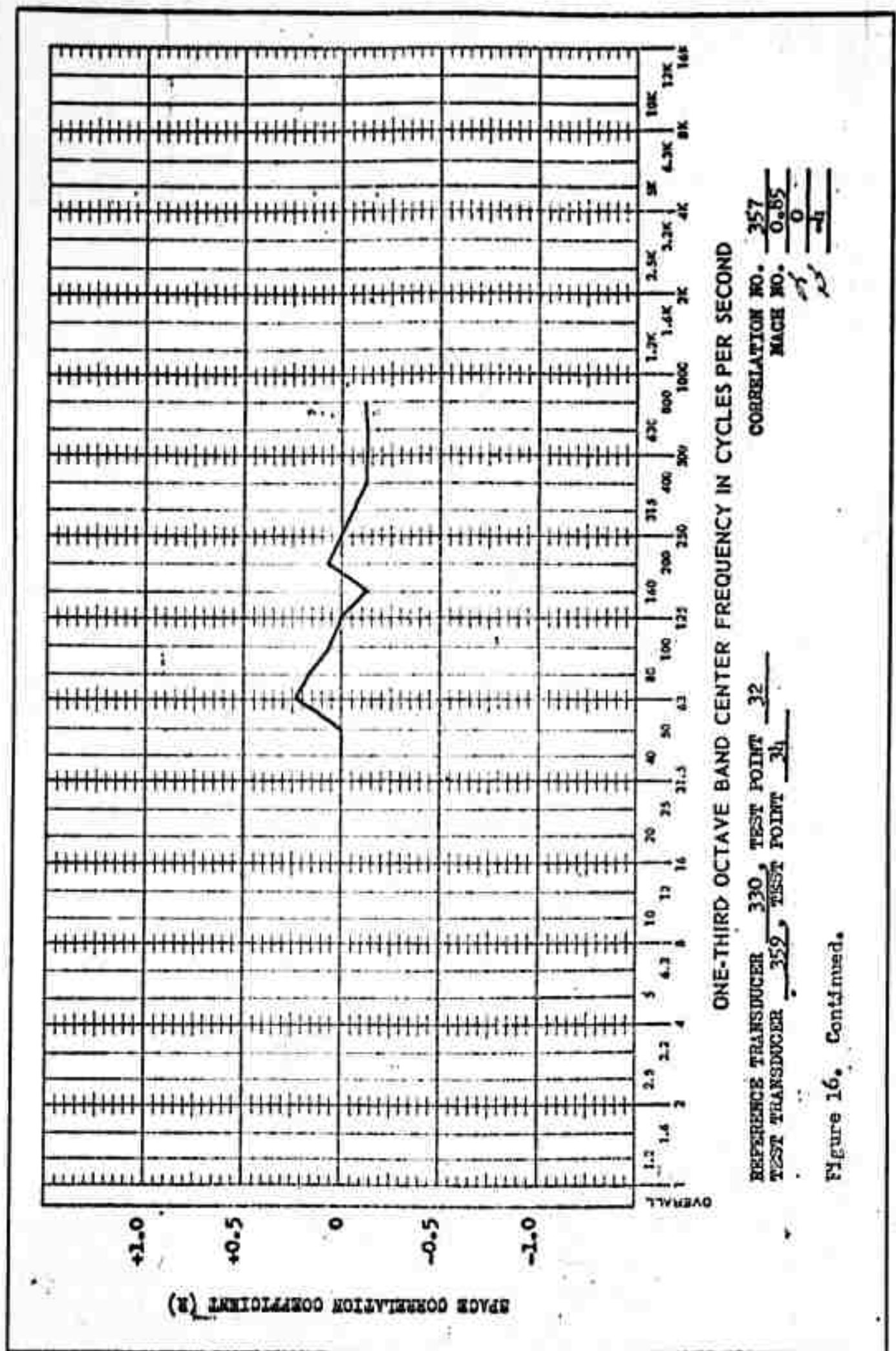
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

CORRELATION NO. 307  
 MACH NO. 0.85  
4  
0

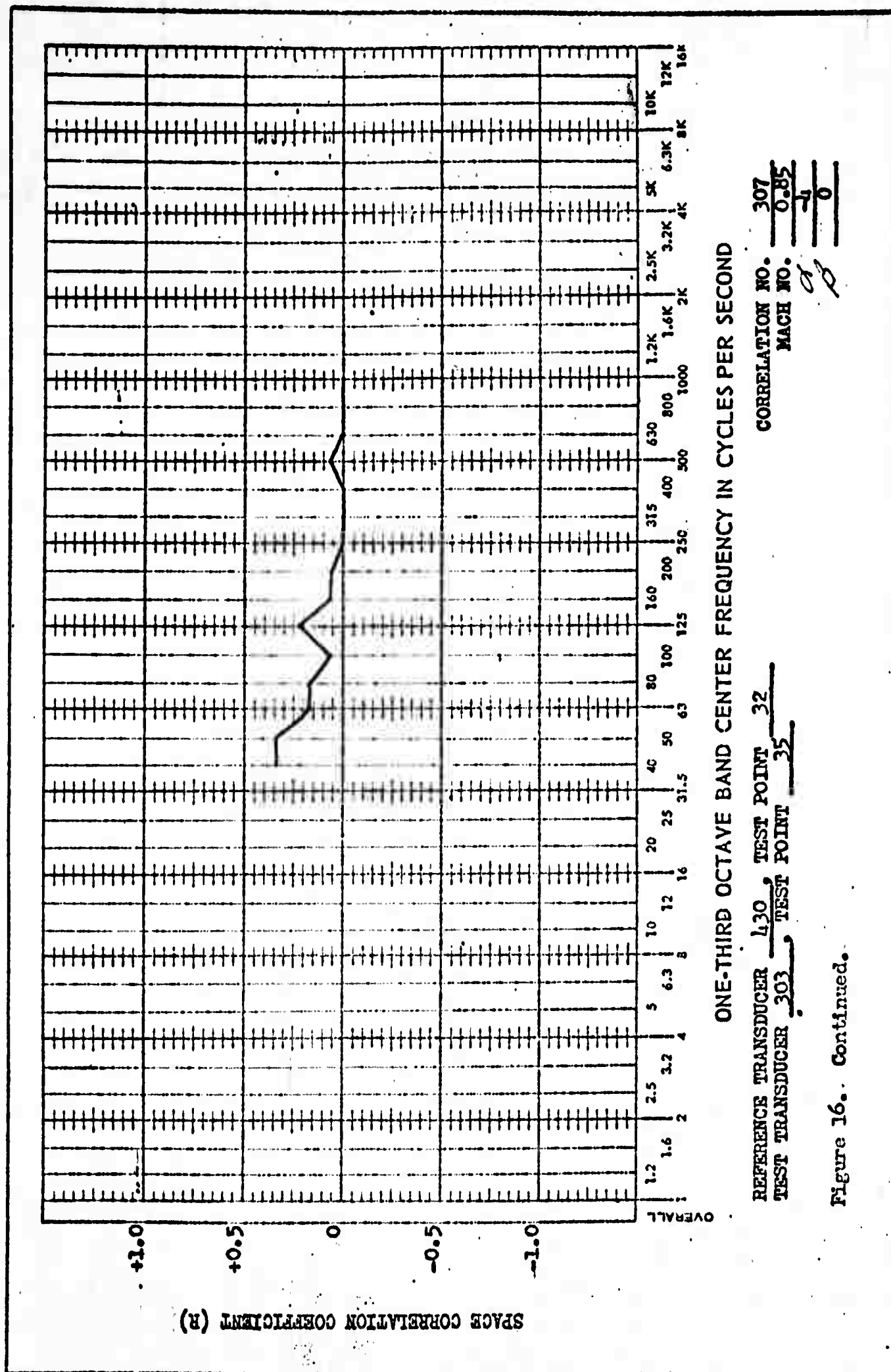
REFERENCE TRANSDUCER 430, TEST POINT 32  
 TEST TRANSDUCER 359, TEST POINT 34

Figure 16. Continued.





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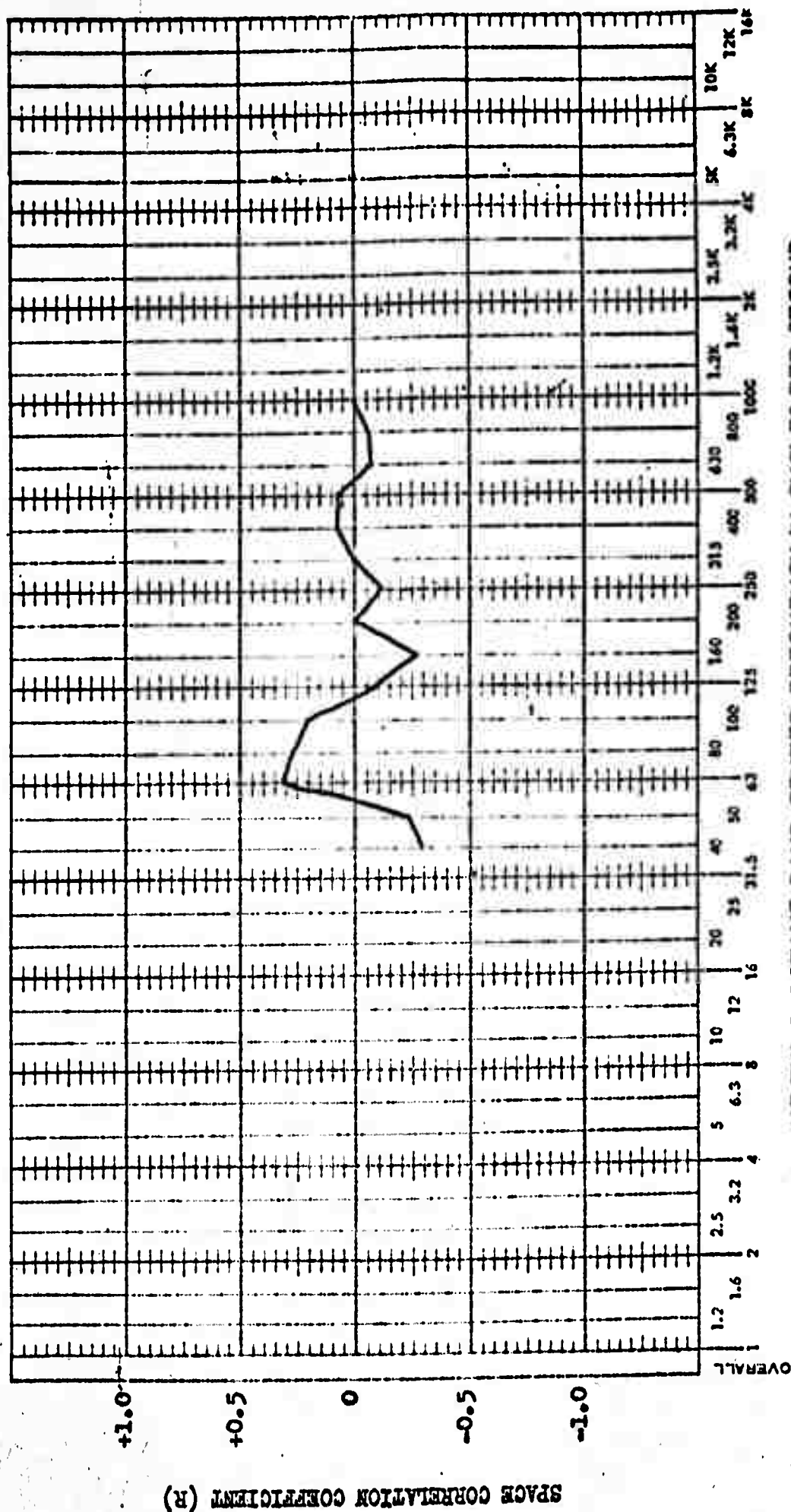
BOEING

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Y





ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

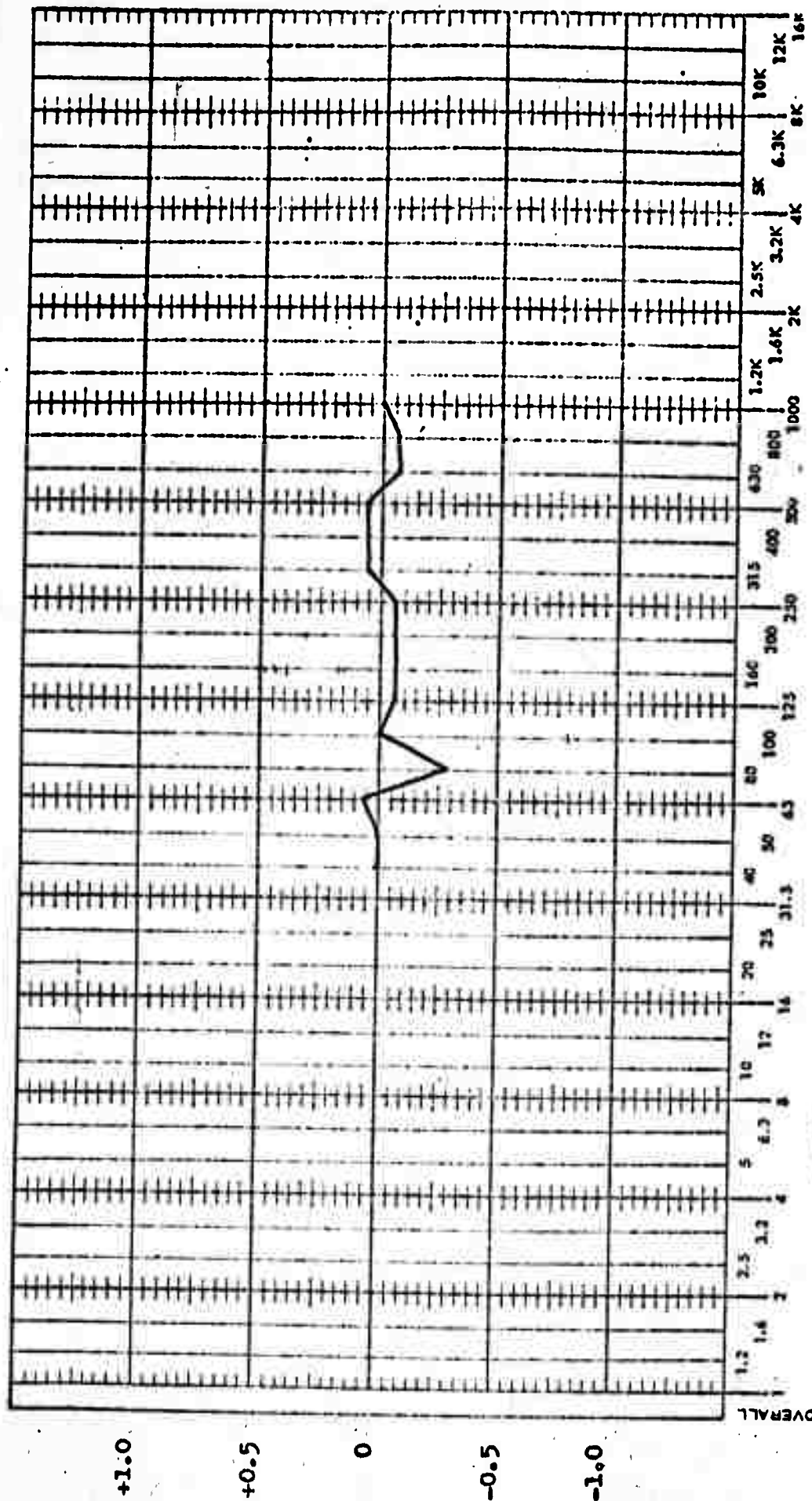
REFERENCE TRANSDUCER 105, TEST POINT 32

TEST TRANSDUCER 303, TEST POINT 35

CORRELATION NO. 397

MACH NO. 0.8

Figure 16. Continued.



ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

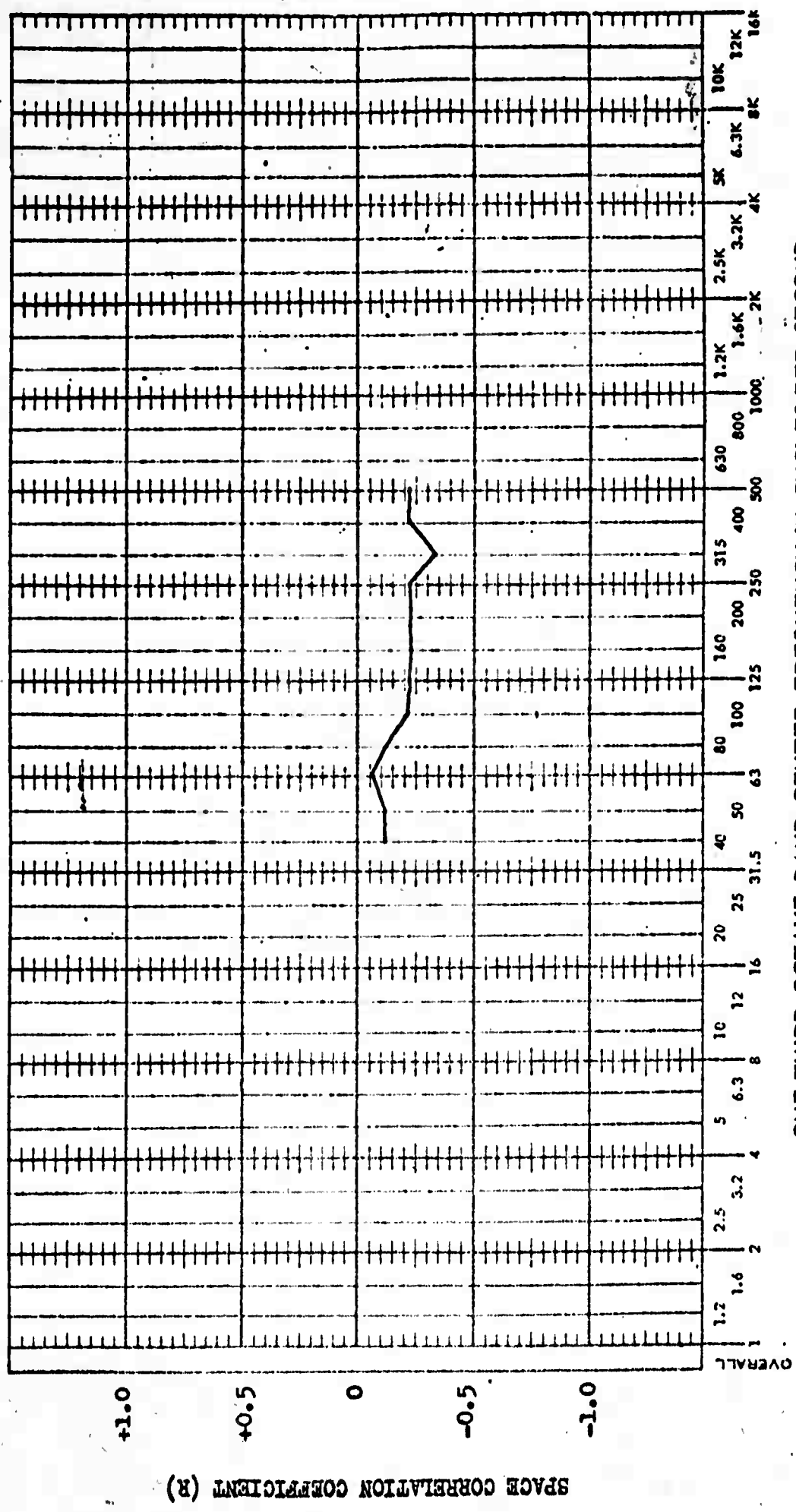
CORRELATION NO. 415  
MACH NO. 0.9

REFERENCE TRANSDUCER 105, TEST POINT 32  
TEST TRANSDUCER 301, TEST POINT 35

Figure 16. Continued.

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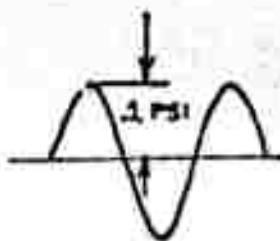
ONE-THIRD OCTAVE BAND CENTER FREQUENCY IN CYCLES PER SECOND

CORRELATION NO. 307  
MACH NO. 0.85

REFERENCE TRANSDUCER 430, TEST POINT 32  
TEST TRANSDUCER 339, TEST POINT 36

Figure 16. Space correlation coefficients (Concluded).

FIG. 17 a



150DB (1000 CPS  
ACOUSTIC  
REFERENCE SIGNAL)

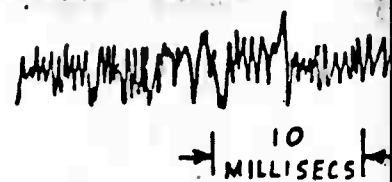
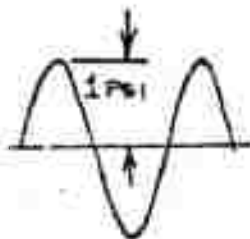


FIG. 17 b



170DB (1000 CPS  
ACOUSTIC  
REFERENCE SIGNAL)

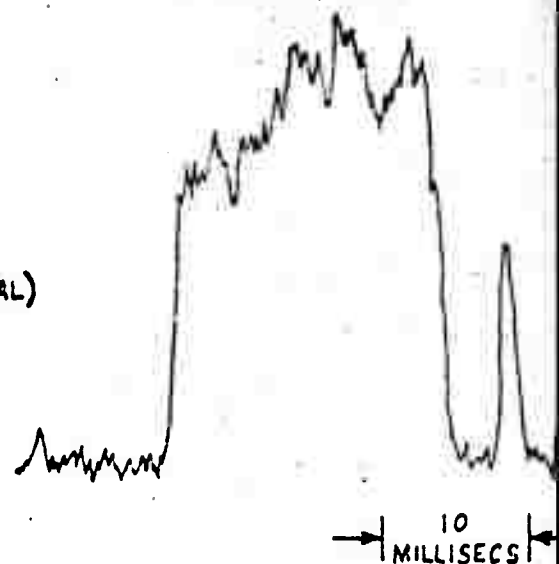
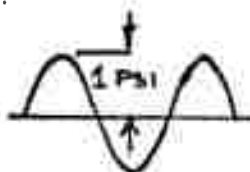
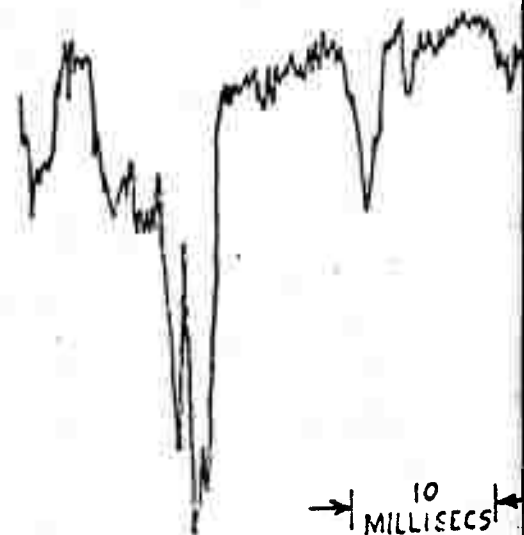


FIG. 17 c



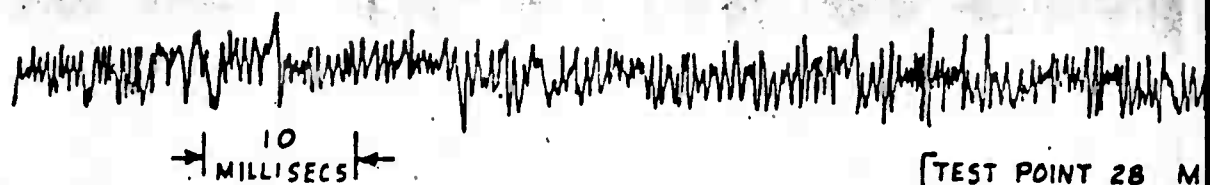
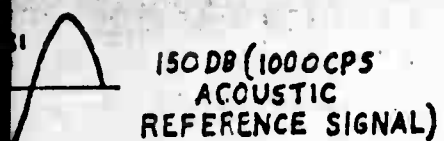
170DB (1000 CPS  
ACOUSTIC  
REFERENCE SIGNAL)



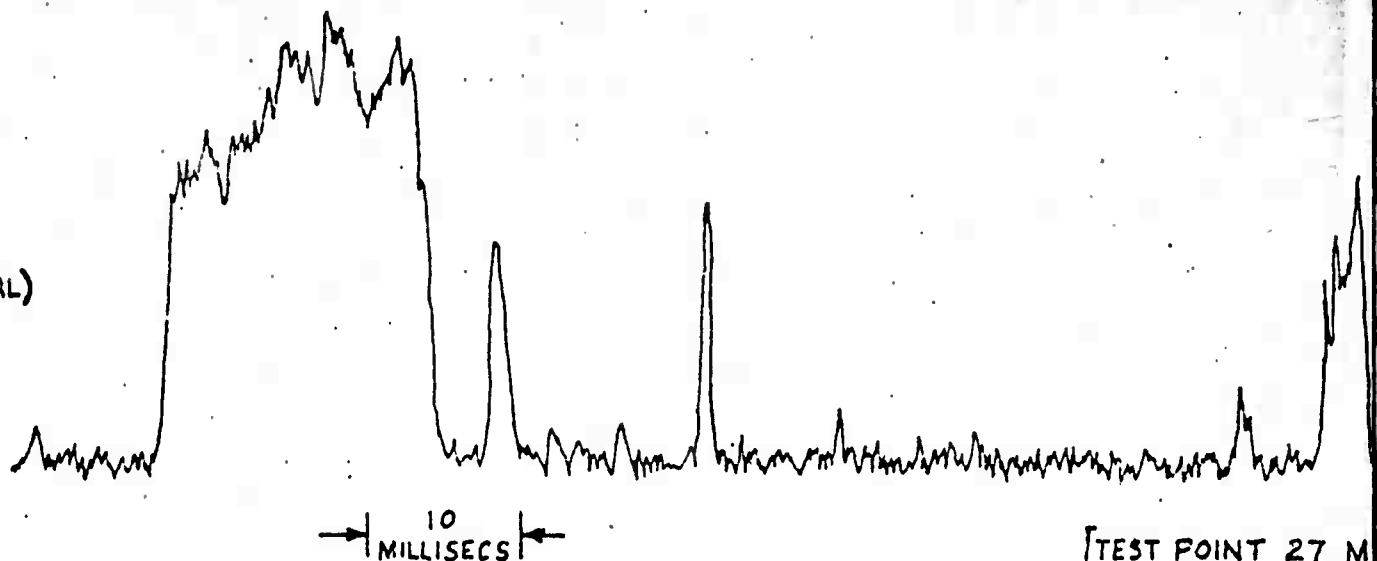
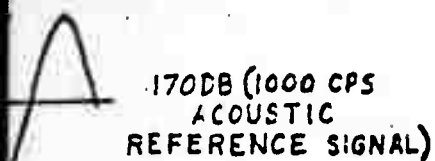
1

TRANSIENT

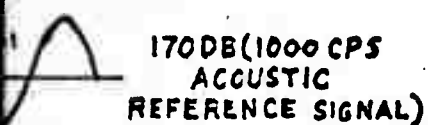
REVISED \_\_\_\_\_



[TEST POINT 28 M



[TEST POINT 27 M



[TEST POINT 32

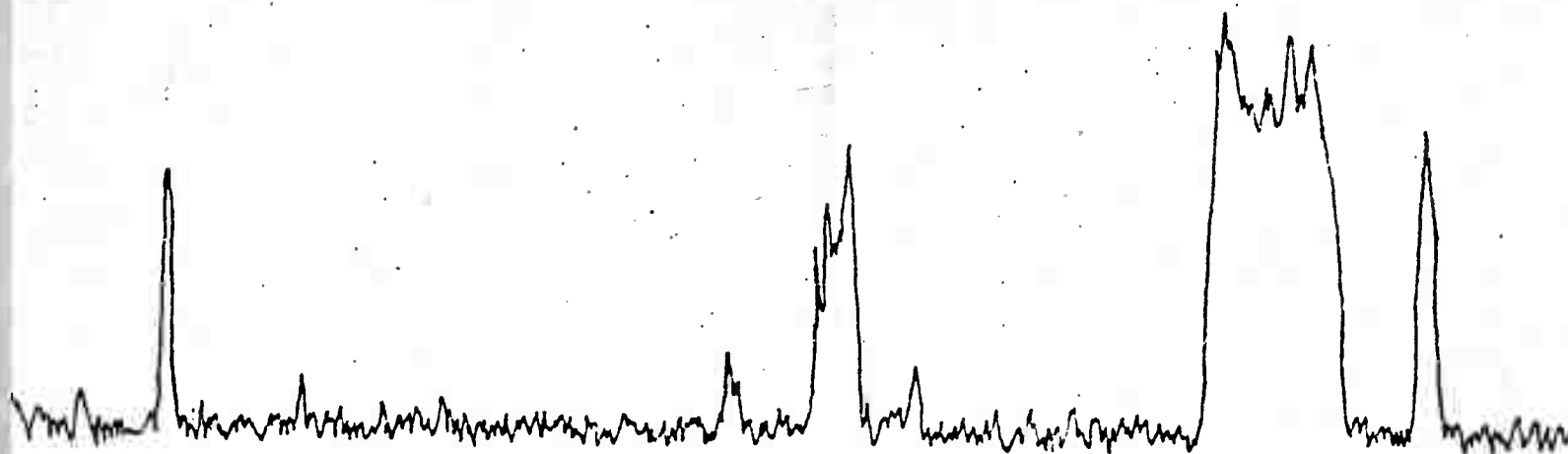
2

TRANSIENT PRESSURE FLUCTUATION





[TEST POINT 28  $M=0.92$ ,  $\alpha=-4$ ,  $\beta=-4$ ]



[TEST POINT 27  $M=0.92$ ,  $\alpha=-4$ ,  $\beta=-4$ ]



[TEST POINT 32  $M=0.9$ ,  $\alpha=+4$ ,  $\beta=-4$ ]

3

PRESSURE FLUCTUATION

FIGURE 17

BOEING

VOL

NO T2-2648

SEC

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UNCLASSIFIED

UNCLASSIFIED